

AD-A049 272

NAVAL POSTGRADUATE SCHOOL MONTEREY CALIF  
LASTOP - A COMPUTER CODE FOR LASER TURRET OPTIMIZATION OF SMALL--ETC(U)  
DEC 77 6 N VANDERPLAATS, A E FUHS

F/G 1/1

UNCLASSIFIED

NPS69-77-004

NL

1 OF 3  
AD  
A049 272



AD A 0 4 9 2 7 2

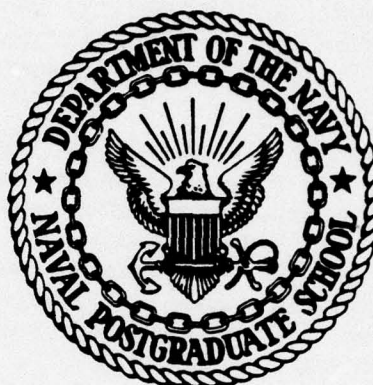
NPS69-77-004

2

# NAVAL POSTGRADUATE SCHOOL

Monterey, California

AD NO. \_\_\_\_\_  
DDC FILE COPY



DDC  
RECEIVED  
FEB 1 1978  
RECEIVED

LASTOP - A COMPUTER CODE FOR LASER TURRET  
OPTIMIZATION OF SMALL PERTURBATION TURRETS  
IN SUBSONIC OR SUPERSONIC FLOW

by

G. N. VANDERPLAATS

A. E. FUHS

Approved for public release; distribution unlimited.



NAVAL POSTGRADUATE SCHOOL  
Monterey, California

Rear Admiral Isham Linder  
Superintendent

J. R. Borsting  
Provost

LASTOP - A COMPUTER CODE FOR LASER TURRET  
OPTIMIZATION OF SMALL PERTURBATION TURRET IN  
SUBSONIC OR SUPERSONIC FLOW

A program has been developed which calculates optical path length and phase distortion arising from the density field surrounding a laser turret. Further, the program finds the optimum turret shape yielding minimum phase distortion. The aerodynamic model is briefly described; however, the optimization and control codes are thoroughly presented. Sample data input and sample output are given. The program is listed. The material is presented in detail so that this report constitutes a user's manual.

*Allen E Fuhs*

---

Allen E. Fuhs  
Distinguished Professor and Chairman  
Department of Mechanical Engineering

Approved by:

*G. J. Haltiner*

---

G. J. Haltiner, acting  
Dean of Science and Engineering

*William M. Tolles*

---

W. M. Tolles, acting  
Dean of Research

REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER 14 NPS69-77-004 ✓	2. GOVT ACCESSION NO.	3. RECIPIENT'S CATALOG NUMBER 9
4. TITLE (and Subtitle) 6 LASTOP - A Computer Code for Laser Turret Optimization of Small Perturbation Turrets in Subsonic or Supersonic Flow.		5. TYPE OF REPORT & PERIOD COVERED Final Report, 1976-1977
7. AUTHOR(s) 10 Garret N. Vanderplaats Allen E. Fuhs		8. PERFORMING ORG. REPORT NUMBER NPS69-77-004
9. PERFORMING ORGANIZATION NAME AND ADDRESS Department of Mechanical Engineering Naval Postgraduate School Monterey, CA 93940 ✓		10. CONTRACT OR GRANT NUMBER(s) AF185 Project Order No. 77-050
11. CONTROLLING OFFICE NAME AND ADDRESS Lt. Col. K. Gilbert AFWL/LRO Kirtland AFB, NM 87117		12. REPORT DATE 11 20 December 1977
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office) 12 238p.		13. NUMBER OF PAGES 231
		15. SECURITY CLASS. (of this report) Unclassified
		16. DECLASSIFICATION/DOWNGRADING SCHEDULE
16. DISTRIBUTION STATEMENT (of this Report) Approved for public release; distribution unlimited.		
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report)		
18. SUPPLEMENTARY NOTES		
19. KEY WORDS (Continue on reverse side if necessary and identify by block number) Laser turrets. Phase Distortion. Optimized turret shape. Aero-optics. Laser aerodynamics.		
20. ABSTRACT (Continue on reverse side if necessary and identify by block number) A program has been developed which calculates optical path length and phase distortion arising from the density field surrounding a laser turret. Further, the program finds the optimum turret shape yielding minimum phase distortion. The aerodynamic model is briefly described; however, the optimization and control codes are thoroughly presented. Sample data input and sample output are given. The program is listed. The material is presented in detail so that this report constitutes a user's manual.		

DD FORM 1473  
1 JAN 73EDITION OF 1 NOV 65 IS OBSOLETE  
S/N 0102-014-6601

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)



# ABSTRACT

A program has been developed which calculates optical path length and phase distortion arising from the density field surrounding a laser turret. Further, the program finds the optimum turret shape yielding minimum phase distortion. The aerodynamic model is briefly described; however, the optimization and control codes are thoroughly presented. Sample data input and sample output are given. The program is listed. The material is presented in detail so that this report constitutes a user's manual.

ACCESSION for	
RTIS	White Section <input checked="" type="checkbox"/>
DDC	Gulf Section <input type="checkbox"/>
UNANNOUNCED	<input type="checkbox"/>
JUSTIFICATION	
BY	
DISTRIBUTION/AVAILABILITY CODES	
Dist. AVAIL. and/or SPECIAL	
A	23 E.S.

DDC  
RECEIVED  
FEB 1 1978  
D

## TABLE OF CONTENTS

LIST OF FIGURES	4
NOMENCLATURE	5
I. INTRODUCTION	6
II. PROGRAM ORGANIZATION	11
III. DATA TRANSFER	13
IV. PROGRAM DATA	16
A. COPES - A CONTROL PROGRAM FOR ENGINEERING SYNTHESSES	17
B. LASER TURRET ANALYSIS	39
V. SAMPLE DATA	54
A. GEOMETRY	54
B. AERODYNAMICS	55
C. MIRROR	55
D. BEAM ORIENTATIONS	55
E. PHASE DISTORTION	56
F. COPES DATA	56
G. TURRET ANALYSIS DATA	62
VI. SAMPLE OUTPUT	66
LIST OF REFERENCES	92
APPENDIX A - PROGRAM FLOW CHARTS AND FORTRAN VARIABLES	93
APPENDIX B - PROGRAM LISTING	103
APPENDIX C - DATA FORMS	231



# LIST OF FIGURES

1. Small Perturbation Laser Turret on a Cylindrical Fuselage-----	6
2. Turret Geometry -----	7
3. Laser Beam Orientation-----	8
4. Phase Distortion Calculation Within the Laser Beam-----	9
5. Program Organization-----	11

I. INTRODUCTION	1
II. PROGRAM ORGANIZATION	11
III. DATA TRANSFER	12
IV. PROGRAM DATA	16
A. CODES - A CONTROL PROGRAM FOR ENGINEERING STUDENTS	17
B. LASER TURRET ANALYSIS	19
V. SAMPLE DATA	24
A. GEOMETRY	24
B. AERODYNAMICS	25
C. MINOR	27
D. BEAM ORIENTATION	28
E. PHASE DISTORTION	28
F. CODES DATA	28
G. TURRET ANALYSIS DATA	31
VI. SAMPLE OUTPUT	32
VII. LIST OF REFERENCES	33
APPENDIX A - PROGRAM FLOW CHARTS AND FORTRAN VARIABLES	34
APPENDIX B - PROGRAM LISTING	103
APPENDIX C - DATA FORMS	121

# NOMENCLATURE

$a_k$	coefficients for the turret shape polynomial in x-direction
$b_p$	coefficients for the turret shape polynomial in the $\theta$ -direction
$\ell$	extent of turret in x-direction; for $ x  > \ell$ radius of fuselage is $R_0$ .
$L$	the distance $2L$ is separation between turrets
OPL	optical path length
PD	phase distortion; nondimensional
$r$	radial distance
$R_D$	fuselage radius
$W_i$	weighting factor for i-th beam direction
$x$	axial distance in cylindrical coordinates
$x_M$	axial location of mirror center
$z$	reference direction to measure angles within beam cross section
$\beta^2$	shorthand notation for $1 - M_\infty^2$
$\gamma$	beam elevation angle
$\epsilon$	nondimensional turret height; $R_0$ is reference length
$\epsilon_M$	radial location of mirror center
$\eta$	polar coordinate used to locate points or rays within the beam
$\theta$	variable in cylindrical coordinates used to describe turret shape
$\phi$	perturbation potential function; also, beam azimuth angle.



## I. INTRODUCTION

A computer program is described here which obtains the optimum shape of a laser turret to minimize optical distortion of a laser beam. The analysis and optimization procedure on which the program is based are described in detail in Ref. 1.

The turret is assumed to be situated on a cylindrical fuselage, as shown in Figure 1. The details of the turret geometry are shown in Figure 2. The shape of the turret is defined by the product of two polynomials, so that

$$r = \epsilon f(x) f(\theta) \quad (1)$$

where

$$f(x) = 1 + \bar{a}_1 x + \bar{a}_2 x^2 + \dots + \bar{a}_k x^k \quad (2)$$

and

$$f(\theta) = 1 + \bar{b}_2 \theta^2 + \dots + \bar{b}_p \theta^p \quad (3)$$

where  $p$  is the sequence of even numbers 2, 4, 6 . . . .

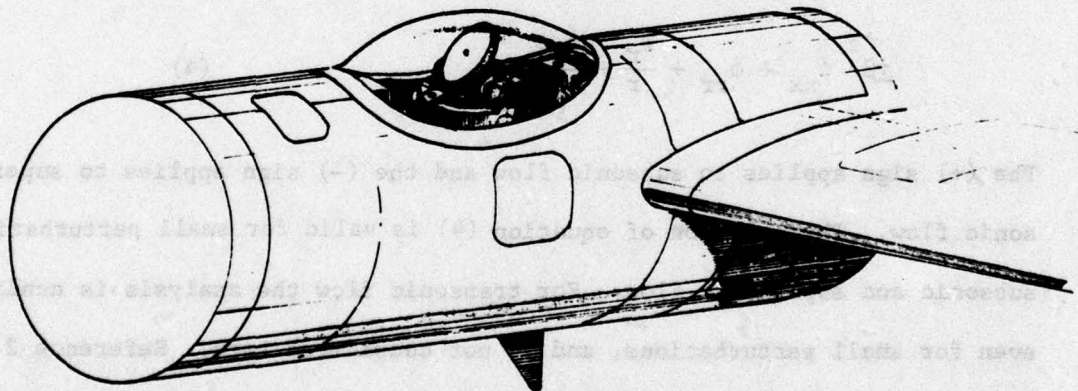


Figure 1. Small Perturbation Laser Turret on a Cylindrical Fuselage.

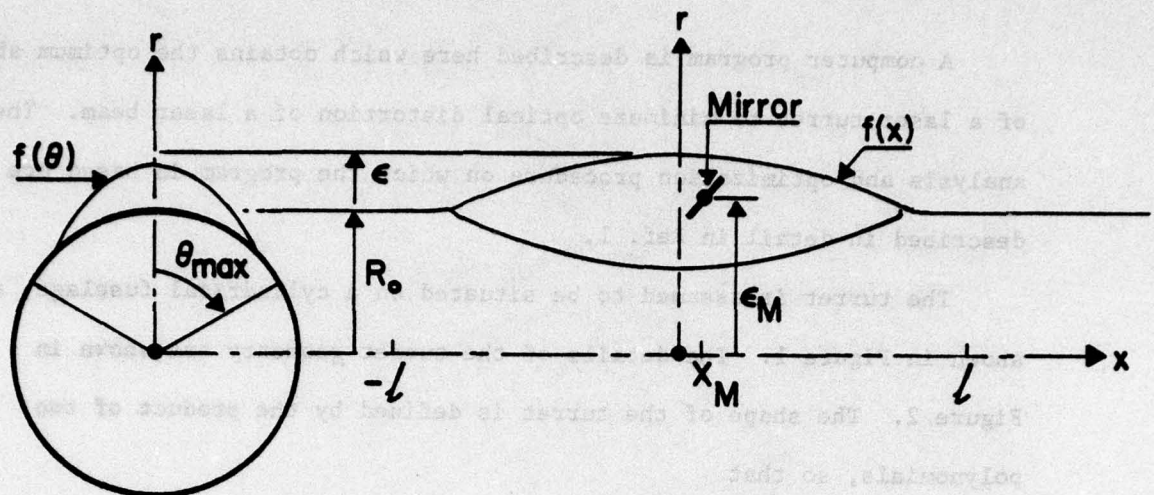


Figure 2. Turret Geometry.

Optional distortion is introduced into a laser beam propagating through the flow field surrounding the turret; see Ref. 2. For purposes of this analysis, the flow is assumed to be compressible and inviscid and is governed by the small perturbation equation

$$\pm \beta^2 \phi_{xx} + \phi_{rr} + \frac{\phi_r}{r} + \frac{\phi_{\theta\theta}}{r^2} = 0 \quad (4)$$

The (+) sign applies to subsonic flow and the (-) sign applies to supersonic flow. The solution of equation (4) is valid for small perturbation subsonic and supersonic flow. For transonic flow the analysis is nonlinear, even for small perturbations, and is not considered here. Reference 2 discusses the formulation of the aerodynamics model for a variety of geometrical shapes and flow regimes.

From the solution of the potential equation, the perturbation velocities,  $u$  and  $v$ , may be calculated anywhere in the flow field. From knowledge of the flow field the optical path length on any ray of a laser



beam is calculated. The laser beam is propagated through the flow field as shown in Figure 3. Taking the center of the beam as the reference ray, the difference in optical path lengths, OPL, between a specified ray and the ray on the beam center is calculated as

$$\Delta OPL = OPL_j - OPL_i \quad (5)$$

where the subscript i corresponds to the reference ray and j corresponds to the particular ray being considered. The phase distortion, PD, is defined as  $\Delta OPL/\lambda$  where  $\lambda$  is the wave length of radiation; Refs. 3 and 4 discuss OPL and PD in more detail.

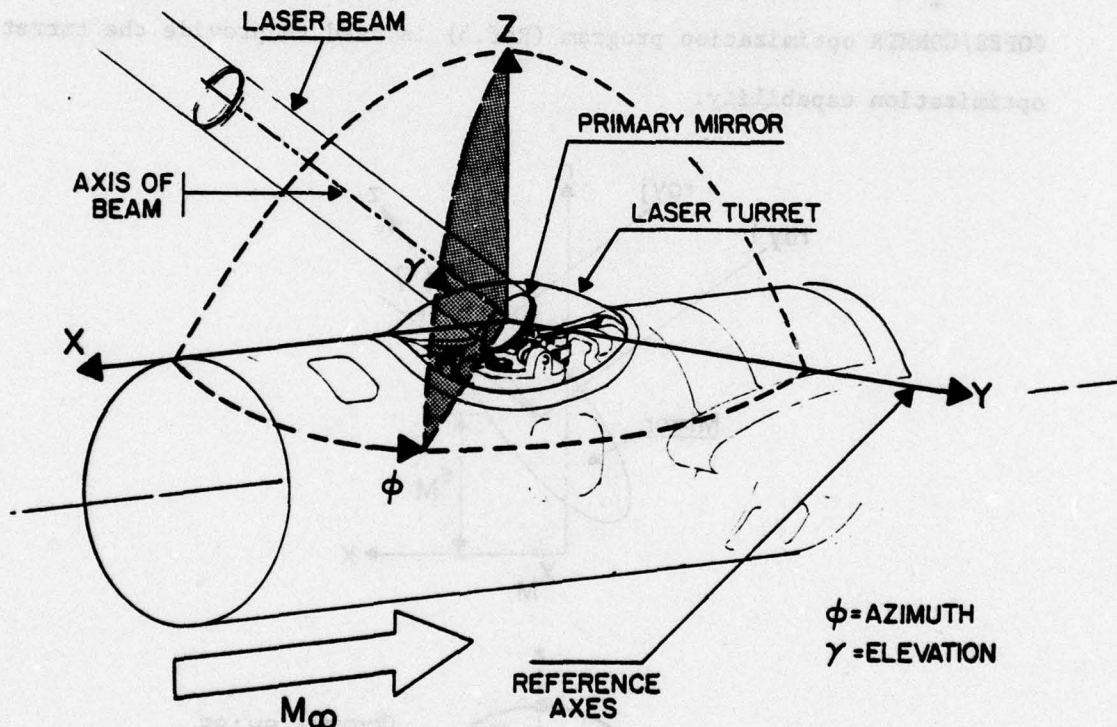


Figure 3. Laser Beam Orientation

Phase distortion, PD, is calculated numerically at several radial and angular locations within the beam as shown in Figure 4. The sum of  $(PD)^2$  over all calculation points for several beam orientations is considered to provide a measure of the "goodness" of the turret design. The coefficients of the turret shape functions of equations 2 and 3 are then determined to minimize

$$SUMPD = \sum_{\text{orientations}} W_i \sum_{\text{radii}} \sum_{\text{angles}} (PD)^2 \quad (6)$$

where  $W_i$  is a weighting factor applied to the  $i$ -th beam orientation. The COPES/CONMIN optimization program (Ref.5) is used to provide the turret optimization capability.

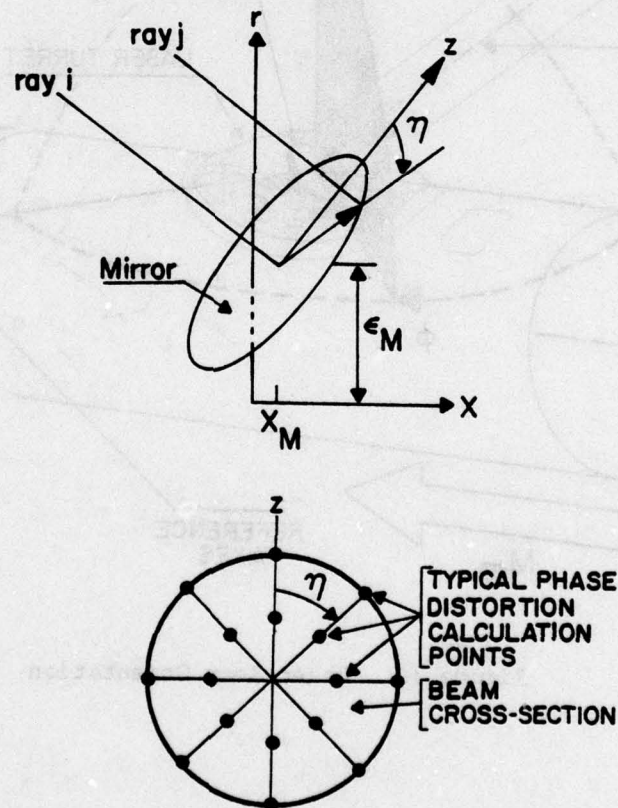
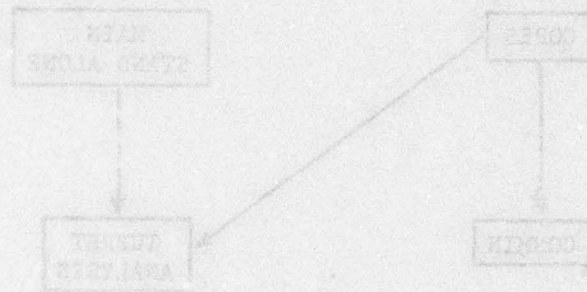


Figure 4. Phase Distortion Calculation Within the Laser Beam.



Finally, the optical aberrations are calculated for each beam orientation in terms of Zernicke coefficients. This provides a measure of the turret design in terms familiar to optical design specialists; see Ref. 6.

In the following sections, the program organization, data transfer mechanism, input data and output are described. Test cases are provided to help in making the program operational. Additional program details and a FORTRAN listing are included in the Appendices.



## II. PROGRAM ORGANIZATION

The basic program organization is shown in block diagram form in Figure 5. The COPES program is the main driver which calls the optimization program, CONMIN, and the turret analysis program; COPES is an acronym for Control Program for Engineering Synthesis, and CONMIN is an acronym for CONstrained function MINimization. Both are general purpose programs which may be applied to a wide variety of engineering design problems (Ref. 7). If only the analysis of a specific turret shape is desired, this may be done without COPES/CONMIN by using a very simple main program. Alternatively, COPES/CONMIN may be used for a single analysis by specifying the proper value of a single control parameter in the input data.

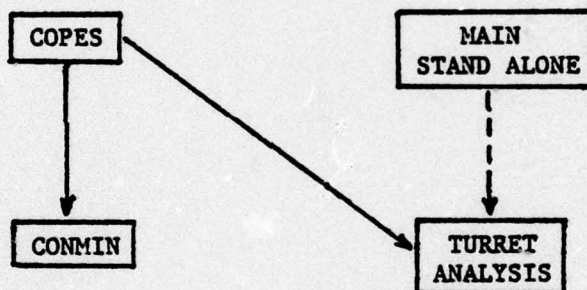


Figure 5. Program Organization.

The combined program containing COPES, CONMIN and laser turret analysis is referred to by the acronym LASTOP, for LASer Turret Optimization.

The entire program is written in FORTRAN IV and has been executed, without modification, on IBM 360/67 and CDC 7600 computers. The program executes in approximately  $50^k$  octal words of storage on a CDC computer.



The program reads from unit 5 and writes on unit 6. Units 20 and 40 are scratch files. (These file numbers may be changed by changing two cards at the beginning of the COPES program.) Execution times on a CDC 7600 computer are approximately 0.3 and 1.0 CPU seconds for subsonic and supersonic flow respectively for the analysis of one beam orientation. In a typical design optimization run, fifteen beam orientations may be considered. Assuming fifty candidate designs are analyzed before the optimum is obtained, the total CPU time is from 200 to 750 seconds.

To execute the turret optimization program, the user must be familiar with the mechanism by which data are transferred between analysis and design programs. This is the subject of the following section.

### III. DATA TRANSFER

To couple the analysis and optimization programs for automated design optimization, pertinent data must be transferred between programs. This is done by means of a single labeled common block. To execute the program, it is necessary for the user to know what information is transferred and the location in common of that information. This section defines the data to be transferred and identifies their location within the common block.

The variables contained in the "GLOBAL" common block are listed below; the terms have the following meaning:

**LOCATION** - The physical location of the variable in the common block.

For example, the polynomial coefficient  $\bar{a}_1$  is in location 2 while  $\bar{a}_2$  is in location 3. The usual design objective (phase distortion), SUMP2, is in global location 169.

**TYPE** - The purpose of the variable in design optimization. D = design variable, S = sensitivity variable, O = objective function and C = constraint function. Note that a sensitivity variable may be a design variable if this is meaningful. For example, the direction of minimum phase distortion may be found by considering only one beam orientation and treating the azimuth angle (location 108) and elevation angle (location 78) as design variables. Similarly, objective and constraint functions are interchangeable. For example, the minimum turret half-length (location 21) may be found with an upper bound on phase distortion (location 169). Under special circumstances, the objective function may also be a design variable. For example, the maximum turret height (location 76) may be sought, subject to a



Because the turret height is intended as a design variable, it must also be a design variable here because it only appears on the right-hand side of equations in the program.

**MATH** - The mathematical symbol for the variable (used in Ref. 1).

**DEFINITION** - Physical meaning of the variable.

<u>LOCATION</u>	<u>TYPE</u>	<u>FORTRAN</u>	<u>MATH</u>	<u>DEFINITION</u>
1-20	D	ABAR(20)	$\bar{a}_1$	Polynomial coefficients on $f(x)$ .
21	S	ACL	L	Turret half-spacing for Fourier analysis.
22	S	AKPRIM	$k'$	Constant in phase distortion calculations.
23	D,S	AL	$l$	Turret half length divided by RFUS.
24-53	S	AMACHI(30)	$M_\infty$	Mach number associated with i-th beam orientation.
54-73	D	BBAR(20)	$\bar{b}_1$	Polynomial coefficients on $f(\theta)$ .
74	S	DENRTO	$\rho/\rho_{SL}$	Density of air divided by density of air at sea level.
75	S	DENGAM	$\gamma$	Exponent in pressure-density relationship.
76	D,S	EPS	$\epsilon$	Turret height divided by RFUS.
77	S	EPSM	$\epsilon_m$	Mirror center height divided by RFUS.
78-107	S	GAMMAI(30)	$\gamma$	Elevation angle of i-th beam orientation.
108-137	S	PHII(30)	$\phi$	Azimuth angle of i-th beam orientation.
138	S	RFUS	$R_0$	Fuselage radius (meters).
139-168	C	SLOPEX(30)	$f'(x)$	Slope of turret surface in stream-wise direction.
169	O	SUMPD2	$\Sigma(PD)^2$	Sum of squares of all calculated phase distortions.
170	D,S	TDENRT	$\rho/\rho_{SL}$	Density of air inside canopy divided by density of air at sea level.
171	D,S	THMAX	$\theta_{MAX}$	Half angle of turret (degrees).
172	S	WAVEL	$\lambda$	Wave length of radiation (meters).
173-202	S	WGHTI(30)	$W_1$	Weighting factor on i-th beam orientation.
203	S	XM	$X_M$	X-coordinate of center of mirror.



#### IV. PROGRAM DATA

The data for laser turret analysis and optimization are separated into two parts. First are the control program (COPES) data which control the analysis and design operations. These are followed by the turret analysis data.

When the program is being made operational or when only analysis is desired, the turret analysis program may be run, stand-alone using a simple driver program given in the subsection on laser turret analysis. In this case, the COPES data are omitted, and only the turret analysis data are provided.

Appendix C contains convenient data forms for both the COPES and the turret analysis data. The reader may want to copy these forms for use in preparing a problem.

#### A. COPEES - A CONTROL PROGRAM FOR ENGINEERING SYNTHESSES

The COPEES program is a general purpose program to aid in design optimization and is not limited to the specific application for which it is used here. The user must provide an analysis program in subroutine form, which in this case is the analysis of a laser turret in subsonic and supersonic flow. The principal requirements are that the analysis program be coded in FORTRAN and be segmented into input, execution and output and that all design information be stored in a single labeled common block called GLOBCM.

The COPEES program provides four specific capabilities:

1. Simple analysis - just as if COPEES was not used.
2. Optimization - minimization or maximization of one calculated function with limits imposed on other functions.
3. Sensitivity analysis - the effect of changing one or more design variables on one or more calculated functions.
4. Two-variable function space - analysis for all specified combinations of two design variables.

COPEES utilizes the general purpose optimization program CONMIN (Ref. 2) for optimization, and this is the capability of primary interest here. Data requirements for options 3 and 4 are included for completeness.

To better understand the COPEES data requirements, the following definitions are useful:

Design Variables - Design variables are those parameters which the optimization program is allowed to change in order to improve the design. Design variables appear only on the right-hand side of an equation in the analysis program. COPEES considers two types of design variables, independent and dependent. If two or more variables are always required to have the



same value or be in a constant ratio, one is the independent variable while the remaining are dependent variables. For example, if the turret shape polynomials are required to be the same in both the x and  $\theta$  directions, the coefficients  $\bar{a}_i$  may be independent variables, and the  $\bar{b}_i$  may be dependent variables. In this example, the total number of design variables will then be twice the number of independent design variables.

Objective Function - The parameter which is to be minimized or maximized during optimization is an objective function. Included are parameters calculated as a function of specified design variables during a sensitivity or two-variable function space study. Objective functions always occur on the left side of an equation unless the objective function is also a design variable. (The turret height may be maximized as an objective function if it is also a design variable. In this way, the maximum height is found for which no constraints are violated.) An objective function may be linear or non-linear and implicit or explicit but must be a continuous function of the design variables to be meaningful.

Constraint - Any parameter which must not exceed specified bounds for the design to be acceptable is a constraint. Constraint functions always appear on the left side of an equation. Just as for objective functions, constraints may be linear or non-linear and implicit or explicit but must be continuous functions of the design variables.

The COPES program reads from unit 5 and writes output on unit 6. Units 20 and 40 are used as scratch files. The scratch file numbers may be changed by changing two cards at the beginning of the COPES program.

The data required to run the COPES program are now defined. All GLOBAL LOCATION NUMBERS refer to the location of the specified variable in the labeled common block, GLOBCM. The pertinent variables and their global locations are listed in the section entitled DATA TRANSFER.

The data are segmented into "blocks" for convenience. All formats are alphanumeric for TITLE, END, and STOP cards; F10 for real data; and I10 for integer data. Comment cards may be inserted anywhere in the data stack prior to the END card and are identified by a dollar sign (\$) in Column 1. The COPES data stack must terminate with an end card containing the word "END" in Columns 1-3.

Data coding forms are provided in Appendix C.



# COPIES

DATA BLOCK     A

DESCRIPTION:    Title Card

## FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
<u>TITLE</u>								20A4
LASER TURRET OPTIMIZATION								

### FIELD

### CONTENTS

1-8

Any 80 character title

### REMARKS

1) Program is terminated by the word 'STOP' in columns 1-4.

COPES

DATA BLOCK     B

DESCRIPTION : Program Control Parameters

FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
NCALC	NDV	NSV	N2VAR	IPNPUT	IPSENS	IP2ZAR		7I10
2	4	3	2	0	0	0		

FIELD

CONTENTS

- |   |  |
|---|--|
| 1 | NCALC: Calculation control<br>0 - Read input and stop. Data of blocks A-B are required. Remaining data are optional.<br>1 - One cycle through program. Data of blocks A-B are required. Remaining data are optional.<br>2 - Optimization. Data of blocks A-I are required. Remaining data are optional.<br>3 - Sensitivity analysis. Data of blocks A-B and J-K are required. Remaining data are optional.<br>4 - Two variable function space. Data of blocks A-B and L-O are required. Remaining data are optional. |
| 2 | NDV: Number of independent design variables in optimization.   |
| 3 | NSV: Number of variables on which sensitivity analysis will be performed.  |
| 4 | N2VAR: Number of objective functions in a two variable function space study.   |
| 5 | IPNPUT: Input print control<br>0 - Print card images plus formatted print of input.<br>1 - Formatted print of input only.<br>2 - No print of input.  |
| 6 | IPSENS: Print control for sensitivity analysis. If IPSENS.GT.0 detailed print will be called for at each step in the sensitivity analysis.<br>DEFAULT = No print.  |
| 7 | IP2VAR: Print control for two variable function space study. If IP2VAR.GT.0 detailed print will be called for at each step (each X-Y combination).<br>DEFAULT = No print.  |



2. JAMES W. TAYLOR, 7

## COPEs

DATA BLOCK    C    Omit if NDV = 0 in Block A

DESCRIPTION : Integer Optimization Control Parameters

### FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
IPRINT	ITMAX	ICNDIR	NSCAL	ITRM	LINOBJ	NACMX1	NFDG	8I10
5	0	0	5	0	0	0	0	

### FIELD

### CONTENTS

- |   |   |
|---|---|
| 1 | IPRINT: Print control used in optimization program, CONMIN.<br>0 - No print during optimization.<br>1 - Print initial and final optimization information.<br>2 - Print above plus function value and design variable values at each iteration.<br>3 - Print above plus constraint values, direction vector and move parameter at each iteration.<br>4 - Print above plus gradient information.<br>5 - Print above plus each proposed design vector, objective function and constraints during the one-dimensional search. |
| 2 | ITMAX: Maximum number of optimization iterations allowed.<br>DEFAULT = 20.  |
| 3 | ICNDIR: Conjugate direction restart parameter.<br>DEFAULT = NDV+1.  |
| 4 | NSCAL: Scaling parameter. GT.0 - Scale design variables to order of magnitude one every NSCAL iterations. LT.0 - Scale design variables according to scaling values input.<br>DEFAULT = No scaling.   |
| 5 | ITRM: Number of subsequent iterations which must satisfy relative or absolute convergence criterion before optimization process is terminated.<br>DEFAULT = 3.  |
| 6 | LINOBJ: Linear objective function identifier. If the optimization objective is known to be a linear function of the design variables, set LINOBJ = 1.<br>DEFAULT = Non-Linear.  |
| 7 | NACMX1: One plus the maximum number of active constraints anticipated.<br>DEFAULT = NDV+2.  |



FIELDCONTENTS

8

- NFDG: Finite difference gradient identifier.
- 0 - All gradient information is computed by finite difference.
  - 1 - Gradient of objective is computed analytically. Gradients of constraints are computed by finite difference.
  - 2 - All gradient information is computed analytically.

REMARKS

- 1) For LASER TURRET OPTIMIZATION, the value of LINOBJ and NFDG should always be zero. The value of NSCAL = 5 is suggested and ITRM = NACMX1 = 0 should be used. The value of IPRINT may be reduced when the user is familiar with the optimization output.

# COPES

DATA BLOCK    D    Omit if NDV = 0 in Block A

DESCRIPTION : Floating Point Optimization Program Parameters

## FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
FDCH	FDCHM	CT	CTMIN	CTL	CTLMIN	THETA	PHI	8F10
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	
DELFUN	DABFUN							2F10
0.0	0.0							

Note: Two cards of data are read here.

### FIELD

### CONTENTS

- |   |  |
|---|--|
| 1 | FDCH: Relative change in design variables in calculating finite difference gradients.<br>DEFAULT = 0.01.   |
| 2 | FDCHM: Minimum absolute step in finite difference gradient calculations.<br>DEFAULT = 0.001.   |
| 3 | CT: Constraint thickness parameter.<br>DEFAULT = -0.05.  |
| 4 | CTMIN: Minimum absolute value of CT considered in the optimization process.<br>DEFAULT = 0.004.  |
| 5 | CTL: Constraint thickness parameter for linear and side constraints.<br>DEFAULT = -0.01.   |
| 6 | CTLMIN: Minimum absolute value of CTL considered in the optimization process.<br>DEFAULT = 0.001.  |
| 7 | THETA: Mean value of push-off factor in the method of feasible directions.<br>DEFAULT = 1.0.   |
| 8 | PHI: Participation coefficient, used if one or more constraints are violated.<br>DEFAULT = 5.0.  |
| 1 | DELFUN: Minimum relative change in objective function to indicate convergence of optimization process.<br>DEFAULT = 0.001.                                       |
| 2 | DABFUN: Minimum absolute change in objective function to indicate convergence of the optimization process.<br>DEFAULT = 0.001 times the initial objective value. |



1) For LASER TURRET OPTIMIZATION default values of these parameters usually work well.

# COPES

DATA BLOCK    E    Omit if NDV = 0 in Block A

DESCRIPTION:    Total Number of Design Variables, Design Objective Identification and Sign on Design Objective.

## FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
NDVTOT	IOBJ	SGNOBJ						2I10,F10
0	163	-1.0						

### FIELD

### CONTENTS

- |   |         |  |
|---|---------|--|
| 1 | NDVTOT: | Total number of variables linked to the design variables. NDVTOT must be greater than or equal to NDV. This option allows two or more parameters to be assigned to a single design variable. The value of each parameter is the value of the design variable times a multiplier which may be different for each parameter.<br>DEFAULT = NDV. |
| 2 | IOBJ:   | Global variable number associated with objective function in optimization.   |
| 3 | SGNOPT: | Sign used on objective of optimization to identify whether function is to be maximized or minimized. +1.0 indicates maximization. -1.0 indicates minimization.<br>DEFAULT = -1.0.  |

### REMARKS

- 1) For LASER TURRET OPTIMIZATION, the numbers used in this example are correct if phase distortion is to be minimized. If phase distortion is to be maximized set SGNOPT = +1.0.



## COPEs

DATA BLOCK    F    Omit if NDV = 0 in Block A

DESCRIPTION: Design variable bounds, initial values and scaling factors.

## FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
VLB	VUB	X	SCAL					4F10
-3.0	3.0	0.0	0.0					

Note: Read one card for each of the NDV independent design variables.

### FIELD

### CONTENTS

- |   |   |
|---|---|
| 1 | VLB: Lower bound on the design variable.  |
| 2 | VUB: Upper bound on the design variable.  |
| 3 | X: Initial value of the design variable.<br>If X is non-zero, this will supercede<br>the value initialized by subroutine<br>ANALIZ. |
| 4 | SCAL: Design variable scale factor. Not used<br>if NSCAL.GE.0 in Block C.   |

### REMARKS

- 1) For LASER TURRET OPTIMIZATION, the values used in this example are suggested.

COPES

DATA BLOCK G Omit if NDV = 0 in Block A.

DESCRIPTION: Design Variable Identification

FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
NDSGN	IDSGN	AMULT						
1	26	1.0						2I10,F10

Note: Read one card for each of the NDVTOT Design Variables.

FIELD

CONTENTS

- |   |   |
|---|---|
| 1 | NDSGN: Design variable number associated with the variable.   |
| 2 | IDSGN: Global variable number associated with the variable.   |
| 3 | AMULT: Constant multiplier on the variable.<br>The value of the variable will be the value of the design variable, NDSGN times AMULT.<br>DEFAULT = 1.0. |



# COPEs

DATA BLOCK H Omit if NDV = 0 in Block A

DESCRIPTION : Number of sets of constrained parameters.

## FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
NCONS								I10
1								

### FIELD

### CONTENTS

1

NCONS: Number of constraint sets in the optimization problem.

### REMARKS

- 1) If two or more adjacent parameters in the Global common block have the same limits imposed, these are part of the same constraint set.

## COPEs

DATA BLOCK I Omit if NDV = 0 in Block A or if NCONS = 0 in Block M.

DESCRIPTION: Constraint Identification and Bounds.

## FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
ICON	JCON	LCON						3I10
224	234	1						
BL	SCAL1	BU	SCAL2					4F10
-.3	.3	.3	.3					

Note: Read two cards for each of the NCONS constraint sets.

<u>FIELD</u>	<u>CONTENTS</u>
1	ICON: First Global number corresponding to the constraint set.
2	ICON: Last Global number corresponding to the constraint set. DEFAULT = ICON.
3	LCON: Linear constraint identifier for this set of constrained variables. LCON = 1 indicates linear constraints. DEFAULT = 0 = Nonlinear constraint.
1	BL: Lower bound on the constrained variables. Value less than $-1.0E+15$ is assumed unbounded.
2	SCAL1: Normalization factor on lower bound. DEFAULT = Max of ABS(BL), 0.1.
3	BU: Upper bound on the constrained variables. Value greater than $1.0E+15$ is assumed unbounded.
4	SCAL2: Normalization factor on upper bound. DEFAULT = Max of ABS(BU), 0.1.

## REMARKS

- 1) The normalization factors should usually be defaulted.



# COPES

DATA BLOCK J Omit if NSV = 0 in Block A

DESCRIPTION : Sensitivity Objectives.

## FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
NSOBJ								I10
4								
NSN1	NSN2	NSN3	NSN4					8I10
26	27	41	42					

Note: Two or more cards are read here.

### FIELD

### CONTENTS

1

NSOBJ: Number of separate objective functions to be calculated as functions of the sensitivity variables.

1-8

NSNI: Global variable number associated with the sensitivity objective functions.

### REMARKS

- 1) More than eight sensitivity objectives are allowed. Add data cards as required to contain data.

## COPES

DATA BLOCK K Omit if NSV = 0 in Block A

DESCRIPTION: Sensitivity Variables

### FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
ISENS	NSENS							2I10
26	4							
SNS1	SNS2	SNS3	SNS4	...	...			8F10
2.0	1.0	3.0	4.0					

Note: Read one set of data for each of the NSV sensitivity variables.

Note: Two or more cards are read here.

#### FIELD

#### CONTENTS

1

ISENS: Global variable number associated with the sensitivity variable.

2

NSENS: Number of values of the sensitivity variable to be considered.

1-8

SNSI: Values of the sensitivity variable, for J = 1, NSENS. J = 1 corresponds to nominal value.

#### REMARKS

- 1) More than eight values of the sensitivity variable are allowed. Add data cards as required to contain data.



# COPEs

DATA BLOCK    L    Omit if N2VAR = 0 in Block A

DESCRIPTION: Two variable function space control parameters.

## FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
N2VX	M2VX	N2VY	M2VY					4I10
26	5	27	5					

### FIELD

### CONTENTS

- |   |   |
|---|---|
| 1 | N2VX: Global location of X-variable in two-variable function space. |
| 2 | M2VX: Number of values of X-variable to be considered.              |
| 3 | N2VY: Global location of Y-variable in two-variable function space. |
| 4 | M2VY: Number of values of Y-variable to be considered.              |

# COPEs

DATA BLOCK M Omit if NZVAR = 0 in Block A

DESCRIPTION: Objective Functions of Two-variable Function Space Study.

## FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
NZ1	NZ2	NZ3	NZ4	. . .	. . .			8I10
7	4	21	67					

### FIELD

1-8

### CONTENTS

NZ1: Global variable location corresponding to ITH function of X and Y in two variable function space.

### REMARKS

I = 1, NZVAR, where NZVAR is read in Block A.

- 1) More than eight objective functions are allowed. Add data cards as required to contain data.



# COPEs

DATA BLOCK N Omit if N2VAR = 0 in Block A

DESCRIPTION: Values of X-variable in Two-variable Function Space Study.

## FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
X1	X2	X3	X4	...	...			8F10
0.5	1.0	1.5	2.0					

### FIELD

1-8

### CONTENTS

XI: Values of X-variable to be considered in two-variable function space.

I = 1, MZVX, where MZVX is read in Block L.

### REMARKS

- 1) More than eight X-values are allowed. Add data cards as required to contain data.

## COPEs

DATA BLOCK 0 Omit if N2VAR = 0 in Block A

DESCRIPTION: Values of Y-variable in two-variable Function Space Study.

### FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
Y1	Y2	Y3	..	..	..			8F10
0.0	-1.0	1.0						

### FIELD

1-8

### CONTENTS

YI: Values of Y-variable to be considered  
in two-variable function space.  
I = 1, MZVY, where MZVY is read in Block L.

### REMARKS

- 1) More than eight Y-values are allowed. Add data cards as required to contain data.



## COPES

### DATA BLOCK P

DESCRIPTION: Copes data 'END' card.

### FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
END								3A1.
END								

### FIELD

### CONTENTS

1

The word 'END' in columns 1-3.

### REMARKS

- 1) This card must appear at the end of the COPES data.
- 2) This ends the COPES input data.

## B. LASER TURRET ANALYSIS

Data for the laser turret analysis follow the COPES data. If the general design capability of COPES is not needed, the analysis program can be run by itself by using the following simple main program.

C    MAIN PROGRAM FOR STAND ALONE LASER TURRET ANALYSIS.

C

C -    INPUT  
      ICALC = 1  
      CALL ANALIZ(ICALC)

C

C -    EXECUTION AND OUTPUT.  
      ICALC = 3  
      CALL ANALIZ(ICALC)  
      STOP  
      END

If this main program is used, the COPES and CONMIN routines are omitted, and the COPES data are not read. This provides simple analysis of a specified turret and allows the turret analysis program to be tested independently.

The turret analysis program reads from unit 5 and writes the output on unit 6.

The input data are segmented into blocks for convenience, just as for the COPES data.

Comment cards are not allowed in the turret analysis data.

Data coding forms are provided in Appendix C.



TURRET

DATA BLOCK    A

DESCRIPTION:    Title Card.

FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
TITLE								20/44
LASER TURRET ANALYSIS								

FIELD

CONTENTS

1-8

Title : Any 80 character title.

TURRET

DATA BLOCK    B

DESCRIPTION : Aerodynamics, Optics constants

FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
AMACH	DENRTO	TDENRT	DENGAM	AKPRIM	WAVEL			6F10
1.25	.25	.25	1.405	.00023	3.4-6			

FIELD

CONTENTS

- |   |  |
|---|--|
| 1 | AMACH: Freestream Mach number.   |
| 2 | DENRTO: Freestream air density/sea level density   |
| 3 | TDENRT: Air density inside turret/sea level density  |
| 4 | DENGAM: Exponent in pressure-density relationship<br>$\frac{p}{p_0} = \left(\frac{\rho}{\rho_0}\right)^\gamma$ |
| 5 | AKPRIM: Phase distortion constant, $k'$  |
| 6 | WAVEL: Wave length of radiation, $\lambda$ (meters)  |

REMARKS

- 1) AMACH is the freestream MACH number for all beam orientations unless specified otherwise in data Block N.



# TURRET

DATA BLOCK c

DESCRIPTION: Turret Geometry

## FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
RFUS	AL	THMAX	ACL	EPS				5F10
2.5	2.0	60.	10.	0.3				

### FIELD

### CONTENTS

- |   |        |  |
|---|--------|--|
| 1 | RFUS:  | Fuselage Radius (meters)   |
| 2 | AL:    | Turret half length divided by RFUS.  |
| 3 | THMAX: | Half angle subtended by turret (deg.)  |
| 4 | ACL:   | Half spacing between turrets divided by RFUS, for Fourier Series calculations. |
| 5 | EPS:   | Turret height divided by RFUS at $x = r = 0$ .                                 |

### REMARKS

- 1) ACL must be much larger for supersonic flow than for subsonic flow to avoid interference between turrets. ACL = 5. is adequate for subsonic flow calculations.

# TURRET

DATA BLOCK    D

DESCRIPTION:    Turret Geometry

## FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
MAXK	MAXP	NXBC	NTHBC					4I10
6	6	2	1					

### FIELD

### CONTENTS

1

MAXK: Order of x-polynomial shape function.  
 $f(x) = 1 + a_1 x + \dots + a_{\text{maxk}} x^{\text{maxk}}$

2

MAXP: Order of polynomial shape function.  
 $f(\theta) = 1 + b_1 \theta + \dots + b_{\text{maxp}} \theta^{\text{maxp}}$

3

NXBC: Number of sets of y and y' boundary conditions in x-direction, externally imposed.

4

NTHBC: Number of sets of  $\theta$  and  $\theta'$  boundary conditions in  $\theta$ -direction, externally imposed.

### REMARKS

- 1) The order plus one of each polynomial must be at least as great as the actual number of externally imposed boundary conditions.



# TURRET

## DATA BLOCK    E

DESCRIPTION: Polynomial coefficients in x-direction.

## FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
1	ABAR1	.	.			ABAR6	.	8F10
1.	0.	-.61111	0.	-.18056	0.	-.00694		

### FIELD

### CONTENTS

1-8

ABAR1    Coefficient of x-polynomial shape  
function,  $f(x) = 1 + \bar{a}_1 x + \dots$   
 $\bar{a}_{\text{maxk}} x^{\text{maxk}}$

### REMARKS

- 1) The total number of coefficients equals  $1 + \text{MAXK}$ . Additional data cards are used as required to contain the data.

# TURRET

## DATA BLOCK    F

DESCRIPTION: Geometric boundary conditions in x-direction.

## FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
X	YBC	YPBC						3F10
-1.	0.	0.						

Note: NXBC cards are required.

### FIELD

### CONTENTS

- |   |  |
|---|--|
| 1 | X: X-location as fraction of turret half-length, AL, where boundary conditions is imposed. |
| 2 | YBC: Required value of $f(x)$ at $x$ .   |
| 3 | YPBC: Required value of $f'(x)$ at $x$ .   |

### REMARKS

- 1) The boundary condition that  $f(x, \theta) = \text{EPS}$  at  $x = \theta = 0$  is automatically imposed.
- 2) If YBC or YPBC is input greater than or equal 200., the corresponding boundary condition is omitted, i.e., if YPBC = 200., no boundary condition is imposed on  $f'(x)$ .



# TURRET

## DATA BLOCK G

DESCRIPTION: Polynomial coefficients in  $\theta$ -directions.

## FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
1	BBAR1	BBAR2	...	...	...	BBAR6	...	8F10
1.	0.	-.61111	0.	-.18056	0.	-.006944		

### FIELD

### CONTENTS

1-8

BBAR1: Coefficient of  $\theta$  polynomial shape  
function,  $f(\theta) = 1 + \bar{b}_1 \theta +$

$$\sum_{n=1}^{b_{\max}} \bar{b}_n \theta^n$$

### REMARKS

- 1) The total number of coefficients equals  $1 + \text{MAXP}$ . Additional data cards are used as required to contain the data.

# TURRET

## DATA BLOCK    H

DESCRIPTION : Geometric boundary conditions in  $\theta$ -direction.

## FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
THETA	YBC	YPBC						3F10
1.	0.	0.						

Note: NTHBC cards are required.

<u>FIELD</u>	<u>CONTENTS</u>
1	THETA: $\theta$ -location divided by turret half angle, THMAX, where the boundary condition is imposed.
2	YBC: Required value of $f(\theta)$ at THETA.
3	YPBC: Required value of $f'(\theta)$ at THETA.

## REMARKS

- 1) The boundary condition that  $f(x, \theta) = \text{EPS}$  at  $x = \theta = 0$  is automatically imposed.
- 2) If YBC or YPBC is input greater than or equal 200., the corresponding boundary condition is omitted, i.e., if YPBC = 200., no boundary condition is imposed on  $f'(\theta)$ .
- 3) Symmetry about  $\theta = 0$  is automatically imposed.



TURRET

DATA BLOCK I

DESCRIPTION : Mirror location.

FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
EPSM	XM							2F10
1.15	0.							

FIELD

CONTENTS

- 1 EPSM: Distance from fuselage axis to mirror center divided by RFUS.
- 2 XM: x-coordinate of mirror center divided by RFUS.

REMARKS

- 1) Mirror is along fuselage centerline,  $\theta = 0$ .

TURRET

DATA BLOCK    J

DESCRIPTION : Number of angular and radial locations on beam where phase distortion is to be calculated.

FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
NETAI	NRBI							2I10
8	2							

FIELD

CONTENTS

- |   |        |   |
|---|--------|---|
| 1 | NETAI: | Number of angular points at which phase distortion is calculated. |
| 2 | NRBI:  | Number of radial points at which phase distortion is calculated.  |



TURRET

DATA BLOCK    K

DESCRIPTION: Angles around beam at which phase distortion is calculated.

FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
ETA1	ETA2	ETA3	...	...	ETA6	...	...	8F10
0.	45.	90.			225.			

FIELD

CONTENTS

1-8                      ETA1: Angle at which phase distortion is calculated in the laser beam

REMARKS

- 1) If more than eight angular locations are considered, use additional data cards to contain the data.
- 2) Phase distortion is calculated at each combination of angular and radial locations.

TURRET

DATA BLOCK L

DESCRIPTION: Radial locations in beam at which phase distortion is calculated.

FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
RB1	RB2	. . .	R24	. .	. .	. .	. .	8F10
0.025	0.05		0.1					

FIELD

CONTENTS

1-8

RBI: Radial location in laser beam at which phase distortion is calculated.

REMARKS

- 1) If more than eight radial locations are considered, use additional data cards to contain the data.
- 2) Phase distortion is calculated at each combination of angular and radial locations.



TURRET

DATA BLOCK M

DESCRIPTION: Number of separate beam orientations to be analyzed.

FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
NBEAM								I10
10								

FIELD

CONTENTS

1

NBEAM: Number of beam orientations considered in the analysis.

# TURRET

DATA BLOCK    N

DESCRIPTION: Beam orientation information.

## FORMAT AND EXAMPLE

1	2	3	4	5	6	7	8	FORMAT
PHI	GAMMA	AMACHI	WGHT	.				4F10
30.	45.	1.4	1.					

Note: NBEAM cards are required.

### FIELD

### CONTENTS

- |   |         |   |
|---|---------|---|
| 1 | PHI:    | Beam Azimuth angle. Measured from aircraft nose positive to the right. (degrees)  |
| 2 | GAMMA:  | Beam elevation angle. Measured from the horizontal plane, positive upward (degrees)   |
| 3 | AMACHI: | Flight Mach number for this beam orientation. May be different then AMACH read in DATA Block B.    DEFAULT = AMACH.   |
| 4 | WGHT:   | Weighting factor which multiplies the phase distortion for this beam orientation. Measure of relative importance to the design objective.    DEFAULT = 1.0. |

### REMARKS

- 1) If AMACHI is read as zero, it is set equal to AMACH read in DATA Block B.
- 2) If WGHT is read as zero, it is set to 1.0.
- 3) This ends the input data for laser turret analysis.



## V. SAMPLE DATA

Assume the turret shown in Figure 1 is to be analyzed or designed. The initial geometry of the turret, together with the aircraft flight and beam orientation information, is listed here.

### A. GEOMETRY

- $R_0 = 2.5$  meters = fuselage radius.
- $\epsilon = 0.3$  = turret height relative to  $R_0$ .
- $l = 2.0$  = turret half-length relative to  $R_0$ .
- $L = 10.0$  = turret half-spacing for Fourier series approx.
- $\theta_{\max} = 60.0$  degrees = turret half angle.
- $f(x) = 1.0 - 0.6111 lx^2 - 0.18056x^4 - 0.006944 x^6$  = shape function in X.
- $f(\theta) = f(x)$  = shape function in  $\theta$  [initially the same as  $f(x)$ ].

Boundary conditions imposed in this example are that  $f(x) = f'(x) = 0$  at  $x/l = \pm 1.0$ , and  $f(\theta) = f'(\theta) = 0$  at  $\theta/\theta_{\max} = 1.0$ . The boundary condition that  $f(x, \theta) = 0.3 = \epsilon$  at  $x = \theta = 0$  is automatically imposed by the program.

A total of five boundary conditions is imposed on  $f(x)$  so that  $\bar{a}_0 - \bar{a}_4$  are computed by the analysis program and may not be design variables. Six boundary conditions are imposed on  $f(\theta)$  (including symmetry requirements) so that only  $\bar{b}_6$  may be treated as a design variable. The three design variables available for optimization in this example are

<u>Variable</u>	<u>Global Location</u>
$\bar{a}_5$	6
$\bar{a}_6$	7
$\bar{b}_6$	60

Because the aerodynamic analysis is based on small perturbation theory, it is only valid if the slope of the turret in the x-direction is small.

Therefore, constraints are imposed on the design so that the turret shape contained in vector SLOPEX is less than 0.3 in magnitude. That is

$$-0.3 \leq \text{SLOPE}(\bar{I}) \leq 0.3 \quad I = 1, 30$$

SLOPEX is stored in global locations 139 - 168 inclusive.

#### B. AERODYNAMICS

The aircraft is assumed to fly at sea level, and the turret is not pressurized so that

$$\text{DENRTO} = \text{TDENRT} = 1.0$$

The aerodynamic and optical constants are

$$\text{DENGGRAM} = 1.405$$

$$\text{AKPRIM} = 0.00023$$

$$\text{WAVEL} = 3.4 \times 10^{-6} \quad \text{infrared radiation}$$

$$\text{AMACH} = 0.7 \quad \text{nominal Mach number}$$

#### C. MIRROR

The mirror is situated at

$$\text{XM} = 0.0$$

$$\text{EPSM} = 1.15$$

#### D. BEAM ORIENTATIONS

Three orientations are considered as follows:

<u>Beam</u>	<u>Azimuth</u> (PHI)	<u>Elevation</u> (GAMMA)
1	0.	50.
2	45.	30.
3	90.	10.



For brevity only three beam orientations are considered here.  
Typically fifteen orientations are used for optimization.

#### E. PHASE DISTORTION

The phase distortion is calculated at all combinations of two radial and eight angular positions.

$R = 0.05, 0.10$  relative to  $R_0$ .

$\eta = 0, 45, 90, 135, 180, 225, 270, 315$  degrees

Note that since the maximum value of  $R$  is 0.10, this is the assumed radius of the mirror.

#### F. COPES DATA

Based on the above requirements, the COPES data are listed here on a data sheet reproduced from APPENDIX C. These data are for a complete optimization. If only a simple analysis is desired, these data may be run by changing NCALC in DATA BLOCK B to 1 instead of NCALC = 2 given here.





**COPEs DATA CONT.**

**DATA BLOCK G - OMIT IF NDV = 0**

[illegible]

DATA BLOCK H - OMIT IF NDV = 0

+ \$BLOCK H - CONSTRAINTS		COMMENT
NCONS		FORMAT
1		I10

DATA BLOCK I - OMIT IF NDV = 0 OR NCONS = 0

[illegible]

COPES DATA CONT.

**DATA BLOCK I - CONT.**

[illegible]

DATA BLOCK J - OMIT IF NSV = 0

[illegible]

DATA BLOCK K - OMIT IF NSV = 0

[illegible][illegible]



COPES DATA - CONT.

DATA BLOCK K - CONT.

+	\$								COMMENT
	ISENS	NSENS							FORMAT
*									2I10
+	\$								COMMENT
	SNS1	SNS2	SNS3	SNS4	SNS5	SNS6	SNS7	SNS8	FORMAT
*									8F10

+	\$								COMMENT
	ISENS	NSENS							FORMAT
*									2I10
+	\$								COMMENT
	SNS1	SNS2	SNS3	SNS4	SNS5	SNS6	SNS7	SNS8	FORMAT
*									8F10

DATA BLOCK L - OMIT IF N2VAR = 0

+	\$								COMMENT
	N2VX	M2VX	N2VY	M2VY					FORMAT
*									4I10

DATA BLOCK M - OMIT IF N2VAR = 0

+	\$								COMMENT
	NZ1	NZ2	NZ3	NZ4	NZ5	NZ6	NZ7	NZ8	FORMAT
*									8I10

DATA BLOCK N - OMIT IF N2VAR = 0

+	\$								COMMENT
	X1	X2	X3	X4	X5	X6	X7	X8	FORMAT
*									8F10

DATA BLOCK O - OMIT IF N2VAR = 0

+	\$								COMMENT
	Y1	Y2	Y3	Y4	Y5	Y6	Y7	Y8	FORMAT
*									8F10

DATA BLOCK P

+	END								FORMAT
*	END								3A1

# SUMMARY OF COPES DATA

## LISTING OF DATA AS IT APPEARS ON PUNCHED CARDS

COL. → 1      10      20      30      40      50      ....

LASER TURRET OPTIMIZATION AT M = 0.7

\$ BLOCK B - CONTROL PARAMETERS

2

3

\$ BLOCK C - CONMIN INTEGER PARAMETERS

5

\$ BLOCK D - CONMIN REAL PARAMETERS. USE ALL DEFAULTS.

0.

0.

\$ BLOCK E - MINIMIZE PHASE DISTORTION

3

169 -1.0

\$ BLOCK F - DESIGN VARIABLE LIMITS

\$ COEFFICIENT A - 5

-3.0 3.0

\$ COEFFICIENT A - 6

-3.0 3.0

\$ COEFFICIENT B - 6

-3.0 3.0

\$ BLOCK G - DESIGN VARIABLE IDENTIFICATION

\$ COEFFICIENT A - 5

1

6 1.0

\$ COEFFICIENT A - 6

2

7 1.0

\$ COEFFICIENT B - 6

3

60 1.0

\$ BLOCK H - CONSTRAINTS

1

\$ BLOCK I - CONSTRAINT ON SLOPE

139

168

1

\$ LIMITED TO SMALL PERTURBATION THEORY

-0.3

0.3

0.3

0.3

\$ BLOCK P - END OF COPES DATA

END



### G. TURRET ANALYSIS DATA

The data required to analyze the laser turret described above are listed here on a data sheet reproduced from APPENDIX C. Note that the Mach number for each beam orientation (BLOCK N) is read as zero so that all beam orientations will be analyzed at the nominal Mach number of 0.7. If another run is desired at a different Mach number, only AMACH (BLOCK B) need be changed. If certain beam orientations are to be analyzed at different Mach numbers, the appropriate value should be read in BLOCK N.

DATA BLOCK 1

ORIENT	YANG	YBNC	YBNC	YBNC	YBNC	YBNC	YBNC	YBNC	YBNC
0110									

DATA BLOCK 2

ORIENT	YANG	YBNC	YBNC	YBNC	YBNC	YBNC	YBNC	YBNC	YBNC
0110									

DATA BLOCK 3

ORIENT	YANG	YBNC	YBNC	YBNC	YBNC	YBNC	YBNC	YBNC	YBNC
0110									

DATA BLOCK 4

ORIENT	YANG	YBNC	YBNC	YBNC	YBNC	YBNC	YBNC	YBNC	YBNC
0110									

DATA BLOCK 5

ORIENT	YANG	YBNC	YBNC	YBNC	YBNC	YBNC	YBNC	YBNC	YBNC
0110									

DATA BLOCK 6

ORIENT	YANG	YBNC	YBNC	YBNC	YBNC	YBNC	YBNC	YBNC	YBNC
0110									

# LASER TURRET ANALYSIS DATA

## DATA BLOCK A

TITLE	FORMAT
* SUBSONIC LASER TURRET AT SEA LEVEL	20A4

## DATA BLOCK B

AMACH	DENRTO	TDENRT	DENGAM	AKPRIM	WAVEL	FORMAT
* 0.7	1.0	1.0	1.405	.00023	3.4 -6	6F10

## DATA BLOCK C

RFUS	AL	THMAX	ACL	EPS	FORMAT
* 1.5	2.0	60.	5.	0.3	5F10

## DATA BLOCK D

MAXK	MAXP	NXBC	NTHBC	FORMAT
* 6	6	2	1	4I10

## DATA BLOCK E

ABAR0	ABAR1	ABAR2	ABAR3	ABAR4	ABAR5	ABAR6	ABAR7	FORMAT
* 1.0	0.	-0.61111	0.	-0.18056	0.	-0.006944		8F10

## DATA BLOCK F

X	YBC	YPBC	FORMAT
* -1.0	0.	0.	3F10
1.0	0.	0.	

## DATA BLOCK G

BBAR0	BBAR 1	BBAR2	BBAR3	BBAR4	BBAR5	BBAR6	BBAR7	FORMAT
* 1.0	0.	-0.61111	0.	-0.18056	0.	-0.006944		8F10

## DATA BLOCK H

THETA	YBC	YPBC	FORMAT
* 1.0	0.	0.	3F10



**LASER TURRET ANALYSIS DATA - CONT.**

## DATA BLOCK I

	EPSP	CM		FORMAT
*	1.15	0.		2F10

**DATA BLOCK J**

DATA DESIGN 3			
NETAI	NRBI		FORMAT
8	2		2I10

**DATA BLOCK K**

	ETA1	ETA2	ETA3	ETA4	ETA5	ETA6	ETA7	ETA8	FORMAT
*	0.	45.	90.	135.	180.	225.	270.	315.	8F10

**DATA BLOCK L**

	RB1	RB2	RB3	RB4	RB5	RB6	RB7	RB8	FORMAT
*	0.05	0.1							8F10

**DATA BLOCK M**

	NBEAM		FORMAT
*	3		Y10

## DATA BLOCK N

[illegible]

# SUMMARY OF TURRET ANALYSIS DATA

## LISTING OF DATA AS IT APPEARS ON PUNCHED CARDS

COL. → BLOCK	1	10	20	30	40	50	60	....
A	SUBSONIC LASER TURRET AT SEA LEVEL							
B	0.7	1.0	1.0	1.405	0.00023		3.4-6	
C	2.5	2.0	60.0	10.0	0.3			
D		6	6	2	1			
E	1.0	0.	-.61111	0.	-.18056	0.		-.006944
F	-1.0	0.	0.					
F	1.0	0.	0.					
G	1.0	0.	-.61111	0.	-.18056	0.		-.006944
H	1.0	0.	0.					
I	1.15	0.						
J		8	2					
K	0.	45.	90.	135.	180.	225.	270.	
							(end of card)	315.
L	0.05	0.1						
M		3						
N	0.	50.						
N	45.	30.						
N	90.	10.						
A	STOP ← (New COPES Data Title Card To Terminate Program After This Run)							



# VI. SAMPLE OUTPUT

CCCCCCC	0000000	ppppppp	EEEEEEE	SSSSSSS
C	0	P	E	S
C	0	P	E	S
C	0	P	E	S
C	0	P	E	S
C	0	P	E	S
CCCCCCC	0000000	ppppppp	EEEEEEE	SSSSSSS

NASA - AMES  
CONTROL PROGRAM  
FOR  
ENGINEERING SYNTHESIS

TITLE  
LASER TURRET OPTIMIZATION AT  $M = 0.7$

# CARD IMAGES OF CONTROL DATA

CARD

IMAGE

```

1) LASER TURRET OPTIMIZATION AT M = 0.7
2) $ BLOCK B - CONTROL PARAMETERS
3)      2      3
4) $ BLOCK C - COMMEN INTEGER PARAMETERS
5)      5
6) $ BLOCK D - COMMEN REAL PARAMETERS, USE ALL DEFAULTS.
7) 0.
8) 0.
9) $ BLOCK E - MINIMIZE PHASE DISTORTION
10)      3      169 -1.0
11) $ BLOCK F - DESIGN VARIABLE LIMITS
12) $ COEFFICIENT A - 5
13) -3.0      3.0
14) $ COEFFICIENT A - 6
15) -3.0      3.0
16) $ COEFFICIENT A - 6
17) -3.0      3.0
18) $ BLOCK G - DESIGN VARIABLE IDENTIFICATION
19) $ COEFFICIENT A - 5
20)      1      6 1.0
21) $ COEFFICIENT A - 6
22)      2      7 1.0
23) $ COEFFICIENT B - 6
24)      3      60 1.0
25) $ BLOCK H - CONSTRAINTS
26)      1
27) $ BLOCK I - CONSTRAINT ON SLOPE
28)      139      168      1
29) $ LIMITED TO SMALL PERTURBATION THEORY
30) -0.3      0.3      0.3      0.3
31) $ BLOCK P - END OF COPES DATA
32) END

```



# CARD IMAGES OF CONTROL DATA

CARD

IMAGE

```

1) LASER TURRET OPTIMIZATION AT M = 0.7
2) $ BLOCK B - CONTROL PARAMETERS
3)      2      3
4) $ BLOCK C - COMMIN INTEGER PARAMETERS
5)      5
6) $ BLOCK D - COMMIN REAL PARAMETERS, USE ALL DEFAULTS.
7) 0.
8) 0.
9) $ BLOCK E - MINIMIZE PHASE DISTORTION
10)      3      169 -1.0
11) $ BLOCK F - DESIGN VARIABLE LIMITS
12) $ COEFFICIENT A - 5
13) -3.0      3.0
14) $ COEFFICIENT A - 6
15) -3.0      3.0
16) $ COEFFICIENT A - 6
17) -3.0      3.0
18) $ BLOCK G - DESIGN VARIABLE IDENTIFICATION
19) $ COEFFICIENT A - 5
20)      1      6 1.0
21) $ COEFFICIENT A - 6
22)      2      7 1.0
23) $ COEFFICIENT B - 6
24)      3      60 1.0
25) $ BLOCK H - CONSTRAINTS
26)      1
27) $ BLOCK I - CONSTRAINT ON SLOPE
28)      139      168      1
29) $ LIMITED TO SMALL PERTURBATION THEORY
30) -0.3      0.3      0.3      0.3
31) $ BLOCK P - END OF COPE DATA
32) END
    
```

TITLE:  
LASER TURRET OPTIMIZATION AT M = 0.7

CONTROL PARAMETERS:  
 CALCULATION CONTROL, NCALC = 2  
 NUMBER OF GLOBAL DESIGN VARIABLES, NDV = 3  
 NUMBER OF SENSITIVITY VARIABLES, NSV = -0  
 NUMBER OF FUNCTIONS IN TWO-SPACE, N2VAR = -0  
 INPUT INFORMATION PRINT CODE, IPNPUT = -0  
 SENSITIVITY PRINT CODE, IPSENS = -0  
 TWO-SPACE PRINT CODE, IP2VAR = -0  
 DEBUG PRINT CODE, IPOBG = -0

CALCULATION CONTROL, NCALC  
 VALUE MEANING  
 1 SINGLE ANALYSIS  
 2 OPTIMIZATION  
 3 SENSITIVITY  
 4 TWO-VARIABLE FUNCTION SPACE

GLOBAL VARIABLE NUMBER OF OBJECTIVE = 169  
 MULTIPLIER (NEGATIVE INDICATES MINIMIZATION) = -.1000E+01

CONMIN PARAMETERS (IF ZERO, CONMIN DEFAULT WILL OVER-RIDE)

IPRINT	ITMAX	ICNDYR	NSCAL	ITRM	LINOBJ	NACMX1	NFDG
5	-0	-0	-0	-0	-0	5	-0
FDCM		FDCMH		CT		CTMIN	
0.		-0.		-0.		-0.	
CTL		CTLMIN		THETA		PHI	
-0.		-0.		-0.		-0.	
DELFUN		DABFUN					
0.		-0.					

DESIGN VARIABLE INFORMATION  
 NON-ZERO INITIAL VALUE WILL OVER-RIDE MODULE INPUT

D. V.	LOWER	UPPER	INITIAL	SCALE
NO.	BOUND	BOUND	VALUE	
1	-.30000E+01	.30000E+01	-0.	-0.
2	-.30000E+01	.30000E+01	-0.	-0.
3	-.30000E+01	.30000E+01	-0.	-0.

DESIGN VARIABLES

ID	D. V.	GLOBAL	MULTIPLYING
	NO.	VAR. NO.	FACTOR
1	1	6	.10000E+01
2	2	7	.10000E+01
3	3	60	.10000E+01

CONSTRAINT INFORMATION

THERE ARE 1 CONSTRAINT SETS



ID	GLOBAL VAR. 1	GLOBAL VAR. 2	LINEAR ID	LOWER BOUND	NORMALIZATION FACTOR	UPPER BOUND	NORMALIZATION FACTOR
1	139	168	1	-.30000E+00	.30000E+00	.30000E+00	.30000E+00

TOTAL NUMBER OF CONSTRAINED PARAMETERS = 30

#### DATA STORAGE REQUIREMENTS

REAL			INTEGER		
INPUT	EXECUTION	AVAILABLE	INPUT	EXECUTION	AVAILABLE
144	407	5000	103	118	1000

# TURRET ANALYSIS INPUT

TITLE  
SUBSONIC LASER TURRET AT SEA LEVEL

## AERO-OPTICS

MACH NUMBER, AMACH = .700  
EXTERNAL DENSITY RATION, DENRTO = 1.000  
INTERNAL DENSITY RATIO, IDENRT = 1.000  
PRESSURE-DENSITY EXPONENT, DENGAM = 1.405  
PHASE DISTORTION CONSTANT, AKPRIM = .2300E-03  
WAVELENGTH, WAVEL = .3400E-05

## GEOMETRY

FUSELAGE RADIUS, RFUR = 2.500  
TURRET HALF-LENGTH, = 2.000  
TURRET HALF-ANGLE, THMAX = 60.000 DEGREES  
TURRET HEIGHT FACTOR, EPS = .300  
TURRET HALF-SPACING, ACL = 10.000

## TURRET POLYNOMIAL SHAPE COEFFICIENTS

X-DIRECTION, ORDER = 6  
COEFFICIENTS  
.10000E+01 0. -.61111E+00 0. -.18056E+00  
0. -.69440E-02

## BOUNDARY CONDITIONS

X/L	Y	Y-PRIME
0.000	.300	200.000
-1.000	-0.000	-0.000
1.000	-0.000	-0.000

## THETA-DIRECTION, ORDER = 6

COEFFICIENTS  
.10000E+01 0. -.61111E+00 0. -.18056E+00  
0. -.69440E-02

## BOUNDARY CONDITIONS

THETA/THMAX	Y	Y-PRIME
0.000	.300	200.000
1.000	-0.000	-0.000

## LOCATION OF CENTER OF MIRROR

XM = -0.000 EPSM = 1.150

## PHASE DISTORTION CALCULATION POINTS

ANGLES	0.000	45.000	90.000	135.000	180.000
	225.000	270.000	315.000		

## RADII

.050 .100

## BEAM ORIENTATIONS



BEAM	PHI	GAMMA	MACH	WEIGHT
1	0.00	50.00	.700	1.000
2	45.00	30.00	.700	1.000
3	90.00	10.00	.700	1.000

# PHASE DISTORTION CALCULATIONS

BEAM ORIENTATION NUMBER = 1  
 AZMUTH ANGLE = 0.00 DEGREES  
 ELEVATION ANGLE = 50.00 DEGREES  
 MACH NUMBER = .70

R	ETA	X	Y	A	N
0.	0.00	0.	0.	.1920E+00	0.
.5000E-01	0.00	0.	.5000E-01	.1523E+00	.7546E+00
.5000E-01	45.00	.3536E-01	.3536E-01	.1629E+00	.5477E+00
.5000E-01	90.00	.5000E-01	.6675E-09	.1898E+00	.1850E-01
.5000E-01	135.00	.3536E-01	-.3536E-01	.2174E+00	-.5351E+00
.5000E-01	180.00	.1335E-08	-.5000E-01	.2291E+00	-.7745E+00
.5000E-01	225.00	-.3536E-01	-.3536E-01	.2174E+00	-.5351E+00
.5000E-01	270.00	-.5000E-01	-.2002E-08	.1898E+00	.1850E-01
.5000E-01	315.00	-.3536E-01	.3536E-01	.1629E+00	.5477E+00
.1000E+00	0.00	0.	.1000E+00	.1109E+00	.1461E+01
.1000E+00	45.00	.7071E-01	.7071E-01	.1307E+00	.1089E+01
.1000E+00	90.00	.1000E+00	.1335E-08	.1829E+00	.7426E-01
.1000E+00	135.00	.7071E-01	-.7071E-01	.2396E+00	-.1070E+01
.1000E+00	180.00	.2670E-08	-.1000E+00	.2645E+00	-.1581E+01
.1000E+00	225.00	-.7071E-01	-.7071E-01	.2396E+00	-.1070E+01
.1000E+00	270.00	-.1000E+00	-.4005E-08	.1829E+00	.7426E-01
.1000E+00	315.00	-.7071E-01	.7071E-01	.1307E+00	.1089E+01

# ZERNICKE COEFFICIENTS/

AVERAGE = .10883E-02  
 TILT, X = .12854E+00 Y = -.10536E-02  
 FOCUS = .30067E-03  
 ASTIG = -.20378E-02 .12277E-04  
 COMA = -.13262E-03 -.21463E-06 -.48326E-03 .25331E-02

# PHASE DISTORTION CALCULATIONS

BEAM ORIENTATION NUMBER = 2  
 AZMUTH ANGLE = 45.00 DEGREES  
 ELEVATION ANGLE = 30.00 DEGREES  
 MACH NUMBER = .70

R	ETA	X	Y	A	N
0.	0.00	0.	0.	.2532E+00	0.
.5000E-01	0.00	0.	.5000E-01	.1943E+00	.7872E+00
.5000E-01	45.00	.3536E-01	.3536E-01	.2166E+00	.2368E+00
.5000E-01	90.00	.5000E-01	.6675E-09	.2601E+00	-.4696E+00
.5000E-01	135.00	.3536E-01	-.3536E-01	.2977E+00	-.8906E+00
.5000E-01	180.00	.1335E-08	-.5000E-01	.3062E+00	-.7552E+00
.5000E-01	225.00	-.3536E-01	-.3536E-01	.2833E+00	-.1882E+00
.5000E-01	270.00	-.5000E-01	-.2002E-08	.2434E+00	.4563E+00
.5000E-01	315.00	-.3536E-01	.3536E-01	.2073E+00	.8422E+00
.1000E+00	0.00	0.	.1000E+00	.1181E+00	.1800E+01

.1000E+00	45.00	.7071E-01	.7071E-01	-.1668E+00	.6386E+00
.1000E+00	90.00	.1000E+00	.1335E-08	.2640E+00	-.9474E+00
.1000E+00	135.00	.7071E-01	-.7071E-01	.3411E+00	-.1821E+01
.1000E+00	180.00	.2670E-08	-.1000E+00	.3547E+00	-.1486E+01
.1000E+00	225.00	-.7071E-01	-.7071E-01	.3077E+00	-.3232E+00
.1000E+00	270.00	-.1000E+00	-.4005E-08	.2309E+00	.8960E+00
.1000E+00	315.00	-.7071E-01	.7071E-01	.1530E+00	.1759E+01

# ZERNICKE COEFFICIENTS/

AVERAGE = .49776E-02  
 TILT, X = .13398E+00  
 FOCUS = .37912E-02  
 ASTIG = .26635E-02  
 COMA = .85058E-03

Y = -.78917E-01  
 .58705E-02  
 .77995E-04  
 .32978E-02  
 .27808E-02

# PHASE DISTORTION CALCULATIONS

BEAM ORIENTATION NUMBER = 3  
 AZMUTH ANGLE = 90.00 DEGREES  
 ELEVATION ANGLE = 10.00 DEGREES  
 MACH NUMBER = .70

R	ETA	X	Y	A	N
0.	0.00	0.	0.	.3440E+00	0.
.5000E-01	0.00	0.	.5000E-01	.2708E+00	.7185E+00
.5000E-01	45.00	.3536E-01	.3536E-01	.2928E+00	.4686E+00
.5000E-01	90.00	.5000E-01	.6675E-09	.3434E+00	-.1814E-01
.5000E-01	135.00	.3536E-01	-.3536E-01	.3901E+00	-.2612E+00
.5000E-01	180.00	.1335E-08	-.5000E-01	.4066E+00	-.4199E+00
.5000E-01	225.00	-.3536E-01	-.3536E-01	.3901E+00	-.2612E+00
.5000E-01	270.00	-.5000E-01	-.2002E-08	.3434E+00	-.1814E-01
.5000E-01	315.00	-.3536E-01	.3536E-01	.2928E+00	.4686E+00
.1000E+00	0.00	0.	.1000E+00	.1783E+00	.1692E+01
.1000E+00	45.00	.7071E-01	.7071E-01	.2361E+00	.9818E+00
.1000E+00	90.00	.1000E+00	.1335E-08	.3413E+00	-.9792E-01
.1000E+00	135.00	.7071E-01	-.7071E-01	.4249E+00	-.6617E+00
.1000E+00	180.00	.2670E-08	-.1000E+00	.4514E+00	-.8736E+00
.1000E+00	225.00	-.7071E-01	-.7071E-01	.4249E+00	-.6617E+00
.1000E+00	270.00	-.1000E+00	-.4005E-08	.3413E+00	-.9792E-01
.1000E+00	315.00	-.7071E-01	.7071E-01	.2361E+00	.9818E+00

# ZERNICKE COEFFICIENTS/

AVERAGE = .19155E-01  
 TILT, X = .99613E-01  
 FOCUS = .62398E-02  
 ASTIG = .81849E-02  
 COMA = .30317E-02

Y = -.74724E-03  
 -.72295E-04  
 .10055E-04  
 .29599E-02  
 .17965E-02

FLOW FIELD FOR THETA = 0.000 DEGREES

MACH NUMBER = .700



X	R	PHI	U	V	CP
-.4000E+01	.1000E+01	-.4757E-02	-.7515E-02	.1878E-02	.1503E-01
-.3600E+01	.1000E+01	-.7151E-02	-.3386E-02	.6275E-02	.6733E-02
-.3200E+01	.1000E+01	-.7146E-02	.2194E-02	.7948E-03	-.4389E-02
-.2800E+01	.1000E+01	-.7275E-02	-.6934E-02	-.1104E-01	.1375E-01
-.2400E+01	.1000E+01	-.1601E-01	-.4028E-01	-.6405E-02	.8052E-01
-.2000E+01	.1000E+01	-.4092E-01	-.8283E-01	.3960E-01	.1641E+00
-.1600E+01	.1028E+01	-.7512E-01	-.9061E-01	.1169E+00	.1675E+00
-.1200E+01	.1103E+01	-.9141E-01	-.3756E-01	.1593E+00	.4976E-01
-.8000E+00	.1197E+01	-.7710E-01	.3703E-01	.1370E+00	-.9283E-01
-.4000E+00	.1272E+01	-.4235E-01	.9030E-01	.7510E-01	-.1862E+00
.4974E-13	.1300E+01	.5391E-14	.1084E+00	-.9533E-14	-.2168E+00
.4000E+00	.1272E+01	.4235E-01	.9030E-01	-.7510E-01	-.1862E+00
.8000E+00	.1197E+01	.7710E-01	.3703E-01	-.1370E+00	-.9283E-01
.1200E+01	.1103E+01	.9141E-01	-.3756E-01	-.1593E+00	.4976E-01
.1600E+01	.1028E+01	.7512E-01	-.9061E-01	-.1169E+00	.1675E+00
.2000E+01	.1000E+01	.4092E-01	-.8283E-01	-.3960E-01	.1641E+00
.2400E+01	.1000E+01	.1601E-01	-.4028E-01	.6405E-02	.8052E-01
.2800E+01	.1000E+01	.7275E-02	-.6934E-02	.1104E-01	.1375E-01
.3200E+01	.1000E+01	.7146E-02	.2194E-02	-.7948E-03	-.4389E-02
.3600E+01	.1000E+01	.7151E-02	-.3386E-02	-.6275E-02	.6733E-02
.4000E+01	.1000E+01	.4757E-02	-.7515E-02	-.1878E-02	.1503E-01

CRITICAL PRESSURE COEFFICIENT ON SURFACE = 41.76395

SURFACE DEFINITION (EPS = .300)

POLYNOMIAL COEFFICIENTS (A(I), I=0,MAX) IN X-DIRECTION

.10000E+01	0.	-.61110E+00	.11805E+00
0.	-.69440E-02		

POLYNOMIAL COEFFICIENTS (B(I), I=0,MAX) IN THETA-DIRECTION

.10000E+01	0.	-.18321E+01	.84677E+00
0.	-.69440E-02		

COORDINATES

X	Z	Z-PRIME
-2.200	0.0000	0.0000
-2.000	.0000	0.0000
-1.800	.0069	.0700
-1.600	.0278	.1375
-1.400	.0610	.1918
-1.200	.1032	.2263
-1.000	.1500	.2375
-.800	.1966	.2249
-.600	.2385	.1904
-.400	.2716	.1377
-.200	.2927	.0722
.000	.3000	-.0000
.200	.2927	-.0722
.400	.2716	-.1377
.600	.2385	-.1904
.800	.1966	-.2249
1.000	.1500	-.2375
1.200	.1032	-.2263
1.400	.0610	-.1918
1.600	.0278	-.1375
1.800	.0069	-.0700
2.000	.0000	0.0000

2.200	0.0000	0.0000	
THETA			
RADIANS	DEGREES	Z	Z-PRIME
-1.152	-66.0000	0.0000	0.0000
-1.047	-60.0000	.0000	-.0000
-.942	-54.0000	.0107	.1947
-.838	-48.0000	.0387	.3286
-.733	-42.0000	.0777	.4082
-.628	-36.0000	.1225	.4399
-.524	-30.0000	.1684	.4302
-.419	-24.0000	.2114	.3859
-.314	-18.0000	.2482	.3139
-.209	-12.0000	.2764	.2209
-.105	-6.0000	.2940	.1139
.000	.0000	.3000	-.0000
.105	6.0000	.2940	-.1139
.209	12.0000	.2764	-.2209
.314	18.0000	.2482	-.3139
.419	24.0000	.2114	-.3859
.524	30.0000	.1684	-.4302
.628	36.0000	.1225	-.4399
.733	42.0000	.0777	-.4082
.838	48.0000	.0387	-.3286
.942	54.0000	.0107	-.1947
1.047	60.0000	.0000	.0000
1.152	66.0000	0.0000	0.0000

SUM OF SQUARES OF PHASE DISTORTION = .36648E+02



C O N M I N

# FORTRAN PROGRAM FOR

## CONSTRAINED FUNCTION MINIMIZATION

NASA/AMES RESEARCH CENTER, MOFFETT FIELD, CALIF.

VERSION II JULY, 1975

## CONTROL PARAMETERS

```
IPRINT  NOV  ITHAX  NCON  NSIDE  ICNDR  NSCAL  NFDG
      5      3      20      60      1      4      -0      -0
```

LINE	OBJ	TRM	N1	N2	N3	N4	N5
-0		3	5	66	5	5	10

CT	CTMIN	CTL	CTLMIN
-.10000E+00	.40000E-02	-.10000E-01	.10000E-02

THETA	PHI	DELFUN	DARFUN
.10000E+01	.50000E+01	.10000E-03	.36648E-01

FDCH FDCH  
.10000E-01 .10000E-01

1) -.30000E+01 -.30000E+01 -.30000E+01

UPPER BOUNDS ON DECISION VARIABLES (VUB)  
1) .30000E+01 .30000E+01 .30000E+01

ALL CONSTRAINTS ARE LINEAR

## INVOLVED FUNCTION INFORMATION

00-1 3444835-03

1) 0, -0.69440E-02 -0.69440E-02

DECISION VARIABLES (X-VECTOR)  
1) 0 = 694405-02 = 694405-02

1)  $-1.00000E+01$   $-1.00000E+01$   $-1.1598E+01$   $-8.4021E+00$   $-1.3218E+01$   $-$

CONSTRAINT VALUES (G-VECTOR)

1) -1.00000E+01 -1.00000E+01 -1.1598E+01 -1.84021E+00 -1.13218E+01 -1.67816E+00

7)	-.14725E+01	-.52753E+00	-.16011E+01	-.39889E+00	-.17001E+01	-.29944E+00
13)	-.13435E+01	-.31734E+00	-.13910E+01	-.30905E+00	-.13384E+01	-.23038E+00

19)	-.17315E+01	-.26850E+00	-.16495E+01	-.35047E+00	-.15380E+01	-.46200E+00
25)	-.14023E+01	-.59771E+00	-.12487E+01	-.75130E+00	-.10841E+01	-.91586E+00
31)	-.91586E+00	-.10841E+01	-.75130E+00	-.12487E+01	-.59771E+00	-.14023E+01
37)	-.46200E+00	-.15380E+01	-.35047E+00	-.16495E+01	-.26850E+00	-.17315E+01
43)	-.22039E+00	-.17796E+01	-.20905E+00	-.17910E+01	-.23576E+00	-.17642E+01
49)	-.29994E+00	-.17001E+01	-.39889E+00	-.16011E+01	-.52753E+00	-.14725E+01
55)	-.67816E+00	-.13218E+01	-.84021E+00	-.11598E+01	-.10000E+01	-.10000E+01

BEGIN ITERATION NUMBER 1

CT = -.10000E+00 CTL = -.10000E-01 PHI = .50000E+01

THERE ARE 0 ACTIVE CONSTRAINTS

THERE ARE 0 VIOLATED CONSTRAINTS

THERE ARE 0 ACTIVE SIDE CONSTRAINTS

GRADIENT OF OBJ

1) .53485E+03 -.70436E+03 .19787E+02

SEARCH DIRECTION (S-VECTOR)

1) -.75935E+00 .10000E+01 -.28092E-01

ONE-DIMENSIONAL SEARCH

INITIAL SLOPE = -.1111E+04 PROPOSED ALPHA = .3298E-02

\* \* CONSTRAINED ONE-DIMENSIONAL SEARCH INFORMATION \* \* \*

PROPOSED DESIGN

ALPHA = .32985E-02

X-VECTOR

-.2505E-02 -.3646E-02 -.7037E-02

OBJ = .33029E+02

CONSTRAINT VALUES

-.1000E+01	-.1000E+01	-.1219E+01	-.7805E+00	-.1407E+01	-.5927E+00	-.1559E+01	-.4405E+00
-.1674E+01	-.3263E+00	-.1749E+01	-.2510E+00	-.1786E+01	-.2144E+00	-.1785E+01	-.2149E+00
-.1750E+01	-.2503E+00	-.1683E+01	-.3171E+00	-.1589E+01	-.4113E+00	-.1472E+01	-.5283E+00
-.1337E+01	-.6630E+00	-.1190E+01	-.8097E+00	-.1037E+01	-.9629E+00	-.8833E+00	-.1117E+01
-.7347E+00	-.1265E+01	-.5968E+00	-.1403E+01	-.4749E+00	-.1525E+01	-.3738E+00	-.1626E+01
-.2978E+00	-.1702E+01	-.2506E+00	-.1749E+01	-.2347E+00	-.1765E+01	-.2521E+00	-.1748E+01
-.3634E+00	-.1697E+01	-.3881E+00	-.1612E+01	-.5043E+00	-.1496E+01	-.6486E+00	-.1351E+01
-.8161E+00	-.1184E+01	-.1000E+01	-.1000E+01				

TWO-POINT INTERPOLATION

PROPOSED DESIGN

ALPHA = .16492E-01



X-VECTOR  
 -.1252E-01 .9548E-02 -.7407E-02

OBJ = .20655E+02

CONSTRAINT VALUES							
-.1000E+01	-.1000E+01	-.1458E+01	-.5418E+00	-.1749E+01	-.2511E+00	-.1907E+01	-.9258E-01
-.1964E+01	-.3596E-01	-.1945E+01	-.5527E-01	-.1871E+01	-.1287E+00	-.1762E+01	-.2384E+00
-.1630E+01	-.3698E+00	-.1489E+01	-.5115E+00	-.1345E+01	-.6548E+00	-.1206E+01	-.7937E+00
-.1076E+01	-.9241E+00	-.9565E+00	-.1043E+01	-.8490E+00	-.1151E+01	-.7531E+00	-.1247E+01
-.6682E+00	-.1332E+01	-.5930E+00	-.1407E+01	-.5263E+00	-.1474E+01	-.4671E+00	-.1535E+01
-.4152E+00	-.1585E+01	-.3713E+00	-.1629E+01	-.3375E+00	-.1662E+01	-.3177E+00	-.1682E+01
-.3175E+00	-.1683E+01	-.3451E+00	-.1655E+01	-.4115E+00	-.1588E+01	-.5306E+00	-.1469E+01
-.7197E+00	-.1280E+01	-.1000E+01	-.1000E+01				

THREE-POINT INTERPOLATION

PROPOSED DESIGN

ALPHA = .18127E-01

X-VECTOR

-.1376E-01 .1118E-01 -.7453E-02

OBJ = .19478E+02

CONSTRAINT VALUES							
-.1000E+01	-.1000E+01	-.1488E+01	-.5122E+00	-.1791E+01	-.2087E+00	-.1951E+01	-.4948E-01
-.2000E+01	.4145E-13	-.1969E+01	-.3103E-01	-.1882E+01	-.1181E+00	-.1759E+01	-.2413E+00
-.1615E+01	-.3846E+00	-.1464E+01	-.5355E+00	-.1315E+01	-.6850E+00	-.1173E+01	-.8268E+00
-.1044E+01	-.9564E+00	-.9276E+00	-.1072E+01	-.8257E+00	-.1174E+01	-.7370E+00	-.1263E+01
-.6600E+00	-.1340E+01	-.5925E+00	-.1407E+01	-.5326E+00	-.1467E+01	-.4786E+00	-.1521E+01
-.4297E+00	-.1570E+01	-.3863E+00	-.1614E+01	-.3503E+00	-.1650E+01	-.3258E+00	-.1674E+01
-.3192E+00	-.1681E+01	-.3398E+00	-.1660E+01	-.4000E+00	-.1600E+01	-.5160E+00	-.1484E+01
-.7078E+00	-.1292E+01	-.1000E+01	-.1000E+01				

\*\*\* END OF ONE-DIMENSIONAL SEARCH

CALCULATED ALPHA = .18127E-01

OBJ = .194784E+02

DECISION VARIABLES (X-VECTOR)

1) -.13765E-01 .11183E-01 -.74532E-02

CONSTRAINT VALUES (G-VECTOR)

1)	-.10000E+01	-.10000E+01	-.14878E+01	-.51220E+00	-.17913E+01	-.20873E+00
7)	-.19505E+01	-.49484E-01	-.20000E+01	.41448E-13	-.19690E+01	-.31029E-01
13)	-.18819E+01	-.11814E+00	-.17587E+01	-.24133E+00	-.16154E+01	-.38459E+00
19)	-.14645E+01	-.53553E+00	-.13150E+01	-.68499E+00	-.11734E+01	-.82660E+00
25)	-.10436E+01	-.95640E+00	-.92757E+00	-.10724E+01	-.82566E+00	-.11743E-01
31)	-.73701E+00	-.12630E+01	-.65999E+00	-.13400E+01	-.59255E+00	-.14075E+01
37)	-.53264E+00	-.14674E+01	-.47864E+00	-.15214E+01	-.42971E+00	-.15703E+01
43)	-.38626E+00	-.16137E+01	-.35028E+00	-.16497E+01	-.32578E+00	-.16742E+01
49)	-.31919E+00	-.16808E+01	-.33978E+00	-.16602E+01	-.40000E+00	-.16000E+01
55)	-.51597E+00	-.14840E+01	-.70778E+00	-.12922E+01	-.10000E+01	-.10000E+01

BEGIN ITERATION NUMBER 2

CT = -.10000E+00 CTL = -.10000E-01 PHI = .50000E+01

THERE ARE 1 ACTIVE CONSTRAINTS  
CONSTRAINT NUMBERS ARE  
10

THERE ARE 0 VIOLATED CONSTRAINTS

THERE ARE 0 ACTIVE SIDE CONSTRAINTS

GRADIENT OF OBJ

1) .12385E+03 -.58717E+03 .80411E+01

GRADIENTS OF ACTIVE AND VIOLATED CONSTRAINTS

CONSTRAINT NUMBER 10.  
1) -.12342E+02 .12633E+02 0.

PUSH-OFF FACTORS, (THETA(I), I=1,NAC)

1) 0.

CONSTRAINT PARAMETER, BETA = .74996E+00

SEARCH DIRECTION (S-VECTOR)

1) .10000E+01 .97698E+00 -.34940E-01

ONE-DIMENSIONAL SEARCH

INITIAL SLOPE = -.4901E+03 PROPOSED ALPHA = .8655E-02

\* \* CONSTRAINED ONE-DIMENSIONAL SEARCH INFORMATION \* \* \*

PROPOSED DESIGN

ALPHA = .86554E-02

X-VECTOR

-.5109E-02 .1964E-01 -.7756E-02

OBJ = .15172E+02

CONSTRAINT VALUES

-.1000E+01	-.1000E+01	-.1534E+01	-.4663E+00	-.1842E+01	-.1580E+00	-.1982E+01	-.1844E-01
-.2000E+01	.5329E-13	-.1937E+01	-.6340E-01	-.1823E+01	-.1770E+00	-.1684E+01	-.3161E+00
-.1538E+01	-.4621E+00	-.1398E+01	-.6022E+00	-.1272E+01	-.7280E+00	-.1164E+01	-.8357E+00
-.1076E+01	-.9245E+00	-.1004E+01	-.9964E+00	-.9446E+00	-.1055E+01	-.8931E+00	-.1107E+01
-.8433E+00	-.1157E+01	-.7894E+00	-.1211E+01	-.7266E+00	-.1273E+01	-.6514E+00	-.1349E+01
-.5629E+00	-.1437E+01	-.4627E+00	-.1537E+01	-.3565E+00	-.1644E+01	-.2541E+00	-.1746E+01
-.1704E+00	-.1830E+01	-.1261E+00	-.1874E+01	-.1485E+00	-.1851E+01	-.2720E+00	-.1728E+01
-.5389E+00	-.1461E+01	-.1000E+01	-.1000E+01				

TWO-POINT INTERPOLATION

PROPOSED DESIGN

ALPHA = .13765E-01

X-VECTOR



-.3525E-14 .2463E-01 -.7934E-02

OBJ = .13638E+02

CONSTRAINT VALUES

-.1000E+01	-.1000E+01	-.1561E+01	-.4392E+00	-.1872E+01	-.1281E+00	-.2000E+01	-.1179E-03
-.2000E+01	.7698E-13	-.1917E+01	-.8250E-01	-.1788E+01	-.2117E+00	-.1640E+01	-.3602E+00
-.1492E+01	-.5079E+00	-.1359E+01	-.6415E+00	-.1247E+01	-.7534E+00	-.1159E+01	-.8411E+00
-.1094E+01	-.9056E+00	-.1049E+01	-.9515E+00	-.1015E+01	-.9852E+00	-.9852E+00	-.1015E+01
-.9515E+00	-.1049E+01	-.9056E+00	-.1094E+01	-.8411E+00	-.1159E+01	-.7534E+00	-.1247E+01
-.6415E+00	-.1359E+01	-.5079E+00	-.1492E+01	-.3602E+00	-.1640E+01	-.2117E+00	-.1788E+01
-.8250E-01	-.1917E+01	0.	-.2000E+01	-.1179E-03	-.2000E+01	-.1281E+00	-.1872E+01
-.4392E+00	-.1561E+01	-.1000E+01	-.1000E+01				

\* \* \* END OF ONE-DIMENSIONAL SEARCH

CALCULATED ALPHA = .13765E-01

OBJ = .136384E+02

DECISION VARIABLES (X-VECTOR)

1) -.35250E-14 .24631E-01 -.79342E-02

CONSTRAINT VALUES (G-VECTOR)

1)	-.10000E+01	-.10000E+01	-.15608E+01	-.43920E+00	-.18719E+01	-.12806E+00
7)	-.19999E+01	-.11790E-03	-.20000E+01	.76975E-13	-.19175E+01	-.82503E-01
13)	-.17883E+01	-.21172E+00	-.16398E+01	-.36016E+00	-.14921E+01	-.50786E+00
19)	-.13585E+01	-.64148E+00	-.12466E+01	-.75345E+00	-.11589E+01	-.84107E+00
25)	-.10944E+01	-.90561E+00	-.10485E+01	-.95146E+00	-.10148E+01	-.98522E+00
31)	-.98522E+00	-.10148E+01	-.95146E+00	-.10485E+01	-.90561E+00	-.10944E+01
37)	-.84107E+00	-.11589E+01	-.75345E+00	-.12466E+01	-.64148E+00	-.13585E+01
43)	-.50786E+00	-.14921E+01	-.36016E+00	-.16398E+01	-.21172E+00	-.17883E+01
49)	-.82503E-01	-.19175E+01	0.	-.20000E+01	-.11790E-03	-.19999E+01
55)	-.12806E+00	-.18719E+01	-.43920E+00	-.15608E+01	-.10000E+01	-.10000E+01

BEGIN ITERATION NUMBER 3

CT = -.10000E+00 CTL = -.10000E-01 PHI = .50000E+01

THERE ARE 4 ACTIVE CONSTRAINTS  
CONSTRAINT NUMBERS ARE  
8 10 51 53

THERE ARE 0 VIOLATED CONSTRAINTS

THERE ARE 0 ACTIVE SIDE CONSTRAINTS

GRADIENT OF OBJ

1) .39138E+03 .66056E+03 .67398E+01

GRADIENTS OF ACTIVE AND VIOLATED CONSTRAINTS  
CONSTRAINT NUMBER 8

1) -.12733E+02 .16704E+02 0.

CONSTRAINT NUMBER 10

1) -.12342E+02 .12633E+02 0.

CONSTRAINT NUMBER 51  
1) .12342E+02 .12633E+02 0.

CONSTRAINT NUMBER 53  
1) .12733E+02 .16704E+02 0.

PUSH-OFF FACTORS, (THETA(I), I=1,NAC).  
1) 0. 0. 0. 0.

CONSTRAINT PARAMETER, BETA = .11624E+01

SEARCH DIRECTION (S-VECTOR)  
1) -.59251E+00 -.10000E+01 -.10203E-01

ONE-DIMENSIONAL SEARCH  
INITIAL SLOPE = -.8925E+03 PROPOSED ALPHA = .3056E-02

\* \* CONSTRAINED ONE-DIMENSIONAL SEARCH INFORMATION \* \* \*

PROPOSED DESIGN  
ALPHA = .3056E-02  
X-VECTOR  
-.1811E-02 .2157E-01 -.7965E-02

OBJ = .14934E+02

CONSTRAINT VALUES							
-.1000E+01	-.1000E+01	-.1535E+01	-.4652E+00	-.1839E+01	-.1611E+00	-.1972E+01	-.2811E-01
-.1984E+01	-.1624E-01	-.1915E+01	-.8459E-01	-.1800E+01	-.2004E+00	-.1662E+01	-.3384E+00
-.1520E+01	-.4799E+00	-.1388E+01	-.6123E+00	-.1272E+01	-.7280E+00	-.1176E+01	-.8237E+00
-.1100E+01	-.8997E+00	-.1041E+01	-.9592E+00	-.9927E+00	-.1007E+01	-.9497E+00	-.1050E+01
-.9050E+00	-.1095E+01	-.8519E+00	-.1148E+01	-.7850E+00	-.1215E+01	-.7009E+00	-.1299E+01
-.5984E+00	-.1402E+01	-.4801E+00	-.1520E+01	-.3527E+00	-.1647E+01	-.2277E+00	-.1772E+01
-.1225E+00	-.1877E+01	-.6096E-01	-.1939E+01	-.7422E-01	-.1926E+01	-.2015E+00	-.1798E+01
-.4909E+00	-.1509E+01	-.1000E+01	-.1000E+01				

TWO-POINT INTERPOLATION

PROPOSED DESIGN  
ALPHA = .10361E-02  
X-VECTOR  
-.6139E-03 .2359E-01 -.7945E-02

OBJ = .14034E+02

CONSTRAINT VALUES							
-.1000E+01	-.1000E+01	-.1552E+01	-.4480E+00	-.1861E+01	-.1393E+00	-.1990E+01	-.9608E-02
-.1994E+01	-.5512E-02	-.1917E+01	-.8321E-01	-.1792E+01	-.2079E+00	-.1647E+01	-.3528E+00
-.1502E+01	-.4984E+00	-.1368E+01	-.6316E+00	-.1255E+01	-.7448E+00	-.1165E+01	-.8352E+00
-.1096E+01	-.9036E+00	-.1046E+01	-.9541E+00	-.1007E+01	-.9927E+00	-.9732E+00	-.1027E+01
-.9357E+00	-.1004E+01	-.8874E+00	-.1113E+01	-.8221E+00	-.1178E+01	-.7356E+00	-.1264E+01
-.6269E+00	-.1373E+01	-.4985E+00	-.1502E+01	-.3576E+00	-.1642E+01	-.2171E+00	-.1783E+01
-.4680E-01	-.1904E+01	-.2067E-01	-.1979E+01	-.2524E-01	-.1975E+01	-.1530E+00	-.1847E+01



-.4567E+00 -.1543E+01 -.1000E+01 -.1000E+01

### THREE-POINT INTERPOLATION

#### PROPOSED DESIGN

ALPHA = .30615E-03

#### X-VECTOR

-.1814E-03 .2432E-01 -.7937E-02

OBJ = .13750E+02

#### CONSTRAINT VALUES

-.1000E+01	-.1000E+01	-.1558E+01	-.4418E+00	-.1869E+01	-.1314E+00	-.1997E+01	-.2922E-02
-.1998E+01	-.1629E-02	-.1917E+01	-.8271E-01	-.1789E+01	-.2106E+00	-.1642E+01	-.3580E+00
-.1495E+01	-.5051E+00	-.1361E+01	-.6386E+00	-.1249E+01	-.7509E+00	-.1161E+01	-.8393E+00
-.1095E+01	-.9050E+00	-.1048E+01	-.9522E+00	-.1013E+01	-.9874E+00	-.9817E+00	-.1018E+01
-.9468E+00	-.1053E+01	-.9002E+00	-.1100E+01	-.8355E+00	-.1165E+01	-.7482E+00	-.1252E+01
-.6372E+00	-.1363E+01	-.5051E+00	-.1495E+01	-.3594E+00	-.1641E+01	-.2133E+00	-.1787E+01
-.8651E-01	-.1913E+01	-.6107E-02	-.1994E+01	-.7541E-02	-.1992E+01	-.1354E+00	-.1865E+01
-.4444E+00	-.1556E+01	-.1000E+01	-.1000E+01				

\*\*\* END OF ONE-DIMENSIONAL SEARCH

CALCULATED ALPHA = .69389E-17

OBJ = .136384E+02 NO CHANGE ON OBJ

#### DECISION VARIABLES (X-VECTOR)

1) -.35258E-14 .24631E-01 -.79342E-02

#### CONSTRAINT VALUES (G-VECTOR)

1)	-.10000E+01	-.10000E+01	-.15608E+01	-.43920E+00	-.18719E+01	-.12806E+00
7)	-.19999E+01	-.11790E-03	-.20000E+01	.76975E-13	-.19175E+01	-.82503E-01
13)	-.17883E+01	-.21172E+00	-.16398E+01	-.36016E+00	-.14921E+01	-.50786E+00
19)	-.13585E+01	-.64148E+00	-.12466E+01	-.75345E+00	-.11589E+01	-.84107E+00
25)	-.10944E+01	-.90561E+00	-.10485E+01	-.95146E+00	-.10148E+01	-.98522E+00
31)	-.98522E+00	-.10148E+01	-.95146E+00	-.10485E+01	-.90561E+00	-.10944E+01
37)	-.84107E+00	-.11589E+01	-.75345E+00	-.12466E+01	-.64148E+00	-.13585E+01
43)	-.50786E+00	-.14921E+01	-.36016E+00	-.16398E+01	-.21172E+00	-.17883E+01
49)	-.82503E-01	-.19175E+01	0.	-.20000E+01	-.11790E-03	-.19999E+01
55)	-.12806E+00	-.18719E+01	-.43920E+00	-.15608E+01	-.10000E+01	-.10000E+01

BEGIN ITERATION NUMBER 4

CT = -.34200E-01 CTL = -.46416E-02 PHI = .50000E+01

THERE ARE 4 ACTIVE CONSTRAINTS

CONSTRAINT NUMBERS ARE

8 10 51 53

THERE ARE 0 VIOLATED CONSTRAINTS

THERE ARE 0 ACTIVE SIDE CONSTRAINTS

#### GRADIENT OF OBJ

1) .39138E+03 .66056E+03 .67398E+01

# GRADIENTS OF ACTIVE AND VIOLATED CONSTRAINTS

CONSTRAINT NUMBER 8  
1) -.60623E+00 .79529E+00 0.

CONSTRAINT NUMBER 10  
1) -.69883E+00 .71529E+00 0.

CONSTRAINT NUMBER 51  
1) .69883E+00 .71529E+00 0.

CONSTRAINT NUMBER 53  
1) .60623E+00 .79529E+00 0.

PUSH-OFF FACTORS, (TWETA(I), I=1, NAC)  
1) 0. 0. 0. 0.

CONSTRAINT PARAMETER, BETA = .11624E+01

SEARCH DIRECTION (S-VECTOR)  
1) -.59251E+00 -.10000E+01 -.10203E+01

ONE-DIMENSIONAL SEARCH  
INITIAL SLOPE = -.8925E+03 PROPOSED ALPHA = .2291E-02

## \* \* CONSTRAINED ONE-DIMENSIONAL SEARCH INFORMATION \* \* \*

PROPOSED DESIGN  
ALPHA = .22907E-02  
X-VECTOR  
-.1357E-02 .2234E-01 -.7958E-02  
OBJ = .14579E+02

CONSTRAINT VALUES							
-.1000E+01	-.1000E+01	-.1541E+01	-.4587E+00	-.1847E+01	-.1528E+00	-.1979E+01	-.2110E-01
-.1988E+01	-.1219E-01	-.1916E+01	-.8407E-01	-.1797E+01	-.2032E+00	-.1656E+01	-.3438E+00
-.1513E+01	-.4869E+00	-.1380E+01	-.6196E+00	-.1266E+01	-.7344E+00	-.1172E+01	-.8281E+00
-.1099E+01	-.9012E+00	-.1043E+01	-.9573E+00	-.9983E+00	-.1002E+01	-.9586E+00	-.1041E+01
-.9166E+00	-.1083E+01	-.8653E+00	-.1135E+01	-.7991E+00	-.1201E+01	-.7140E+00	-.1286E+01
-.6092E+00	-.1391E+01	-.4871E+00	-.1513E+01	-.3546E+00	-.1645E+01	-.2237E+00	-.1776E+01
-.1125E+00	-.1888E+01	-.4569E-01	-.1954E+01	-.5566E-01	-.1944E+01	-.1831E+00	-.1817E+01
-.4779E+00	-.1522E+01	-.1000E+01	-.1000E+01				

## TWO-POINT INTERPOLATION

PROPOSED DESIGN  
ALPHA = .78455E-03  
X-VECTOR  
-.4649E-03 .2385E-01 -.7942E-02  
OBJ = .13977E+02



CONSTRAINT VALUES

-.1000E+01	-.1000E+01	-.1554E+01	-.4459E+00	-.1863E+01	-.1365E+00	-.1993E+01	-.7304E-02
-.1996E+01	-.4174E-02	-.1917E+01	-.8304E-01	-.1791E+01	-.2088E+00	-.1645E+01	-.3546E+00
-.1499E+01	-.5007E+00	-.1366E+01	-.6340E+00	-.1253E+01	-.7469E+00	-.1163E+01	-.8366E+00
-.1096E+01	-.9041E+00	-.1047E+01	-.9535E+00	-.1009E+01	-.9909E+00	-.9761E+00	-.1024E+01
-.9395E+00	-.1060E+01	-.8918E+00	-.1108E+01	-.8267E+00	-.1173E+01	-.7399E+00	-.1260E+01
-.6304E+00	-.1370E+01	-.5007E+00	-.1499E+01	-.3582E+00	-.1642E+01	-.2158E+00	-.1784E+01
-.9277E-01	-.1907E+01	-.1565E-01	-.1984E+01	-.1914E-01	-.1981E+01	-.1469E+00	-.1853E+01
-.4525E+00	-.1548E+01	-.1000E+01	-.1000E+01				

### THREE-POINT INTERPOLATION

PROPOSED DESIGN  
 ALPHA = .22044E-03  
 X-VECTOR  
 -.1306E-03 .2441E-01 -.7936E-02  
 OBJ = .13716E+02

CONSTRAINT VALUES

-.1000E+01	-.1000E+01	-.1559E+01	-.4411E+00	-.1870E+01	-.1304E+00	-.1998E+01	-.2137E-02
-.1999E+01	-.1173E-02	-.1917E+01	-.8265E-01	-.1789E+01	-.2109E+00	-.1641E+01	-.3586E+00
-.1494E+01	-.5058E+00	-.1361E+01	-.6394E+00	-.1248E+01	-.7516E+00	-.1160E+01	-.8398E+00
-.1095E+01	-.9052E+00	-.1048E+01	-.9520E+00	-.1013E+01	-.9868E+00	-.9827E+00	-.1017E+01
-.9481E+00	-.1052E+01	-.9017E+00	-.1098E+01	-.8370E+00	-.1163E+01	-.7497E+00	-.1250E+01
-.6384E+00	-.1362E+01	-.5059E+00	-.1494E+01	-.3596E+00	-.1640E+01	-.2129E+00	-.1787E+01
-.8539E-01	-.1915E+01	-.4397E-02	-.1996E+01	-.5463E-02	-.1995E+01	-.1334E+00	-.1867E+01
-.8429E+00	-.1557E+01	-.1000E+01	-.1000E+01				

\*\*\* END OF ONE-DIMENSIONAL SEARCH

CALCULATED ALPHA = .43368E-17

OBJ = .136384E+02 NO CHANGE ON OBJ

DECISION VARIABLES (X-VECTOR)  
 1) -.35267E-14 .24631E-01 -.79342E-02

CONSTRAINT VALUES (G-VECTOR)

1)	-.10000E+01	-.10000E+01	-.15608E+01	-.43920E+00	-.18719E+01	-.12806E+00
7)	-.19999E+01	-.11790E-03	-.20000E+01	.76975E-13	-.19175E+01	-.82503E-01
13)	-.17883E+01	-.21172E+00	-.16398E+01	-.36016E+00	-.14921E+01	-.50786E+00
19)	-.13585E+01	-.64148E+00	-.12466E+01	-.75345E+00	-.11589E+01	-.84107E+00
25)	-.10944E+01	-.90561E+00	-.10485E+01	-.95146E+00	-.10148E+01	-.98522E+00
31)	-.98522E+00	-.10148E+01	-.95146E+00	-.10485E+01	-.90561E+00	-.10944E+01
37)	-.84107E+00	-.11589E+01	-.75345E+00	-.12466E+01	-.64148E+00	-.13585E+01
43)	-.50786E+00	-.14921E+01	-.36016E+00	-.16398E+01	-.21172E+00	-.17883E+01
49)	-.82503E-01	-.19175E+01	0.	-.20000E+01	-.11790E-03	-.19999E+01
55)	-.12806E+00	-.18719E+01	-.43920E+00	-.15608E+01	-.10000E+01	-.10000E+01

BEGIN ITERATION NUMBER 5

CT = -.11696E-01 CTL = -.21544E-02 PHI = .50000E+01

THERE ARE 0 ACTIVE CONSTRAINTS  
 CONSTRAINT NUMBERS ARE

8 10 51 53

THERE ARE 0 VIOLATED CONSTRAINTS

THERE ARE 0 ACTIVE SIDE CONSTRAINTS

GRADIENT OF OBJ

1) .39138E+03 .06056E+03 .67398E+01

GRADIENTS OF ACTIVE AND VIOLATED CONSTRAINTS

CONSTRAINT NUMBER 8

1) -.12733E+02 .16704E+02 -.11842E-11

CONSTRAINT NUMBER 10

1) -.12342E+02 .12633E+02 0.

CONSTRAINT NUMBER 51

1) .12342E+02 .12633E+02 -.11842E-11

CONSTRAINT NUMBER 53

1) .12733E+02 .16704E+02 -.29606E-11

PUSH-OFF FACTORS, (THETA(I), I=1,NAC)

1) 0. 0. 0. 0.

CONSTRAINT PARAMETER, BETA = .11624E+01

SEARCH DIRECTION (S-VECTOR)

1) -.59251E+00 -.10000E+01 -.10203E-01

ONE-DIMENSIONAL SEARCH

INITIAL SLOPE = -.8925E+03 PROPOSED ALPHA = .1528E-05

\* \* CONSTRAINED ONE-DIMENSIONAL SEARCH INFORMATION \* \* \*

PROPOSED DESIGN

ALPHA = .15281E-05

X-VECTOR

-.9054E-06 .2463E-01 -.7934E-02

OBJ = .13639E+02

CONSTRAINT VALUES

-.1000E+01	-.1000E+01	-.1561E+01	-.4392E+00	-.1872E+01	-.1281E+00	-.2000E+01	-.1319E-05
-.2000E+01	-.8130E-05	-.1917E+01	-.8250E-01	-.1788E+01	-.2117E+00	-.1640E+01	-.3602E+00
-.1492E+01	-.5078E+00	-.1359E+01	-.6415E+00	-.1247E+01	-.7534E+00	-.1159E+01	-.8411E+00
-.1094E+01	-.9056E+00	-.1049E+01	-.9515E+00	-.1015E+01	-.9852E+00	-.9852E+00	-.1015E+01
-.9514E+00	-.1049E+01	-.9056E+00	-.1094E+01	-.8410E+00	-.1159E+01	-.7534E+00	-.1247E+01
-.6415E+00	-.1359E+01	-.5078E+00	-.1492E+01	-.3602E+00	-.1640E+01	-.2117E+00	-.1788E+01
-.8250E-01	-.1917E+01	-.3048E-04	-.2000E+01	-.1550E-03	-.2000E+01	-.1281E+00	-.1872E+01
-.4392E+00	-.1561E+01	-.1000E+01	-.1000E+01				

TWO-POINT INTERPOLATION



PROPOSED DESIGN  
 ALPHA = .53016E-06  
 X-VECTOR  
 -.3141E-06 .2463E-01 -.7934E-02

OBJ = .13639E+02

CONSTRAINT VALUES

-.1000E+01	-.1000E+01	-.1561E+01	-.4392E+00	-.1872E+01	-.1281E+00	-.2000E+01	-.1228E-03
-.2000E+01	-.2821E-05	-.1917E+01	-.8250E-01	-.1788E+01	-.2117E+00	-.1640E+01	-.3602E+00
-.1492E+01	-.5079E+00	-.1359E+01	-.6415E+00	-.1247E+01	-.7534E+00	-.1159E+01	-.8411E+00
-.1094E+01	-.9056E+00	-.1049E+01	-.9515E+00	-.1015E+01	-.9852E+00	-.9852E+00	-.1015E+01
-.9515E+00	-.1049E+01	-.9056E+00	-.1094E+01	-.8411E+00	-.1159E+01	-.7534E+00	-.1247E+01
-.6415E+00	-.1359E+01	-.5079E+00	-.1492E+01	-.3602E+00	-.1640E+01	-.2117E+00	-.1788E+01
-.8251E-01	-.1917E+01	-.1057E-04	-.2000E+01	-.1308E-03	-.2000E+01	-.1281E+00	-.1872E+01
-.4392E+00	-.1561E+01	-.1000E+01	-.1000E+01				

### THREE-POINT INTERPOLATION

PROPOSED DESIGN  
 ALPHA = .15380E-06  
 X-VECTOR  
 -.9113E-07 .2463E-01 -.7934E-02

OBJ = .13638E+02

CONSTRAINT VALUES

-.1000E+01	-.1000E+01	-.1561E+01	-.4392E+00	-.1872E+01	-.1281E+00	-.2000E+01	-.1193E-03
-.2000E+01	-.8183E-06	-.1917E+01	-.8250E-01	-.1788E+01	-.2117E+00	-.1640E+01	-.3602E+00
-.1492E+01	-.5079E+00	-.1359E+01	-.6415E+00	-.1247E+01	-.7534E+00	-.1159E+01	-.8411E+00
-.1094E+01	-.9056E+00	-.1049E+01	-.9515E+00	-.1015E+01	-.9852E+00	-.9852E+00	-.1015E+01
-.9515E+00	-.1049E+01	-.9056E+00	-.1094E+01	-.8411E+00	-.1159E+01	-.7534E+00	-.1247E+01
-.6415E+00	-.1359E+01	-.5079E+00	-.1492E+01	-.3602E+00	-.1640E+01	-.2117E+00	-.1788E+01
-.8250E-01	-.1917E+01	-.3068E-05	-.2000E+01	-.1216E-03	-.2000E+01	-.1281E+00	-.1872E+01
-.4392E+00	-.1561E+01	-.1000E+01	-.1000E+01				

\*\*\* END OF ONE-DIMENSIONAL SEARCH

CALCULATED ALPHA = .25411E-20

OBJ = .136384E+02 NO CHANGE ON OBJ

DECISION VARIABLES (X-VECTOR)  
 1) -.35267E-14 .24631E-01 -.79342E-02

CONSTRAINT VALUES (G-VECTOR)

1)	-.10000E+01	-.10000E+01	-.15608E+01	-.43920E+00	-.18719E+01	-.12806E+00
7)	-.19999E+01	-.11790E-03	-.20000E+01	.76975E-13	-.19175E+01	-.82503E-01
13)	-.17883E+01	-.21172E+00	-.16398E+01	-.36016E+00	-.14921E+01	-.50786E+00
19)	-.13585E+01	-.64148E+00	-.12466E+01	-.75345E+00	-.11589E+01	-.84107E+00
25)	-.10944E+01	-.90561E+00	-.10485E+01	-.95146E+00	-.10148E+01	-.98522E+00
31)	-.98522E+00	-.10148E+01	-.95146E+00	-.10485E+01	-.90561E+00	-.10944E+01
37)	-.84107E+00	-.11589E+01	-.75345E+00	-.12466E+01	-.64148E+00	-.13585E+01
43)	-.50786E+00	-.14921E+01	-.36016E+00	-.16398E+01	-.21172E+00	-.17883E+01
49)	-.82503E-01	-.19175E+01	0	-.20000E+01	-.11790E-03	-.19999E+01
55)	-.12806E+00	-.18719E+01	-.43920E+00	-.15608E+01	-.10000E+01	-.10000E+01

# FINAL OPTIMIZATION INFORMATION

OBJ = .136384E+02

## DECISION VARIABLES (X-VECTOR)

1) -.35267E-14 .24631E-01 -.79342E-02

## CONSTRAINT VALUES (G-VECTOR)

1)	-.10000E+01	-.10000E+01	-.15608E+01	-.43920E+00	-.18719E+01	-.12806E+00
7)	-.19999E+01	-.11790E-03	-.20000E+01	.76975E-13	-.19175E+01	-.82503E-01
13)	-.17883E+01	-.21172E+00	-.16398E+01	-.36016E+00	-.14921E+01	-.50786E+00
19)	-.13585E+01	-.64148E+00	-.12466E+01	-.75345E+00	-.11589E+01	-.84107E+00
25)	-.10944E+01	-.90561E+00	-.10485E+01	-.95146E+00	-.10148E+01	-.98522E+00
31)	-.98522E+00	-.10148E+01	-.95146E+00	-.10485E+01	-.90561E+00	-.10944E+01
37)	-.84107E+00	-.11589E+01	-.75345E+00	-.12466E+01	-.64148E+00	-.13585E+01
43)	-.50786E+00	-.14921E+01	-.36016E+00	-.16398E+01	-.21172E+00	-.17883E+01
49)	-.82503E-01	-.19175E+01	0.	-.20000E+01	-.11790E-03	-.19999E+01
55)	-.12806E+00	-.18719E+01	-.43920E+00	-.15608E+01	-.10000E+01	-.10000E+01

THERE ARE 4 ACTIVE CONSTRAINTS

CONSTRAINT NUMBERS ARE

8 10 51 53

THERE ARE 0 VIOLATED CONSTRAINTS

THERE ARE 0 ACTIVE SIDE CONSTRAINTS

## TERMINATION CRITERION

ABS(1-OBJ(I-1)/OBJ(I)) LESS THAN DELFUN FOR 3 ITERATIONS

ABS(OBJ(I)-OBJ(I-1)) LESS THAN DABFUN FOR 3 ITERATIONS

NUMBER OF ITERATIONS = 5

OBJECTIVE FUNCTION WAS EVALUATED 27 TIMES

CONSTRAINT FUNCTIONS WERE EVALUATED 27 TIMES

## PHASE DISTORTION CALCULATIONS

BEAM ORIENTATION NUMBER = 1

AZMUTH ANGLE = 0.00 DEGREES

ELEVATION ANGLE = 50.00 DEGREES

MACH NUMBER = .70

R	ETA	X	Y	A	N
0.	0.00	0.	0.	.1951E+00	0.
.5000E-01	0.00	0.	.5000E-01	.1536E+00	.4060E+00
.5000E-01	45.00	.3536E-01	.3536E-01	.1646E+00	.2946E+00
.5000E-01	90.00	.5000E-01	.6675E-09	.1928E+00	.1179E-01
.5000E-01	135.00	.3536E-01	-.3536E-01	.2230E+00	-.2882E+00
.5000E-01	180.00	.1335E-08	-.5000E-01	.2361E+00	-.4180E+00
.5000E-01	225.00	-.3536E-01	-.3536E-01	.2230E+00	-.2882E+00
.5000E-01	270.00	-.5000E-01	-.2002E-08	.1928E+00	.1179E-01
.5000E-01	315.00	-.3536E-01	.3536E-01	.1646E+00	.2946E+00
.1000E+00	0.00	0.	.1000E+00	.1117E+00	.8038E+00



.1000E+00	45.00	.7071E-01	.7071E-01	.1316E+00	.5981E+00
.1000E+00	90.00	.1000E+00	.1335E-08	.1858E+00	.4740E-01
.1000E+00	135.00	.7071E-01	-.7071E-01	.2483E+00	-.5666E+00
.1000E+00	180.00	.2670E-08	-.1000E+00	.2766E+00	-.8476E+00
.1000E+00	225.00	-.7071E-01	-.7071E-01	.2483E+00	-.5666E+00
.1000E+00	270.00	-.1000E+00	-.4005E-08	.1858E+00	.4740E-01
.1000E+00	315.00	-.7071E-01	.7071E-01	.1316E+00	.5981E+00

#### ZERNICKE COEFFICIENTS/

AVERAGE = .13222E-02  
 TILT, X = .69462E-01  
 FOCUS = .74027E-03  
 ASTIG = -.10778E-02  
 COMA = .42196E-04

Y = -.56737E-03  
 .76826E-05  
 -.20757E-07  
 -.16505E-03  
 .13640E-02

#### PHASE DISTORTION CALCULATIONS

BEAM ORIENTATION NUMBER = 2  
 AZMUTH ANGLE = 45.00 DEGREES  
 ELEVATION ANGLE = 30.00 DEGREES  
 MACH NUMBER = .70

R	ETA	X	Y	A	N
0.	0.00	0.	0.	.2591E+00	0.
.5000E-01	0.00	0.	.5000E-01	.1973E+00	.4980E+00
.5000E-01	45.00	.3536E-01	.3536E-01	.2219E+00	.2349E+00
.5000E-01	90.00	.5000E-01	.6675E-09	.2695E+00	-.1833E+00
.5000E-01	135.00	.3536E-01	-.3536E-01	.3090E+00	-.4811E+00
.5000E-01	180.00	.1335E-08	-.5000E-01	.3157E+00	-.4716E+00
.5000E-01	225.00	-.3536E-01	-.3536E-01	.2894E+00	-.1923E+00
.5000E-01	270.00	-.5000E-01	-.2002E-08	.2466E+00	.1797E+00
.5000E-01	315.00	-.3536E-01	.3536E-01	.2093E+00	.4600E+00
.1000E+00	0.00	0.	.1000E+00	.1195E+00	.1151E+01
.1000E+00	45.00	.7071E-01	.7071E-01	.1724E+00	.5715E+00
.1000E+00	90.00	.1000E+00	.1335E-08	.2777E+00	-.3700E+00
.1000E+00	135.00	.7071E-01	-.7071E-01	.3590E+00	-.9916E+00
.1000E+00	180.00	.2670E-08	-.1000E+00	.3676E+00	-.9181E+00
.1000E+00	225.00	-.7071E-01	-.7071E-01	.3134E+00	-.3451E+00
.1000E+00	270.00	-.1000E+00	-.4005E-08	.2323E+00	.3574E+00
.1000E+00	315.00	-.7071E-01	.7071E-01	.1538E+00	.9660E+00

#### ZERNICKE COEFFICIENTS/

AVERAGE = .43769E-02  
 TILT, X = .84736E-01  
 FOCUS = .29497E-02  
 ASTIG = .17674E-02  
 COMA = .38912E-03

Y = -.31425E-01  
 .39176E-02  
 .92258E-04  
 .17111E-02  
 .16210E-02

#### PHASE DISTORTION CALCULATIONS

BEAM ORIENTATION NUMBER = 3  
 AZMUTH ANGLE = 90.00 DEGREES  
 ELEVATION ANGLE = 10.00 DEGREES

MACH NUMBER	R	ETA	X	Y	A	N
0.	0.00	0.00	0.	0.	.3440E+00	0.
.5000E-01	0.00	0.	0.	.5000E-01	.2707E+00	.5017E+00
.5000E-01	45.00	.3536E-01	.3536E-01	.2931E+00	.3247E+00	
.5000E-01	90.00	.5000E-01	.6675E-09	.3439E+00	-.1160E-02	
.5000E-01	135.00	.3536E-01	-.3536E-01	.3903E+00	-.1608E+00	
.5000E-01	180.00	.1335E-08	-.5000E-01	.4066E+00	-.2919E+00	
.5000E-01	225.00	-.3536E-01	-.3536E-01	.3903E+00	-.1608E+00	
.5000E-01	270.00	-.5000E-01	-.2002E-08	.3439E+00	-.1160E-02	
.5000E-01	315.00	-.3536E-01	.3536E-01	.2931E+00	.3247E+00	
.1000E+00	0.00	0.	.1000E+00	.1783E+00	.1295E+01	
.1000E+00	45.00	.7071E-01	.7071E-01	.2374E+00	.8055E+00	
.1000E+00	90.00	.1000E+00	.1335E-08	.3435E+00	-.4875E-02	
.1000E+00	135.00	.7071E-01	-.7071E-01	.4256E+00	-.4397E+00	
.1000E+00	180.00	.2670E-08	-.1000E+00	.4514E+00	-.6368E+00	
.1000E+00	225.00	-.7071E-01	-.7071E-01	.4256E+00	-.4397E+00	
.1000E+00	270.00	-.1000E+00	-.4005E-08	.3435E+00	-.4875E-02	
.1000E+00	315.00	-.7071E-01	.7071E-01	.2374E+00	.8055E+00	

# ZERNICKE COEFFICIENTS/

AVERAGE = .18672E-01  
TILT, X = .73271E-01  
FOCUS = .78043E-02  
ASTIG = .53594E-02  
COMA = .23076E-02

Y = -.50955E-03  
.45822E-04  
.10126E-04  
.41076E-02  
.12251E-02

# FLOW FIELD FOR THETA = 0.000 DEGREES

MACH NUMBER	R	PHI	U	V	CP
.4000E+01	.1000E+01	-.1254E-01	-.1820E-01	.2255E-01	.3588E-01
.3600E+01	.1000E+01	-.1319E-01	.1550E-01	.2079E-01	-.3144E-01
.3200E+01	.1000E+01	-.3774E-02	.2254E-01	-.1946E-01	-.4547E-01
.2800E+01	.1000E+01	-.3857E-02	-.3198E-01	-.4196E-01	.6221E-01
.2400E+01	.1000E+01	-.3242E-01	-.1075E+00	.1389E-01	.2147E+00
.2000E+01	.1000E+01	-.8165E-01	-.1236E+00	.1328E+00	.2295E+00
.1600E+01	.1078E+01	-.1032E+00	-.4248E-01	.1824E+00	.5167E-01
.1200E+01	.1193E+01	-.8471E-01	.3381E-01	.1370E+00	-.8639E-01
.8000E+00	.1265E+01	-.5571E-01	.6501E-01	.7915E-01	-.1363E+00
.4000E+00	.1294E+01	-.2681E-01	.6833E-01	.3319E-01	-.1378E+00
.4974E-13	.1300E+01	.7817E-14	.6565E-01	-.1226E-13	-.1313E+00
.4000E+00	.1294E+01	.2681E-01	.6833E-01	-.3319E-01	-.1378E+00
.8000E+00	.1265E+01	.5571E-01	.6501E-01	-.7915E-01	-.1363E+00
.1200E+01	.1193E+01	.8471E-01	.3381E-01	.1370E+00	-.8639E-01
.1600E+01	.1078E+01	.1032E+00	-.4248E-01	-.1824E+00	.5167E-01
.2000E+01	.1000E+01	.8165E-01	-.1236E+00	-.1328E+00	.2295E+00
.2400E+01	.1000E+01	.3242E-01	-.1075E+00	-.1389E-01	.2147E+00
.2800E+01	.1000E+01	.3857E-02	-.3198E-01	.4196E-01	.6221E-01
.3200E+01	.1000E+01	.3774E-02	.2254E-01	.1946E-01	-.4547E-01
.3600E+01	.1000E+01	.1319E-01	.1550E-01	-.2079E-01	-.3144E-01
.4000E+01	.1000E+01	.1254E-01	-.1820E-01	-.2255E-01	.3588E-01

CRITICAL PRESSURE COEFFICIENT ON SURFACE = 41.76395



SURFACE DEFINITION (EPS = .300)  
 POLYNOMIAL COEFFICIENTS (A(I), I=0,MAXK) IN X-DIRECTION  
 .10000E+01 -.56843E-13 -.10591E+00 .28422E-13 -.13454E+00  
 -.35267E-14 .24631E-01

POLYNOMIAL COEFFICIENTS (B(I), I=0,MAXP) IN THETA-DIRECTION  
 .10000E+01 0. -.16333E+01 0. .84895E+00  
 0. -.79342E-02

# COORDINATES

X	Z	Z-PRIME
-2.200	0.0000	0.0000
-2.000	.0000	-.0000
-1.800	.0247	.2182
-1.600	.0781	.2981
-1.400	.1383	.2935
-1.200	.1926	.2449
-1.000	.2353	.1807
-.800	.2651	.1190
-.600	.2837	.0696
-.400	.2939	.0353
-.200	.2987	.0140
.000	.3000	-.0000
.200	.2987	-.0140
.400	.2939	-.0353
.600	.2837	-.0696
.800	.2651	-.1190
1.000	.2353	-.1807
1.200	.1926	-.2449
1.400	.1383	-.2935
1.600	.0781	-.2981
1.800	.0247	-.2182
2.000	.0000	.0000
2.200	0.0000	0.0000

THETA		Z	Z-PRIME
RADIANS	DEGREES		
-1.152	-66.0000	0.0000	0.0000
-1.047	-60.0000	.0000	-.0000
-.942	-54.0000	.0107	.1945
-.838	-48.0000	.0386	.3284
-.733	-42.0000	.0776	.4081
-.628	-36.0000	.1224	.4398
-.524	-30.0000	.1683	.4303
-.419	-24.0000	.2113	.3861
-.314	-18.0000	.2482	.3140
-.209	-12.0000	.2764	.2210
-.105	-6.0000	.2940	.1140
.000	.0000	.3000	-.0000
.105	6.0000	.2940	-.1140
.209	12.0000	.2764	-.2210
.314	18.0000	.2482	-.3140
.419	24.0000	.2113	-.3861
.524	30.0000	.1683	-.4303
.628	36.0000	.1224	-.4398
.733	42.0000	.0776	-.4081
.838	48.0000	.0386	-.3284

.942	54.0000	.0107	-.1945
1.047	60.0000	.0000	.0000
1.152	66.0000	0.0000	0.0000

SUM OF SQUARES OF PHASE DISTORTION = .13638E+02

WAVELENGTH	PHASE	AMPLITUDE	PHASE
0000.0	0000.0	000.0	000.0
0000.1	0000.1	000.1	000.1
0000.2	0000.2	000.2	000.2
0000.3	0000.3	000.3	000.3
0000.4	0000.4	000.4	000.4
0000.5	0000.5	000.5	000.5
0000.6	0000.6	000.6	000.6
0000.7	0000.7	000.7	000.7
0000.8	0000.8	000.8	000.8
0000.9	0000.9	000.9	000.9
0001.0	0001.0	001.0	001.0
0001.1	0001.1	001.1	001.1
0001.2	0001.2	001.2	001.2
0001.3	0001.3	001.3	001.3
0001.4	0001.4	001.4	001.4
0001.5	0001.5	001.5	001.5
0001.6	0001.6	001.6	001.6
0001.7	0001.7	001.7	001.7
0001.8	0001.8	001.8	001.8
0001.9	0001.9	001.9	001.9
0002.0	0002.0	002.0	002.0
0002.1	0002.1	002.1	002.1

WAVELENGTH	PHASE	AMPLITUDE	PHASE
0002.2	0002.2	002.2	002.2
0002.3	0002.3	002.3	002.3
0002.4	0002.4	002.4	002.4
0002.5	0002.5	002.5	002.5
0002.6	0002.6	002.6	002.6
0002.7	0002.7	002.7	002.7
0002.8	0002.8	002.8	002.8
0002.9	0002.9	002.9	002.9
0003.0	0003.0	003.0	003.0
0003.1	0003.1	003.1	003.1
0003.2	0003.2	003.2	003.2
0003.3	0003.3	003.3	003.3
0003.4	0003.4	003.4	003.4
0003.5	0003.5	003.5	003.5
0003.6	0003.6	003.6	003.6
0003.7	0003.7	003.7	003.7
0003.8	0003.8	003.8	003.8
0003.9	0003.9	003.9	003.9
0004.0	0004.0	004.0	004.0



#### LIST OF REFERENCES

1. Garret N. Vanderplaats, "CONMIN - A FORTRAN Program for Constrained Function Minimization," NASA TM X-62, 282, August 1973.
2. A. E. Fuhs and S. E. Fuhs, "Phase Distortion Due to Airflow Over a Hemispherical Laser Turret," Naval Postgraduate School Report, NPS-69Fu 76101, Sept. 1976.
3. A. E. Fuhs and S. E. Fuhs, "Phase Distortion at High Subsonic Mach Numbers for a Small Perturbation Laser Turret," Proceedings of the Electro-Optics/Laser Conference - 1976, pp 9 - 19. Proceedings published by Industrial and Scientific Conference Management, Inc., Chicago, 1976.
4. A. E. Fuhs, "Distortion of Laser Turret Optics Due to Aircraft Mainstream Flow," Journal of the Optical Society of America, 66, p 1137, October 1976.
5. Garret N. Vanderplaats, "Inviscid Flow Over Turrets; Optimum Turret Shape," Lecture 5A, Short Course on Laser Aerodynamics presented at AFWL, April, 1977.
6. C. Barry Hogge and R. Russell Butts, "Frequency Spectra for the Geometrical Representation of Wavefront Distortions Due to Atmospheric Turbulence," IEEE Transactions on Antennas and Propagation, AP-24, 2, pp 144-154, 1976.
7. Garret N. Vanderplaats, "The Computer for Design and Optimization," Computing in Applied Mechanics, AMD Vol. 18, ASME, Dec. 1976.

1. Carter N. Vandergriff, "PORTMAN - A PORTMAN Program for Controlled  
 Positioning Mechanism," NASA TM X-52, August 1973.

2. A. E. Fuchs and S. E. Fuchs, "These Discovered Due to Similar Over a  
 Realistic Laser Laser," Naval Postgraduate School Report,  
 NPS-67-101, Sept. 1976.

3. A. E. Fuchs and S. E. Fuchs, "These Discovered at High Subsonic Mach  
 Numbers for a Small Portman Laser Laser," Proceedings of the  
 Electric and Electronic Engineers Conference - 1976, pp. 1-12, Washington,  
 published by Industrial and Scientific Consultants Management, Inc.,  
 Chicago, 1976.

# APPENDIX A

## PROGRAM FLOW CHARTS AND PORTMAN VARIABLES

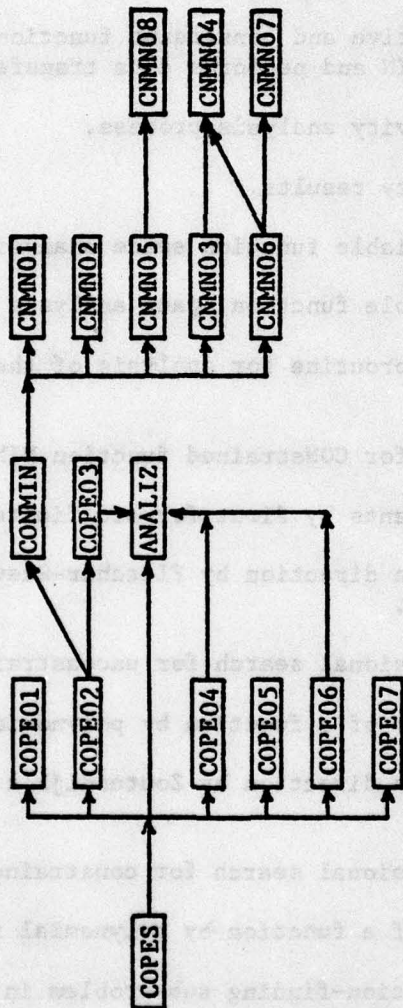
4. A. E. Fuchs, "PORTMAN - A PORTMAN Program for Controlled  
 Positioning Mechanism," Journal of the Optical Society of America, 66,  
 1979, October 1979.

5. Carter N. Vandergriff, "PORTMAN - A PORTMAN Program for Controlled  
 Positioning Mechanism," Naval Postgraduate School Report, NPS-67-101,  
 Sept. 1976.

6. C. Barry Hodge and R. James Hodge, "Frequency Spectra for the  
 Nonlinear Representation of Waveform Distortions due to Atmospheric  
 Turbulence," IEEE Transactions on Antennas and Propagation, AP-28,  
 1980, pp. 124-134, 1980.

7. Carter N. Vandergriff, "The Computer for Design and Simulation,"  
 Computing in Applied Mechanics, Vol. 10, 1976, Dec. 1976.





AD-A049 272

NAVAL POSTGRADUATE SCHOOL MONTEREY CALIF  
LASTOP - A COMPUTER CODE FOR LASER TURRET OPTIMIZATION OF SMALL--ETC(U)  
DEC 77 G N VANDERPLAATS, A E FUHS

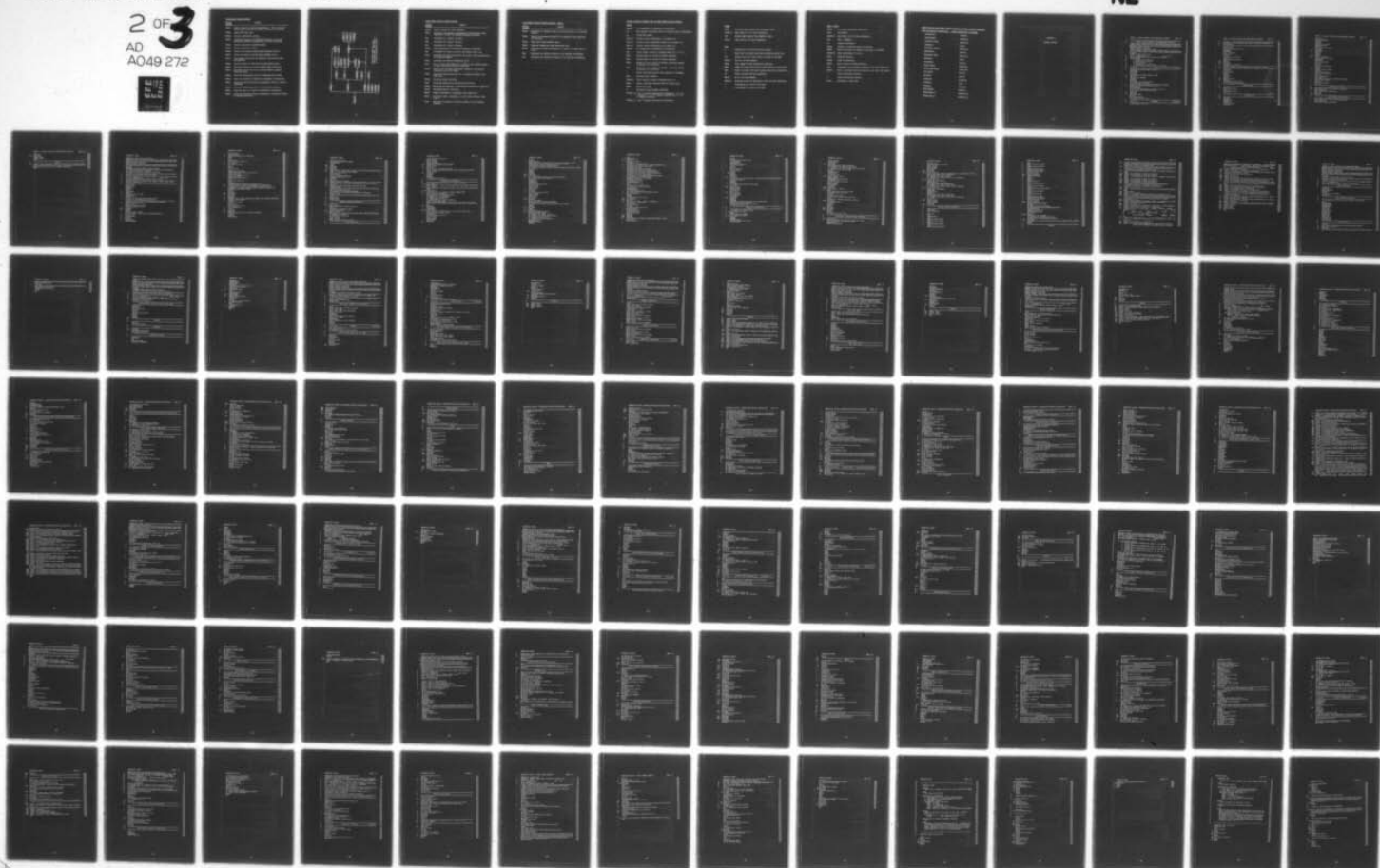
F/G 1/1

UNCLASSIFIED

NPS69-77-004

NL

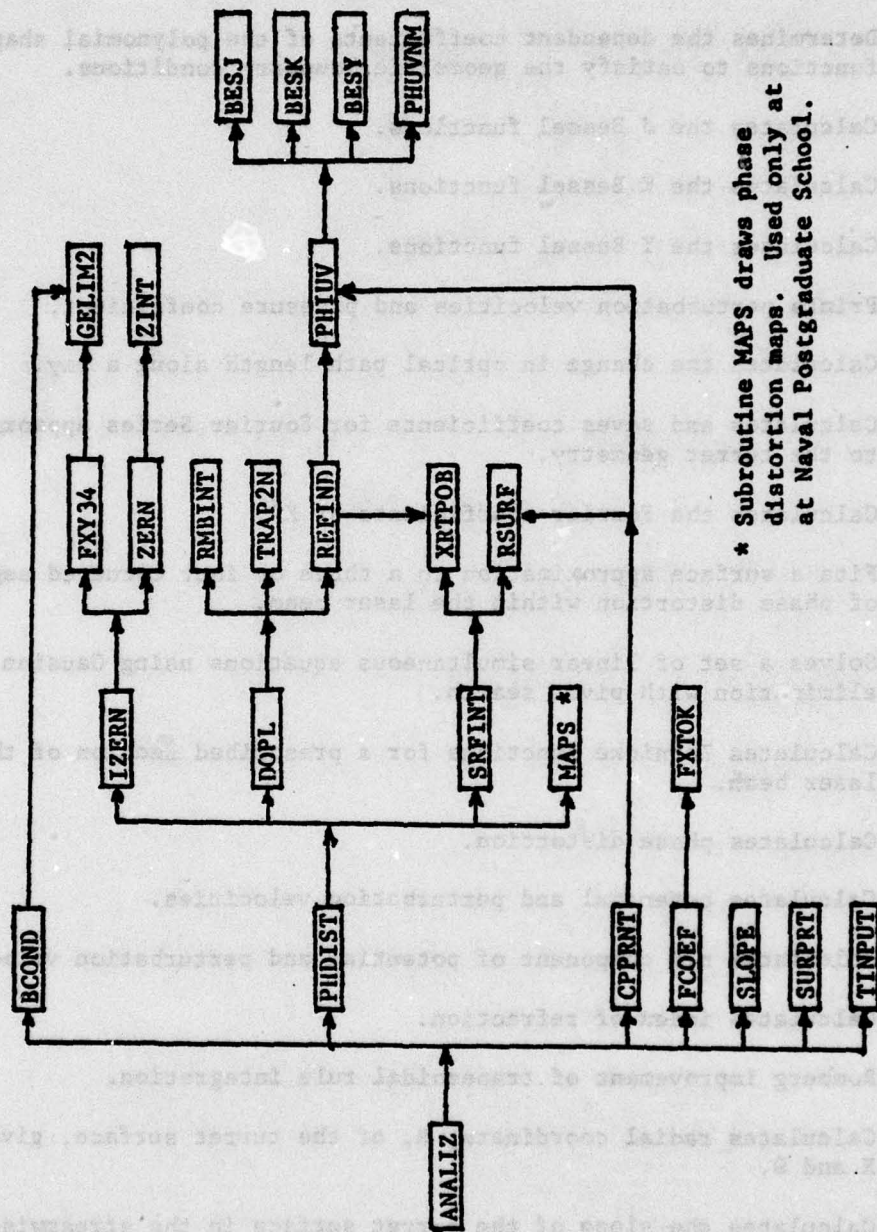
2 OF 3  
AD  
A049 272





## COPEs/CONMIN FORTRAN ROUTINES

<u>FORTRAN ROUTINE</u>	<u>PURPOSE</u>
COPEs	<u>C</u> ONTrol Program for <u>E</u> ngineering <u>S</u> ynthesis. This is the main program which organizes all design and analysis operations.
COPE01	Reads COPEs input data.
COPE02	Controls optimization process.
COPE03	Calculates objective and constraint functions in the form required by CONMIN and performs data transfer operations.
COPE04	Controls sensitivity analysis process.
COPE05	Prints sensitivity results.
COPE06	Controls two-variable function space analysis process.
COPE07	Prints two-variable function space analysis results.
ANALIZ	User supplied subroutine for analysis of the problem under consideration.
CONMIN	Control routine for CONstrained function MINimization.
CNMN01	Calculates gradients by first forward finite difference.
CNMN02	Calculates search direction by Fletcher-Reeves Conjugate Direction Method.
CNMN03	Solves one-dimensional search for unconstrained problems.
CNMN04	Finds the minimum of a function by polynomial interpolation.
CNMN05	Calculates search direction by Zoutendijk's Method of Feasible Directions.
CNMN06	Solves one-dimensional search for constrained problems.
CNMN07	Finds the zero of a function by polynomial interpolation.
CNMN08	Solves the direction-finding sub-problem in Zoutendijk's Method of Feasible Directions.



\* Subroutine MAPS draws phase distortion maps. Used only at Naval Postgraduate School.



# LASER TURRET ANALYSIS FORTRAN ROUTINES

<u>FORTRAN ROUTINE</u>	<u>PURPOSE</u>
ANALIZ	Control routine for turret analysis.
BCOND	Determines the dependent coefficients of the polynomial shape functions to satisfy the geometric boundary conditions.
BESJ	Calculates the J Bessel functions.
BESK	Calculates the K Bessel functions.
BESY	Calculates the Y Bessel functions.
CPPRNT	Prints perturbation velocities and pressure coefficient.
DOPL	Calculates the change in optical path length along a ray.
FCOEF	Calculates and saves coefficients for Fourier Series approximation to the turret geometry.
FXTOK	Calculates the Fourier coefficients of $X^k$ .
FX34	Fits a surface approximation to a three or four cornered segment of phase distortion within the laser beam.
GELIM2	Solves a set of linear simultaneous equations using Gaussian elimination with pivot search.
IZERN	Calculates Zernicke functions for a prescribed section of the laser beam.
PHDIST	Calculates phase distortion.
PHIUV	Calculates potential and perturbation velocities.
PHUVNM	Calculates n,m component of potential and perturbation velocities.
REFIND	Calculates index of refraction.
RMBINT	Romberg improvement of trapezoidal rule integration.
RSURF	Calculates radial coordinate, R, of the turret surface, given X and $\theta$ .
SLOPE	Calculates the slope of the turret surface in the streamwise direction.

# LASER TURRET ANALYSIS FORTRAN ROUTINES - CONCLD.

FORTRAN ROUTINE	PURPOSE
SRFINT	Calculates the distance along a ray from the mirror to the turret surface.
SURPRT	Prints the coordinates defined by the geometric shape functions, $f(X)$ and $f(\theta)$ .
TINPUT	Reads laser turret analysis input.
TRAP2N	Numerical integration using trapezoidal rule.
XRTPOB	Calculates the polar coordinates, $X$ , $R$ and $\theta$ of a given point on a ray.
ZERN	Calculates the definite integral of the Zernicke coefficients.
ZINT	Calculates the indefinite integral of the Zernicke coefficients.



# FORTRAN VARIABLES COMMONLY USED IN LASER TURRET ANALYSIS PROGRAM

## TURRET

ABAR(I) I-1 coefficient of polynomial in x-direction.

ACL Half spacing of periodic turret for Fourier series approximation.

AL Turret half length.

AMX(I,m) Fourier a-sub-m coefficient on I-1 power of x.

ANT(I,J) Fourier a-sub-n coefficient ( $J=n+1$ ) on I-1 power of x.

BMX(I,m) Fourier b-sub-m coefficient on I-1 power of x.

BBAR(I) I-1 coefficient of polynomial in  $\theta$ -direction.

EPS Turret height relative to fuselage radius at  $x = \theta = 0$ .

MMAX Maximum number of m-terms in Fourier expansion.

NMAX Maximum number of n-terms in Fourier expansion.

NTHBC Number of f and f' pairs of boundary conditions imposed on geometry in  $\theta$ -direction.

NXBC Number of f and f' pairs of boundary conditions imposed on geometry in x-direction.

R Radial coordinate measured from centerline of fuselage.

RFUS Fuselage radius (meters).

SLOPEX(I) Turret slope at various x-locations for  $\theta = 0$ .

THETA Angular coordinate measured from the vertical axis.

THMAX Turret half angle.

X Coordinate along fuselage centerline.

YYPXBC(I,J) f and f' boundary conditions in x-direction. J = 1 is x location, J = 2 is f boundary condition and J = 3 is f' boundary condition.

YYPTBC(I,J) f and f' boundary conditions in  $\theta$ -direction.

## MIRROR

**GAMMA** Elevation angle measured from horizontal plane.  
**GAMMAI(I)** Angle GAMMA for I-th beam orientation.  
**PHI** Azimuth angle measured from negative x-axis..  
**PHII(I)** Angle PHI for I-th beam orientation.

## BEAM

**A** Intercept of a ray with the turret surface.  
**B** Upper limit for phase distortion calculations along a ray.  
**ETA** Angular point from local z-axis to a point on the beam.  
**ETAI(I)** ETA for I-th beam element.  
**NBEAM** Total number of beam orientations considered.  
**NETAI** Number of values of ETA used in phase distortion calculations.  
**NRBI** Number of values of RB used in phase distortion calculations.  
**RB** Radial distance from beam centerline.  
**RBI(I)** RB for I-th beam element.  
**WGHTI(I)** Weighting factor for importance of the I-th beam orientation.  
**Y** Y-coordinate of a point on the beam.  
**Z** Z-coordinate of a point on the beam.



# AERO - OPTICS

AKPRIM	$k'$ in phase distortion relationship.
AMACH	Mach number.
AMACHI(I)	Mach number for I-th beam orientation.
BETA	$\text{ABS}(1 - \text{AMACH}^{**2})$
CP	Pressure coefficient.
DENGAM	Exponent in pressure-density relationship.
DENTRO	Ratio of external air density to sea level air density.
PDISTI(I)	Phase distortion if I-th ray.
PHIPP	Potential function.
RINDEX	Index of refraction.
SUMPD2	Sum of squares of phase distortion.
T(I)	Trapezoidal rule or Romberg integration for phase distortion.
TDENRT	Ratio of internal turret air density to sea level air density.
U	Axial perturbation velocity.
V	Radial perturbation velocity.
WAVEL	Wavelength of laser beam.

ARRAYS USED IN LASER TURRET ANALYSIS PROGRAM AND THEIR REQUIRED DIMENSIONS

ARRAY AND REQUIRED DIMENSION(S)      ACTUAL DIMENSION(S) IN PROGRAM

ABAR(MAXK+1)	ABAR(20)
AMACHI(NBEAM)	AMACHI(30)
AMX(MAXK+1,MMAX)	AMX(10,15)
AN(MAXK+1)	AN(10)
ANT(MAXP+1,NMAX+1)	ANT(10,15)
BBAR(MAXP+1)	BBAR(20)
BMX(MAXK+1,MMAX)	BMX(10,15)
BN(MAXK+1)	BN(10)
ETAI(NETAI)	ETAI(16)
GAMMAI(NBEAM)	GAMMAI(30)
PDISTI(NRBI*NETAI)	PDISTI(200)
PHII(NBEAM)	PHII(30)
RBI(NRBI)	RBI(10)
SLOPEX(30)	SLOPEX(30)
T(KTRAP+1)	T(10)
TITLE(20)	TITLE(20)
WGHTI(NBEAM)	WGHTI(30)
YYPTBC(NTHBC,3)	YYPTBC(10,3)
YYPXBC(NXBC,3)	YYPXBC(10,3)



## PROGRAM LISTING

103

COPEs - A CONTROL PROGRAM FOR ENGINEERING SYNTHESIS

SEPT. 77

```

C *****
C      COPEs - CONTROL PROGRAM FOR ENGINEERING SYNTHESIS.
C *****
COMMON /CMN1/ IPRINT,NDV,ITMAX,NCON,NSIDE,ICNDR,NSCAL,NFDG,FDCH,
IFDCHM,CT,CTMIN,CTL,CTLMIN,THETA,PHI,NAC,DELFUN,DARFUN,LINOBJ,ITRM,
ZITER,INFOG,IGOTO,INFO,OBJ
COMMON /COPEs1/ ATITLE(20)
COMMON /COPEs2/ RA(5000),IA(1000)
COMMON /COPEs3/ SGNOPT,NCALC,IOBJ,NSV,NSORJ,NCONA,N2VX,N2VY,M
12VY,N2VAR,IPSENS,IP2VAR,IPDRG,NACMX1,NDVTOT,LOCR(25),LOCI(25),ISCR
*1,ISCR2
COMMON /G1ORCH/ ARRAY(1500)
BY G. N. VANDERPLAATS                      OCT., 1974.
NASA-AMES RESEARCH CENTER, MOFFETT FIELD, CALIF.
NCALC OPTIONS:
0. READ ALL INPUT AND STOP.
1. SINGLE PASS ANALYSIS.
2. OPTIMIZATION.
3. SENSITIVITY -  $Z = F(X)$ .
4. TWO VARIABLE FUNCTION SPACE -  $Z = F(X,Y)$ .
-----
C *****          INPUT          *****
C -----
C DIMENSIONS OF ARRAYS ARRAY, RA AND IA.
NARRAY=1500
NDRA=5000
NDIA=1000
C READ GENERAL SYNTHESIS CONTROL INPUT.
10 CONTINUE
C SCRATCH TAPE NUMBERS.
ISCR1=20
ISCR2=40
CALL COPE01 (RA,IA,NDRA,NDIA)
IF (NCALC.LT.0) GO TO 140
C CHECK TO INSURE STORAGE REQUIREMENTS DO NOT EXCEED
C DIMENSIONED SIZES OF ARRAYS RA AND IA.
NDRA1=LOCR(25)
NDIA1=LOCI(25)
IF (NDRA1.LE.NDRA.AND.NDIA1.LE.NDIA) GO TO 20
WRITE (6,150) NDRA,NURA1,NDIA,NDIA1
GO TO 140
20 CONTINUE
C READ USER INPUT.
ICALC=1
CALL ANALYZ(ICALC)
IF (NCALC.LE.0) GO TO 10
C -----
C *****          EXECUTION          *****
C -----
IF (NCALC.NE.2) GO TO 50

```



```

C -----
C IF ABS(X(I)).GT.0 OVER-RIDE USER INPUT OF DECISION VARIABLES FOR
C OPTIMIZATION.
C -----
DO 40 I=1,NDV
XX=ARS(RA,I)
IF (XX.LT.1.0E-10) GO TO 40
N5=LOCN(5)
M2=LOCI(2)
DO 30 J=1,NDVTOT
NN1=IA(M2)
M2=M2+1
IF (NN1.NF.I) GO TO 30
NN1=IA(J)
ARRAY(NN1)=RA(I)*RA(N5)
30 N5=N5+1
40 CONTINUE
50 CONTINUE
IF(NCALC.NE.3) GO TO 70
C -----
C TRANSFER NOMINAL VALUES OF SENSITIVITY VARIABLES TO ARRAY.
C -----
M6=LOCI(6)
M7=LOCI(7)
DO 60 I=1,NSV
N=IA(M7)
M7=M7+1
NN=IA(M6)
M6=M6+1
60 ARRAY(NN)=RA(N)
70 CONTINUE
IF(NCALC.GT.4) GO TO 140
GO TO (80,90,120,130),NCALC
C -----
C ONE ANALYSIS
C -----
80 ICALC=2
CALL ANALIZ(ICALC)
ICALC=3
CALL ANALIZ(ICALC)
GO TO 10
C -----
C OPTIMIZATION
C -----
90 CONTINUE
N2=LOCN(2)
N3=LOCN(3)
N4=LOCN(4)
DO 100 I=1,NOV
C X=VECTOR.

```

COPEs - A CONTROL PROGRAM FOR ENGINEERING SYNTHESIS

SEPT. 77

	M2=LOC1(2)	1010
	DO 91 J=1,NDVTOT	1020
	N=IA(M2)	1030
	M2=M2+1	1040
	IF(N,NE,1) GO TO 91	1050
	N5=LOC1(5)+J-1	1060
	N=IA(J)	1070
	RA(I)=ARRAY(N)/RA(N5)	1080
	GO TO 92	1090
91	CONTINUE	1100
92	CONTINUE	1110
	N2=N2+1	1120
	N3=N3+1	1130
100	N4=N4+1	1140
C	INITIAL ANALYSIS.	1150
C	DESIGN VARIABLE VALUES.	1160
	M2=LOC1(2)	1170
	N5=LOC1(5)	1180
	DO 111 I=1,NDVTOT	1190
	N=IA(M2)	1200
	M=IA(I)	1210
	ARRAY(M)=RA(N)*RA(N5)	1220
	N5=N5+1	1230
111	M2=M2+1	1240
C	ANALYZE INITIAL DESIGN.	1250
	ICALC=2	1260
	CALL ANALYZ(ICALC)	1270
C	OUTPUT INITIAL DESIGN.	1280
	ICALC=3	1290
	CALL ANALYZ(ICALC)	1300
C	OPTIMIZATION.	1310
	CALL COPE02 (ARRAY,RA,IA,NARRAY,NDRA,NDIA)	1320
C	OUTPUT FINAL DESIGN.	1330
	ICALC=3	1340
	CALL ANALYZ(ICALC)	1350
	GO TO 10	1360
C	-----	1370
C	SENSITIVITY ANALYSIS	1380
C	-----	1390
120	CALL COPE04 (ARRAY,RA,IA,NARRAY,NDRA,NDIA)	1400
C	OUTPUT RESULTS.	1410
	CALL COPE05 (RA,IA,NDRA,NDIA)	1420
	GO TO 10	1430
130	CONTINUE	1440
C	-----	1450
C	TWO VARIABLE FUNCTION SPACE	1460
C	-----	1470
	CALL COPE06 (ARRAY,RA,IA,NARRAY,NDRA,NDIA)	1480
C	OUTPUT RESULTS.	1490
	CALL COPE07 (RA,IA,NDRA,NDIA)	1500





## SUBROUTINE COPE01

SEPT. 77

```

SUBROUTINE COPE01 (RA,IA,NORA,NDIA)
COMMON /CNMNI/ IPRINT,NDV,ITMAX,NCON,NSIDE,ICNDR,NSCAL,NFOD,FOCH,
1FOCHM,CT,CTMIN,CTL,CTLMIN,THETA,PHI,NAC,DELFUN,DABFUN,LINOBJ,ITRM,
2ITER,INFUG,IGOTO,INFO,OBJ
COMMON /COPE1/ ATITLF(20)
COMMON /COPE3/ SGNOPT,NCALC,IOBJ,NSV,NSOBJ,NCONA,N2VX,M2VX,N2VY,M
12VY,N2VAR,IPSENS,IP2VAR,IPDBG,NACHX1,NDVTOT,LOCR(25),LOC1(25),ISCR
*1,ISCR2
DIMENSION RA(NORA),IA(NDIA),CC(10),TITLE(20)
DATA STOP1/1HS/,STOP2/1HT/,STOP3/1HO/,STOP4/1HP/,STOP5/4HSTOP/
DATA END1/1HE/,END2/1HN/,END3/1HO/
DATA COM/1H/,COMMA/1H/,BLANK/1H/,ZERO/1H0/
C *****
C ROUTINE TO READ CONTROL INPUT FOR COPE.
C *****
C BY G. N. VANDERPLAATS MAR., 1973.
C NASA-AMES RESEARCH CENTER, MOFFETT FIELD, CALIF.
C -----
C READ CARD IMAGES AND STORE ON UNIT ISCR2. STORE ON UNIT ISCR1
C WITHOUT COMMENT CARDS
C -----
C REWIND ISCR1
C REWIND ISCR2
C NCARDS=0
C LOCI(25)=0
C NCOM=0
2 FORMAT(80A1)
C ICARD=0
10 READ(5,2)(RA(I),I=1,80)
C ICARD=ICARD+1
C IFORM=0
C IS THIS THE TITLE CARD OR A COMMENT CARD?
C IF(RA(1).EQ.COM.OR.NCOM.EQ.0) GO TO 27
C IF(RA(1).EQ.END1.AND.(RA(2).EQ.END2.AND.RA(3).EQ.END3)) GO TO 27
C UNFORMATTED INPUT CHECK. USE RA FOR TEMP. STORAGE.
C CHECK FOR FORMATTED INPUT.
C DO 25 J=1,80
C IF(RA(J).EQ.COMMA) GO TO 26
C IF(RA(J).EQ.COM) GO TO 27
25 CONTINUE
27 CONTINUE
C IFORM=1
C IF(RA(1).NE.COM) NCON=1
C NO COMMA FOUND. THIS DATA IS ALREADY FORMATTED.
C DO 21 J=1,80
21 RA(J+80)=RA(J)
C GO TO 18
26 CONTINUE
C ICARD=ICARD+1
C BLANK B=VECTOR.

```



## SUBROUTINE COPE01

SEPT. 77

	DO 11 I=1,80	510
11	RA(I+80)=BLANK	520
C	CONVERT UNFORMATTED TO FORMATTED.	530
	I2=10	540
	LI=1	550
	DO 12 I=1,8	560
C	BLANK WORKING VECTOR, CC.	570
	DO 13 J=1,10	580
13	CC(J)=BLANK	590
C	PUT FIELD I IN CC.	600
	K=0	610
	NFLG=0	620
	DO 14 J=LI,80	630
	JJ=J	640
C	IGNORE LEADING BLANKS.	650
	IF(RA(J).EQ.BLANK.AND.K.LT.1) GO TO 14	660
C	CHECK FOR COMMA.	670
	IF(RA(J).EQ.COMMA) GO TO 16	680
C	CHECK FOR COMMENT.	690
	IF(RA(J).EQ.COM) GO TO 17	700
	K=K+1	710
	IF(K.LE.10) GO TO 29	720
	K=K-1	730
	IF(NFLG.GT.0) GO TO 14	740
	WRITE(6,2A)(RA(L),L=1,80),I,(CC(L),L=1,10)	750
28	FORMAT(/5X,37H* * INPUT FIELD EXCEEDS 10 CHARACTERS/5X,	760
	* 13MCARD INPUT IS/5X,80A1/5X,17HERROR IS IN FIELD,15/5X,	770
	* 45HFIRST 10 NON-BLANK CHARACTERS ARE RETAINED AS,2X,10A1/5X,	780
	* 24HRESULTS MAY NOT BE VALID)	790
	NFLG=1	800
	GO TO 14	810
29	CC(K)=RA(I)	820
14	CONTINUE	830
	GO TO 18	840
17	CONTINUE	850
C	COMMENT FOUND. STORE BEGINNING IN FIELD I OR IN ACTUAL LOCATION,	860
C	WHICHEVER IS GREATER.	870
	I1=I2-10	880
	IF(I1.LT.IJ) I1=JJ	890
	I1=I1+1	900
	DO 19 J=J,1,79	910
	IF(I1.GT.A0) GO TO 18	920
	RA(I1+80)=RA(J+1)	930
19	I1=I1+1	940
	GO TO 18	950
16	CONTINUE	960
C	STORE CONTENTS OF CC IN B, RIGHT JUSTIFIED.	970
	LI=JJ+1	980
	J1=I2+80	990
	DO 22 J=1,10	1000

	SEPT. 77	
SUBROUTINE COPE01		
IF(K.EQ.0) GO TO 23		1010
IF(CC(K).EQ.BLANK) CC(K)=ZERO		1020
RA(J)=CC(K)		1030
J1=J1-1		1040
22 K=K-1		1050
23 CONTINUE		1060
I2=I2+10		1070
12 CONTINUE		1080
C CHECK TO SEE IF MORE THAN 8 FIELDS OF INPUT ARE CONTAINED ON THIS		1090
C CARD. IF YES, PRINT ERROR MESSAGE.		1100
IF(L1.GT.A0) GO TO 18		1110
DO 32 J=L1,A0		1120
IF(RA(J).EQ.COMMA) GO TO 33		1130
IF(RA(J).EQ.COM) GO TO 18		1140
32 CONTINUE		1150
GO TO 18		1160
33 WRITE(6,34)(RA(J),J=1,80)		1170
34 FORMAT(/5X,51H* * INPUT DATA CARD CONTAINS MORE THAN EIGHT FIELDS/		1180
* 5X,13HCARD INPUT IS/5X,80A1/5X,24HRESULTS MAY NOT BE VALID)		1190
18 CONTINUE		1200
IF(RA(1).NE.COM) WRITE(1SCR1,2)(RA(I),I=81,160)		1210
NCARDS=NCARDS+1		1220
IF((RA(1).EQ.STOP1.AND,RA(2).EQ.STOP2).AND,(RA(3).EQ.STOP3.AND,		1230
* RA(4).EQ.STOP4)) GO TO 20		1240
WRITE(1SCR2,4)NCARDS,(RA(I),I=1,A0)		1250
IF(IFORM.EQ.0) WRITE(1SCR2,4)NCARDS,(RA(I),I=81,60)		1260
41 FORMAT(15/80A1)		1270
IF(RA(1).EQ.END1.AND,(RA(2).EQ.END2.AND,RA(3).EQ.END3)) GO TO 20		1280
GO TO 10		1290
20 REWIND 1SCR1		1300
REWIND 1SCR2		1310
C -----		1320
C GENERAL SYNTHESIS INFORMATION		1330
C -----		1340
C TITLE.		1350
C ---- DATA BLOCK A.		1360
READ (1SCR1,750) (ATITLE(I),I=1,20)		1370
NCALC=-1		1380
IF(ATITLE(1).EQ.STOP5) RETURN		1390
C CONTROL PARAMETERS.		1400
C ---- DATA BLOCK B.		1410
READ (1SCR1,770) NCALC,NOV,NSV,N2VAR,IPNPUT,IPSENS,IP2VAR,IPDBG		1420
IF (NCALC.LT.0) RETURN		1430
IF (IPNPUT.GT.1) GO TO 50		1440
WRITE (6,540)		1450
WRITE (6,550)		1460
WRITE (6,560) (ATITLE(I),I=1,20)		1470
C -----		1480
C CARD IMAGE PRINT		1490
C -----		1500



## SUBROUTINE COPE01

SEPT. 77

	IF (IPNPUT.GT.0) GO TO 40	1510
	WRITE (6,830)	1520
	WRITE (6,840)	1530
	DO 30 I=1,ICARD	1540
	READ (ISCR2,41) NCARDS, (RA(J), J=1,80)	1550
30	WRITE (6,450) NCARDS, (RA(J), J=1,80)	1560
	REWIND ISCR2	1570
40	CONTINUE	1580
	WRITE (6,570) (ATITLE(I), I=1,20)	1590
	WRITE (6,580) NCALC,NDV,NSV,N2VAR,IPNPUT,IPSENS,IP2VAR,IPDBG	1600
	WRITE (6,480)	1610
50	NACHX1=0	1620
	NDVTOT=0	1630
	NCONA=0	1640
	NACHX2=0	1650
	IF (NDV.LE.0) GO TO 200	1660
C	-----	1670
C	OPTIMIZATION INFORMATION	1680
C	-----	1690
C	OPTIMIZATION CONTROL VARIABLES. - CONMIN DEPENDENT,	1700
C	---- DATA BLOCK C.	1710
	READ (ISCR1,770) IPRINT,ITMAX,ICNDIR,NSCAL,ITRM,LINOBJ,NACHX1,NFDG	1720
C	---- DATA BLOCK D.	1730
	READ (ISCR1,780) FDCH,FDCHM,CT,CTMIN,CTL,CTLMIN,THETA,PHI,DELFUN,D	1740
	IABFUN	1750
C	---- DATA BLOCK E.	1760
C	TOTAL NO. OF D. V., OBJECTIVE GLOBAL NUMBER, SIGN	1770
C	ON OPTIMIZATION OBJECTIVE.	1780
	READ (ISCR1,490) NDVTOT,I0BJ,SGNOPT	1790
	IF (NDVTOT.LT.NDV) NDVTOT=NDV	1800
	IF (NACHX1.LE.0) NACHX1=NDV+2	1810
	IF (IPNPUT.GE.2) GO TO 60	1820
	IF (ABS(SGNOPT).LT.1.0E-10) SGNOPT=-1.	1830
	WRITE (6,830) I0BJ,SGNOPT	1840
	WRITE (6,310) IPRINT,ITMAX,ICNDIR,NSCAL,ITRM,LINOBJ,NACHX1,NFDG	1850
	WRITE (6,320) FDCH,FDCHM,CT,CTMIN,CTL,CTLMIN,THETA,PHI,DELFUN,DABF	1860
	IUN	1870
60	N2=NDV+3	1880
	N3=N2+NDV+2	1890
	N4=N3+NDV+2	1900
C	---- DATA BLOCK F.	1910
C	DESIGN VARIABLE INFORMATION, LB, UB, INITIAL VALUE, SCAL.	1920
	IF (IPNPUT.LT.2) WRITE (6,640)	1930
	NS=N4+NDV+2	1940
	IF (NS.LE.NDRA) GO TO 70	1950
	WRITE (6,330)	1960
	WRITE (6,340)	1970
	LOC(25)=NS	1980
	GO TO 300	1990
70	CONTINUE	

## SUBROUTINE COPE01

SEPT. 77

```

2010 NSIDE=0
2020 DO 80 I=1,NDV
2030 READ (ISCR1,620) RA(N2),RA(N3),RA(I),RA(N4),(TITLE(J),J=1,5)
2040 IF (RA(N2).GT.-1.0E+15.OR.RA(N3).LT.1.0E+15) NSIDE=1
2050 IF (RA(N2).LE.-1.0E+15) RA(N2)=-1.1E+15
2060 IF (RA(N3).GE.1.0E+15) RA(N3)=1.1E+15
2070 IF (IPNPUT.LT.2) WRITE (6,650) I,RA(N2),RA(N3),RA(I),RA(N4),(TITLE
2080 I(J),J=1,5)
2090 N2=N2+1
2100 N3=N3+1
2110 N4=N4+1
2120 80 CONTINUE
2130 C ---- DATA BLOCK G.
2140 C D. V. NO., GLOBAL LOCATION, MULTIPLYING FACTOR.
2150 IF (IPNPUT.LT.2) WRITE (6,500)
2160 N5=4*NDV+9
2170 M2=NDVTOT+1
2180 N6=N5+NDVTOT
2190 M3=M2+NDVTOT
2200 IF (N6.LE.NORA) GO TO 90
2210 WRITE (6,730)
2220 WRITE (6,750)
2230 LOC(25)=N5
2240 GO TO 300
2250 90 CONTINUE
2260 IF (M3.LE.NDIA) GO TO 100
2270 WRITE (6,740)
2280 WRITE (6,750)
2290 LOC(25)=M3
2300 GO TO 300
2310 100 CONTINUE
2320 DO 110 I=1,NDVTOT
2330 READ (ISCR1,490) IA(M2),IA(I),RA(N5)
2340 IF (ABS(RA(N5)).LT.1.0E-20) RA(N5)=1.0
2350 IF (IPNPUT.LT.2) WRITE (6,510) I,IA(M2),IA(I),RA(N5)
2360 M2=M2+1
2370 N5=N5+1
2380 110 CONTINUE
2390 NCON=0
2400 C ---- DATA BLOCK H.
2410 C NUMBER OF CONSTRAINT SETS.
2420 READ (ISCR1,490) NCONS
2430 IF (IPNPUT.LT.2) WRITE (6,670)
2440 IF (IPNPUT.LT.2) WRITE (6,680) NCONS
2450 IF (NCONS.EQ.0) GO TO 200
2460 IF (IPNPUT.LT.2) WRITE (6,690)
2470 N6=4*NDV+NDVTOT+9
2480 M3=2*NDVTOT+1
2490 M4=2*NDVTOT+NCONS
2500 M4A=M4+1

```



## SUBROUTINE COPE01

SEPT. 77

	L=1	2510
C	DATA BLOCK I,	2520
	NCONA=0	2530
	DO 170 I=1,NCONS	2540
	NNN=N6+3	2550
	IF (NNN.GT.NDRA) GO TO 180	2560
C	GLOBAL NO. 1, GLOBAL NO. 2, LINEAR CONSTRAINT ID.	2570
	READ(ISCRT,770) ICONI,JCONI,LCONI	2580
C	LB, NORM, UR, NORM	2590
	READ(ISCRT,780)(RA(J),J=N6,NNN)	2600
	IF(RA(N6).LE.-1.0E+15) RA(N6)=-1.1E+15	2610
	IF(RA(N6+2).GE.1.0E+15) RA(N6+2)=1.1E+15	2620
	IF(RA(N6+1).LT.1.0E-20) RA(N6+1)=ABS(RA(N6))	2630
	IF(RA(N6+1).LT.0.1) RA(N6+1)=0.1	2640
	IF(RA(N6+3).LT.1.0E-20) RA(N6+3)=ABS(RA(N6+2))	2650
	IF(RA(N6+3).LT.0.1) RA(N6+3)=0.1	2660
C	NUMBER OF VARIABLES IN THIS SET.	2670
	NVAR=JCONI-ICONI+1	2680
	IF (NVAR.LT.1) NVAR=1	2690
	NCONA=NCONA+NVAR	2700
C	HOW MANY CONSTRAINTS?	2710
	J1=0	2720
	IF (RA(N6).GE.-1.0E+15) J1=1	2730
	IF(RA(N6+2).LT.1.0E+15) J1=J1+1	2740
	NCONI=J1+NVAR	2750
	NCON=NCONI,NCONI	2760
	IF (J1.EQ.0) GO TO 130	2770
C	ADD LINEAR CONSTRAINT IDENTIFIERS TO ISC.	2780
	DO 120 J=1,NCONI	2790
	M4=M4+1	2800
	MMH=M4	2810
	IF (MMH.GT.NDIA) GO TO 190	2820
120	IA(M4)=LCONI	2830
130	CONTINUE	2840
C	ADD LB, UR AND SCAL TO BLU IF NVAR.GT.1.	2850
	IF (NVAR.EQ.1) GO TO 150	2860
	NVAR1=NVAR-1	2870
	DO 140 J=1,NVAR1	2880
	NNN=N6+7	2890
	IF (NNN.GT.NDRA) GO TO 180	2900
	RA(N6+4)=RA(N6)	2910
	RA(N6+5)=RA(N6+1)	2920
	RA(N6+6)=RA(N6+2)	2930
	RA(N6+7)=RA(N6+3)	2940
	N6=N6+4	2950
140	CONTINUE	2960
150	CONTINUE	2970
C	ADD CONSTRAINED VARIABLE GLOBAL IDENTIFIERS TO ICONI.	2980
	ICONI=ICONI	2990
	DO 160 J=1,NVAR	3000

## SUBROUTINE COPE01

SEPT. 77

```

      MMM=M3
      IF (MMM.GT.NDIA) GO TO 190
      IA(M3)=ICON1
      ICON1=ICON1+1
      IF (J.EQ.1) GO TO 160
C     SHIFT ISC VECTOR.
      L1=M4+1
      L2=M4
      DO 165 K=M4A,M4
      IA(L1)=IA(L2)
      L1=L1+1
165  L2=L2+1
      M4=M4+1
      M4A=M4A+1
160  M3=M3+1
      IF (IPNPUT.LT.2) WRITE (6,660) L,ICON1,JCON1,LCON1,RA(N6),RA(N6+1)
      1,RA(N6+2),RA(N6+3)
      N6=N6+4
      L=NCON+1
      CONTINUE
170  IF (IPNPUT.LT.2) WRITE (6,470) NCONA
      GO TO 200
180  WRITE (6,430)
      WRITE (6,470)
      LOCR(25)=NCON
      GO TO 300
190  WRITE (6,460)
      WRITE (6,470)
      LOCI(25)=MMM
      GO TO 300
200  CONTINUE
      NSOBJ=0
      NSVTOT=0
C     STARTING LOCATIONS FOR SENSITIVITY INFORMATION.
      NSVR=4*NDV,NDVTOT+4*NCONA+9
      NSVI=2*(NDV+NCONA)+2*NDVTOT+NCONA+1
      IF (NSV.LE.0) GO TO 240
C     -----
C     SENSITIVITY INFORMATION
C     -----
      IF (IPNPUT.LT.2) WRITE (6,590)
C     ---- DATA BLOCK J, PART 1.
C     NSOBJ.
      READ (ISCD,770) NSOBJ
C     ---- DATA BLOCK J, PART 2.
C     NSENSZ.
      M5=NSVI
      MM5=M5+NSOBJ-1
      IF (MM5.LE.NDIA) GO TO 210
      WRITE (6,460)

```



## SUBROUTINE COPE01

183003 4117 SEPT. 77

	WRITE (6,780)	3510
	LOC(25)=MM5	3520
	GO TO 300	3530
210	CONTINUE	3540
	READ (ISCR1,770) (IA(I),I=M5,MM5)	3550
	IF (IPNPUT.LT.2) WRITE (6,530) NSOBJ	3560
	IF (IPNPUT.LT.2) WRITE (6,520) (IA(I),I=M5,MM5)	3570
	IF (IPNPUT.LT.2) WRITE (6,600)	3580
	N7=NSVR	3590
	M6=NSVI+NSOBJ	3600
	M7=M6+NSV	3610
	DO 230 I=1,NSV	3620
C ----	DATA BLOCK K, PART 1.	3630
C	ISENS, NSENS.	3640
	READ (ISCR1,770) IA(M6),NN1	3650
	NN7=NN7+NN1-1	3660
	IF (NN7.LE.NDRA) GO TO 220	3670
	WRITE (6,730)	3680
	WRITE (6,790)	3690
	LOC(25)=NN7	3700
	GO TO 300	3710
220	CONTINUE	3720
C ----	DATA BLOCK K, PART 2.	3730
C	SENS.	3740
	READ (ISCR1,780) (RA(J),J=N7,NN7)	3750
	IF (IPNPUT.GE.2) GO TO 225	3760
	JJ=N7+5	3770
	IF (JJ.GT.NN7) JJ=NN7	3780
	WRITE (6,610) I,IA(M6),(RA(J),J=N7,JJ)	3790
	JJ=JJ+1	3800
	IF (JJ.LE.NN7) WRITE (6,615) (RA(J),J=JJ,NN7)	3810
225	CONTINUE	3820
	NSVTOT=NSVTOT+NN1	3830
	IA(M7)=N7	3840
	N7=NN7+1	3850
	M6=M6+1	3860
	M7=M7+1	3870
230	CONTINUE	3880
240	CONTINUE	3890
	M2VX=0	3900
	M2VY=0	3910
	IF (M2VAR.LE.0) GO TO 270	3920
C	-----	3930
C	TWO-VARIABLE FUNCTION SPACE INFORMATION	3940
C	-----	3950
C ----	DATA BLOCK L.	3960
C	VARIABLE NUMBERS AND NUMBER OF VALUES OF X AND Y.	3970
	READ (ISCR1,770) N2VX,M2VX,N2VY,M2VY	3980
	N8=NSVR+NSVTOT	3990
	M8=NSVI+NSOBJ+2+NSV	4000

## SUBROUTINE COPE01

SEPT. 77

2100	MM8=MM8+N2VAR-1	4010
2110	IF (MM8.LF.NDIA) GO TO 250	4020
2120	WRITE (6,760)	4030
2130	WRITE (6,400)	4040
2140	LOC(25)=MM8	4050
2150	GO TO 300	4060
250	CONTINUE	4070
260	C ---- DATA BLOCK M.	4080
270	C GLOBAL VARIABLE NUMBERS CORRESPONDING TO FUNCTIONS OF X AND Y.	4090
280	READ (ISCR1,770) (IA(I),I=MM8,MM8)	4100
290	IF (IPNPUT.LT.2) WRITE (6,730)	4110
300	IF (IPNPUT.LT.2) WRITE (6,740) (IA(I),I=MM8,MM8)	4120
310	C ---- DATA BLOCK N.	4130
320	C VALUES OF X COMPONENTS.	4140
330	NN8=NR+M2VX-1	4150
340	IF (NN8.GT.NDRA) GO TO 260	4160
350	READ (ISCR1,780) (RA(I),I=NN8,NN8)	4170
360	IF (IPNPUT.LT.2) WRITE (6,700) N2VX	4180
370	IF (IPNPUT.LT.2) WRITE (6,720) (RA(I),I=NN8,NN8)	4190
380	C ---- DATA BLOCK O.	4200
390	C VALUES OF Y COMPONENTS.	4210
400	N9=NR+M2VY	4220
410	NN9=NR+M2VY-1	4230
420	NN8=NN9	4240
430	READ (ISCR1,780) (RA(I),I=N9,NN9)	4250
440	IF (IPNPUT.LT.2) WRITE (6,710) N2VY	4260
450	IF (IPNPUT.LT.2) WRITE (6,720) (RA(I),I=N9,NN9)	4270
460	GO TO 270	4280
470	WRITE (6,330)	4290
480	WRITE (6,400)	4300
490	LOC(25)=NNA	4310
500	GO TO 300	4320
270	CONTINUE	4330
510	C -----	4340
520	C DYNAMIC STORAGE ALLOCATION	4350
530	C -----	4360
540	NDV2=NDV+2	4370
550	C REAL VARIABLES.	4380
560	C X.	4390
570	LOC(1)=1	4400
580	C VLB.	4410
590	LOC(2)=NNDV+3	4420
600	C VUB.	4430
610	LOC(3)=LOC(2)+NDV2	4440
620	C SCAL.	4450
630	LOC(4)=LOC(3)+NDV2	4460
640	C AMULT.	4470
650	LOC(5)=LOC(4)+NDV2	4480
660	C BLU.	4490
670	LOC(6)=LOC(5)+NDVTOT	4500



## SUBROUTINE COPE01

SEPT. 77

C	SENS.	4510
	LOCR(7)=LOCR(6)+4*NCONA	4520
C	XM2V.	4530
	LOCR(8)=LOCR(7)+NSVTOT	4540
C	YM2V.	4550
	LOCR(9)=LOCR(8)+M2VX	4560
C	EXECUTION LEVEL ARRAYS.	4570
	LOCR(10)=LOCR(9)+M2VY	4580
	DO 280 I=1,25	4590
280	LOCR(I)=LOCR(10)	4600
C	INTEGER VARIABLES.	4610
C	IDSGN.	4620
	LOCI(1)=1	4630
C	NDSGN.	4640
	LOCI(2)=NDVTOT+1	4650
C	ICON.	4660
	LOCI(3)=LOCI(2)+NDVTOT	4670
C	ISC.	4680
	LOCI(4)=LOCI(3)+NCONA	4690
C	NSENSZ	4700
	LOCI(5)=LOCI(4)+2*(NDV+NCONA)	4710
C	ISENS.	4720
	LOCI(6)=LOCI(5)+NSOBJ	4730
C	NSENS.	4740
	LOCI(7)=LOCI(6)+NSV	4750
C	N2VZ.	4760
	LOCI(8)=LOCI(7)+NSV	4770
C	EXECUTION LEVEL ARRAYS.	4780
	LOCI(9)=LOCI(8)+N2VAR	4790
	DO 290 I=10,25	4800
290	LOCI(I)=LOCI(9)	4810
C	STORAGE FOR CONMIN ARRAYS.	4820
	IF(NCALC.NE.2) GO TO 295	4830
	NR1=NDV	4840
	IF(NACMX1.GT.NR1) NR1=NACMX1	4850
	NR=3*NCON+8*NDV+NACMX1*(NDV2+NACMX1)+NR1+4	4860
	NI=NACMX1+2*NR1	4870
	LOCR(25)=LOCR(10)+NR	4880
	LOCI(25)=LOCI(9)+NI	4890
	GO TO 300	4900
295	NR=NSV	4910
	IF(NSOBJ.GT.NR) NR=NSOBJ	4920
	IF(NCALC.EQ.3) LOCR(25)=LOCR(10)+NR	4930
	IF(NCALC.EQ.4) LOCR(25)=LOCR(10)+N2VAR	4940
300	CONTINUE	4950
	IF(IPNPUT.LT.2) WRITE(6,410)LOCR(10),LOCR(25),NDRA,LOCI(9),LOCI(25	4960
	),NDIA	4970
	RETURN	4980
C	-----	4990
C	FORMATS	5000

## SUBROUTINE COPE01

SEPT. 77

```

C ----- 5010
310 FORMAT (/5X,58HCONMIN PARAMETERS (IF ZERO, CONMIN DEFAULT WILL OVE 5020
1R-RIDE))//5X,6HPRINT,2X,5HITMAX,3X,6HICNDIR,3X,5HNSCAL,3X,4HITRM,3 5030
2X,6HLINEOB,2X,6HNACMX1,3X,4HNF0G/A18) 5040
320 FORMAT (/4X,4HFDCH,12X,5HFDCHM,11X,2HCT,14X,5HCTMIN/1X,4(2X,E14.5) 5050
*//6X,3HCT,13X,6HCTLMIN,10X,5HTHETA,11X,3HPHI/1X,4(2X,E14.5)// 5060
* 6X,6HDELFUN,10X,6HDAEFUN/1X,2(2X,E14.5)) 5070
330 FORMAT (/5X,54HREQUIRED STORAGE IN ARRAY RA EXCEEDS AVAILABLE STO 5080
IRAGE) 5090
340 FORMAT (/5X,27HUNABLE TO READ DATA BLOCK F) 5100
350 FORMAT (/5X,27HUNABLE TO READ DATA BLOCK G) 5110
360 FORMAT (/5X,54HREQUIRED STORAGE IN ARRAY IA EXCEEDS AVAILABLE STO 5120
IRAGE) 5130
370 FORMAT (/5X,27HUNABLE TO READ DATA BLOCK I) 5140
380 FORMAT (/5X,27HUNABLE TO READ DATA BLOCK J) 5150
390 FORMAT (/5X,27HUNABLE TO READ DATA BLOCK K) 5160
400 FORMAT (/5X,27HUNABLE TO READ DATA BLOCK L) 5170
410 FORMAT(/5X,25HDATA STORAGE REQUIREMENTS//17X,4HREAL,26X, 5180
* 7HINTEGER/7X,27HINPUT EXECUTION AVAILABLE,5X, 5190
* 27HINPUT EXECUTION AVAILABLE/1X,3I10,2X,3I10) 5200
420 FORMAT (A1,A2,A1,19A4) 5210
430 FORMAT (1H1,4X,27HCARD IMAGES OF CONTROL DATA//5X,4HCARD,20X,5HIM 5220
1AGE) 5230
440 FORMAT (1H0) 5240
450 FORMAT(18,1H),2X,80A1) 5250
470 FORMAT (/5X,40HTOTAL NUMBER OF CONSTRAINED PARAMETERS =,15) 5260
480 FORMAT (/5X,26HCALCULATION CONTROL, NCALC/5X,5HVALUE,3X,7HMEANING 5270
1/7X,1H1,5X,15HSINGLE ANALYSIS/7X,1H2,5X,12HOPTIMIZATION/7X,1H3,5X, 5280
2 11HSENSITIVITY/7X,1H4,5X,27HTWO-VARIABLE FUNCTION SPACE) 5290
490 FORMAT(2I10,F10.2) 5300
500 FORMAT (/5X,16HDESIGN VARIABLES//11X,5HD. V.,5X,6HGLOBAL,4X,11HMUL 5310
1TIPLYING/5X,2HID,5X,3HNO.,5X,8HVAR. NO.,5X,6HFACTOR) 5320
510 FORMAT (2I7,5X,15,6X,E12.5) 5330
520 FORMAT (5X,16I5) 5340
530 FORMAT (/5X,34HNUMBER OF SENSITIVITY OBJECTIVES =,15/5X,53HGLOBAL 5350
1NUMBERS ASSOCIATED WITH SENSITIVITY OBJECTIVES) 5360
540 FORMAT (1H1,//////,5X,47HCCCCCCC 0000000 PPPPPPP EEEEEEE S 5370
1SSSSSS/5X,47HC 0 0 P P E S /5X,47 5380
2HC 0 0 P P E S /5X,47HC 0 0 P 5390
3 0 PPPPPPP EEEE SSSSSSS/5X,47HC 0 0 P 5400
4 E S/5X,47HC 0 0 P E 5410
5 S/5X,47HCCCCCCC 0000000 P EEEEEEE SSSSSSS 5420
6) 5430
550 FORMAT (//////,18X,19H A S A = A M E S//14X,29HC O N T R O L P 5440
1 R O G R A M//26X,5HF O R//8X,41H E N G I N E E R I N G S Y N T H 5450
2 E S I S) 5460
560 FORMAT (//////24X,9HT I T L F//5X,20A4) 5470
570 FORMAT (1H1,4X,6HTITLF:/5X,20A4) 5480
580 FORMAT (////5X,19HCONTROL PARAMETERS//5X,42HCALCULATION CONTROL, 5490
1 NCALC =,15/5X,42HNUMBER OF GLOBAL DESIGN VARIABLES, 5500

```



## SUBROUTINE COPE01

SEPT. 77

```

2NDV =,15/5X,42HNUMBER OF SENSITIVITY VARIABLES,      NSV =,15/5X,42      5510
3HNUMBER OF FUNCTIONS IN TWO-SPACE, N2VAR =,15/5X,42HINPUT INFORMA      5520
TION PRINT CODE,      IPNPUT =,15/5X,42HSENSITIVITY PRINT CODE,      5530
5      IPSENS =,15/5X,42HTWO-SPACE PRINT CODE,      IP2VAR      5540
6=,15/5X,42HDEBUG PRINT CODE,      IPDBG =,15)      5550
590      FORMAT (/5X,27H* * SENSITIVITY INFORMATION)      5560
600      FORMAT (/14X,6HGLOBAL,4X,7HNOMINAL/5X,6HNUMBER,2X,8HVARIA      5570
        BLE,6X,1AHUFF-NOMINAL VALUES)      5580
610      FORMAT (5X,14,I8,5X,E12.5,1X,5E11.4)      5590
615      FORMAT(35X,5E11.4)      5600
620      FORMAT (4F10.2,10A4)      5610
630      FORMAT(/5X,35HGLOBAL VARIABLE NUMBER OF OBJECTIVE,10X,1H=,15/5X,      5620
        1 46HMULTIPLIER (NEGATIVE INDICATES MINIMIZATION) =,E12.4)      5630
640      FORMAT (/5X,27HDESIGN VARIABLE INFORMATION/5X,50HNON-ZERO INITIAL      5640
        1VALUE WILL OVEH-RIDE MODULE INPUT/5X,5HD. V.,5X,5HLOWER,10X,5HUPPE      5650
        2R,9X,7HINITIAL/5X,3HNO.,7X,5HBOUND,10X,5HBOUND,10X,5HVALUE,10X,5H      5660
        SCALE)      5670
650      FORMAT (1R,4X,E12.5,3X,E12.5,3X,E12.5,3X,E12.5,5A4)      5680
660      FORMAT (1R,17,210,5X,E12.5,3X,E12.5,3X,E12.5,3X,E12.5)      5690
670      FORMAT (/5X,22HCONSTRAINT INFORMATION)      5700
680      FORMAT (/5X,9HTHERE ARE,13,16H CONSTRAINT SETS)      5710
690      FORMAT (11X,6HGLOBAL,2X,6HGLOBAL,2X,6HLINEAR,6X,5HLOWER,6X,      5720
        * 13HNORMALIZATION,7X,5HUPPER,6X,13HNORMALIZATION/6X,2HID,3X,      5730
        * 6HVAR. 1,2X,6HVAR. 2,4X,2HID,8X,5HBOUND,9X,6HFACTOR,10X,      5740
        * 5HBOUND,9X,6HFACTOR)      5750
700      FORMAT (/5X,49HGLOBAL VARIABLE NUMBER CORRESPONDING TO X, N2VX =,      5760
        115/5X,20HVALUES OF X-VARIABLE)      5770
710      FORMAT (/5X,49HGLOBAL VARIABLE NUMBER CORRESPONDING TO Y, N2VY =,      5780
        115/5X,20HVALUES OF Y-VARIABLE)      5790
720      FORMAT (3X,5E12.4)      5800
730      FORMAT (/5X,51H* * TWO-VARIABLE FUNCTION SPACE MAPPING INFORMATI      5810
        1ON/5X,52HGLOBAL VARIABLE NUMBERS ASSOCIATED WITH F(X,Y), M2VZ)      5820
740      FORMAT (5X,10I5)      5830
750      FORMAT (20A4)      5840
770      FORMAT(8I10)      5850
780      FORMAT (8F10.2)      5860
      END      5870

```

## SUBROUTINE COPE02

SEPT. 77

	SUBROUTINE COPE02 (ARRAY, RA, IA, NARRAY, NDRA, NDIA)	10
	COMMON /COMMON1/ IPRINT, NDV, ITMAX, NCON, NSIDE, ICNDR, NSCAL, NFDG, FDCH,	20
	IFDCH, CT, CTMIN, CTL, CTLMIN, THETA, PHI, NAC, DELFUN, DABFUN, LINOBJ, ITRM,	30
	2ITER, INFOG, IGOTO, INFO, OBJ	40
	COMMON /COMMON2/ ATITLE(20)	50
	COMMON /COMMON3/ SGNOPT, NCALC, IUBJ, NSV, NSOBJ, NCONA, N2VX, M2VX, N2VY, M	60
	12VY, N2VAR, IPSENS, IP2VAR, IPORG, NACMX1, NDVTOT, LOCR(25), LOCI(25), ISCR	70
	*1, ISCR2	80
	DIMENSION ARRAY(NARRAY), RA(NDRA), IA(NDIA)	90
C	*****	100
C	ROUTINE TO CONTROL OPTIMIZATION.	110
C	*****	120
C	BY G. N. VANDERPLAATS MAR., 1973.	130
C	NASA-AMES RESEARCH CENTER, MOFFETT FIELD, CALIF.	140
C	-----	150
C	ARRAY DIMENSIONS	160
C	-----	170
	NN1=NDV+2	180
	NN2=2*NDV+NCON	190
	NN3=NACMX1	200
	NN4=NN3	210
	IF (NDV.GT.NN4) NN4=NDV	220
	NN5=2*NN4	230
C	-----	240
C	ARRAY STARTING LOCATIONS	250
C	-----	260
C	X, VLB, VUB, DF, A, S, G1, G2, C, B, SCAL, ISC, IC, MS1	270
	NX=1	280
	NVLB=LOC(2)	290
	NVUB=LOC(3)	300
	NNSCAL=LOC(4)	310
	NDF=LOC(10)	320
	NG=NDF+NN1	330
	NA=NG+NN2	340
	NS=NA+NN1+NN3	350
	NG1=NS+NN1	360
	NG2=NG1+NN2	370
	NC=NG2+NN3	380
	NB=NC+NN4	390
	NISC=LOC(4)	400
	NIC=LOC(10)	410
	NMS1=NIC+NN3	420
C	-----	430
C	OPTIMIZATION	440
C	-----	450
	IGOTO=0	460
C	CALL COMMIN (X, DF, G, ISC, IC, A, S, G1, G2, C, MS1, B, VLB, VUB,	470
C	*SCAL, N1, N2, N3, N4, N5)	480
50	CONTINUE	490
	CALL COMMIN (RA(NX), RA(NDF), RA(NG), IA(NISC), IA(NIC), RA(NA), RA(NS),	500



SUBROUTINE COPE02

SEPT. 77

```

1 RA(NG1),RA(NG2),RA(NG),IA(NMS1),RA(NB),RA(NVLB),RA(NVUB),RA(NNSCAL
2),NN1,NN2,NN3,NN4,NN5)
C
  ANALIZE.
  CALL COPE03 (ARRAY,NARRAY,RA(NX),RA(NDF),RA(NG),IA(NIC),RA(NA),NN1
1,NN2,NN3,RA,IA,NORA,NORA)
  IF (IGOTO.GT.0) GO TO 50
  RETURN
END

```

510  
520  
530  
540  
550  
560  
570  
580

## SUBROUTINE COPE03

SEPT. 77

```

SUBROUTINE COPE03 (ARRAY,NARRAY,X,DF,G,IC,A,NN1,NN2,NN3,HA,IA,NDRA
1,NDIA)
COMMON /CNMNI/ IPRINT,NDV,ITMAX,NCON,NSIDE,ICNDIR,NSCAL,NFDG,FDCH,
IFDCHM,CT,CTMIN,CTL,CTLMIN,THETA,PHI,NAC,DELFUN,DABFUN,LINOBJ,ITRM,
ZITER,INFOG,IGOTO,INFO,OBJ
COMMON /COPE03/ SGNOPT,NCALC,I OBJ,NSV,NSOBJ,NCONA,N2VX,M2VX,N2VY,M
12VY,N2VAR,IPSENS,IP2VAR,IPURG,NACHX1,NDVTOT,LOCR(25),LOCI(25),ISCR
*1,ISCR2
DIMENSION ARRAY(NARRAY),RA(NDRA),IA(NDIA)
DIMENSION X(NN1),DF(NN1),G(NN2),IC(NN3),A(NN3,NN1)
*****
BUFFER BETWEEN COMMIN AND COPE03 FUNCTION EVALUATION.
*****
BY G. N. VANDERPLAATS MAR., 1973.
NASA-AMES RESEARCH CENTER, MOFFETT FIELD, CALIF.
INITIAL ANALYSIS HAS BEEN DONE. IF ITER = 0, GO EVALUATE
OBJECTIVE AND CONSTRAINTS.
IF(ITER.LT.1) GO TO 25
-----
TRANSFER DESIGN VARIABLE VALUES TO USER ARRAY
-----
N5=LOCR(5)
M2=LOCI(2)
DO 20 I=1,NDVTOT
N=IA(M2)
M=IA(I)
ARRAY(M)=RA(N)*RA(N5)
N5=N5+1
M2=M2+1
M9=M9+1
20 CONTINUE
-----
ANALYZE
-----
ICALC=2
CALL ANALYZ(ICALC)
-----
OBJECTIVE
-----
CONTINUE
OBJ=-SGNOPT*ARRAY(I OBJ)
IF (NCON.EQ.0) RETURN
-----
CONSTRAINT VALUES
-----
M3=LOCI(3)
M6=LOCR(6)
N=0
DO 40 I=1,NCONA
PARAMETER IDENTIFIER.

```



SEP1. 77

123

## SUBROUTINE COPE04

SEPT. 77

```

SUBROUTINE COPE04 (ARRAY, RA, IA, NARRAY, NDRA, NDIA)
COMMON /CNMNI/ IPRINT, NDV, ITHAX, NCON, NSIDE, ICNDR, NSCAL, NFOG, FDCM,
1FOCHM, CT, CTMIN, CTL, CTLMIN, THETA, PHI, NAC, DELFUN, DABFUN, LINOBJ, ITRM,
2ITER, INFOG, IGOTO, INFO, ORJ
COMMON /CNPE1/ ATITLE(20)
COMMON /CNPE3/ SGNOPT, NCALC, IOBJ, NSV, NSOBJ, NCONA, N2VX, M2VX, N2VY, M
12VY, N2VAR, IPSENS, IP2VAR, IPDBG, NACMX1, NDVTOT, LOCR(25), LUCI(25), ISCR
*1, ISCR2
DIMENSION ARRAY(NARRAY), RA(NDRA), IA(NDIA)
*****
ROUTINE TO PROVIDE SENSITIVITY INFORMATION WITH RESPECT TO
A PRESCRIBED SET OF DESIGN VARIABLES.
*****
BY G. N. VANDERPLAATS
MAR., 1973.
STORE OUTPUT ON UNIT ISCR1.
REWIND ISCR1
-----
WRITE BASIC INFORMATION ON UNIT ISCR1
-----
TITLE.
WRITE (ISCR1, 330) (ATITLE(I), I=1, 20)
NCALC, NSV, NSOBJ
WRITE (ISCR1, 340) NCALC, NSV, NSOBJ
ISENS(I), I=1, NSV.
M6=LOC1(6)
M7=M6+NSV-1
WRITE (ISCR1, 340) (IA(I), I=M6, M7)
NSENSZ(I), I=1, NSOBJ.
M5=LOC1(5)
M6=M5+NSOBJ-1
WRITE (ISCR1, 340) (IA(I), I=M5, M6)
JCALC=3
ICALC=2
-----
***** NOMINAL *****
-----
CALL ANALIZ(ICALC)
IF (IPSENS.GT.0) CALL ANALIZ(JCALC)
-----
WRITE NOMINAL RESULTS ON UNIT ISCR1
-----
SENS(I, 1)
M7=LOC1(7)
N10=LOC1(10)
N11=N10
DO 160 I=1, NSV
N=M7+I-1
N=IA(N)
NA(N11)=RA(N)
160 N11=N11+1

```



## SUBROUTINE C0PE04

SEPT. 77

	N11=N10+NSV-1	510
	WRITE (ISCR1,350) (RA(I), I=N10,N11)	520
C	SENSITIVITY OBJECTIVES, OBJZ.	530
	M5=LOCI(5)	540
	N10=LOCR(10)	550
	N11=N10	560
	DO 170 I=1, NSOBJ	570
	M=M5+I-1	580
	M=IA(M)	590
	RA(N11)=ARRAY(M)	600
170	N11=N11+1	610
	N11=N10+NSOBJ-1	620
	WRITE (ISCR1,350) (RA(I), I=N10,N11)	630
C	-----	640
C	***** SENSITIVITIES *****	650
C	-----	660
	NSVAL=LOCR(8)-LOCR(7)-NSV	670
	NSVAL1=0	680
	DO 320 II=1, NSV	690
C	SENSITIVITY VARIABLE NUMBER.	700
	M6=LOCI(6)+II-1	710
	ISENS=IA(M6)	720
C	STARTING LOCATION OF SENSITIVITY VALUES IN RA (M7).	730
	M7=LOCI(7)+II-1	740
	M8=IA(M7+1)	750
	M7=IA(M7)	760
C	NUMBER OF SENSITIVITY VARIABLES, NSENS.	770
	NSENS=M8-M7	780
	IF (II.EQ. NSV) NSENS=NSVAL-NSVAL1+1	790
	IF (NSENS.LE.1) GO TO 320	800
C	WRITE ISENS AND NSENS ON UNIT ISCR1.	810
	NSENSI=NSENS-1	820
	WRITE (ISCR1,340) ISENS, NSENSI	830
C	-----	840
C	VARY THE VALUE OF THE SENSITIVITY PARAMETER	850
C	-----	860
	DO 310 JJ=2, NSENS	870
	NSVAL1=NSVAL1+1	880
	K=M7+JJ-1	890
	ARRAY (ISENS)=RA(K)	900
C	WRITE SENS(T,J) ON UNIT ISCR1.	910
	WRITE (ISCR1,350) ARRAY (ISENS)	920
C	ANALYZE.	930
	CALL ANALYZ (ICALC)	940
	IF (IPSENS.GT.0) CALL ANALYZ (JCALC)	950
C	-----	960
C	WRITE SENSITIVITY RESULTS ON UNIT ISCR1	970
C	-----	980
C	OBJZ.	990
	M5=LOCI(5)	1000

SUBROUTINE COPE04

SEPT. 77

01	N10=LOC(70)	1010
02	N11=N10	1020
03	DO 300 I=1, NS0BJ	1030
04	M=M5+1-1	1040
05	M=IA(M)	1050
06	RA(N11)=ARRAY(M)	1060
07	N11=N11+1	1070
08	N11=N10+NS0BJ-1	1080
09	WRITE(ISCRI,350)(RA(I),I=N10,N11)	1090
10	310 CONTINUE	1100
11	ARRAY(ISENS)=RA(M7)	1110
12	320 CONTINUE	1120
13	RETURN	1130
14	C -----	1140
15	C FORMATS	1150
16	C -----	1160
17	330 FORMAT (2A4)	1170
18	340 FORMAT (1A15)	1180
19	350 FORMAT (5F15.8)	1190
20	END	1200



## SUBROUTINE COPE05

SEPT. 77

```

SUBROUTINE COPE05 (RA,IA,NDRA,NDIA)
COMMON /CNMNI/ IPRINT,NDV,ITMAX,NCON,NSIDE,ICNOIR,NSCAL,NFOG,FOCH,
IFDCHM,CT,CTMIN,CTL,CTLMIN,THETA,PHI,NAC,DELFUN,DARFUN,LINOBJ,ITRM,
ZITER,INFO,IGOTO,INFO,ORJ
COMMON /COPE05/ ATITLE(20)
COMMON /COPE05/ SGNOPT,NCALC,I0BJ,NSV,NSOBJ,NCONA,N2VX,M2VX,N2VY,M
I2VY,N2VAR,IPSENS,IP2VAR,IPUBG,NACHX1,NOVTOT,LOCR(25),LOC(25),ISCR
*1,ISCR2
DIMENSION RA(NDRA),IA(NDIA)
*****
ROUTINE TO PRINT SENSITIVITY INFORMATION STORED ON UNIT ISCR1.
*****
BY G. N. VANDERPLAATS                                JULY, 1974.
NASA-AMES RESEARCH CENTER, MOFFETT FIELD, CALIF.
REWIND ISCR1
-----
GENERAL INFORMATION
-----
TITLE.
READ (ISCR1,60) (ATITLE(I),I=1,20)
NCALC, NSV, NSOBJ
READ (ISCR1,70) NCALC,NSV,NSOBJ
IF(NCALC.NF.3) RETURN
WRITE(6,80)
WRITE (6,90) (ATITLE(I),I=1,20)
WRITE (6,90) NSV,NSOBJ
ISENS(I),I=1,NSV.
READ (ISCR1,70) (IA(I),I=1,NSV)
WRITE (6,110)
WRITE (6,120) (IA(I),I=1,NSV)
NSENSZ(I),I=1,NSOBJ.
READ (ISCR1,70) (IA(I),I=1,NSOBJ)
WRITE (6,130)
WRITE (6,120) (IA(I),I=1,NSOBJ)
-----
NOMINAL INFORMATION
-----
SENS(I),I=1,NSV.
READ (ISCR1,140) (RA(I),I=1,NSV)
WRITE (6,150)
WRITE (6,160) (RA(I),I=1,NSV)
ORJZ(I),I=1,NSOBJ.
READ (ISCR1,140) (RA(I),I=1,NSOBJ)
WRITE (6,170)
WRITE (6,160) (RA(I),I=1,NSOBJ)
-----
SENSITIVITY INFORMATION
-----
WRITE (6,180)
DO 40 ISENS=1,NSV

```

## SUBROUTINE CUPE05

SEP. 77

C	ISENSI, NSENSI	510
	READ (ISCR1,70) ISENSI,NSENSI	520
	WRITE (6,190) ISENSI	530
	IF (NSENSI.EQ.0) WRITE (6,200)	540
	IF (NSENSI.EQ.0) GO TO 40	550
	DO 30 JJ=1,NSENSI	560
C	SENS(I,J)	570
	READ (ISCR1,140) SENSJ	580
C	OBJZ(I),I=1,NSOBJ	590
	READ (ISCR1,140) (RA(I),I=1,NSOBJ)	600
	N=MINO(4,NSOBJ)	610
	WRITE (6,210) SENSJ,(RA(I),I=1,N)	620
	N=(NSOBJ-1)/4	630
	IF (N.LT.1) GO TO 20	640
	L1=5	650
	DO 10 I=1,N	660
	L2=L1+3	670
	L2=MINO(L2,NSOBJ)	680
	WRITE (6,220) (RA(J),J=L1,L2)	690
10	L1=L1+4	700
20	CONTINUE	710
30	CONTINUE	720
40	CONTINUE	730
	RETURN	740
C	-----	750
C	FORMATS	760
C	-----	770
50	FORMAT (//5X,5HTITLE/5X,20A4)	780
60	FORMAT (20A4)	790
70	FORMAT (16I5)	800
80	FORMAT (14I,4X,47HSTANDARD SENSITIVITY ANALYSIS RESULTS (NCALC=3))	810
90	FORMAT (//5X,36HNUMBER OF SENSITIVITY VARIABLES, NSV,9X,1H=,15/5X,	820
	139HNUMBER OF SENSITIVITY OBJECTIVES, NSOBJ,6X,1H=,15)	830
110	FORMAT (//5X,52HGLOBAL NUMBERS ASSOCIATED WITH SENSITIVITY VARIABLE	840
	IES)	850
120	FORMAT (5X,10I5)	860
130	FORMAT (//5X,53HGLOBAL NUMBERS ASSOCIATED WITH SENSITIVITY OBJECTI	870
	IVES)	880
140	FORMAT (5F15.6)	890
150	FORMAT (//5X,26HNOMINAL DESIGN INFORMATION//5X,31HVALUES OF SENS	900
	ITIVITY VARIABLES)	910
160	FORMAT (5X,5E13.5)	920
170	FORMAT (//5X,81HVALUES OF SENSITIVITY OBJECTIVE FUNCTIONS)	930
180	FORMAT (//5X,28HSENSITIVITY ANALYSIS RESULTS)	940
190	FORMAT (//5X,15HGLOBAL VARIABLE,15//10X,1HX,20X,4HF(X))	950
200	FORMAT (//5X,35HTHE NOMINAL VALUE IS THE ONLY VALUE/5X,27HSPECIFIED	960
	1 FOR THIS VARIABLE)	970
210	FORMAT (//1X,E12.4,3X,4E13.4)	980
220	FORMAT (1AX,4E13.4)	990
	END	1000



## SUBROUTINE COPE06

SEPT. 77

```

SUBROUTINE COPE06 (ARRAY,RA,IA,NARRAY,NDRA,NDIA)
COMMON /CMMN1/ IPRINT,NDV,ITMAX,NCON,NSIDE,ICNDIR,NSCAL,NFDG,FDCH,
1FDCHM,CT,CTMIN,CTL,CTLMIN,THETA,PHI,NAC,DELFUN,DARFUN,LINOBJ,ITRM,
2ITER,INFO,IGOTO,INFO,OBJ
COMMON /COPE06/ ATITLE(20)
COMMON /COPE06/ SCNOPT,NCALC,I0BJ,NSV,NSOBJ,NCONA,N2VX,M2VX,N2VY,M
12VY,N2VAR,TPSENS,IP2VAR,IP0BG,NACMX1,NDVTOT,LOCR(25),LOC1(25),ISCR
*1,ISCR2
DIMENSION ARRAY(NARRAY),RA(NDRA),IA(NDIA)
C *****
C ROUTINE TO CALCULATE FUNCTIONS OF TWO DESIGN VARIABLES FOR ALL
C COMBINATIONS OF A SET OF PRESCRIBED VALUES OF THESE VARIABLES.
C *****
C WRITE OUTPUT INFORMATION ON SCRATCH UNIT ISCR1.
C BY G. N. VANDERPLAATS AUG., 1974.
C NASA-AMES RESEARCH CENTER, MOFFETT FIELD, CALIF.
C REWIND ISCR1
C -----
C UNIT ISCR1 WRITE
C -----
WRITE (ISCR1,160) (ATITLE(I),I=1,20)
WRITE (ISCR1,170) NCALC,N2VAR,M2VX,N2VX,M2VY,N2VY
N2VZ,
M8=LOC1(8),
M9=LOC1(9)-1
WRITE (ISCR1,170) (IA(I),I=M8,M9)
C -----
C TWO-VARIABLE FUNCTION SPACE
C -----
ICALC=2
KCALC=3
ISIGN=1
N8=LOCR(8),
N9=LOCR(9)-1
DO 150 I=1,M2VX
ARRAY(N2VX)=RA(N8)
DO 140 J=1,M2VY
N9=N9+ISIGN
ARRAY(N2VY)=RA(N9)
C ANALIZE.
110 CALL ANALIZ(ICALC)
120 CONTINUE
IF(IP2VAR.GT.0) CALL ANALIZ(KCALC)
C -----
C UNIT ISCR1 WRITE
C -----
WRITE X, Y.
WRITE (ISCR1,180) RA(N8),RA(N9)
F(X,Y) VALUES.
C N10=LOCR(10)

```

## SUBROUTINE COPE06

SEPT. 77

```

N11=N10
M8=LOC1(8)
DO 130 K=1,N2VAR
  N=IA(M8)
  RA(N11)=A0RAY(N)
  N11=N11+1
  M8=M8+1

```

130 CONTINUE

N11=N10+N2VAR-1

WRITE (ISCR1,180) (RA(K),K=N10,N11)

140 CONTINUE

N9=N9+ISIGN

M8=M8+1

ISIGN=-ISIGN

150 CONTINUE

RETURN

C

C

C

## FORMATS

160 FORMAT (2A4)

170 FORMAT (1A15)

180 FORMAT (5F15.8)

END

510

520

530

540

550

560

570

580

590

600

610

620

630

640

650

660

670

680

690

700

710

720

730



## SUBROUTINE COPE07

SEPT. 77

```

SUBROUTINE COPE07 (RA,IA,NDRA,NDIA)
COMMON /CNHNI/ IPRINT,NDV,ITMAX,NCUN,NSIDE,ICNDR,NSCAL,NFDG,FDCH,
IFDCHM,CT,CTMIN,CTL,CTLMIN,THETA,PHI,NAC,DELFUN,DABFUN,LINOBJ,ITRM,
ZITER,INFO,IGOTO,INFO,OBJ
COMMON /COPE07/ ATITLE(20)
COMMON /COPE07/ SGNOPT,NCALC,IUBJ,NSV,NSOBJ,NCONA,N2VX,M2VX,N2VY,M
12VY,N2VAR,IPSENS,IP2VAR,IP0BG,NACMX1,NDVTOT,LOCR(25),LOC1(25),ISCR
*1,ISCR2
DIMENSION RA(NDRA),IA(NDIA)
*****
ROUTINE TO PRINT TWO VARIABLE FUNCTION SPACE INFORMATION STORED ON
UNIT ISCR1.
*****
BY G. N. VANDERPLAATS AUG., 1974.
NASA-AMES RESEARCH CENTER, MOFFETT FIELD, CALIF.
REWIND ISCR1
-----
GENERAL INFORMATION
-----
TITLE.
HEAD (ISCR1,80) (ATITLE(I),I=1,20)
HEAD (ISCR1,90) NCALC,N2VAR,M2VX,M2VY,M2VY,N2VY
IF(NCALC.NE.4) RETURN
N2VZ(I),I=1,N2VAR.
C READ (ISCR1,90) (IA(I),I=1,N2VAR)
WRITE(6,50)
C WRITE (6,80) (ATITLE(I),I=1,20)
N2VX, N2VY.
C WRITE (6,100) N2VX,N2VY
N2VZ.
C WRITE (6,150)
WRITE (6,100) (IA(I),I=1,N2VAR)
-----
TWO-VARIABLE FUNCTION SPACE INFORMATION
-----
DO 30 I=1,M2VX
WRITE (6,160)
DO 30 J=1,M2VY
C X, Y.
READ (ISCR1,170) XX,YY
C F(X,Y).
N10=LOCR(10)
N11=N10+N2VAR-1
READ(ISCR1,170)(RA(K),K=N10,N11)
N=4
IF (N2VAR,LT.4) N=N2VAR
N11=N10+N-1
IF(J.EQ.1) WRITE(6,120)XX,YY,(RA(K),K=N10,N11)
IF(J.GT.1) WRITE(6,110)YY,(RA(K),K=N10,N11)
IF (N.LE.N2VAR) GO TO 20

```

**SEPT. 77**

132



SUBROUTINE CONMIN - CONSTRAINED FUNCTION MINIMIZATION      SEPT. 77

	SUBROUTINE CONMIN (X,DF,G,ISC,IC,A,S,G1,G2,C,MS1,B,VLB,VUB,SCAL,N1	10
	1,N2,N3,N4,N5)	20
	COMMON /CONMIN/ IPRINT,NDV,ITMAX,NCON,NSIDE,ICNDIR,NSCAL,NFDG,FDCH,	30
	1FDCMH,CT,CTMIN,CTL,CILMIN,THETA,PHI,NAC,DELFUN,DARFUN,LINOBJ,ITRM,	40
	2ITER,INFO,IGOTO,INFO,OBJ	50
	COMMON /CONSAV/DM1,DM2,DM3,DM4,DM5,DM6,DM7,DM8,DM9,DM10,DCT,DCTL,P	60
	1M11,AROBJ,AROBJ1,ALPHAX,CTA,CTAM,CTBM,OBJ1,SLOPE,DX,DX1,F1,XI,DFID	70
	2F1,ALP,FFF,D1(21),RSPACE,DM1,DM2,DM3,JDIP,	80
	4IOBJ,KOBJ,KCOUNT,NCAL(2),NFEAS,MSCAL,NCOBJ,NVC,ID1(7)	90
	*,III,NLNC,JGOTO,ISPACF(2)	100
	DIMENSION X(N1),DF(N1),G(N2),ISC(N2),IC(N3),A(N3,N1),S(N1),G1(N2),	110
	IG2(N2),C(N4),MS1(N5),B(N3,N3),VLB(N1),VUB(N1),SCAL(N1)	120
	ROUTINE TO SOLVE CONSTRAINED OR UNCONSTRAINED FUNCTION	130
	MINIMIZATION.	140
	BY G. N. VANDERPLAATS      APRIL, 1972.	150
	NASA-AMES RESEARCH CENTER, MOFFETT FIELD, CALIF.	160
	REFERENCE: CONMIN - A FORTRAN PROGRAM FOR CONSTRAINED FUNCTION	170
	MINIMIZATION: USER'S MANUAL, BY G. N. VANDERPLAATS,	180
	NASA TM X-62,282, AUGUST, 1973.	190
	STORAGE REQUIREMENTS:	200
	PROGRAM - 7000 DECIMAL WORDS (CDC COMPUTER)	210
	ARRAYS - APPROX. 2*(NDV**2)+26*NDV+4*NCON,	220
	WHERE N3 = NDV+2.	230
	RE-SCALE VARIABLES IF REQUIRED.	240
	IF (NSCAL.EQ.0.OR.IGOTO.EQ.0) GO TO 20	250
	DO 10 I=1,NDV	260
10	X(I)=C(I)	270
20	CONTINUE	280
	CONSTANTS:	290
	NDV1=NDV+1	300
	NDV2=NDV+2	310
	IF (IGOTO.EQ.0) GO TO 30	320
	GO TO (150,370,360,650,670), IGOTO	330
	-----	340
	SAVE INPUT CONTROL PARAMETERS	350
	-----	360
30	CONTINUE	370
	IF (IPRINT.GT.0) WRITE (6,1230)	380
	IF (LINOBJ.EQ.0.OR.(NCON.GT.0.OR.NSIDE.GT.0)) GO TO 40	390
	TOTALLY UNCONSTRAINED FUNCTION WITH LINEAR OBJECTIVE.	400
	SOLUTION IS UNBOUNDED.	410
	WRITE (6,670) LINOBJ,NCON,NSIDE	420
	RETURN	430
40	CONTINUE	440
	IDM1=ITRM	450
	IDM2=ITMAX	460
	IDM3=ICNDIR	470
	DM1=DELFUN	480
	DM2=DARFUN	490
	DM3=CT	500

## SUBROUTINE CONMIN - CONSTRAINED FUNCTION MINIMIZATION

SEPT. 77

```

510 DM4=CTMIN
520 DM5=CTL
530 DM6=CTLMIN
540 DM7=THETA
550 DM8=PHI
560 DM9=FDCH
570 DM10=FDCHM
580 -----
590
600
610
620
630
640
650
660
670
680
690
700
710
720
730
740
750
760
770
780
790
800
810
820
830
840
850
860
870
880
890
900
910
920
930
940
950
960
970
980
990
1000

```

-----

DEFAULTS

-----

```

IF (ITRM,LE.0) ITRM=5
IF (ITMAX,LE.0) ITMAX=20
NDV1=NDV+1
IF (ICNDIR,EQ.0) ICNDIR=NDV1
IF (DELFUN,LE.0) DELFUN=.0001
CT=-ABS(CT)
IF (CT,GE.0) CT=-.1
CTMIN=ABS(CTMIN)
IF (CTMIN,LE.0) CTMIN=.004
CTL=-ABS(CTL)
IF (CTL,GE.0) CTL=-0.01
CTLMIN=ABS(CTLMIN)
IF (CTLMIN,LE.0) CTLMIN=.001
IF (THETA,LE.0) THETA=1.
IF (PHI,LE.0) PHI=5.
IF (FDCH,LE.0) FDCH=.01
IF (FDCHM,LE.0) FDCHM=.01

```

-----

INITIALIZE INTERNAL PARAMETERS

-----

```

INFOG=0
ITER=0
JDIR=0
IORJ=0
KORJ=0
NDV2=NDV+2
KCOUNT=0
NCAL(1)=0
NCAL(2)=0
NAC=0
NFEAS=0
MSCAL=NSCAL
CT1=ITRM
CT1=1./CT1
DCT=(CTMIN/ABS(CT))*CT1
DCTL=(CTLMIN/ABS(CTL))*CT1
PHI1=PHI
ABOBJ1=.1
ABOBJ1=.1
ALPHAX=.1

```



## SUBROUTINE CONMIN - CONSTRAINED FUNCTION MINIMIZATION

SEPT. 77

	NCOBJ=0	1010
	CTAM=ABS(CTMIN)	1020
	CTBM=ABS(CTLMIN)	1030
C	CALCULATE NUMBER OF LINEAR CONSTRAINTS, NLNC.	1040
	NLNC=0	1050
	IF (NCON.EQ.0) GO TO 60	1060
	DO 50 I=1,NCON	1070
	IF (ISC(I).GT.0) NLNC=NLNC+1	1080
50	CONTINUE	1090
60	CONTINUE	1100
C	-----	1110
C	CHECK TO BE SURE THAT SIDE CONSTRAINTS ARE SATISFIED	1120
C	-----	1130
	IF (NSIDE.EQ.0) GO TO 100	1140
	DO 90 I=1,NDV	1150
	IF (VLB(I).LE.VUB(I)) GO TO 70	1160
	XX=.5*(VLR(I)+VUB(I))	1170
	X(I)=XX	1180
	VLB(I)=XX	1190
	VUB(I)=XX	1200
	WRITE (6,1120) I	1210
70	CONTINUE	1220
	XX=X(I)-VLB(I)	1230
	IF (XX.GE.0.) GO TO 80	1240
C	LOWER BOUND VIOLATED.	1250
	WRITE (6,1130) X(I),VLB(I),I	1260
	X(I)=VLB(I)	1270
	GO TO 90	1280
80	CONTINUE	1290
	XX=VUB(I)-X(I)	1300
	IF (XX.GE.0.) GO TO 90	1310
	WRITE (6,1140) X(I),VUB(I),I	1320
	X(I)=VUB(I)	1330
90	CONTINUE	1340
100	CONTINUE	1350
C	-----	1360
C	INITIALIZE SCALING VECTOR, SCAL	1370
C	-----	1380
	IF (NSCAL.EQ.0) GO TO 140	1390
	IF (NSCAL.LT.0) GO TO 120	1400
	DO 110 I=1,NDV	1410
110	SCAL(I)=1.	1420
	GO TO 140	1430
120	CONTINUE	1440
	DO 130 I=1,NDV	1450
	SI=ABS(SCAL(I))	1460
	IF (SI.LT.1.0E-20) SI=1.0E-5	1470
	SCAL(I)=SI	1480
	SI=1./SI	1490
	X(I)=X(I)+SI	1500

SUBROUTINE CONMIN - CONSTRAINED FUNCTION MINIMIZATION      SEPT. 77

	IF (NSIDE.EQ.0) GO TO 130	1510
	VLB(I)=VLR(I)*SI	1520
	VUB(I)=VUR(I)*SI	1530
130	CONTINUE	1540
140	CONTINUE	1550
C	-----	1560
C	***** CALCULATE INITIAL FUNCTION AND CONSTRAINT VALUES *****	1570
C	-----	1580
	INFO=1	1590
	NCAL(1)=1	1600
	IGOTO=1	1610
	GO TO 950	1620
150	CONTINUE	1630
	OBJ1=OBJ	1640
	IF (DABFUN.LE.0.) DABFUN=.001*ABS(OBJ)	1650
	IF (DABFUN.LT.1.0E-10) DABFUN=1.0E-10	1660
	IF (IPRINT.LE.0) GO TO 260	1670
C	-----	1680
C	PRINT INITIAL DESIGN INFORMATION	1690
C	-----	1700
	IF (IPRINT.LE.1) GO TO 220	1710
	IF (NSIDE.EQ.0.AND.NCON.EQ.0) WRITE (6,1300)	1720
	IF (NSIDE.NE.0.OR.NCON.GT.0) WRITE (6,1240)	1730
	WRITE (6,1250) IPRINT,NDV,ITMAX,NCON,NSIDE,ICNDIR,NSCAL,NFDG,LINOB	1740
	IJ,ITRM,N1,N2,N3,N4,N5	1750
	WRITE (6,1270) CT,CTMIN,CTL,CTLMIN,THETA,PHI,DELFUN,DABFUN	1760
	WRITE (6,1260) FDCM,FDCM	1770
	IF (NSIDE.EQ.0) GO TO 180	1780
	WRITE (6,1280)	1790
	DO 160 I=1,NDV,6	1800
	M1=MINO(NDV,I+5)	1810
160	WRITE (6,1010) I,(VLB(J),J=I,M1)	1820
	WRITE (6,1290)	1830
	DO 170 I=1,NDV,6	1840
	M1=MINO(NDV,I+5)	1850
170	WRITE (6,1010) I,(VUB(J),J=I,M1)	1860
180	CONTINUE	1870
	IF (NSCAL.GE.0) GO TO 190	1880
	WRITE (6,1310)	1890
	WRITE (6,1470) (SCAL(I),I=1,NDV)	1900
190	CONTINUE	1910
	IF (NCON.FQ.0) GO TO 220	1920
	IF (NLNC.FQ.0.OR.NLNC.EQ.NCON) GO TO 210	1930
	WRITE (6,1020)	1940
	DO 200 I=1,NCON,15	1950
	M1=MINO(NCON,I+14)	1960
200	WRITE (6,1030) I,(ISC(J),J=I,M1)	1970
	GO TO 220	1980
210	IF (NLNC.FQ.NCON) WRITE (6,1040)	1990
	IF (NLNC.FQ.0) WRITE (6,1050)	2000



## SUBROUTINE COMMIN - CONSTRAINED FUNCTION MINIMIZATION

SEPT. 77

220	CONTINUE	2010
	WRITE (6,1450) OBJ	2020
	WRITE (6,1460)	2030
	DO 230 I=1,NDV	2040
	X1=1.	2050
	IF (NSCAL.NE.0) X1=SCAL(I)	2060
230	G1(I)=X(I)*X1	2070
	DO 240 I=1,NDV,6	2080
	M1=MIN0(NDV,I+5)	2090
240	WRITE (6,1010) I,(G1(J),J=I,M1)	2100
	IF (NCUN.EQ.0) GO TO 260	2110
	WRITE (6,1440)	2120
	DO 250 I=1,NCON,6	2130
	M1=MIN0(NCON,I+5)	2140
250	WRITE (6,1010) I,(G(J),J=I,M1)	2150
260	CONTINUE	2160
	IF (IPRINT.GT.1) WRITE (6,1370)	2170
C	-----	2180
C	***** BEGIN MINIMIZATION *****	2190
C	-----	2200
270	CONTINUE	2210
	ITER=ITER+1	2220
	IF (ABOBJ.LT..0001) ABOBJ=.0001	2230
	IF (ABOBJ.GT..2) ABOBJ=.2	2240
	IF (ALPHAX.GT.1.) ALPHAX=1.	2250
	IF (ALPHAX.LT..01) ALPHAX=.01	2260
	IF (IPRINT.GT.2) WRITE (6,1320) ITER	2270
	NFEAS=NFEAS+1	2280
	IF (NFEAS.GT.10) GO TO 790	2290
	IF (IPRINT.GT.3.AND.NCON.GT.0) WRITE (6,1330) CT,CTL,PHI	2300
	CTA=ABS(CT)	2310
	IF (NCOBJ.EQ.0) GO TO 310	2320
	-----	2330
C	NO MOVE ON LAST ITERATION, DELETE CONSTRAINTS THAT ARE NO	2340
C	LONGER ACTIVE.	2350
C	-----	2360
	NNAC=NAC	2370
	DO 300 I=1,NNAC	2380
	NIC=IC(I)	2390
	IF (NIC.GT.NCON) NAC=NAC-1	2400
	IF (NIC.GT.NCON) GO TO 300	2410
	CT1=CT	2420
	IF (ISC(NIC).GT.0) CT1=CTL	2430
	IF (G(NIC).GT.CT1) GO TO 300	2440
	NAC=NAC-1	2450
	IF (I.EQ.NNAC) GO TO 300	2460
	IP1=I+1	2470
	DO 290 K=IP1,NNAC	2480
	II=K-1	2490
	DO 280 J=1,NDV2	2500

## SUBROUTINE COMMIN - CONSTRAINED FUNCTION MINIMIZATION

SEPT. 77

280	A(I1,J)=A(K,J)	2510
290	IC(I1)=IC(K)	2520
300	CONTINUE	2530
	GO TO 400	2540
310	CONTINUE	2550
	IF (MSCAL.LT.MSCAL.OR.MSCAL.EQ.0) GO TO 330	2560
	IF (MSCAL.LT.0.AND.KCOUNT.LT.ICNDIR) GO TO 330	2570
	MSCAL=0	2580
	KCOUNT=0	2590
C	-----	2600
C	SCALE VARIABLES	2610
C	-----	2620
	DO 320 I=1,NDV	2630
	SI=SCAL(I)	2640
	XI=SI*X(I)	2650
	SIB=SI	2660
	IF (MSCAL.GT.0) SI=ABS(XI)	2670
	IF (SI.LT.1.0E-10) GO TO 320	2680
	SCAL(I)=SI	2690
	SI=1./SI	2700
	X(I)=XI*SI	2710
	IF (NSIDE.EQ.0) GO TO 320	2720
	VLR(I)=SIB*SI*VLR(I)	2730
	VUB(I)=SIB*SI*VUB(I)	2740
320	CONTINUE	2750
	IF (IPRINT.LT.4.OR.(MSCAL.LT.0.AND.ITER.GT.1)) GO TO 330	2760
	WRITE (6,1340)	2770
	WRITE (6,1470) (SCAL(I),I=1,NDV)	2780
330	CONTINUE	2790
	MSCAL=MSCAL+1	2800
	NAC=0	2810
C	-----	2820
C	OBTAIN GRADIENTS OF OBJECTIVE AND ACTIVE CONSTRAINTS	2830
C	-----	2840
	INFO=2	2850
	NCAL(2)=NVAL(2)+1	2860
	IF (NFDG.IT.2) GO TO 350	2870
	IGOTO=2	2880
	GO TO 950	2890
350	CONTINUE	2900
	JGOTO=0	2910
360	CONTINUE	2920
	CALL CNMNO1 (JGOTO,X,DF,G,ISC,IC,A,G1,VLB,VUB,SCAL,C,NCAL,DX,DX1,F	2930
	11,X1,III,N1,N2,N3,N4)	2940
	IGOTO=3	2950
	IF (JGOTO.GT.0) GO TO 950	2960
370	CONTINUE	2970
	INFO=1	2980
	IF (NAC.GT.N3) GO TO 790	2990
	IF (MSCAL.EQ.0.OR.NFDG.EQ.0) GO TO 400	3000



**SEPT. 77**

### 3. CALCULATE GRADIENTS OF ACTIVE SIDE CONSTRAINTS

## SUBROUTINE CONMIN - CONSTRAINED FUNCTION MINIMIZATION

SEPT. 77

```

C -----
C IF (NSIDE.EQ.0) GO TO 510
MCN1=MCN
M1=0
DO 490 I=1,NDV
C LOWER BOUND.
X1=X(I)
XID=VLB(I)
X12=ABS(XID)
IF (X12.LT.1.) X12=1.
GI=(XID-X1)/X12
IF (GI.LT.-1.0E-6) GO TO 470
M1=M1+1
MS1(M1)=I
NAC=NAC+1
IF (NAC.GF.N3) GO TO 790
MCN1=MCN1+1
DO 460 J=1,NDV
460 A(NAC,J)=0.
A(NAC,I)=-1.
IC(NAC)=MCN1
G(MCN1)=GI
ISC(MCN1)=1
C UPPER BOUND.
470 XID=VUB(I)
X12=ABS(XID)
IF (X12.LT.1.) X12=1.
GI=(X1-XID)/X12
IF (GI.LT.-1.0E-6) GO TO 490
M1=M1+1
MS1(M1)=I
NAC=NAC+1
IF (NAC.GF.N3) GO TO 790
MCN1=MCN1+1
DO 480 J=1,NDV
480 A(NAC,J)=0.
A(NAC,I)=-1.
IC(NAC)=MCN1
G(MCN1)=GI
ISC(MCN1)=1
490 CONTINUE
C -----
C PRINT
C -----
C PRINT ACTIVE SIDE CONSTRAINT NUMBERS.
IF (IPRINT.LT.3) GO TO 510
WRITE (6,1090) M1
IF (M1.EQ.0) GO TO 510
WRITE (6,1100)
DO 500 I=1,M1,15

```



## SUBROUTINE CONMIN - CONSTRAINED FUNCTION MINIMIZATION

SEPT. 77

```

      M2=MIN0(M1,I+14)
      WRITE (6,1490) (MS1(J),J=1,M2)
500  CONTINUE
      PRINT GRADIENTS OF ACTIVE AND VIOLATED CONSTRAINTS.
      IF (IPRINT.LT.4) GO TO 550
      WRITE (6,1350)
      DO 520 I=1,NDV,6
      M1=MIN0(NDV,I+5)
520  WRITE (6,1010) I,(UF(J),J=1,M1)
      IF (NAC.EQ.0) GO TO 550
      WRITE (6,1360)
      DO 540 I=1,NAC
      M1=IC(I)
      M2=M1-NCON
      M3=0
      IF (M2.GT.0) M3=IABS(MS1(M2))
      IF (M2.LE.0) WRITE (6,990) M1
      IF (M2.GT.0) WRITE (6,1000) M3
      DO 530 K=1,NDV,6
      M1=MIN0(NDV,K+5)
530  WRITE (6,1010) K,(A(I,J),J=K,M1)
540  WRITE (6,1370)
550  CONTINUE
      C -----
      C ***** DETERMINE SEARCH DIRECTION *****
      C -----
      ALP=1.0E+20
      IF (NAC.GT.0) GO TO 560
      C -----
      C UNCONSTRAINED FUNCTION
      C -----
      C FIND DIRECTION OF STEEPEST DESCENT OR CONJUGATE DIRECTION.
      NVC=0
      NFEAS=0
      KCOUNT=KCOUNT+1
      IF KCOUNT.GT.ICNDIR RESTART CONJUGATE DIRECTION ALGORITHM.
      IF (KCOUNT.GT.ICNDIR.OR.IOBJ.EQ.2) KCOUNT=1
      IF (KCOUNT.EQ.1) JDIR=0
      C IF JDIR = 0 FIND DIRECTION OF STEEPEST DESCENT.
      CALL CNMND2 (JDIR,SLOPE,OFD0F1,OF,S,M1)
      GO TO 610
560  CONTINUE
      C -----
      C CONSTRAINED FUNCTION
      C -----
      C FIND USABLE-FEASIBLE DIRECTION.
      KCOUNT=0
      JDIR=0
      PHI=0.0*PHI
      IF (PHI.GT.1000.) PHI=1000.

```

## SUBROUTINE CONMIN - CONSTRAINED FUNCTION MINIMIZATION

SEPT. 77

```

      IF (NFEAS.EQ.1) PHI=PHI1
      CALCULATE DIRECTION, S.
C     CALL CNMNO5 (NVC,SLOPE,DF,G,ISC,IC,A,S,C,MS1,B,N1,N2,N3,N4,N5)
C     IF THIS DESIGN IS FEASIBLE AND LAST ITERATION WAS INFEASIBLE,
C     SET AROBJ1=.05 (5 PERCENT).
      IF (NVC.EQ.0.AND.NFEAS.GT.1) AROBJ1=.05
      IF (NVC.EQ.0) NFEAS=0
      IF (IPRINT.LT.3) GO TO 580
      WRITE (6,1380)
      DO 570 I=1,NAC,6
      M1=MINO(NAC,I,5)
570  WRITE (6,1410) I,(A(J,NDV1),J=1,M1)
      WRITE (6,1220) S(NDV1)
580  CONTINUE
C     -----
C     ***** ONE-DIMENSIONAL SEARCH *****
C     -----
      IF (S(NDV1).LT.1.0E-6.AND.NVC.EQ.0) GO TO 690
C     -----
C     FIND ALPHA TO OBTAIN A FEASIBLE DESIGN
C     -----
      IF (NVC.EQ.0) GO TO 610
      ALP=-1.
      DO 600 I=1,NAC
      NCI=IC(I)
      C1=G(NCI)
      CTC=CTAM
      IF (ISC(NCI).GT.0) CTC=CTBM
      IF (C1.LE.CTC) GO TO 600
      ALP1=0.
      DO 590 J=1,NDV
590  ALP1=ALP1+S(J)*A(I,J)
      ALP1=ALP1+A(I,NDV2)
      IF (ABS(A(I,P1)).LT.1.0E-20) GO TO 600
      ALP1=-C1/ALP1
      IF (ALP1.GT.ALPH) ALP=ALP1
600  CONTINUE
610  CONTINUE
C     -----
C     LIMIT CHANCE TO AROBJ1*OBJJ
C     -----
      ALP1=1.0E+20
      SI=ABS(OBJJ)
      IF (SI.LT..01) SI=.01
      IF (ABS(SLOPE).GT.1.0E-20) ALP1=AROBJ1*SI/SLOPE
      ALP1=ABS(ALP1)
      IF (NVC.GT.0) ALP1=10.*ALP1
      IF (ALP1.LT.ALPH) ALP=ALP1
C     -----
C     LIMIT CHANGE IN VARIABLE TO ALPHAX

```



## SUBROUTINE CONMIN - CONSTRAINED FUNCTION MINIMIZATION

SEPT. 77

```

C -----
ALP11=1.0E+20
DO 620 I=1,NDV
SI=ABS(S(I))
XI=ABS(X(I))
IF (SI.LT.1.0E-10.OR.XI.LT.0.1) GO TO 620
ALP1=ALPHAX*XI/SI
IF (ALP1.LT.ALP11) ALP11=ALP1
620 CONTINUE
IF (NVC.GT.0) ALP11=10.*ALP11
IF (ALP11.LT.ALPH) ALP=ALP11
IF (ALP.GT.1.0E+20) ALP=1.0E+20
IF (ALP.LE.1.0E-20) ALP=1.0E-20
IF (IPRINT.LT.3) GO TO 640
WRITE (6,1390)
DO 630 I=1,NDV,6
M1=MINO(NDV,I+5)
630 WRITE (6,1010) I,(S(J),J=I,M1)
WRITE (6,1110) SLOPE,ALP
640 CONTINUE
IF (NCON.GT.0.OR.NSIDE.GT.0) GO TO 660
C -----
C DO ONE-DIMENSIONAL SEARCH FOR UNCONSTRAINED FUNCTION
C -----
JGOTO=0
CONTINUE
650 CALL CMHN03 (X,S,DF,G,A,IC,SCAL,C,N1,N2,N3,N4)
IGOTO=4
IF (JGOTO.GT.0) GO TO 950
JDIR=1
PROCEED TO CONVERGENCE CHECK.
GO TO 680
C -----
C SOLVE ONE-DIMENSIONAL SEARCH PROBLEM FOR CONSTRAINED FUNCTION
C -----
660 CONTINUE
JGOTO=0
670 CONTINUE
CALL CMHN06 (X,DF,G,ISC,S,G1,G2,VLB,VUB,SCAL,N1,N2)
IGOTO=5
IF (JGOTO.GT.0) GO TO 950
IF (NAC.EQ.0) JDIR=1
C -----
C ***** UPDATE ALPHAX *****
C -----
680 CONTINUE
690 CONTINUE
IF (ALP.GT.1.0E+19) ALP=0.
UPDATE ALPHAX TO BE AVERAGE OF MAXIMUM CHANGE IN X(I)
AND ALPHAX.

```

```

SUBROUTINE CONMIN - CONSTRAINED FUNCTION MINIMIZATION      SEPT. 77
ALP11=0.                                                    5510
DO 700 I=1,NDV                                              5520
SI=ABS(S(I))                                                5530
XI=ABS(X(I))                                                5540
IF (XI.LT.1.0E-10) GO TO 700                              5550
ALP1=ALP+SI/XI                                              5560
IF (ALP1.GT.ALP11) ALP11=ALP1                              5570
700 CONTINUE                                                5580
ALP11=.5*(ALP11+ALPHAX)                                     5590
ALP12=.5*ALPHAX                                             5600
IF (ALP11.GT.ALP12) ALP11=ALP12                            5610
ALPHAX=ALP11                                                5620
NCOBJ=NCOBJ+1                                              5630
C ABSOLUTE CHANGE IN OBJECTIVE.                             5640
OBJD=OBJ1-NRJ                                              5650
OBJB=ABS(NRJ)                                              5660
IF (OBJB.LT.1.0E-10) OBJB=0.                               5670
IF (NAC.EQ.0.OR.OBJB.GT.0.) NCOBJ=0                        5680
IF (NCOBJ.GT.1) NCOBJ=0                                    5690
C -----                                                  5700
C PRINT                                                      5710
C -----                                                  5720
C PRINT MOVE PARAMETER, NEW X-VECTOR AND CONSTRAINTS.     5730
IF (IPRINT.LT.3) GO TO 710                                 5740
WRITE (6,1400) ALP                                          5750
710 IF (IPRINT.LT.2) GO TO 780                              5760
IF (OBJB.GT.0.) GO TO 720                                  5770
IF (IPRINT.EQ.2) WRITE (6,1410) ITER,OBJ                  5780
IF (IPRINT.GT.2) WRITE (6,1420) OBJ                        5790
GO TO 740                                                    5800
720 IF (IPRINT.EQ.2) GO TO 730                              5810
WRITE (6,1430) OBJ                                          5820
GO TO 740                                                    5830
730 WRITE (6,1440) ITER,OBJ                                  5840
740 WRITE (6,1460)                                           5850
DO 750 I=1,NDV                                              5860
FF1=1.                                                      5870
IF (NSCAL.NE.0) FF1=SCAL(I)                                5880
750 G1(I)=FF1*X(I)                                          5890
DO 760 I=1,NUV,6                                           5900
M1=M1NO(NDV,I+5)                                           5910
760 WRITE (6,1010) I,(G1(J),J=I,M1)                       5920
IF (NCON.EQ.0) GO TO 780                                   5930
WRITE (6,1480)                                              5940
DO 770 I=1,NCON,6                                           5950
M1=M1NO(NCON,I+5)                                           5960
770 WRITE (6,1010) I,(G(J),J=I,M1)                         5970
780 CONTINUE                                                5980
C -----                                                  5990
C CHECK CONVERGENCE                                         6000

```



## SUBROUTINE CONMIN - CONSTRAINED FUNCTION MINIMIZATION

SEPT. 77

```

C ----- 6010
C STOP IF ITER EQUALS ITMAX. 6020
C IF (ITER,GE,ITMAX) GO TO 790 6030
C ----- 6040
C ABSOLUTE CHANGE IN OBJECTIVE 6050
C ----- 6060
C OBJB=ABS(OBJD) 6070
C KOBJ=KOBJ+1 6080
C IF (OBJB,GE,DABFUN,OR,NFEAS,GT,0) KOBJ=0 6090
C ----- 6100
C RELATIVE CHANGE IN OBJECTIVE 6110
C ----- 6120
C IF (ABS(OBJ1),GT,1.0E-10) OBJD=OBJD/ABS(OBJ1) 6130
C ABOBJ=.5*(ABS(ABOBJ)+ABS(OBJD)) 6140
C ABOBJ=ABS(OBJD) 6150
C IOBJ=IOBJ+1 6160
C IF (NVC,GT,0,OR,OBJD,GE,DELFUN) IOBJ=0 6170
C IF (IOBJ,GE,ITRM,OR,KOBJ,GE,ITRM) GO TO 790 6180
C OBJ1=OBJ 6190
C ----- 6200
C REDUCE CT IF OBJECTIVE FUNCTION IS CHANGING SLOWLY 6210
C ----- 6220
C IF (IOBJ,LT,1,OR,NAC,EQ,0) GO TO 270 6230
C CT=DCT*CT 6240
C CTL=CTL+DCTL 6250
C IF (ABS(CT),LT,CTMIN) CT=-CTMIN 6260
C IF (ABS(CTL),LT,CTLMIN) CTL=-CTLMIN 6270
C ----- 6280
C CHECK FOR UNBOUNDED SOLUTION 6290
C ----- 6300
C STOP IF OBJ IS LESS THAN -1.0E+40. 6310
C IF (OBJ,GT,-1.0E+40) GO TO 270 6320
C WRITE (6,980) 6330
790 CONTINUE 6340
C IF (NAC,GE,N3) WRITE (6,1500) 6350
C ----- 6360
C ***** FINAL FUNCTION INFORMATION ***** 6370
C ----- 6380
C IF (NSCAL,EQ,0) GO TO 820 6390
C UN-SCALE THE DESIGN VARIABLES. 6400
C DO 810 I=1,NDV 6410
C XI=SCAL(I) 6420
C IF (NSIDE,EQ,0) GO TO 810 6430
C VLB(I)=XI*VLB(I) 6440
C VUB(I)=XI*VUB(I) 6450
810 X(I)=XI*X(I) 6460
C ----- 6470
C PRINT FINAL RESULTS 6480
C ----- 6490
820 IF (IPRINT,EQ,0,OR,NAC,GE,N3) GO TO 940 6500

```

## SUBROUTINE CONMIN - CONSTRAINED FUNCTION MINIMIZATION

SEPT. 77

WRITE (6,1510)	6510
WRITE (6,1430) OBJ	6520
WRITE (6,1460)	6530
DO 830 I=1,NDV,6	6540
M1=MIN0(NDV,1+5)	6550
830 WRITE (6,1010) I,(X(J),J=I,M1)	6560
IF (NCON,EQ,0) GO TO 890	6570
WRITE (6,1440)	6580
DO 840 I=1,NCON,6	6590
M1=MIN0(NCON,1+5)	6600
840 WRITE (6,1010) I,(G(J),J=I,M1)	6610
C DETERMINE WHICH CONSTRAINTS ARE ACTIVE AND PRINT.	6620
NAC=0	6630
NVC=0	6640
DO 860 I=1,NCON	6650
CTA=CTAM	6660
IF (ISC(I),GT,0) CTA=CTBM	6670
GI=G(I)	6680
IF (GI,GT,CTA) GO TO 850	6690
IF (GI,LT,CT,AND,ISC(I),EQ,0) GO TO 860	6700
IF (GI,LT,CTL,AND,ISC(I),GT,0) GO TO 860	6710
NAC=NAC+1	6720
IC(NAC)=I	6730
GO TO 860	6740
850 NVC=NVC+1	6750
MS1(NVC)=I	6760
860 CONTINUE	6770
WRITE (6,1060) NAC	6780
IF (NAC,EQ,0) GO TO 870	6790
WRITE (6,1070)	6800
WRITE (6,1490) (IC(J),J=1,NAC)	6810
870 WRITE (6,1040) NVC	6820
IF (NVC,EQ,0) GO TO 880	6830
WRITE (6,1070)	6840
WRITE (6,1490) (MS1(J),J=1,NVC)	6850
880 CONTINUE	6860
890 CONTINUE	6870
IF (NSIDE,EQ,0) GO TO 920	6880
C DETERMINE WHICH SIDE CONSTRAINTS ARE ACTIVE AND PRINT.	6890
NAC=0	6900
DO 910 I=1,NDV	6910
XI=X(I)	6920
XID=VLB(I)	6930
X12=ABS(XID)	6940
IF (X12,LT,1.) X12=1.	6950
GI=(XID-XI)/X12	6960
IF (GI,LT,-1.0E-6) GO TO 900	6970
NAC=NAC+1	6980
MS1(NAC)=I	6990
900 XID=VUB(I)	7000



## SUBROUTINE CONMIN - CONSTRAINED FUNCTION MINIMIZATION

SEPT, 77

	X12=ABS(X1D)	7010
	IF (X12.LT.1.) X12=1.	7020
	GI=(X1-X1D)/X12	7030
	IF (GI.LT.-1.0E-6) GO TO 910	7040
	NAC=NAC+1	7050
	MS1(NAC)=I	7060
910	CONTINUE	7070
	WRITE (6,1090) NAC	7080
	IF (NAC.EQ.0) GO TO 920	7090
	WRITE (6,1100)	7100
	WRITE (6,1490) (MS1(J),J=1,NAC)	7110
920	CONTINUE	7120
	WRITE (6,1150)	7130
	IF (ITER.GE.ITMAX) WRITE (6,1160)	7140
	IF (NFEAS.GE.10) WRITE (6,1170)	7150
	IF (IOBJ.GE.ITRM) WRITE (6,1190) ITRM	7160
	IF (KOBJ.GE.ITRM) WRITE (6,1200) ITRM	7170
	WRITE (6,1210) ITER	7180
	WRITE (6,1520) NCAL(1)	7190
	IF (NCON.GT.0) WRITE (6,1530) NCAL(1)	7200
	IF (NFDG.NE.0) WRITE (6,1540) NCAL(2)	7210
	IF (NCON.GT.0.AND.NFDG.EQ.2) WRITE (6,1550) NCAL(2)	7220
C	-----	7230
C	RE-SET BASIC PARAMETERS TO INPUT VALUES	7240
C	-----	7250
940	ITRM=IDM1	7260
	ITMAX=IDM2	7270
	ICNDIR=IDM3	7280
	DELFUN=DM1	7290
	DABFUN=DM2	7300
	CT=DM3	7310
	CTMIN=DM4	7320
	CTL=DM5	7330
	CTLMIN=DM6	7340
	THETA=DM7	7350
	PHI=DM8	7360
	FDCH=DM9	7370
	FDCHM=DM10	7380
	IGOTO=0	7390
950	CONTINUE	7400
	IF (NSCAL.EQ.0.OR.IGOTO.EQ.0) RETURN	7410
C	UN-SCALE VARIABLES.	7420
	DO 960 I=1,NDV	7430
	C(I)=X(I)	7440
960	X(I)=X(I)+SCAL(I)	7450
	RETURN	7460
C	-----	7470
C	FORMATS	7480
C	-----	7490
C		7500

## SUBROUTINE CONMIN - CONSTRAINED FUNCTION MINIMIZATION

SEPT. 77

```

970  FORMAT (//,5X,72H A COMPLETELY UNCONSTRAINED FUNCTION WITH A LINEAR      7510
1  OBJECTIVE IS SPECIFIED//10X,8H LNOBJ =,15/10X,8H NCON =,15/10X,8      7520
2H NSIDE =,15/5X,35H CONTROL RETURNED TO CALLING PROGRAM)              7530
980  FORMAT (//,5X,56H CONMIN HAS ACHIEVED A SOLUTION OF OBJ LESS THAN -    7540
11.0E+40/5X,32H SOLUTION APPEARS TO BE UNBOUNDED/5X,26H OPTIMIZATION     7550
2 IS TERMINATED)                                                         7560
990  FORMAT (5X,17H CONSTRAINT NUMBER,15)                                7570
1000 FORMAT (5X,27H SIDE CONSTRAINT ON VARIABLE,15)                     7580
1010 FORMAT (3X,15,1H),2X,6E13,5)                                         7590
1020 FORMAT (/5X,35H LINEAR CONSTRAINT IDENTIFIERS (ISC)/5X,36H NON-ZERO   7600
1 INDICATES LINEAR CONSTRAINT)                                           7610
1030 FORMAT (3X,15,1H),2X,15I5)                                           7620
1040 FORMAT (/5X,26H ALL CONSTRAINTS ARE LINEAR)                          7630
1050 FORMAT (/5X,30H ALL CONSTRAINTS ARE NON-LINEAR)                      7640
1060 FORMAT (/5X,9H THERE ARE,15,19H ACTIVE CONSTRAINTS)                 7650
1070 FORMAT (5X,22H CONSTRAINT NUMBERS ARE)                               7660
1080 FORMAT (/5X,9H THERE ARE,15,21H VIOLATED CONSTRAINTS)               7670
1090 FORMAT (/5X,9H THERE ARE,15,24H ACTIVE SIDE CONSTRAINTS)             7680
1100 FORMAT (5X,43H DECISION VARIABLES AT LOWER OR UPPER BOUNDS,30H (MIN   7690
1 US INDICATES LOWER BOUND))                                              7700
1110 FORMAT (/5X,22H ONE-DIMENSIONAL SEARCH/5X,15H INITIAL SLOPE =,E12.4,  7710
12X,16H PROPOSED ALPHA =,E12.4)                                           7720
1120 FORMAT (//,5X,35H * CONMIN DETECTS VLB(I).GT,VUB(I)/5X,57H FIX IS    7730
1 SET X(I)=VLR(I)=VUB(I) =.5*(VLB(I)+VUB(I)) FOR I =,15)                 7740
1130 FORMAT (//,5X,41H * CONMIN DETECTS INITIAL X(I).LT,VLB(I)/5X,6HX(    7750
1 I) =,E12.4,2X,8H VLB(I) =,E12.4/5X,35HX(I) IS SET EQUAL TO VLB(I) F    7760
2 OR I =,15)                                                              7770
1140 FORMAT (//,5X,41H * CONMIN DETECTS INITIAL X(I).GT,VUB(I)/5X,6HX(    7780
1 I) =,E12.4,2X,8H VUB(I) =,E12.4/5X,35HX(I) IS SET EQUAL TO VUB(I) F    7790
2 OR I =,15)                                                              7800
1150 FORMAT (/5X,21H TERMINATION CRITERION)                               7810
1160 FORMAT (10X,17H ITER EQUALS ITMAX)                                    7820
1170 FORMAT (10X,62H TEN CONSECUTIVE ITERATIONS FAILED TO PRODUCE A FEAS   7830
1 BLE DESIGN)                                                             7840
1190 FORMAT (10X,43H ABS(1-OBJ(I-1)/OBJ(I)) LESS THAN DELFUN FOR,13,11H    7850
1 ITERATIONS)                                                            7860
1200 FORMAT (10X,43H ABS(OBJ(I)-OBJ(I-1)) LESS THAN DABFUN FOR,13,11H     7870
1 ITERATIONS)                                                            7880
1210 FORMAT (/5X,22H NUMBER OF ITERATIONS =,15)                           7890
1220 FORMAT (/5X,24H CONSTRAINT PARAMETER, BETA =,E14.5)                  7900
1230 FORMAT (1H1,///12X,27(2H* )/12X,1H*,51X,1H*/12X,1H*,20X,11H C O N    7910
1 H I N,20X,1H*/12X,1H*,51X,1H*/12X,1H*,15X,21H FORTRAN PROGRAM FOR     7920
2,15X,1H*/12X,1H*,51X,1H*/12X,1H*,9X,33H CONSTRAINED FUNCTION MINIMI    7930
3 ZATION,9X,1H*/12X,1H*,51X,1H*/12X,1H*,2X,48H NASA/AMES RESEARCH CEN    7940
4 TER, MOFFETT FIELD, CALIF.,1X,1H*/12X,1H*,51X,1H*/12X,1H*,13X,25H V    7950
S E R S I O N  I I      J U L Y , 1 9 7 5 , 1 3 X , 1 H * / 1 2 X , 1 H * , 5 1 X , 1 H * / 1 2 X , 2 7 ( 2 H * ) ) 7960
1240 FORMAT (//,5X,33H CONSTRAINED FUNCTION MINIMIZATION//5X,18H CONTROL  7970
1 PARAMETERS)                                                            7980
1250 FORMAT (/5X,60H PRINT NOV      ITMAX      NCON      NSIDE ICNDR      NSC  7990
1 AL      NFDG/818//5X,12H LNOBJ      ITRM,5X,2HN1,6X,2HN2,6X,2HN3,6X,2HN4, 8000

```



## SUBROUTINE CONMIN - CONSTRAINED FUNCTION MINIMIZATION

SEPT. 77

```

26X,2HN5/87A)
1260 FORMAT (/9X,4HFDCH,12X,5HFDCHM/3X,2E14.5) 8010
1270 FORMAT (/9X,2HCT,14X,5HCTMIN,11X,3HCTL,13X,6HCTLMIN/1X,4(2X,E14.5) 8020
1//9X,5HTHETA,11X,3HPhi,13X,6HDELFIN,10X,6HDABFUN/1X,4(2X,E14.5)) 8030
1280 FORMAT (/5X,40HLOWER BOUNDS ON DECISION VARIABLES (VLB)) 8040
1290 FORMAT (/5X,40HUPPER BOUNDS ON DECISION VARIABLES (VUB)) 8050
1300 FORMAT (///5X,35HUNCONSTRAINED FUNCTION MINIMIZATION//5X,18HCONTR 8060
10L PARAMETERS) 8070
1310 FORMAT (/5X,21HSCALING VECTOR (SCAL)) 8080
1320 FORMAT (///5X,22HBEGIN ITERATION NUMBER,IS) 8090
1330 FORMAT (/5X,4HCT =,E14.5,5X,5HCTL =,E14.5,5X,5HPhi =,E14.5) 8100
1340 FORMAT (/5X,25HNEW SCALING VECTOR (SCAL)) 8110
1350 FORMAT (/5X,15HGRADIENT OF OBJ) 8120
1360 FORMAT (/5X,40HGRADIENTS OF ACTIVE AND VIOLATED CONSTRAINTS) 8130
1370 FORMAT (14) 8140
1380 FORMAT (/5X,37HPUSH-OFF FACTORS, (THETA(I), I=1,NAC)) 8150
1390 FORMAT (/5X,27HSEARCH DIRECTION (S=VECTOR)) 8160
1400 FORMAT (/5X,18HCALCULATED ALPHA =,E14.5) 8170
1410 FORMAT (///5X,6HITER =,IS,5X,5HOBJ =,E14.5,5X,16HNO CHANGE IN OBJ 8180
1) 8190
1420 FORMAT (/5X,5HOBJ =,E15.6,5X,16HNO CHANGE ON OBJ) 8200
1430 FORMAT (/5X,5HOBJ =,E15.6) 8210
1440 FORMAT (///5X,6HITER =,IS,5X,5HOBJ =,E14.5) 8220
1450 FORMAT (/5X,28HINITIAL FUNCTION INFORMATION//5X,5HOBJ =,E15.6) 8230
1460 FORMAT (/5X,29HDECISION VARIABLES (X=VECTOR)) 8240
1470 FORMAT (3Y,7E13.4) 8250
1480 FORMAT (/5X,28HCUNSTRAINT VALUES (G=VECTOR)) 8260
1490 FORMAT (5X,15I5) 8270
1500 FORMAT (/5X,59HTHE NUMBER OF ACTIVE AND VIOLATED CONSTRAINTS EXCEE 8280
1DS N3-1,5X,66HDIMENSIONED SIZE OF MATRICES A AND B AND VECTOR IC 8290
2IS INSUFFICIENT/5X,61HOPTIMIZATION TERMINATED AND CONTROL RETURNED 8300
3 TO MAIN PROGRAM.) 8310
1510 FORMAT (141,///4X,30HFINAL OPTIMIZATION INFORMATION) 8320
1520 FORMAT (/5X,32HOBJECTIVE FUNCTION WAS EVALUATED,8X,IS,2X,5HTIMES) 8330
1530 FORMAT (/5X,35HCUNSTRAINT FUNCTIONS WERE EVALUATED,110,2X,5HTIMES) 8340
1540 FORMAT (/5X,36HGRADIENT OF OBJECTIVE WAS CALCULATED,19,2X,5HTIMES) 8350
1550 FORMAT (/5X,40HGRADIENTS OF CONSTRAINTS WERE CALCULATED,15,2X,5HTI 8360
MES) 8370
END 8380
8390

```

## SUBROUTINE CNMNO1

SEPT. 77

```

SUBROUTINE CNMNO1 (JGOTO,X,DF,G,ISC,IC,A,G1,VLB,VUB,SCAL,C,NCAL,DX
1,DX1,FI,XI,II,N1,N2,N3,N4)
COMMON /CNMNO1/ IPRINT,NDV,ITHAX,NCON,NSIDE,ICNDR,NSCAL,NFDG,FDCH,
IFOCMM,CT,CTMIN,CTL,CTLMIN,THETA,PHI,NAC,DELFUN,DARFUN,LINOBJ,ITRM,
2ITER,INFOR,IGOTO,INFO,OHJ
DIMENSION X(N1),DF(N1),G(N2),ISC(N2),IC(N3),A(N3,N1),G1(N2),VLB(N1
1),VUB(N1),SCAL(N1),NCAL(2),C(N4)
C ROUTINE TO CALCULATE GRADIENT INFORMATION BY FINITE DIFFERENCE.
C BY G. N. VANDERPLAATS JUNE, 1972.
C NASA-AMES RESEARCH CENTER, MOFFETT FIELD, CALIF.
IF (JGOTO,EQ.1) GO TO 10
IF (JGOTO,EQ.2) GO TO 70
INFOG=0
INF=INFO
NAC=0
IF (LINOBJ,NE.0,AND,ITER,GT.1) GO TO 10
C -----
C GRADIENT OF LINEAR OBJECTIVE
C -----
IF (NFDG,EQ.1) JGOTO=1
IF (NFDG,EQ.1) RETURN
10 CONTINUE
JGOTO=0
IF (NFDG,EQ.1,AND,NCON,EQ.0) RETURN
IF (NCON,EQ.0) GO TO 40
C -----
C * * * DETERMINE WHICH CONSTRAINTS ARE ACTIVE OR VIOLATED * * *
C -----
DO 20 I=1,NCON
IF (G(I),LT,CT) GO TO 20
IF (ISC(I),GT.0,AND,G(I),LT,CTL) GO TO 20
NAC=NAC+1
IF (NAC,GE,N3) RETURN
IC(NAC)=I
20 CONTINUE
IF (NFDG,EQ.1,AND,NAC,EQ.0) RETURN
IF ((LINOBJ,GT.0,AND,ITER,GT.1),AND,NAC,EQ.0) RETURN
C -----
C STORE VALUES OF CONSTRAINTS IN G1
C -----
DO 30 I=1,NCON
G1(I)=G(I)
30 CONTINUE
40 JGOTO=0
IF (NAC,EQ.0,AND,NFDG,EQ.1) RETURN
C -----
C CALCULATE GRADIENTS
C -----
INFOG=1
INFO=1

```



## SUBROUTINE CNMNO1

SEPT. 77

	FI=OBJ	510
	III=0	520
50	III=III+1	530
	XI=X(III)	540
	DX=FDCH*X <sub>i</sub>	550
	DX=ABS(DX)	560
	FDCH1=FDCHM	570
	IF (NSCAL.NE.0) FDCH1=FDCHM/SCAL(III)	580
	IF (DX.LT.FDCH1) DX=FDCH1	590
	XI=XI+DX	600
	IF (NSIDE.EQ.0) GO TO 60	610
	IF (X1.LT.VLB(III).AND.DX.LT.0.) XI=XI-DX	620
	IF (X1.GT.VUB(III).AND.DX.GT.0.) XI=XI-DX	630
60	DX1=1./DX	640
	X(III)=XI*DX	650
	NCAL(1)=NCAL(1)+1	660
C	-----	670
C	FUNCTION EVALUATION	680
C	-----	690
	JGOTO=2	700
	RETURN	710
70	CONTINUE	720
	X(III)=XI	730
	IF (NFDG.EQ.0) DF(III)=DX1*(OBJ-FI)	740
	IF (NAC.EQ.0) GO TO 90	750
C	-----	760
C	DETERMINE GRADIENT COMPONENTS OF ACTIVE CONSTRAINTS	770
C	-----	780
	DO 80 J=1,NAC	790
	I1=IC(J)	800
80	A(J,III)=DX1*(G(I1)-G1(I1))	810
90	CONTINUE	820
	IF (III.LT.NDV) GO TO 50	830
	INFOG=0	840
	INFO=INF	850
	JGOTO=0	860
	OBJ=FI	870
	IF (NCON.EQ.0) RETURN	880
C	-----	890
C	STORE CURRENT CONSTRAINT VALUES BACK IN G-VECTOR	900
C	-----	910
	DO 100 I=1,NCON	920
100	G(I)=G1(I)	930
	RETURN	940
	END	950

## SUBROUTINE CNMN02

SEPT. 77

```

SUBROUTINE CNMN02 (NCALC,SLOPE,DFTDF1,DF,S,N1)
COMMON /CNMN1/ IPRINT,NDV,ITMAX,NCON,NSIDE,ICNOIR,NSCAL,NFOG,FDCH,
IFDCHM,CT,CTMIN,CTL,CTLMIN,THETA,PHI,NAC,DELFUN,DABFUN,LINOBJ,ITHM,
ZITER,INFOG,IGOTO,INFO,OBJ
DIMENSION DF(N1),S(N1)
ROUTINE TO DETERMINE CONJUGATE DIRECTION VECTOR OR DIRECTION
OF STEEPEST DESCENT FOR UNCONSTRAINED FUNCTION MINIMIZATION.
BY G. N. VANDERPLAATS APRIL, 1972.
NASA-AMES RESEARCH CENTER, MOFFETT FIELD, CALIF.
NCALC = CALCULATION CONTROL.
NCALC = 0, S = STEEPEST DESCENT.
NCALC = 1, S = CONJUGATE DIRECTION.
CONJUGATE DIRECTION IS FOUND BY FLETCHER-REEVES ALGORITHM.
-----
CALCULATE NORM OF GRADIENT VECTOR
-----
DFTDF=0.
DO 10 I=1,NDV
DFI=DF(I)
DFTDF=DFTDF+DFI*DFI
-----
FIND DIRECTION S
-----
IF (NCALC.NE.1) GO TO 30
IF (DFTDF.LT.1.0E-20) GO TO 30
-----
FIND FLETCHER-REEVES CONJUGATE DIRECTION
-----
BETA=DFTDF/DFTDF1
SLOPE=0.
DO 20 I=1,NDV
DFI=DF(I)
SI=BETA*S(I)-DFI
SLOPE=SLOPE+SI*DFI
20 S(I)=SI
GO TO 50
30 CONTINUE
NCALC=0
-----
CALCULATE DIRECTION OF STEEPEST DESCENT
-----
DO 40 I=1,NDV
S(I)=-DF(I)
SLOPE=-DFTDF
50 CONTINUE
-----
NORMALIZE S TO MAX ABS VALUE OF UNITY
-----
SI=0.
DO 60 I=1,NDV

```



SUBROUTINE CMN02

SEPT. 77

```

S2=ABS(S(I))
IF (S2.GT.S1) S1=S2
60 CONTINUE
IF (S1.LT.1.0E-20) S1=1.0E-20
S1=1./S1
DFTDF1=DFTDF*S1
70 DO 70 I=1,N
S(I)=S1*S(I)
SLOPE=S1*SLOPE
RETURN
END

```

510  
520  
530  
540  
550  
560  
570  
580  
590  
600  
610

## SUBROUTINE CNMN03

SEPT. 77

```

SUBROUTINE CNMN03 (X,S,DF,G,A,IC,SCAL,C,N1,N2,N3,N4)      10
COMMON /CNMNI/ IPRINT,NDV,ITMAX,NCON,NSIDE,ICNDIR,NSCAL,NFDG,FDCH,  20
IFDCHM,CT,CTMIN,CTL,CTLMIN,THETA,PHI,NAC,DELFUN,DARFUN,LINOBJ,ITRM,  30
ZITER,INFO,IGOTO,INFO,OBJ
COMMON /CONSAR/ D1(20),SLOPE,D2(3),XI,D3,                40
ZALP,FFF,A1,A2,A3,A4,F1,F2,F3,F4,D4(4),APP,              50
D5(8),RSPACE,ID1(6),KCOUNT,NCAL(2),ID2(4),KOUNT,ID3(8),  60
JGOTO,ISPACE(2)                                           70
DIMENSION X(N1),S(N1),DF(N1),G(N2),A(N3,N1),IC(N3),SCAL(N1),C(N4)  80
ROUTINE TO SOLVE ONE-DIMENSIONAL SEARCH IN UNCONSTRAINED  90
MINIMIZATION USING 2-POINT QUADRATIC INTERPOLATION, 3-POINT 100
CUBIC INTERPOLATION AND 4-POINT CUBIC INTERPOLATION.      110
BY G. N. VANDERPLAATS                                     120
APRIL, 1972.                                              130
NASA-AMES RESEARCH CENTER, MOFFETT FIELD, CALIF.        140
ALP = PROPOSED MOVE PARAMETER.                            150
SLOPE = INITIAL FUNCTION SLOPE = S-TRANSPOSE TIMES DF.  160
SLOPE MUST BE NEGATIVE.                                   170
OBJ = INITIAL FUNCTION VALUE.                             180
ZRO=0.                                                     190
IF (JGOTO.EQ.0) GO TO 10                                  200
GO TO (50,80,110,140,180,220,270), JGOTO                 210
-----
INITIAL INFORMATION (ALPHA=0)                             220
-----
IF (SLOPE.LT.0.) GO TO 20                                 250
ALP=0.                                                     260
RETURN                                                     270
CONTINUE                                                  280
IF (IPRINT.GT.4) WRITE (6,360)                           290
FFF=OBJ                                                    300
AP1=0.                                                     310
A1=0.                                                      320
F1=OBJ                                                     330
A2=ALP                                                     340
A3=0.                                                      350
F3=0.                                                      360
AP=A2                                                      370
KOUNT=0                                                    380
-----
MOVE A DISTANCE AP*S AND UPDATE FUNCTION VALUE          400
-----
CONTINUE                                                  410
KOUNT=KOUNT+1                                             420
DO 40 I=1,NDV                                            430
X(I)=X(I)+AP*S(I)                                         440
IF (IPRINT.GT.4) WRITE (6,370) AP                        450
IF (IPRINT.GT.4) WRITE (6,380) (X(I),I=1,NDV)           460
NCAL(1)=NCAL(1)+1                                         470
JGOTO=1                                                   480
RETURN                                                    490

```



## SUBROUTINE CNMN03

SEPT. 77

50	CONTINUE	510
	F2=OBJ	520
	IF (IPRINT.GT.4) WRITE (6,390) F2	530
	IF (F2.LT.F1) GO TO 120	540
C	-----	550
C	CHECK FOR ILL-CONDITIONING	560
C	-----	570
	IF (KOUNT.GT.5) GO TO 60	580
	FF=2.*ABS(F1)	590
	IF (F2.LT.FF) GO TO 90	600
	FF=5.*ABS(F1)	610
	IF (F2.LT.FF) GO TO 60	620
	A2=.5*A2	630
	AP=A2	640
	ALP=A2	650
	GO TO 30	660
60	F3=F2	670
	A3=A2	680
	A2=.5*A2	690
C	-----	700
C	UPDATE DESIGN VECTOR AND FUNCTION VALUE	710
C	-----	720
	AP=A2-ALP	730
	ALP=A2	740
	DO 70 I=1,NDV	750
70	X(I)=X(I)+AP*S(I)	760
	IF (IPRINT.GT.4) WRITE (6,370) A2	770
	IF (IPRINT.GT.4) WRITE (6,380) (X(I),I=1,NDV)	780
	NCAL(1)=NCAL(1)+1	790
	JGOTO=2	800
	RETURN	810
80	CONTINUE	820
	F2=OBJ	830
	IF (IPRINT.GT.4) WRITE (6,390) F2	840
C	PROCEED TO CUBIC INTERPOLATION.	850
	GO TO 160	860
90	CONTINUE	870
C	-----	880
C	***** 2-POINT QUADRATIC INTERPOLATION *****	890
C	-----	900
	JJ=1	910
	II=1	920
	CALL CNMN04 (II,APP,ZRO,A1,F1,SLOPE,A2,F2,ZRO,ZRO,ZRO,ZRO)	930
	IF (APP.LT.ZRO.OR.APP.GT.A2) GO TO 120	940
	F3=F2	950
	A3=A2	960
	A2=APP	970
	JJ=0	980
C	-----	990
C	UPDATE DESIGN VECTOR AND FUNCTION VALUE	1000

## SUBROUTINE CNMN03

SEPT. 77

```

C -----
C AP=A2-ALP
C ALP=A2
C DO 100 I=1,NDV
100 X(I)=X(I)+AP*S(I)
C IF (IPRINT.GT.4) WRITE (6,370) A2
C IF (IPRINT.GT.4) WRITE (6,380) (X(I),I=1,NDV)
C NCAL(1)=NCAL(1)+1
C JGOTO=3
C RETURN
110 CONTINUE
C F2=OBJ
C IF (IPRINT.GT.4) WRITE (6,390) F2
C GO TO 150
120 A3=2.*A2
C -----
C UPDATE DESIGN VECTOR AND FUNCTION VALUE
C -----
C AP=A3-ALP
C ALP=A3
C DO 130 I=1,NDV
130 X(I)=X(I)+AP*S(I)
C IF (IPRINT.GT.4) WRITE (6,370) A3
C IF (IPRINT.GT.4) WRITE (6,380) (X(I),I=1,NDV)
C NCAL(1)=NCAL(1)+1
C JGOTO=4
C RETURN
140 CONTINUE
C F3=OBJ
C IF (IPRINT.GT.4) WRITE (6,390) F3
150 CONTINUE
C IF (F3.LT.F2) GO TO 190
160 CONTINUE
C -----
C ***** 3-POINT CUBIC INTERPOLATION *****
C -----
C II=3
C CALL CNMN04 (II,APP,ZRO,A1,F1,SLOPE,A2,F2,A3,F3,ZRO,ZRU)
C IF (APP.LT.ZRO.OR.APP.GT.A3) GO TO 190
C -----
C UPDATE DESIGN VECTOR AND FUNCTION VALUE.
C -----
C AP=APP
C AP=APP-ALP
C ALP=APP
C DO 170 I=1,NDV
170 X(I)=X(I)+AP*S(I)
C IF (IPRINT.GT.4) WRITE (6,370) ALP
C IF (IPRINT.GT.4) WRITE (6,380) (X(I),I=1,NDV)
C NCAL(1)=NCAL(1)+1

```



## SUBROUTINE CNHM03

SEPT. 77

	JGOTO=5	1510
	RETURN	1520
180	CONTINUE	1530
	IF (IPRINT.GT.4) WRITE (6,390) OBJ	1540
C	-----	1550
C	CHECK CONVERGENCE	1560
C	-----	1570
	AA=1.-APP/A2	1580
	AB2=ABS(F2)	1590
	AB3=ABS(Obj)	1600
	AB=AB2	1610
	IF (AB3.GT.AB) AB=AB3	1620
	IF (AB.LT.1.0E-15) AB=1.0E-15	1630
	AB=(AB2-AB3)/AB	1640
	IF (ABS(AA).LT.1.0E-15.AND.ABS(AA).LT..001) GO TO 330	1650
	A4=A3	1660
	F4=F3	1670
	A3=APP	1680
	F3=Obj	1690
	IF (A3.GT.A2) GO TO 230	1700
	A3=A2	1710
	F3=F2	1720
	A2=APP	1730
	F2=Obj	1740
	GO TO 230	1750
190	CONTINUE	1760
C	-----	1770
C	***** 4-POINT CUBIC INTERPOLATION *****	1780
C	-----	1790
200	CONTINUE	1800
	A4=2.*A3	1810
C	UPDATE DESIGN VECTOR AND FUNCTION VALUE.	1820
	AP=A4-ALP	1830
	ALP=A4	1840
	DO 210 I=1,NDV	1850
210	X(I)=X(I)+AP*S(I)	1860
	IF (IPRINT.GT.4) WRITE (6,370) ALP	1870
	IF (IPRINT.GT.4) WRITE (6,380) (X(I),I=1,NDV)	1880
	NCAL(1)=NCAL(1)+1	1890
	JGOTO=6	1900
	RETURN	1910
220	CONTINUE	1920
	F4=Obj	1930
	IF (IPRINT.GT.4) WRITE (6,390) F4	1940
	IF (F4.GT.F3) GO TO 230	1950
	A1=A2	1960
	F1=F2	1970
	A2=A3	1980
	F2=F3	1990
	A3=A4	2000

## SUBROUTINE CNMN03

SEPT. 77

	F3=F4	2010
	GO TO 200	2020
230	CONTINUE	2030
	II=4	2040
	CALL CNMNO3 (II,APP,A1,A1,F1,SLOPE,A2,F2,A3,F3,A4,F4)	2050
	IF (APP.GT.A1) GO TO 250	2060
	AP=A1-ALP	2070
	ALP=A1	2080
	OBJ=F1	2090
	DO 240 I=1,NDV	2100
240	X(I)=X(I)+AP*S(I)	2110
	GO TO 280	2120
250	CONTINUE	2130
C	-----	2140
C	UPDATE DESIGN VECTOR AND FUNCTION VALUE	2150
C	-----	2160
	AP=APP-ALP	2170
	ALP=APP	2180
	DO 260 I=1,NDV	2190
260	X(I)=X(I)+AP*S(I)	2200
	IF (IPRINT.GT.4) WRITE (6,370) ALP	2210
	IF (IPRINT.GT.4) WRITE (6,380) (X(I),I=1,NDV)	2220
	NCAL(1)=NCAL(1)+1	2230
	JGOTO=7	2240
	RETURN	2250
270	CONTINUE	2260
	IF (IPRINT.GT.4) WRITE (6,390) OBJ	2270
280	CONTINUE	2280
C	-----	2290
C	CHECK FOR ILL-CONDITIONING	2300
C	-----	2310
	IF (OBJ.GT.F2.OR.OBJ.GT.F3) GO TO 290	2320
	IF (OBJ.LF.F1) GO TO 330	2330
	AP=A1-ALP	2340
	ALP=A1	2350
	OBJ=F1	2360
	GO TO 310	2370
290	CONTINUE	2380
	IF (F2.LT.F3) GO TO 300	2390
	OBJ=F3	2400
	AP=A3-ALP	2410
	ALP=A3	2420
	GO TO 310	2430
300	OBJ=F2	2440
	AP=A2-ALP	2450
	ALP=A2	2460
310	CONTINUE	2470
C	-----	2480
C	UPDATE DESIGN VECTOR	2490
C	-----	2500



## SUBROUTINE CNMN03

SEPT. 77

	DO 320 I=1,NDV	2510
320	X(I)=X(I)+AP*S(I)	2520
330	CONTINUE	2530
C	-----	2540
C	CHECK FOR MULTIPLE MINIMA	2550
C	-----	2560
	IF (OBJ.LF.FFF) GO TO 350	2570
C	INITIAL FUNCTION IS MINIMUM.	2580
	DO 340 I=1,NDV	2590
340	X(I)=X(I)-ALP*S(I)	2600
	ALP=0.	2610
	OBJ=FFF	2620
350	CONTINUE	2630
	JGOTO=0	2640
	RETURN	2650
C	-----	2660
C	FORMATS	2670
C	-----	2680
C	-----	2690
360	FORMAT (////5X,60H* * * UNCONSTRAINED ONE-DIMENSIONAL SEARCH INFO	2700
	IRMATION * * *)	2710
370	FORMAT (/5X,7HALPHA =,E14,5/5X,8HX=VECTOR)	2720
380	FORMAT (5X,6E13,5)	2730
390	FORMAT (/5X,5H0BJ =,E14,5)	2740
	END	2750

## SUBROUTINE CNMN04

SEPT. 77

```

C SUBROUTINE CNMN04 (II,XBAR,EPS,X1,Y1,SLOPE,X2,Y2,X3,Y3,X4,Y4) 10
C ROUTINE TO FIND FIRST XBAR,GE,EPS CORRESPONDING TO A MINIMUM 20
C OF A ONE-DIMENSIONAL REAL FUNCTION BY POLYNOMIAL INTERPOLATION. 30
C BY G. N. VANDERPLAATS APRIL, 1972. 40
C NASA-AMES RESEARCH CENTER, MOFFETT FIELD, CALIF. 50
C 60
C II = CALCULATION CONTROL. 70
C 1: 2-POINT QUADRATIC INTERPOLATION, GIVEN X1, Y1, SLOPE, 80
C X2 AND Y2. 90
C 2: 3-POINT QUADRATIC INTERPOLATION, GIVEN X1, Y1, X2, Y2, 100
C X3 AND Y3. 110
C 3: 3-POINT CUBIC INTERPOLATION, GIVEN X1, Y1, SLOPE, X2, Y2, 120
C X3 AND Y3. 130
C 4: 4-POINT CURIC INTERPOLATION, GIVEN X1, Y1, X2, Y2, X3, 140
C Y3, X4 AND Y4. 150
C EPS MAY BE NEGATIVE. 160
C IF REQUIRED MINIMUM ON Y DOES NOT EXIST, OR THE FUNCTION IS 170
C ILL-CONDITIONED, XBAR = EPS-1.0 WILL BE RETURNED AS AN ERROR 180
C INDICATOR. 190
C IF DESIRED INTERPOLATION IS ILL-CONDITIONED, A LOWER ORDER 200
C INTERPOLATION, CONSISTANT WITH INPUT DATA, WILL BE ATTEMPTED, 210
C AND II WILL BE CHANGED ACCORDINGLY. 220
C XBAR1=EPS-1. 230
C XBAR=XBAR1 240
C X21=X2-X1 250
C IF (ABS(X21).LT.1.0E-20) RETURN 260
C NSLOP=MOD(II,2) 270
C GO TO (10,20,40,50), II 280
10 CONTINUE 290
C ----- 300
C II=1: 2-POINT QUADRATIC INTERPOLATION 310
C ----- 320
C II=1 330
C DX=X1-X2 340
C IF (ABS(DX).LT.1.0E-20) RETURN 350
C AA=(SLOPE*(Y2-Y1)/DX)/DX 360
C IF (AA.LT.1.0E-20) RETURN 370
C BB=SLOPE-2.*AA*X1 380
C XBAR=-.5*BB/AA 390
C IF (XBAR.LT.EPS) XBAR=XBAR1 400
C RETURN 410
20 CONTINUE 420
C ----- 430
C II=2: 3-POINT QUADRATIC INTERPOLATION 440
C ----- 450
C II=2 460
C X21=X2-X1 470
C X31=X3-X1 480
C X32=X3-X2 490
C QQ=X21*X31-X32 500

```



## SUBROUTINE CNMNO4

SEPT. 77

```

IF (ABS(QQ),LT.1.0E-20) RETURN 510
AA=(Y1*X3-Y2*X31+Y3*X21)/QQ 520
IF (AA,LT.1.0E-20) GO TO 30 530
BB=(Y2-Y1)/X21-AA*(X1+X2) 540
XBAR=-.5*RR/AA 550
IF (XBAR,LT.EPS) XBAR=XBAR1 560
RETURN 570
30 CONTINUE 580
IF (NSLOP,EQ.0) RETURN 590
GO TO 10 600
40 CONTINUE 610
C ----- 620
C II=3: 3-POINT CUBIC INTERPOLATION 630
C ----- 640
II=3 650
X21=X2-X1 660
X31=X3-X1 670
X32=X3-X2 680
QQ=X21*X31*X32 690
IF (ABS(QQ),LT.1.0E-20) RETURN 700
X11=X1-X1 710
DNOM=X2*X31-X11*X32-X3*X3*X21 720
IF (ABS(DNOM),LT.1.0E-20) GO TO 20 730
AA=((X31*X31*(Y2-Y1)-X21*X21*(Y3-Y1))/(X31*X21)-SLOPE*X32)/DNOM 740
IF (ABS(AA),LT.1.0E-20) GO TO 20 750
BB=((Y2-Y1)/X21-SLOPE-AA*(X2*X2+X1*X2-2.*X11))/X21 760
CC=SLOPE-.5*AA*X11-2.*BB*X1 770
BAC=BB+BB-.3*AA*CC 780
IF (BAC,LT.0.) GO TO 20 790
BAC=SQRT(BAC) 800
XBAR=(BAC-BB)/(3.*AA) 810
IF (XBAR,LT.EPS) XBAR=EPS 820
RETURN 830
50 CONTINUE 840
C ----- 850
C II=4: 4-POINT CUBIC INTERPOLATION 860
C ----- 870
X21=X2-X1 880
X31=X3-X1 890
X41=X4-X1 900
X32=X3-X2 910
X42=X4-X2 920
X11=X1-X1 930
X22=X2-X2 940
X33=X3-X3 950
X44=X4-X4 960
X111=X1*X11 970
X222=X2*X22 980
Q2=X31*X21*X32 990
IF (ABS(Q2),LT.1.0E-30) RETURN 1000

```

## SUBROUTINE CNMN04

SEPT. 77

```

Q1=X111*X32-X222*X31+X3*X33*X21      1010
Q4=X111*X42-X222*X41+X4*X44*X21      1020
Q5=X41*X21+X42      1030
DNOM=Q2*Q4-Q1*Q5      1040
IF (ABS(DNOM).LT.1.0E-30) GO TO 60      1050
Q3=Y3*X21-Y2*X31+Y1*X32      1060
Q6=Y4*X21-Y2*X41+Y1*X42      1070
AA=(Q2*Q6-Q3*Q5)/DNOM      1080
BB=(Q3-Q1*AA)/Q2      1090
CC=(Y2-Y1-AA*(X222-X111))/X21-BB*(X1+X2)      1100
BAC=BB-BB-3.*AA*CC      1110
IF (ABS(AA).LT.1.0E-20.OR.BAC.LT.0.) GO TO 60      1120
BAC=SQRT(BAC)      1130
XBAR=(BAC-BB)/(3.*AA)      1140
IF (XBAR.LE.T.EPS) XBAR=XBAR1      1150
RETURN      1160
60 CONTINUE      1170
IF (NSLOP.EQ.1) GO TO 40      1180
GO TO 20      1190
END      1200

```



## SUBROUTINE CNMN05

SEPT. 77

```

SUBROUTINE CNMN05 (NVC,SLOPE,DF,G,ISC,IC,A,S,C,MS1,B,N1,N2,N3,N4,N
15)
COMMON /CNMN1/ IPRINT,NDV,ITMAX,NCON,NSIDE,ICNDIR,NSCAL,NFDG,FDCH,
1FDCHM,CT,CTMIN,CTL,CTLMIN,THETA,PHI,NAC,DELFUN,DARFUN,LINOBJ,ITRM,
2ITER,INFO,IGOTO,INFO,OBJ
DIMENSION DF(N1),G(N2),ISC(N2),IC(N3),A(N3,N1),S(N1),C(N4),MS1(N5)
1,B(N3,N3)
ROUTINE TO SOLVE DIRECTION FINDING PROBLEM IN MODIFIED METHOD OF
C FEASIBLE DIRECTIONS.
C BY G. N. VANDERPLAATS MAY, 1972.
C NASA-AMES RESEARCH CENTER, MOFFETT FIELD, CALIF.
C NORM OF S VECTOR USED HERE IS S-TRANSPOSE TIMES S, LE. 1.
C IF NVC = 0 FIND DIRECTION BY ZOUTENDIJK'S METHOD, OTHERWISE
C FIND MODIFIED DIRECTION.
C -----
C *** NORMALIZE GRADIENTS, CALCULATE THETA'S AND DETERMINE NVC ***
C -----
NDV1=NDV+1
NDV2=NDV+2
NAC1=NAC+1
NVC=0
THMAX=0.
CTA=ABS(CT)
CT1=1./CTA
CTAM=ABS(CTMIN)
CTR=ABS(CTL)
CT2=1./CTR
CTBM=ABS(CTLMIN)
A1=1.
DO 40 I=1,NAC
C CALCULATE THETA
NCI=IC(I)
NCJ=1
IF (NCI,LF,NCON) NCJ=ISC(NCI)
C1=G(NCI)
CTD=CT1
CTC=CTAM
IF (NCJ,LF,0) GO TO 10
CTC=CTBM
CTD=CT2
10 IF (C1,GT,CTC) NVC=NVC+1
THT=0.
GG=1.+CTD*C1
IF (NCJ,EQ,0.OR,C1,GT,CTC) THT=THETA*GG*GG
IF (NCJ,GT,0.AND,C1,GT,CTC) THT=THT-3.*THETA
IF (THT,GT,50.) THT=50.
IF (THT,GT,THMAX) THMAX=THT
A(I,NOV1)=THT
C -----
C NORMALIZE GRADIENTS OF CONSTRAINTS

```

## SUBROUTINE CNMN05

SEP. 77

```

C ----- 510
A(I,NDV2)=1. 520
IF (NCI,GT,NCON) GO TO 40 530
A1=0. 540
DO 20 J=1,NDV 550
A1=A1+A(I,J)**2 560
20 CONTINUE 570
IF (A1,LT,1.0E-20) A1=1.0E-20 580
A1=SQRT(A1) 590
A(I,NDV2)=A1 600
A1=1./A1 610
DO 30 J=1,NDV 620
A(I,J)=A1*A(I,J) 630
30 CONTINUE 640
C ----- 650
C NORMALIZE GRADIENT OF OBJECTIVE FUNCTION AND STORE IN NAC+1 660
C ROW OF A 670
C ----- 680
A1=0. 690
DO 50 I=1,NDV 700
A1=A1+DF(I)**2 710
50 CONTINUE 720
IF (A1,LT,1.0E-20) A1=1.0E-20 730
A1=SQRT(A1) 740
A1=1./A1 750
DO 60 I=1,NDV 760
A(NAC1,I)=A1*DF(I) 770
60 BUILD C VECTOR. 780
IF (NVC,GT,0) GO TO 80 790
C ----- 800
C BUILD C FOR CLASSICAL METHOD 810
C ----- 820
NDB=NAC1 830
A(NDB,NDV1)=1. 840
DO 70 I=1,NDB 850
C(I)=-A(I,NDV1) 860
70 GO TO 110 870
80 CONTINUE 880
C ----- 890
C BUILD C FOR MODIFIED METHOD 900
C ----- 910
NDB=NAC 920
A(NAC1,NDV1)=-PHI 930
C ----- 940
C SCALE THETA'S SO THAT MAXIMUM THETA IS UNITY 950
C ----- 960
IF (THMAX,GT,0.00001) THMAX=1./THMAX 970
DO 90 I=1,NDB 980
NCI=IC(I) 990
CI=CTA 1000

```



## SUBROUTINE CNMN05

SEPT. 77

```

      IF (ISC(NC1),GT,0) C1=CTB                      1010
      A(I,NDV1)=A(I,NDV1)+THMAX                      1020
90    CONTINUE                                         1030
      DO 100 I=1,NDB                                1040
      C(I)=0.                                          1050
      DO 100 J=1,NDV1                                1060
100    C(I)=C(I)+A(I,J)*A(NAC1,J)                    1070
110    CONTINUE                                         1080
C      -----                                         1090
C      BUILD B MATRIX                                  1100
C      -----                                         1110
      DO 120 I=1,NDB                                  1120
      DO 120 J=1,NDB                                  1130
      B(I,J)=0.                                       1140
      DO 120 K=1,NDV1                                1150
120    B(I,J)=B(I,J)+A(I,K)*A(J,K)                  1160
C      -----                                         1170
C      SOLVE SPECIAL L. P. PROBLEM                    1180
C      -----                                         1190
      CALL CNMN08 (NDB,NER,C,MS1,B,N3,N4,N5)          1200
      IF (IPRINT.GT,1.AND,NER.GT,0) WRITE (6,180)    1210
C      CALCULATE RESULTING DIRECTION VECTOR, S.      1220
      SLOPE=0.                                         1230
C      -----                                         1240
C      USABLE=FEASIBLE DIRECTION                      1250
C      -----                                         1260
      DO 140 I=1,NDV                                  1270
      S1=0.                                           1280
      IF (NVC,GT,0) S1=-A(NAC1,I)                    1290
      DO 130 J=1,NDB                                  1300
130    S1=S1-A(J,I)*C(J)                             1310
      SLOPE=SLOPE+S1*DF(I)                           1320
140    S(I)=S1                                         1330
      S(NDV1)=1.                                       1340
      IF (NVC,GT,0) S(NDV1)=-A(NAC1,NDV1)            1350
      DO 150 J=1,NDB                                  1360
150    S(NDV1)=S(NDV1)+A(J,NDV1)*C(J)                1370
C      -----                                         1380
C      NORMALIZE S TO MAX ABS OF UNITY                1390
C      -----                                         1400
      S1=0.                                           1410
      DO 160 I=1,NDV                                  1420
      A1=ABS(S(I))                                     1430
      IF (A1.GT,S1) S1=A1                             1440
160    CONTINUE                                         1450
      IF (S1.LT,1.0E-10) S1=1.0E-10                 1460
      S1=1./S1                                         1470
      DO 170 I=1,NDV                                  1480
170    S(I)=S1*S(I)                                   1490
      SLOPE=S1*SLOPE                                  1500

```

SUBROUTINE CNMN05

SEPT. 77

S(NDV1)=S1+S(NDV1)  
RETURN

C

100

FORMAT (/5X,46H\* \* DIRECTION FINDING PROCESS DID NOT CONVERGE/5X,  
129H\* \* S-VECTOR MAY NOT BE VALID)  
END

1510  
1520  
1530  
1540  
1550  
1560



## SUBROUTINE CNMN06

SEPT. 77

```

SUBROUTINE CNMN06 (X,DF,G,ISC,S,G1,G2,VLB,VUB,SCAL,N1,N2)      10
COMMON /CNMN1/ IPRINT,NDV,ITMAX,NCON,NSIDE,ICNDIR,NSCAL,NFDG,FDCH, 20
1FDCHM,CT,CTMIN,CTL,CTLMIN,THETA,PHI,NAC,DELFUN,DABFUN,LINOBJ,ITRM, 30
2ITER,INFUC,IGOTO,INFO,OBJ
COMMON /CONSAV/D1(16),CTA,CTAM,CTRM,D2,SLOPE,D3(3),X1, 50
2D4,ALP,D5(2),A2,A3,A4,D6,F2,F3,F4,CV1,CV2,CV3,CV4,D7,ALPCA,A 60
3LPFES,ALPIN,ALPMIN,ALPNC,ALPSAV,ALPSID,ALPTOT,RSPACE,IO1(7), 70
*NCAL(2),IO2(3),NVC,IO3,ICOUNT, 80
SIGOOD1,IGOOD2,IGOOD3,IGOOD4,IBEST,III,NLNC,JGOTO,ISPACE(2) 90
DIMENSION X(N1),DF(N1),G(N2),ISC(N2),S(N1),G1(N2),G2(N2),VLB(N1),V 100
IUB(N1),SCAL(N1) 110
ROUTINE TO SOLVE ONE-DIMENSIONAL SEARCH PROBLEM FOR CONSTRAINED 120
FUNCTION MINIMIZATION. 130
BY G. N. VANDERPLAATS AUG., 1974. 140
NASA-AMES RESEARCH CENTER, MOFFETT FIELD, CALIF. 150
OBJ = INITIAL AND FINAL FUNCTION VALUE. 160
ALP = MOVE PARAMETER. 170
SLOPE = INITIAL SLOPE. 180
ALPSID = MOVE TO SIDE CONSTRAINT. 190
ALPFES = MOVE TO FEASIBLE REGION. 200
ALPNC = MOVE TO NEW NON-LINEAR CONSTRAINT. 210
ALPLN = MOVE TO LINEAR CONSTRAINT. 220
ALPCA = MOVE TO RE-ENCOUNTER CURRENTLY ACTIVE CONSTRAINT. 230
ALPMIN = MOVE TO MINIMIZE FUNCTION. 240
ALPTOT = TOTAL MOVE PARAMETER. 250
ZRO=0. 260
IF (JGOTO.EQ.0) GO TO 10 270
GO TO (140,310,520), JGOTO 280
IF (IPRINT.GE.5) WRITE (6,730) 290
ALPSAV=ALP 300
ICOUNT=0 310
ALPTOT=0. 320
TOLERANCES. 330
CTAM=ABS(CTMIN) 340
CTRM=ABS(CTLMIN) 350
PROPOSED MOVE. 360
CONTINUE 370
***** BEGIN SEARCH OR IMPOSE SIDE CONSTRAINT MODIFICATION ***** 380
A2=ALPSAV 390
ICOUNT=ICOUNT+1 400
ALPSID=1.0E+20 410
INITIAL ALPHA AND OBJ. 420
ALP=0. 430
F1=OBJ 440
KSID=0 450
IF (NSIDE.EQ.0) GO TO 70 460

```

## SUBROUTINE CNMN06

SEPT. 77

C	FIND MOVE TO SIDE CONSTRAINT AND INSURE AGAINST VIOLATION OF	510
C	SIDE CONSTRAINTS	520
C	-----	530
	DO 60 I=1,NDV	540
	SI=S(I)	550
	IF (ABS(SI).GT.1.0E-20) GO TO 30	560
C	CALCULATE ALPHA TO MINIMIZE FUNCTION	570
C	-----	580
	II=3	590
	IF (A2.GT.A3.AND.(IGOOD2.EQ.0.AND.IBEST.EQ.2)) II=2	600
	CALL CNMN04 (II,ALPMIN,ZRO,ZRO,F1,SLOPE,A2,F2,A3,F3,ZRO,ZRO)	610
450	CONTINUE	620
C	-----	630
C	PROPOSED MOVE	640
C	-----	650
C	MOVE AT LEAST ENOUGH TO OVERCOME CONSTRAINT VIOLATIONS.	660
	A4=ALPFES	670
C	MOVE TO MINIMIZE FUNCTION.	680
	IF (ALPMIN.GT.A4) A4=ALPMIN	690
C	IF A4.LE.0, SET A4 = ALPSID.	700
	IF (A4.LE.0.) A4=ALPSID	710
C	LIMIT MOVE TO NEW CONSTRAINT ENCOUNTER.	720
	IF (A4.GT.ALPLN) A4=ALPLN	730
	IF (A4.GT.ALPMC) A4=ALPMC	740
C	LIMIT MOVE TO RE-ENCOUNTER CURRENTLY ACTIVE CONSTRAINT.	750
	IF (A4.GT.ALPCA) A4=ALPCA	760
C	LIMIT A4 TO 5.*A3.	770
	IF (A4.GT.(5.*A3)) A4=5.*A3	780
C	UPDATE DESIGN.	790
	IF (IBEST.NE.3.OR.NCON.EQ.0) GO TO 470	800
C	STORE CONSTRAINT VALUES IN G2. F3 IS BEST. F2 IS NOT.	810
	DO 460 I=1,NCON	820
	G2(I)=G(I)	830
460	CONTINUE	840
470	CONTINUE	850
C	IF A4=A3 AND IGOOD1=0 AND IGOOD3=1, SET A4=.9*A3.	860
	ALP=A4-A3	870
	IF ((IGOOD1.EQ.0.AND.IGOOD3.EQ.1).AND.ABS(ALP).LT.1.0E-20) A4=.9*A	880
13		890
C	-----	900
C	MOVE A DISTANCE A4*A3	910
C	-----	920
	ALP=A4-A3	930
	ALPTOT=ALPTOT+ALP	940
	DO 480 I=1,NDV	950
	X(I)=X(I)+ALP*S(I)	960
480	CONTINUE	970
	IF (IPRINT.LT.5) GO TO 510	980
	WRITE (6,720)	990
	WRITE (6,740) A4	1000



## SUBROUTINE CNMN06

SEPT. 77

	IF (NSCAL.EQ.0) GO TO 500	1010
	DO 490 I=1,NDV	1020
490	G(I)=SCAL(I)*X(I)	1030
	WRITE (6,750) (G(I),I=1,NDV)	1040
	GO TO 510	1050
500	WRITE (6,750) (X(I),I=1,NDV)	1060
510	CONTINUE	1070
C	-----	1080
C	UPDATE FUNCTION AND CONSTRAINT VALUES	1090
C	-----	1100
	NCAL(1)=NCAL(1)+1	1110
	JGOTO=3	1120
	RETURN	1130
520	CONTINUE	1140
	F4=OBJ	1150
	IF (IPRINT.GE.5) WRITE (6,760) F4	1160
	IF (IPRINT.LT.5.OR.NCON.EQ.0) GO TO 530	1170
	WRITE (6,770)	1180
	WRITE (6,750) (G(I),I=1,NCON)	1190
530	CONTINUE	1200
C	DETERMINE ACCEPTABILITY OF F4.	1210
	IGOOD4=0	1220
	CV4=0.	1230
	IF (NCON.EQ.0) GO TO 550	1240
	DO 540 I=1,NCON	1250
	CC=CTAM	1260
	IF (ISC(I).GT.0) CC=CTBM	1270
	C1=G(I)-CC	1280
	IF (C1.GT.CV4) CV4=C1	1290
540	CONTINUE	1300
	IF (CV4.GT.0.) IGOOD4=1	1310
550	CONTINUE	1320
	ALP=AA	1330
	OBJ=F4	1340
C	-----	1350
C	DETERMINE BEST DESIGN	1360
C	-----	1370
	GO TO (560,610,660), IBEST	1380
560	CONTINUE	1390
C	CHOOSE BETWEEN F1 AND F4.	1400
	IF (IGOOD1.EQ.0.AND.IGOOD4.EQ.0) GO TO 570	1410
	IF (CV1.GT.CV4) GO TO 710	1420
	GO TO 580	1430
570	CONTINUE	1440
	IF (F4.LE.F1) GO TO 710	1450
580	CONTINUE	1460
C	F1 IS BEST.	1470
	ALPTOT=ALPTOT-AA	1480
	OBJ=F1	1490
	DO 590 I=1,NDV	1500

## SUBROUTINE CNMNO6

SEPT. 77

	X(I)=X(I)-A4*S(I)	1510
590	CONTINUE	1520
	IF (NCON.EQ.0) GO TO 710	1530
	DO 600 I=1,NCON	1540
	G(I)=G1(I)	1550
600	CONTINUE	1560
	GO TO 710	1570
610	CONTINUE	1580
C	CHOOSE BETWEEN F2 AND F4.	1590
	IF (IGOOD2.EQ.0.AND.IGOOD4.EQ.0) GO TO 620	1600
	IF (CV2.GT.CV4) GO TO 710	1610
	GO TO 630	1620
620	CONTINUE	1630
	IF (F4.LE.F2) GO TO 710	1640
630	CONTINUE	1650
C	F2 IS BEST.	1660
	OBJ=F2	1670
	A2=A4-A2	1680
	ALPTOT=ALPTOT-A2	1690
	DO 640 I=1,NDV	1700
	X(I)=X(I)-A2*S(I)	1710
640	CONTINUE	1720
	IF (NCON.EQ.0) GO TO 710	1730
	DO 650 I=1,NCON	1740
	G(I)=G2(I)	1750
650	CONTINUE	1760
	GO TO 710	1770
660	CONTINUE	1780
C	CHOOSE BETWEEN F3 AND F4.	1790
	IF (IGOOD3.EQ.0.AND.IGOOD4.EQ.0) GO TO 670	1800
	IF (CV3.GT.CV4) GO TO 710	1810
	GO TO 680	1820
670	CONTINUE	1830
	IF (F4.LE.F3) GO TO 710	1840
680	CONTINUE	1850
C	F3 IS BEST.	1860
	OBJ=F3	1870
	A3=A4-A3	1880
	ALPTOT=ALPTOT-A3	1890
	DO 690 I=1,NDV	1900
	X(I)=X(I)-A3*S(I)	1910
690	CONTINUE	1920
	IF (NCON.EQ.0) GO TO 710	1930
	DO 700 I=1,NCON	1940
	G(I)=G2(I)	1950
700	CONTINUE	1960
710	CONTINUE	1970
	ALP=ALPTOT	1980
	IF (IPRINT.GE.5) WRITE (6,790)	1990
	JGOTO=0	2000



SUBROUTINE CNMN06

SEPT. 77

	RETURN	2010
C	-----	2020
C	FORMATS	2030
C	IF COMPONENT OF S IS SMALL, SET TO ZERO.	2040
	S(I)=0.	2050
	SLOPE=SLOPE-SI*DF(I)	2060
	GO TO 60	2070
30	CONTINUE	2080
	XI=X(I)	2090
	SI=1./SI	2100
	IF (SI.GT.0.) GO TO 40	2110
C	LOWER BOUND.	2120
	XI2=VLH(I)	2130
	XI1=ABS(XI2)	2140
	IF (XI1.LT.1.) XI1=1.	2150
C	CONSTRAINT VALUE.	2160
	GI=(XI2-XI)/XI1	2170
	IF (GI.GT.-1.0E-6) GO TO 50	2180
C	PROPOSED MOVE TO LOWER BOUND.	2190
	ALPA=(XI2-XI)*SI	2200
	IF (ALPA.LT.ALPSID) ALPSID=ALPA	2210
	GO TO 60	2220
40	CONTINUE	2230
C	UPPER BOUND.	2240
	XI2=VUH(I)	2250
	XI1=ABS(XI2)	2260
	IF (XI1.LT.1.) XI1=1.	2270
C	CONSTRAINT VALUE.	2280
	GI=(XI-XI2)/XI1	2290
	IF (GI.GT.-1.0E-6) GO TO 50	2300
C	PROPOSED MOVE TO UPPER BOUND.	2310
	ALPA=(XI-XI2)*SI	2320
	IF (ALPA.LT.ALPSID) ALPSID=ALPA	2330
	GO TO 60	2340
50	CONTINUE	2350
C	MOVE WILL VIOLATE SIDE CONSTRAINT. SET S(I)=0.	2360
	SLOPE=SLOPE-S(I)*DF(I)	2370
	S(I)=0.	2380
	KSID=KSID+1	2390
60	CONTINUE	2400
C	ALPSID IS UPPER BOUND ON ALPHA.	2410
	IF (A2.GT.ALPSID) A2=ALPSID	2420
70	CONTINUE	2430
C	-----	2440
C	CHECK ILL-CONDITIONING	2450
C	-----	2460
	IF (KSID.FO.NDV.UH.ICOUNT.GT.10) GO TO 710	2470
	IF (NVC.EQ.0.AND.SLOPE.GT.0.) GO TO 710	2480
	ALPFES=1	2490
	ALPHINZ=1	2500

## SUBROUTINE CNHNO6

SEPT. 77

	ALPLN=1,1+ALPSID	2510
	ALPNC=ALPSID	2520
	ALPCA=ALPSID	2530
	IF (NCON.FQ.0) GO TO 90	2540
C	STORE CONSTRAINT VALUES IN G1.	2550
	DO 80 I=1,NCON	2560
	G1(I)=G(I)	2570
80	CONTINUE	2580
90	CONTINUE	2590
C	-----	2600
C	MOVE A DISTANCE A2*S	2610
C	-----	2620
	ALPTNT=ALPTNT+A2	2630
	DO 100 I=1,NDV	2640
	X(I)=X(I)+A2*S(I)	2650
100	CONTINUE	2660
	IF (IPRINT.LI.5) GO TO 130	2670
	WRITE (6,740) A2	2680
	IF (NSCAL.EQ.0) GO TO 120	2690
	DO 110 I=1,NDV	2700
110	G(I)=SCAL(I)*X(I)	2710
	WRITE (6,750) (G(I),I=1,NDV)	2720
	GO TO 130	2730
120	WRITE (6,750) (X(I),I=1,NDV)	2740
C	-----	2750
C	UPDATE FUNCTION AND CONSTRAINT VALUES	2760
C	-----	2770
130	NCAL(1)=NCAL(1)+1	2780
	JGOTO=1	2790
	RETURN	2800
140	CONTINUE	2810
	F2=OBJ	2820
	IF (IPRINT.GE.5) WRITE (6,760) F2	2830
	IF (IPRINT.LI.5.OR.NCON.EQ.0) GO TO 150	2840
	WRITE (6,770)	2850
	WRITE (6,750) (G(I),I=1,NCON)	2860
150	CONTINUE	2870
C	-----	2880
C	IDENTIFY ACCEPTABILITY OF DESIGNS F1 AND F2	2890
C	-----	2900
C	IGOOD = 0 IS ACCEPTABLE.	2910
C	CV = MAXIMUM CONSTRAINT VIOLATION.	2920
	IGOOD1=0	2930
	IGOOD2=0	2940
	CV1=0.	2950
	CV2=0.	2960
	NVC1=0	2970
	IF (NCON.FQ.0) GO TO 170	2980
	DO 160 I=1,NCON	2990
	CC=CTAM	3000



## SUBROUTINE CNMN06

SEPT, 77

```

      IF (ISC(I).GT.0) CC=CTBM
      C1=G1(I)-CC
      C2=G(I)-C1
      IF (C2.GT.0.) NVC1=NVC1+1
      IF (C1.GT.CV1) CV1=C1
      IF (C2.GT.CV2) CV2=C2
160  CONTINUE
      IF (CV1.GT.0.) IG00D1=1
      IF (CV2.GT.0.) IG00D2=1
170  CONTINUE
      ALP=A2
      UBJ=F2
      -----
      C  IF F2 VIOLATES FEWER CONSTRAINTS THAN F1 BUT STILL HAS CONSTRAINT
      C  VIOLATIONS RETURN
      C  -----
      C  IF (NVC1.LT.NVC.AND.NVC1.GT.0) GO TO 710
      C  -----
      C  IDENTIFY REST OF DESIGNS F1 AND F2
      C  -----
      C  IBEST CORRESPONDS TO MINIMUM VALUE DESIGN.
      C  IF CONSTRAINTS ARE VIOLATED, IBEST CORRESPONDS TO MINIMUM
      C  CONSTRAINT VIOLATION.
      C  IF (IG00D1.EQ.0.AND.IG00D2.EQ.0) GO TO 180
      C  VIOLATED CONSTRAINTS. PICK MINIMUM VIOLATION.
      C  IBEST=1
      IF (CV1.GT.CV2) IBEST=2
      GO TO 190
180  CONTINUE
      C  NO CONSTRAINT VIOLATION. PICK MINIMUM F.
      C  IBEST=1
      IF (F2.LE.F1) IBEST=2
190  CONTINUE
      II=1
      IF (NCON.EQ.0) GO TO 230
      C  -----
      C  ***** 2 - POINT INTERPOLATION *****
      C  -----
      C  III=0
200  III=III+1
      C1=G1(III)
      C2=G(III)
      IF (ISC(III).EQ.0) GO TO 210
      C  -----
      C  LINEAR CONSTRAINT
      C  -----
      IF (C1.GE.1.0E-5.AND.C1.LE.CTBM) GO TO 220
      CALL CNMN07 (II,ALP,ZRO,ZRO,C1,A2,C2,ZRO,ZRO)
      IF (ALP.LE.0.) GO TO 220
      IF (C1.GT.CTBM.AND.ALPGT.ALPPES) ALPPES=ALP

```

3010  
 3020  
 3030  
 3040  
 3050  
 3060  
 3070  
 3080  
 3090  
 3100  
 3110  
 3120  
 3130  
 3140  
 3150  
 3160  
 3170  
 3180  
 3190  
 3200  
 3210  
 3220  
 3230  
 3240  
 3250  
 3260  
 3270  
 3280  
 3290  
 3300  
 3310  
 3320  
 3330  
 3340  
 3350  
 3360  
 3370  
 3380  
 3390  
 3400  
 3410  
 3420  
 3430  
 3440  
 3450  
 3460  
 3470  
 3480  
 3490  
 3500

## SUBROUTINE CNMNO6

SEPT. 77

	IF (C1.LT.CTL.AND.ALPLN) ALPLN=ALP	3510
	GO TO 220	3520
210	CONTINUE	3530
C	-----	3540
C	NON-LINEAR CONSTRAINT	3550
C	-----	3560
	IF (C1.GE.1.0E-5.AND.C1.LE.CTAM) GO TO 220	3570
	CALL CNMNO7 (II,ALP,ZRO,ZRO,C1,A2,C2,ZRO,ZRO)	3580
	IF (ALP.LE.0.) GO TO 220	3590
	IF (C1.GT.CTAM.AND.ALPGT.ALPFES) ALPFES=ALP	3600
	IF (C1.LT.CT.AND.ALPLT.ALPMC) ALPMC=ALP	3610
220	CONTINUE	3620
	IF (III.LT.NCON) GO TO 200	3630
230	CONTINUE	3640
	IF (LINOB.GT.0.UR.SLOPE.GE.0.) GO TO 240	3650
C	CALCULATE ALPHA TO MINIMIZE FUNCTION.	3660
	CALL CNMNO8 (II,ALPMIN,ZRO,ZRO,F1,SLOPE,A2,F2,ZRO,ZRO,ZRO)	3670
240	CONTINUE	3680
C	-----	3690
C	PROPOSED MOVE	3700
C	-----	3710
C	MOVE AT LFAST FAR ENOUGH TO OVERCOME CONSTRAINT VIOLATIONS.	3720
	A3=ALPFES	3730
C	MOVE TO MINIMIZE FUNCTION.	3740
	IF (ALPMIN.GT.A3) A3=ALPMIN	3750
C	IF A3.LE.0. SET A3 = ALPSID.	3760
	IF (A3.LE.0.) A3=ALPSID	3770
C	LIMIT MOVE TO NEW CONSTRAINT ENCOUNTER.	3780
	IF (A3.GT.ALPMC) A3=ALPMC	3790
	IF (A3.GT.ALPLN) A3=ALPLN	3800
C	MAKE A3 NON-ZERO.	3810
	IF (A3.LE.1.0E-20) A3=1.0E-20	3820
C	IF A3=A2=ALPSID AND F2 IS BEST, GO INVOKE SIDE CONSTRAINT	3830
C	MODIFICATION.	3840
	ALPB=1.-A2/A3	3850
	ALPA=1.-ALPSID/A3	3860
	JBEST=0	3870
	IF (ABS(ALPB).LT.1.0E-10.AND.ABS(ALPA).LT.1.0E-10) JBEST=1	3880
	IF (JBEST.EQ.1.AND.IBEST.EQ.2) GO TO 20	3890
C	SIDE CONSTRAINT CHECK NOT SATISFIED.	3900
	IF (NCON.EQ.0) GO TO 260	3910
C	STORE CONSTRAINT VALUES IN G2.	3920
	DO 250 I=1,NCON	3930
	G2(I)=G(I)	3940
250	CONTINUE	3950
260	CONTINUE	3960
C	IF A3=A2, SET A3=.9*A2.	3970
	IF (ABS(ALPB).LT.1.0E-10) A3=.9*A2	3980
C	MOVE AT LFAST .01*A2.	3990
	IF (A3.LT.(.01*A2)) A3=.01*A2	4000



## SUBROUTINE CNMN06

SEPT. 77

C	LIMIT MOVE TO 5.*A2.	4010
	IF (A3.GT.(5.*A2)) A3=5.*A2	4020
C	LIMIT MOVE TO ALPSID.	4030
	IF (A3.GT.ALPSID) A3=ALPSID	4040
C	MOVE A DISTANCE A3*S.	4050
	ALP=A3-A2	4060
	ALPTOT=ALPTOT+ALP	4070
	DO 270 I=1,NDV	4080
	X(I)=X(I)+ALP*S(I)	4090
270	CONTINUE	4100
	IF (IPRINT.LT.5) GO TO 300	4110
	WRITE (6,740)	4120
	WRITE (6,740) A3	4130
	IF (NSCAL.EQ.0) GO TO 290	4140
	DO 280 I=1,NDV	4150
280	G(I)=SCAL(I)*X(I)	4160
	WRITE (6,750) (G(I),I=1,NDV)	4170
	GO TO 300	4180
290	WRITE (6,750) (X(I),I=1,NDV)	4190
300	CONTINUE	4200
C	-----	4210
C	UPDATE FUNCTION AND CONSTRAINT VALUES	4220
C	-----	4230
	NCAL(I)=NCAL(I)+1	4240
	JGOTO=2	4250
	RETURN	4260
310	CONTINUE	4270
	F3=OBJ	4280
	IF (IPRINT.GE.5) WRITE (6,760) F3	4290
	IF (IPRINT.LT.5.OR.NCON.EQ.0) GO TO 320	4300
	WRITE (6,770)	4310
	WRITE (6,750) (G(I),I=1,NCON)	4320
320	CONTINUE	4330
C	-----	4340
C	CALCULATE MAXIMUM CONSTRAINT VIOLATION AND PICK BEST DESIGN	4350
C	-----	4360
	CV3=0.	4370
	IGOOD3=0	4380
	NVC1=0	4390
	IF (NCON.EQ.0) GO TO 340	4400
	DO 330 I=1,NCON	4410
	CC=CTAM	4420
	IF (ISC(I).GT.0) CC=CTBM	4430
	C1=G(I)-CC	4440
	IF (C1.GT.CV3) CV3=C1	4450
	IF (C1.GT.0.) NVC1=NVC1+1	4460
330	CONTINUE	4470
	IF (CV3.GT.0.) IGOOD3=1	4480
340	CONTINUE	4490
C	DETERMINE BEST DESIGN.	4500

## SUBROUTINE CNMNO6

SEPT. 77

	IF (IBEST.EQ.2) GO TO 360	4510
C	CHOOSE BETWEEN F1 AND F3.	4520
	IF (IGOOD1.EQ.0.AND.IGOOD3.EQ.0) GO TO 350	4530
	IF (CV1.GF.CV3) IBEST=3	4540
	GO TO 380	4550
350	IF (F3.LE.F1) IBEST=3	4560
	GO TO 380	4570
360	CONTINUE	4580
C	CHOOSE BETWEEN F2 AND F3.	4590
	IF (IGOOD2.EQ.0.AND.IGOOD3.EQ.0) GO TO 370	4600
	IF (CV2.GF.CV3) IBEST=3	4610
	GO TO 380	4620
370	IF (F3.LE.F2) IBEST=3	4630
380	CONTINUE	4640
	ALP=A3	4650
	OBJ=F3	4660
C	IF F3 VIOLATES FEWER CONSTRAINTS THAN F1 RETURN.	4670
	IF (NVC1.LT.NVC) GO TO 710	4680
C	IF OBJECTIVE AND ALL CONSTRAINTS ARE LINEAR, RETURN.	4690
	IF (LINOB.NE.0.AND.NLNC.EQ.NCON) GO TO 710	4700
C	IF A3 = ALPLN AND F3 IS BOTH GOOD AND BEST RETURN.	4710
	ALPB=1.-ALPLN/A3	4720
	IF ((ABS(ALPB).LT.1.0E-20.AND.IBEST.EQ.3).AND.(IGOOD3.EQ.0)) GO TO	4730
	710	4740
C	IF A3 = ALPSID AND F3 IS BEST, GO INVOKE SIDE CONSTRAINT	4750
C	MODIFICATION.	4760
	ALPA=1.-ALPSID/A3	4770
	IF (ABS(ALPA).LT.1.0E-20.AND.IBEST.EQ.3) GO TO 20	4780
C	-----	4790
C	***** 3 = POINT INTERPOLATION *****	4800
C	-----	4810
	ALPNC=ALPSID	4820
	ALPCA=ALPSID	4830
	ALPFES=-1.	4840
	ALPHIN=-1.	4850
	IF (NCON.EQ.0) GO TO 440	4860
	III=0	4870
390	III=III+1	4880
	C1=G1(III)	4890
	C2=G2(III)	4900
	C3=G3(III)	4910
	IF (ISC(III).EQ.0) GO TO 400	4920
C	-----	4930
C	LINEAR CONSTRAINT, FIND ALPFES ONLY, ALPLN SAME AS BEFORE.	4940
C	-----	4950
	IF (C1.LE.CTBM) GO TO 430	4960
	II=1	4970
	CALL CNMNO7 (II,ALP,ZRO,ZRO,C1,A3,C3,ZRO,ZRO)	4980
	IF (ALP.GT.ALPFES) ALPFES=ALP	4990
	GO TO 430	5000



## SUBROUTINE CNMN06

SEPT. 77

400	CONTINUE	5010
C	-----	5020
C	NON-LINEAR CONSTRAINT	5030
C	-----	5040
	II=2	5050
	CALL CNMN07 (II,ALP,ZRO,ZRO,C1,A2,C2,A3,C3)	5060
	IF (ALP,LF,ZRO) GO TO 430	5070
	IF (C1,GE,CT,AND,C1,LE,0.) GO TO 410	5080
	IF (C1,GT,CTAM,OR,C1,LT,0.) GO TO 420	5090
C	ALP IS MINIMUM MOVE. UPDATE FOR NEXT CONSTRAINT ENCOUNTER.	5100
410	ALPA=ALP	5110
	CALL CNMN07 (II,ALP,ALPA,ZRO,C1,A2,C2,A3,C3)	5120
	IF (ALP,LT,ALPCA,AND,ALP,GE,ALPA) ALPCA=ALP	5130
	GO TO 430	5140
420	CONTINUE	5150
	IF (ALP,GT,ALPFES,AND,C1,GT,CTAM) ALPFES=ALP	5160
	IF (ALP,LT,ALPNC,AND,C1,LT,0.) ALPNC=ALP	5170
430	CONTINUE	5180
	IF (III,LT,NCON) GO TO 390	5190
440	CONTINUE	5200
	IF (LINOB,GT,0,OR,SLOPE,GT,0.) GO TO 450	5210
C	-----	5220
C	-----	5230
C	-----	5240
720	FORMAT (/5X,25HTHREE-POINT INTERPOLATION)	5250
730	FORMAT (////58H* * * CONSTRAINED ONE-DIMENSIONAL SEARCH INFORMATI	5260
	ION * * *)	5270
740	FORMAT (/5X,15HPROPOSED DESIGN/5X,7HALPHA =,E12.5/5X,8HMX-VECTOR)	5280
750	FORMAT (1X,8E12.4)	5290
760	FORMAT (/5X,5H0BJ =,E13.5)	5300
770	FORMAT (/5X,17HCONSTRAINT VALUES)	5310
780	FORMAT (/5X,23HTWO-POINT INTERPOLATION)	5320
790	FORMAT (/5X,35H* * * END OF ONE-DIMENSIONAL SEARCH)	5330
	END	5340

## SUBROUTINE CNMN07

SEPT. 77

```

SUBROUTINE CNMN07 (II,XBAR,EPS,X1,Y1,X2,Y2,X3,Y3)
ROUTINE TO FIND FIRST XBAR,GE,EPS CORRESPONDING TO A REAL ZERO
OF A ONE-DIMENSIONAL FUNCTION BY POLYNOMIAL INTERPOLATION.
BY G. N. VANDERPLAATS
NASA-AMES RESEARCH CENTER, MOFFETT FIELD, CALIF.
II = CALCULATION CONTROL.
1: 2-POINT LINEAR INTERPOLATION, GIVEN X1, Y1, X2 AND Y2.
2: 3-POINT QUADRATIC INTERPOLATION, GIVEN X1, Y1, X2, Y2,
X3 AND Y3.
EPS MAY BE NEGATIVE.
IF REQUIRED ZERO ON Y DOES NOT EXIST, OR THE FUNCTION IS
ILL-CONDITIONED, XBAR = EPS-1.0 WILL BE RETURNED AS AN ERROR
INDICATOR.
IF DESIRED INTERPOLATION IS ILL-CONDITIONED, A LOWER ORDER
INTERPOLATION, CONSISTANT WITH INPUT DATA, WILL BE ATTEMPTED AND
II WILL BE CHANGED ACCORDINGLY.
XBAR1=EPS-1.
XBAR=XBAR1
JJ=0
X21=X2-X1
IF (ABS(X21).LT.1.0E-20) RETURN
IF (II.EQ.2) GO TO 30
CONTINUE
-----
II=1: 2-POINT LINEAR INTERPOLATION
-----
II=1
YY=Y1*Y2
IF (JJ.EQ.0.OR.YY.LT.0.) GO TO 20
INTERPOLATE BETWEEN X2 AND X3.
DY=Y3-Y2
IF (ABS(DY).LT.1.0E-20) GO TO 20
XBAR=X2+Y2*(X2-X3)/DY
IF (XBAR.LT.EPS) XBAR=XBAR1
RETURN
DY=Y2-Y1
INTERPOLATE BETWEEN X1 AND X2.
IF (ABS(DY).LT.1.0E-20) RETURN
XBAR=X1+Y1*(X1-X2)/DY
IF (XBAR.LT.EPS) XBAR=XBAR1
RETURN
CONTINUE
-----
II=2: 3-POINT QUADRATIC INTERPOLATION
-----
JJ=1
X31=X3-X1
X32=X3-X2
QQ=X21*X31*X32

```



SUBROUTINE CNMN07

SEPT, 77

```

IF (ABS(QQ).LT.1.0E-20) RETURN
AA=(Y1+X3-Y2+X31+Y3+X21)/QQ
IF (ABS(AA).LT.1.0E-20) GO TO 10
BB=(Y2-Y1)/X21-AA*(X1+X2)
CC=Y1-X1*(AA*X1+BB)
BAC=HR*HB-4.*AA*CC
IF (BAC.LT.0.) GO TO 10
BAC=SQRT(BAC)
AA=5/AA
XBAR=AA*(BAC-BB)
XB2=-AA*(BAC+BB)
IF (XBAR.I.T.EPS) XBAR=XB2
IF (XB2.LT.XBAR.AND.XB2.GT.EPS) XBAR=XB2
IF (XBAR.I.T.EPS) XBAR=XBAR1
RETURN
END

```

510  
520  
530  
540  
550  
560  
570  
580  
590  
600  
610  
620  
630  
640  
650  
660

## SUBROUTINE CNMNO8

SEPT. 77

```

SUBROUTINE CNMNO8 (NDB,NER,C,MS1,B,N3,N4,N5)
DIMENSION C(N4),MS1(N5),B(N3,N3)
ROUTINE TO SOLVE SPECIAL LINEAR PROBLEM FOR IMPOSING S-TRANSPOSE
TIMES S.L.F.1 ROUNDS IN THE MODIFIED METHOD OF FEASIBLE DIRECTIONS.
BY G. N. VANDERPLAATS
NASA-AMES RESEARCH CENTER, MOFFETT FIELD, CALIF.
REF. 'STRUCTURAL OPTIMIZATION BY METHODS OF FEASIBLE DIRECTIONS',
G. N. VANDERPLAATS AND F. MOSES, JOURNAL OF COMPUTERS
AND STRUCTURES, VOL 3, PP 739-755, 1973.
FORM OF L' P. IS  $Rx=c$  WHERE 1ST NDB COMPONENTS OF X CONTAIN VECTOR
U AND LAST NDB COMPONENTS CONTAIN VECTOR V. CONSTRAINTS ARE
 $U \geq 0$ ,  $V \geq 0$ , AND U-TRANSPOSE TIMES V = 0.
NER = ERROR FLAG. IF NER.NE.0 ON RETURN, PROCESS HAS NOT
CONVERGED IN 5*NDB ITERATIONS.
VECTOR MS1 IDENTIFIES THE SET OF BASIC VARIABLES.
-----
CHOOSE INITIAL BASIC VARIABLES AS V, AND INITIALIZE VECTOR MS1
-----
NER=1
N2=2*NDB
C CALCULATE CBMIN AND EPS AND INITIALIZE MS1.
EPS=-1.0E+10
CBMIN=0.
DO 10 I=1,NDB
BI=B(I,I)
CBMAX=0.
IF (BI.LT.-1.0E-6) CBMAX=C(I)/BI
IF (BI.GT.EPS) EPS=BI
IF (CBMAX.GT.CBMIN) CBMIN=CBMAX
10 MS1(I)=0
EPS=.0001*EPS
IF (EPS.LT.-1.0E-10) EPS=-1.0E-10
IF (EPS.GT.-.0001) EPS=-.0001
CBMIN=CBMIN*1.0E-6
IF (CBMIN.LT.1.0E-10) CBMIN=1.0E-10
ITER1=0
NMAX=5*NDB
-----
C ***** BEGIN NEW ITERATION *****
C
20 ITER1=ITER1+1
IF (ITER1.GT.NMAX) RETURN
FIND MAX. C(I)/B(I,I) FOR I=1,NDB.
CBMAX=.9*CBMIN
ICM=0
DO 30 I=1,NDB
C1=C(I)
BI=B(I,I)
IF (BI.GT.EPS.OR.C1.GT.0.) GO TO 30
CB=C1/BI

```



## SUBROUTINE CNMNOB

SEPT. 77

	IF (CB.LE.CBMAX) GO TO 30	510
	ICLK=I	520
	CBMAX=CB	530
30	CONTINUE	540
	IF (CBMAX.LT.CBMIN) GO TO 70	550
	IF (ICLK.EQ.0) GO TO 70	560
C	UPDATE VECTOR MS1.	570
	JJ=ICLK	580
	IF (MS1(JJ).EQ.0) JJ=ICLK+NDB	590
	KK=JJ+NDB	600
	IF (KK.GT.M2) KK=JJ-NDB	610
	MS1(KK)=ICLK	620
	MS1(JJ)=0	630
C	-----	640
C	PIVOT OF B(ICLK,ICLK)	650
C	-----	660
	BB=1./B(ICLK,ICLK)	670
	DO 40 J=1,NDB	680
40	B(ICLK,J)=BB*B(ICLK,J)	690
	C(ICLK)=CBMAX	700
	B(ICLK,ICLK)=BB	710
C	ELIMINATE COEFFICIENTS ON VARIABLE ENTERING BASIS AND STORE	720
C	COEFFICIENTS ON VARIABLE LEAVING BASIS IN THEIR PLACE.	730
	DO 60 I=1,NDB	740
	IF (I.EQ.ICLK) GO TO 60	750
	BB1=B(I,ICLK)	760
	B(I,ICLK)=0.	770
	DO 50 J=1,NDB	780
50	B(I,J)=B(I,J)-BB1*B(ICLK,J)	790
	C(I)=C(I)-BB1*CBMAX	800
60	CONTINUE	810
	GO TO 20	820
70	CONTINUE	830
	NER=0	840
C	-----	850
C	STORE ONLY COMPONENTS OF U-VECTOR IN 'C'. USE B(I,1) FOR	860
C	TEMPORARY STORAGE	870
C	-----	880
	DO 80 I=1,NDB	890
	B(I,1)=C(I)	900
80	CONTINUE	910
	DO 90 I=1,NDB	920
	C(I)=0.	930
	J=MS1(I)	940
	IF (J.GT.n) C(I)=B(J,1)	950
	IF (C(I).LT.0.) C(I)=0.	960
90	CONTINUE	970
	RETURN	980
	END	990

## SUBROUTINE ANALIZ - LASER TURRET ANALYSIS

SEPT. 77

```

SUBROUTINE ANALIZ(ICALC)
ROUTINE TO PERFORM LASER TURRET ANALYSIS IN SUBSONIC AND
SUPERSONIC FLOW.
BY G. N. VANDERPLAATS
NAVAL POSTGRADUATE SCHOOL, MONTEREY, CALIF.
COMMON /G1 ORCH/ ABAR(20),ACL,AKPRIM,AL,AMACHI(30),BBAR(20),DENRTO,
* DENGAM,EPS,EPSM,GAMMAI(30),PHII(30),RFUS,SLOPEX(30),SUMP02,
* TDENRT,THMAX,WAVEL,NGHTI(30),XM
COMMON /CHLOC/ ETAI(16),MAXK,MAXP,NBEAM,NETAI,NRBI,NTHBC,NXBC,
* RBI(10),TITLE(20),YYPXRC(10,3),YYPTBC(10,3)
COMMON /CHLOC2/AMX(10,15),BMX(10,15),ANT(10,15)
DIMENSION T(10),AN(10),BN(10),PDISTI(200)
FOURIER EXPANSION.
NMAX=10
MMAX=10
OPTICAL PATH LENGTH.
KTRAP=3
B=4.
NPRINT=0
IF (ICALC.GT.1) GO TO 10
CALL TINPIT
CALCULATE FOURIER COEFFICIENTS.
CALL FCOEF(AL,ACL,THMAX,AN,BN,MAXK,MAXP,NMAX,MMAX)
RETURN
10 CONTINUE
YYPXBC(1,2)=EPS
YYPTBC(1,2)=EPS
IPRINT=0
IF (ICALC.EQ.3.OR,NPRINT.GT.0) IPRINT=1
IPLOT=0
IF (ICALC.EQ.3) IPLOT=1
SUMP02=0.
BOUNDARY CONDITIONS.
X-DIRECTION.
NSYM=0
AMULT=EPS*ABAR(1)
CALL BCOND(NSYM,NXBC,YYXBC,ABAR,MAXK,AL,AMULT)
THETA-DIRECTION.
NSYM=1
AMULT=EPS*ABAR(1)
CALL RCOND(NSYM,NTHBC,YYPTBC,BBAR,MAXP,THMAX,AMULT)
DO 30 IBEAM=1,NBEAM
AMACH=AMACHI(IBEAM)
CALL PHDIST(X,R,THETA,EPSM,XM,PHI,GAMMA,RHO,Y,Z,PHIPP,U,V,CP,ABAR
1,BBAR,AL,ACL,THMAX,EPS,RINDEX,RR,ETA,AN,BN,MAXK,MAXP,NMAX,MMAX,KTN
2AP,A,B,T,DFLOPL,IBEAM,REFDPL,WAVEL,RFUS,ETAI,RBI,GAMMAI,PHII,NETAI
3,NRBI,TDENRT,PDISTI,DENGAM,AMACH,DENRTO,AKPRIM,IPRINT,IPLOT)
SUM OF SQUARES OF PHASE DISTORTION.
NN=NRBI*NETAI

```



SUBROUTINE ANALIZ - LASER TURRET ANALYSIS

SEPT. 77

	SMP1=0.	510
	DO 20 I=1,NN	520
	SMP1=SMP1+PNIST1(I)**2	530
20	SUMPD2=SUMPD2+WGHT1(IBEAM)*SMP1	540
	30 CONTINUE	550
	THETA=0.	560
	N=20	570
	XMAX=2.*A	580
	XMIN=-XMAX	590
	R=0.	600
	IF(IPRINT.EQ.0) GO TO 50	610
	DO 60 I=1,NBEAM	620
	AMACH=AMACH1(I)	630
	IF(I.EQ.1) GO TO 80	640
	IM1=I-1	650
	DO 70 J=1,IM1	660
	DMACH=AMACH1(J)-AMACH	670
	IF(ABS(DMACH).LT.0.001) GO TO 60	680
70	CONTINUE	690
80	CONTINUE	700
	CALL CPRINT(THETA,AMACH,AL,ACL,THMAX,MAXK,MAXP,NMAX,MMAX,ABAR,	710
	* RBAR,EPS,AN,RN,N,XMIN,XMAX,R,DENGAM)	720
60	CONTINUE	730
	CALL SURPT(ABAR,BBAR,MAXK,MAXP,EPS,AL,THMAX)	740
	WRITE(6,40)SUMPD2	750
	CALCULATE TURRET SLOPE AT 30 POINTS.	760
C	NVAL=30	770
50	AMULT=EPS*RBAR(1)	780
	CALL SLOP (MAXK,ABAR,AL,SLOPEX,NVAL,AMULT)	790
	RETURN	800
	40 FORMAT (//,5X,36HSUM OF SQUARES OF PHASE DISTORTION =,E12.5)	810
	END	820
		830

## SUBROUTINE RCOND

SEPT. 77

```

SUBROUTINE RCOND (NSYM,NRC,YYPBC,ABAR,MAXE,XREF,AMULTS)
DIMENSION YYPRC(10,3),ABAR(1),A(10,10),B(10)
ROUTINE TO IMPOSE POLYNOMIAL BOUNDARY CONDITIONS.
THE FIRST NBCT COEFFICIENTS OF ABAR ARE CALCULATED WHERE NBCT IS
THE TOTAL NUMBER OF B. C. S.
TOTAL NUMBER OF BOUNDARY CONDITIONS.
NBCT=0
DO 10 I=1,NRC
IF (ABS(YYPRC(I,2)).LT.100.) NBCT=NBCT+1
IF (ABS(YYPRC(I,3)).LT.100.) NBCT=NBCT+1
10 CONTINUE
IF (NBCT.EQ.0) RETURN
MAXE1=MAXE+1
IMPOSE SYMMETRY IF REQUIRED.
NSYM1=1
IF (NSYM.EQ.0) GO TO 30
NSYM1=2
DO 20 I=2,MAXE1,2
20 ABAR(I)=0.
30 CONTINUE
NUMBER OF COEFFICIENTS ELIMINATED.
N1=NBCT*NSYM1
SET UP COEFFICIENT MATRIX AND RHS.
N=0
JJ=NSYM1+1
DO 70 I=1,NRC
X=YYPRC(I,1)*XREF
IF (ABS(YYPRC(I,2)).GE.100.) GO TO 50
Y BOUNDARY CONDITION.
N=N+1
B(N)=YYPRC(I,2)/AMULTS
L=1
AA=1.
DO 40 J=1,MAXE1,NSYM1
IF (J.GT.N1) B(N)=B(N)-ABAR(J)*AA
IF (J.LE.N1) A(N,L)=AA
L=L+1
AA=AA*X
IF (NSYM1.EQ.2) AA=AA*X
40 CONTINUE
50 CONTINUE
IF (ABS(YYPRC(I,3)).GE.100.) GO TO 70
Y-PRIME BOUNDARY CONDITION.
N=N+1
B(N)=YYPRC(I,3)/AMULTS
L=2
A(N,1)=0.
AA=1.
IF (NSYM1.EQ.2) AA=X
DO 60 J=J,MAXE1,NSYM1

```



SUBROUTINE BCOND

SEPT. 77

BB=FLOAT(1)=1.  
IF (J.GT.N1) B(N)=B(N)-ABAR(J)\*BB\*AA  
IF (J.LE.N1) A(N,L)=AA\*BB  
L=L+1  
AA=AA\*X  
IF (NSYM1.EQ.2) AA=AA\*X

60 CONTINUE

70 CONTINUE

DETERMINE COEFFICIENTS.

M1=10

M2=10

M3=10

M4=1

NLC=1

CALL GELIM2 (A,B,N,NLC,M1,M2,M3,M4,NER)

STORE RESULTS IN ABAR.

J=1-NSYM1

DO 80 I=1,N

J=J+NSYM1

80 ABAR(J)=B(I)

RETURN

END

510  
520  
530  
540  
550  
560  
570  
580  
590  
600  
610  
620  
630  
640  
650  
660  
670  
680  
690  
700  
710  
720

## SUBROUTINE BESJ

SEPT. 77

C	10
C	20
C	30
C	40
C	50
C	60
C	70
C	80
C	90
C	100
C	110
C	120
C	130
C	140
C	150
C	160
C	170
C	180
C	190
C	200
C	210
C	220
C	230
C	240
C	250
C	260
C	270
C	280
C	290
C	300
C	310
C	320
C	330
C	340
C	350
C	360
C	370
C	380
C	390
C	400
C	410
C	420
C	430
C	440
C	450
C	460
C	470
C	480
C	490
C	500

```

.....
SUBROUTINE BESJ
PURPOSE
  COMPUTE THE J BESSEL FUNCTION FOR A GIVEN ARGUMENT AND ORDER
USAGE
  CALL BESJ(X,N,BJ,D,IER)
DESCRIPTION OF PARAMETERS
  X -THE ARGUMENT OF THE J BESSEL FUNCTION DESIRED
  N -THE ORDER OF THE J BESSEL FUNCTION DESIRED
  BJ -THE RESULTANT J BESSEL FUNCTION
  D -REQUIRED ACCURACY
  IER-RESULTANT ERROR CODE WHERE
      IER=0 NO ERROR
      IER=1 N IS NEGATIVE
      IER=2 X IS NEGATIVE OR ZERO
      IER=3 REQUIRED ACCURACY NOT OBTAINED
      IER=4 RANGE OF N COMPARED TO X NOT CORRECT (SEE REMARKS)
REMARKS
  N MUST BE GREATER THAN OR EQUAL TO ZERO, BUT IT MUST BE
  LESS THAN
      20+10*X-X** 2/3   FOR X LESS THAN OR EQUAL TO 15
      90+X/2           FOR X GREATER THAN 15
SUBROUTINES AND FUNCTION SUBPROGRAMS REQUIRED
  NONE
METHOD
  RECURRENCE RELATION TECHNIQUE DESCRIBED BY H. GOLDSTEIN AND
  R.M. THALER, 'RECURRENCE TECHNIQUES FOR THE CALCULATION OF
  BESSEL FUNCTIONS', M.T.A.C., V.13, PP.102-108 AND I.A. STEGUN
  AND M. ABRAMOWITZ, 'GENERATION OF BESSEL FUNCTIONS ON HIGH
  SPEED COMPUTERS', M.T.A.C., V.11, 1957, PP.255-257
.....
SUBROUTINE BESJ(X,N,BJ,D,IER)
  BJ=.0
  IF(N)10,20,20
10 IER=1
  RETURN
20 IF(X)30,30,31
30 IER=2
  RETURN

```



SUBROUTINE BESJ

SEPT. 77

31	IF(X=15.) 32,32,34	510
32	NTEST=20.+10.*X-X** 2/3	520
	GO TO 36	530
34	NTEST=90.+X/2.	540
36	IF(N-NTEST) 40,38,38	550
38	IER=4	560
	RETURN	570
40	IER=0	580
	N1=N+1	590
	BPREV=.0	600
C	COMPUTE STARTING VALUE OF M	610
C	IF(X=5.) 50,60,60	620
50	MA=X+6.	630
	GO TO 70	640
60	MA=1.4*X+40./X	650
70	MB=N+IFIX(X)/4+2	660
	MZERO=MAX0(MA,MB)	670
C	SET UPPER LIMIT OF M	680
C	MMAX=NTEST	690
100	DO 190 M=MZERO,MMAX,3	700
C	SET F(M), F(M-1)	710
C	FM1=1.0E-28	720
	FM=.0	730
	ALPHA=.0	740
	IF(M-(M/2)+2) 120,110,120	750
110	JT=-1	760
	GO TO 130	770
120	JT=1	780
130	M2=M-2	790
	DO 160 K=1,M2	800
	MK=M-K	810
	BMK=2.*FLNAT(MK)*FM1/X-FM	820
	FM=FM1	830
	FM1=BMK	840
	IF(MK=N-1,150,140,150)	850
140	BJ=BMK	860
150	JT=-JT	870
	S=1+JT	880
160	ALPHA=ALPHA+BMK*S	890
	BMK=2.*FM1/X-FM	900
	IF(N) 180,170,180	910
170	BJ=BMK	920
180	ALPHA=ALPHA+BMK	930
	BJ=BJ/ALPHA	940
		950
		960
		970
		980
		990
		1000

# SUBROUTINE BESJ

```

190 IF (ABS(BJ_RPREV)-ABS(D.BJ))200,200,190
    BPREV=BJ
    IER=3
200 RETURN
    END

```

SEPT. 77

1010  
1020  
1030  
1040  
1050



## SUBROUTINE BESK

SEPT. 77

C	SUBROUTINE BESK	10
C		20
C	COMPUTE THE K BESSEL FUNCTION FOR A GIVEN ARGUMENT AND ORDER	30
C		40
C	USAGE	50
C	CALL BESK(X,N,BK,IER)	60
C		70
C	DESCRIPTION OF PARAMETERS	80
C	X THE ARGUMENT OF THE K BESSEL FUNCTION DESIRED	90
C	N THE ORDER OF THE K BESSEL FUNCTION DESIRED	100
C	BK THE RESULTANT K BESSEL FUNCTION	110
C	IER RESULTANT ERROR CODE WHERE	120
C	IER=0 NO ERROR	130
C	IER=1 N IS NEGATIVE	140
C	IER=2 X IS ZERO OR NEGATIVE	150
C	IER=3 X .GT. 170, MACHINE RANGE EXCEEDED	160
C	IER=4 BK .GT. 10**70	170
C		180
C	REMARKS	190
C	N MUST BE GREATER THAN OR EQUAL TO ZERO	200
C		210
C	SUBROUTINES AND FUNCTION SUBPROGRAMS REQUIRED	220
C	NONE	230
C		240
C	METHOD	250
C	COMPUTES ZERO ORDER AND FIRST ORDER BESSEL FUNCTIONS USING	260
C	SERIES APPROXIMATIONS AND THEN COMPUTES N TH ORDER FUNCTION	270
C	USING RECURRENCE RELATION.	280
C	RECURRENCE RELATION AND POLYNOMIAL APPROXIMATION TECHNIQUE	290
C	AS DESCRIBED BY A.J.M.HITCHCOCK, 'POLYNOMIAL APPROXIMATIONS	300
C	TO BESSEL FUNCTIONS OF ORDER ZERO AND ONE AND TO RELATED	310
C	FUNCTIONS', M.T.A.C., V.11, 1957, PP.86-88, AND G.N. WATSON,	320
C	'A TREATISE ON THE THEORY OF BESSEL FUNCTIONS', CAMBRIDGE	330
C	UNIVERSITY PRESS, 1958, P. 62	340
C		350
C	.....	360
C		370
C	SUBROUTINE BESK (X,N,BK,IER)	380
C	DIMENSION T(12)	390
C	BK=.0	400
C	IF (N) 10,20,20	410
C	10 IER=1	420
C	RETURN	430
C	20 IF (X) 30,30,40	440
C	30 IER=2	450
C	RETURN	460
C	40 IF (X-170.0) 60,60,50	470
C	50 IER=3	480
C	RETURN	490
C	60 IER=0	500

SUBROUTINE RESK

SEPT. 77

	IF (X=1.) 180,180,70	510
70	A=EXP(-X)	520
	B=1./X	530
	C=SQRT(B)	540
	T(1)=B	550
	DO 80 L=2,12	560
80	T(L)=T(L-1)*B	570
	IF (N=1) 90,110,90	580
C		590
C	COMPUTE KN USING POLYNOMIAL APPROXIMATION	600
C		610
90	G0=A*(1.2533141-.1566642*T(1)+.08811128*T(2)-.09139095*T(3)+.13445	620
	196*T(4)-.2299850*T(5)+.3792410*T(6)-.5247277*T(7)+.5575368*T(8)-.4	630
	2262633*T(9)+.2184518*T(10)-.06080977*T(11)+.009189383*T(12))*C	640
	IF (N) 40,100,110	650
100	BK=G0	660
	RETURN	670
C		680
C	COMPUTE K <sub>T</sub> USING POLYNOMIAL APPROXIMATION	690
C		700
110	G1=A*(1.2533141+.4699927*T(1)-.1468583*T(2)+.1280427*T(3)-.1730452	710
	*T(4)+.2847618*T(5)-.4594342*T(6)+.6283381*T(7)-.6632295*T(8)+.505	720
	20239*T(9)-.2581304*T(10)+.07880001*T(11)-.01082418*T(12))*C	730
	IF (N=1) 40,120,130	740
120	BK=G1	750
	RETURN	760
C		770
C	FROM K <sub>0</sub> ,K <sub>T</sub> COMPUTE K <sub>N</sub> USING RECURRENCE RELATION	780
C		790
130	DO 160 J=2,N	800
	GJ=2.*(FLNAT(J)-1.)*G1/X+G0	810
	IF (GJ-1.0E70) 150,150,140	820
140	IER=4	830
	GO TO 170	840
150	G0=G1	850
160	G1=GJ	860
170	BK=GJ	870
	RETURN	880
180	B=X/2.	890
	A=.5772157+ALOG(B)	900
	C=B*A	910
	IF (N=1) 190,220,190	920
C		930
C	COMPUTE K <sub>N</sub> USING SERIES EXPANSION	940
C		950
190	G0=-A	960
	X2J=1.	970
	FACT=1.	980
	HJ=.0	990
	DO 200 J=1,6	1000



AD-A049 272

NAVAL POSTGRADUATE SCHOOL MONTEREY CALIF  
LASTOP - A COMPUTER CODE FOR LASER TURRET OPTIMIZATION OF SMALL--ETC(U)  
DEC 77 G N VANDERPLAATS, A E FUHS  
NPS69-77-004

F/G 1/1

UNCLASSIFIED

NL

3 OF 3

AD  
A049 272



END

DATE  
FILMED

2 - 78

DDC

## SUBROUTINE BESK

SEPT. 77

```

1012 RJ=1./FLOAT(J)
1013 X2J=X2J+C
1014 FACT=FACT+RJ+RJ
1015 MJ=MJ+RJ
200 G0=G0+X2J,FACT*(MJ-A)
    IF (N) 220,210,220
210 BK=G0
    RETURN

```

```

C
C COMPUTE Ki USING SERIES EXPANSION
C

```

```

220 X2J=B
    FACT=1.
    MJ=1.
    G1=1./X+X2J*(.5+A-MJ)
    DO 230 J=2,8
    X2J=X2J+C
    RJ=1./FLOAT(J)
    FACT=FACT+RJ+RJ
    MJ=MJ+RJ
230 G1=G1+X2J,FACT*(.5+(A-MJ)*FLOAT(J))
    IF (N-1) 130,240,130
240 BK=G1
    RETURN
    END

```

```

1010
1020
1030
1040
1050
1060
1070
1080
1090
1100
1110
1120
1130
1140
1150
1160
1170
1180
1190
1200
1210
1220
1230
1240
1250

```



## SUBROUTINE RESY

SEPT. 77

C	.....	10
C		20
C	SUBROUTINE RESY	30
C		40
C	PURPOSE	50
C	COMPUTE THE Y BESSEL FUNCTION FOR A GIVEN ARGUMENT AND ORDER	60
C		70
C	USAGE	80
C	CALL RESY(X,N,BY,IER)	90
C		100
C	DESCRIPTION OF PARAMETERS	110
C	X THE ARGUMENT OF THE Y BESSEL FUNCTION DESIRED	120
C	N THE ORDER OF THE Y BESSEL FUNCTION DESIRED	130
C	BY THE RESULTANT Y BESSEL FUNCTION	140
C	IER RESULTANT ERROR CODE WHERE	150
C	IER=0 NO ERROR	160
C	IER=1 N IS NEGATIVE	170
C	IER=2 X IS NEGATIVE OR ZERO	180
C	IER=3 BY HAS EXCEEDED MAGNITUDE OF 10**70	190
C		200
C	REMARKS	210
C	VERY SMALL VALUES OF X MAY CAUSE THE RANGE OF THE LIBRARY	220
C	FUNCTION ALOG TO BE EXCEEDED	230
C	X MUST BE GREATER THAN ZERO	240
C	N MUST BE GREATER THAN OR EQUAL TO ZERO	250
C		260
C	SUBROUTINES AND FUNCTION SUBPROGRAMS REQUIRED	270
C	NONE	280
C		290
C	METHOD	300
C	RECURRENCE RELATION AND POLYNOMIAL APPROXIMATION TECHNIQUE	310
C	AS DESCRIBED BY A.J.M. HITCHCOCK, 'POLYNOMIAL APPROXIMATIONS	320
C	TO BESSEL FUNCTIONS OF ORDER ZERO AND ONE AND TO RELATED	330
C	FUNCTIONS', M.T.A.C., V.11, 1957, PP.86-88, AND G.N. WATSON,	340
C	'A TREATISE ON THE THEORY OF BESSEL FUNCTIONS', CAMBRIDGE	350
C	UNIVERSITY PRESS, 1958, P. 62	360
C		370
C	.....	380
C		390
C	SUBROUTINE RESY(X,N,BY,IER)	400
C		410
C	CHECK FOR ERRORS IN N AND X	420
C		430
C	IF(N)140,10,10	440
C	10 IER=0	450
C	IF(X)190,190,20	460
C		470
C	BRANCH IF X LESS THAN OR EQUAL 4	480
C		490
C		500

## SUBROUTINE RESY

SEPT. 77

```

20 IF(X=4.0)40,40,30
C
C      COMPUTE Y0 AND Y1 FOR X GREATER THAN 4
C
30 T1=4.0/X
   T2=T1*T1
   P0=((((-0.000037043*T2+.0000173565)*T2-.0000487613)*T2
1   +.00017343)*T2-.001753062)*T2+.3989423
   Q0=((((-0.000032312*T2-.0000142078)*T2+.0000342468)*T2
1   -.0000869791)*T2+.0004564324)*T2-.01246694
   P1=((((-0.000042414*T2-.0000200920)*T2+.0000580759)*T2
1   -.000223203)*T2+.002921826)*T2+.3989423
   Q1=((((-0.000036594*T2+.00001622)*T2-.0000398708)*T2
1   +.0001064741)*T2-.0006390400)*T2+.03740084
   A=2.0/SQRT(X)
   B=A*T1
   C=X-.7853982
   Y0=A*P0*SIN(C)+B*Q0*COS(C)
   Y1=-A*P1*COS(C)+B*Q1*SIN(C)
   GO TO 90
C
C      COMPUTE Y0 AND Y1 FOR X LESS THAN OR EQUAL TO 4
C
40 XX=X/2.
   X2=XX*XX
   T=ALOG(XX)+.5772157
   SUM=0.
   TERM=T
   Y0=T
   DO 70 L=1,15
   IF(L=1)50,60,50
50 SUM=SUM+1./FLOAT(L-1)
60 FL=L
   TS=T-SUM
   TERM=(TERM*(-X2)/FL**2)*(1.-1./(FL*TS))
70 Y0=Y0+TERM
   TERM = XX*(T-.5)
   SUM=0.
   Y1=TERM
   DO 80 L=2,16
   SUM=SUM+1./FLOAT(L-1)
   FL=L
   FL1=FL-1.
   TS=T-SUM
   TERM=(TERM*(-X2)/(FL1*FL))*((TS-.5/FL)/(TS+.5/FL1))
80 Y1=Y1+TERM
   PI2=.6366198
   Y0=PI2*Y0
   Y1=-PI2/X+PI2*Y1
C

```



## SUBROUTINE RESY

SEPT. 77

C	CHECK IF ONLY Y0 OR Y1 IS DESIRED	1010
C		1020
	90 IF(N-1)100,100,130	1030
C		1040
C	RETURN EITHER Y0 OR Y1 AS REQUIRED	1050
C		1060
	100 IF(N)110,120,110	1070
	110 BY=Y1	1080
	GO TO 170	1090
	120 BY=Y0	1100
	GO TO 170	1110
C		1120
C	PERFORM RECURRENCE OPERATIONS TO FIND YN(X)	1130
C		1140
	130 YA=Y0	1150
	YB=Y1	1160
	K=1	1170
	140 T=FLOAT(2*K)/X	1180
	YC=T*YB-YA	1190
	IF(ABS(YC)-1.0E70)145,145,141	1200
	141 IER=3	1210
	RETURN	1220
	145 K=K+1	1230
	IF(K-N)150,160,150	1240
	150 YA=YB	1250
	YB=YC	1260
	GO TO 140	1270
	160 BY=YC	1280
	170 RETURN	1290
	180 IER=1	1300
	RETURN	1310
	190 IER=2	1320
	RETURN	1330
	END	1340

SUBROUTINE CPPRNT

SEPT. 77

	SUBROUTINE CPPRNT (THETA,AMACH,AL,ACL,THMAX,MAXK,MAXP,NMAX,HMAX,AB	10
	IAR,BBAR,EPS,AN,BN,N,XMIN,XMAX,R,DENGAM)	20
	DIMENSION ABAR(1),BBAR(1),AN(1),BN(1)	30
C	ROUTINE TO PRINT PHI,UMV,CP AT N+1 LOCATIONS ALONG X FOR SPECIFIED	40
C	THETA	50
C	IF R = 0 IS INPUT, R IS CALCULATED AS TURRET SURFACE.	60
C	IF R.GT.0 IS INPUT, THAT R IS USED IN CALCULATIONS.	70
	IR=0	80
	IF(R.GT.0) IR=1	90
	WRITE (6,20) THETA,AMACH	100
	DX=(XMAX-XMIN)/FLOAT(N)	110
	X=XMIN+DX	120
	NP1=N+1	130
	DO 10 I=1,NP1	140
	X=X+DX	150
	IF(IR.EQ.0) CALL RSURF(ABAR,BBAR,EPS,MAXK,MAXP,X,THETA,AL,THMAX,R)	160
	CALL PHIUV (X,THETA,R,AMACH,AL,ACL,THMAX,MAXK,MAXP,NMAX,HMAX,ABAR,	170
	BBAR,EPS,AN,BN,PHI,U,V)	180
	CP=2.*U-V**2	190
10	WRITE(6,30) X,R,PHI,U,V,CP	200
C	CRITICAL PRESSURE COEFFICIENT.	210
	CPSTAR=2.*(1+.5*(DENGAM-1.)*AMACH*AMACH)/(DENGAM+1.)	220
	EX1=DENGAM/(DENGAM-1.)	230
	CPSTAR=2.*(CPSTAR*EX1-1.)/(DENGAM*AMACH*AMACH)	240
	WRITE(6,40) CPSTAR	250
40	FORMAT(/5X,42HCRITICAL PRESSURE COEFFICIENT ON SURFACE =,F10.5)	260
	RETURN	270
C		280
20	FORMAT(/,5X,22HFLOW FIELD FOR THETA =,F7.3,8H DEGREES//	290
	* 5X,22HMACH NUMBER =,F7.3//10X,1HX,	300
	110X,1HR,9V,3HPHI,11X,1HU,11X,1HV,10X,2HCP)	310
30	FORMAT(5X,6E11.4)	320
	END	330



## SUBROUTINE DOPL

SEPT. 77

	SUBROUTINE DOPL (X,R,THETA,EPsm,XM,PHI,GAMMA,RHO,Y,Z,PHIPP,U,V,CP,	10
	1ABAR,BBAR,AL,ACL,THMAX,EPs,RINDEX,RB,ETA,AN,BN,MAXK,MAXP,NMAX,HMAX	20
	2,KTRAP,A,R,T,DELOPL,TDENRT,DENGAM,AMACH,DENRTO,AKPRIM,DELPLA)	30
	DIMENSION ABAR(1),BBAR(1),AN(1),BN(1),T(1)	40
C	ROUTINE TO CALCULATE CHANGE IN OPTICAL PATH LENGTH BY INTEGRATING	50
C	THE INDEX OF REFRACTION = 1.0 FROM 0.0 TO A AND A TO B.	60
C	BY G. N. VANDENPLAATS	70
C	NOV., 1976.	80
C	NAVAL POSTGRADUATE SCHOOL, MONTEREY, CALIF.	90
C	INTEGRATE FROM ZERO TO A FOR CONSTANT PRESSURE. DENSITY	100
C	RATIO = TDENRT.	110
	DELOPL=AKPRIM*TDENRT*A	120
	DELPLA=DELOPL	130
C	KTRAP = MAX. NUMBER OF TRAPEZOIDAL SOLUTIONS. MAX NO. OF INTERVAL	140
C	IS 2*(KTRAP-1)	150
	N2=1	160
	DO 30 K=1,KTRAP	170
	IGOTO=0	180
10	CALL THAP,N (IGOTO,A,B,N2,RHO,RINDEX)	190
	IF (IGOTO.EQ.0) GO TO 20	200
C	INDEX OF REFRACTION =1.	210
	CALL REFINO (X,R,THETA,EPsm,XM,PHI,GAMMA,RHO,Y,Z,PHIPP,U,V,CP,ABAR	220
	1,BBAR,AL,ACL,THMAX,EPs,RINDEX,RB,ETA,AN,BN,MAXK,MAXP,NMAX,HMAX,DEN	230
	2GAM,AMACH,DENRTO,AKPRIM)	240
	GO TO 10	250
20	T(K)=RINDEX	260
30	N2=2*N2	270
C	ROMBERG INTEGRATION.	280
	K1=1	290
	CALL RMINT (T,KTRAP,K1)	300
	DELOPL=DELOPL+T(1)	310
	RETURN	320
	END	330

## SUBROUTINE FCOEF

SEPT. 77

	SUBROUTINE FCOEF(AL,ACL,THMAX,AN,BN,MAXK,MAXP,NMAX,MHAX)	10
	COMMON /CM/NC2, AMX(10,15),BMX(10,15),ANT(10,15)	20
	DIMENSION AN(1),BN(1)	30
C	ROUTINE TO CALCULATE FOURIER COEFFICIENTS FOR EXPANSION OF	40
C	POLYNOMIAL SURFACE IN X AND THETA.	50
C	BY G. N. VANDERPLAATS	60
C	NAVAL POSTGRADUATE SCHOOL, MONTEREY, CALIF.	70
C	COEFFICIENTS ON X.	80
	MAXKP1=MAXK+1	90
	DO 10 M=1,MHAX	100
	CALL FXTOK(M,MAXK,AL,ACL,AN,BN)	110
	DO 20 I=1,MAXKP1	120
	AMX(I,M)=AN(I)	130
20	BMX(I,M)=BN(I)	140
10	CONTINUE	150
C	COEFFICIENTS ON THETA.	160
	MAXPP1=MAXP+1	170
	PI=3.1415927	180
	NHAXP1=NMAX+1	190
	DO 30 NP1=1,NHAXP1	200
	N=NP1-1	210
	CALL FXTOK(N,MAXP,THMAX,PI,AN,BN)	220
	DO 40 I=1,MAXPP1	230
40	ANT(I,NP1)=AN(I)	240
30	CONTINUE	250
	RETURN	260
	END	270



## SUBROUTINE FXTOK

SEPT. 77

```

SUBROUTINE FXTOK (N,K,X1,X2,AN,BN)
DIMENSION AN(1),BN(1)
ROUTINE TO CALCULATE THE NTH FOURIER COEFFICIENTS FOR THE
EXPANSION OF 1, X, X**2, . . . X**K,
FORM OF FOURIER SERIES IS
Y = SUM, (AN(K+1)*COS(NX) + BN(K+1)*SIN(NX)), N = 0,1,2.. INF.
BY G. N. VANDERPLAATS
NAVAL POSTGRADUATE SCHOOL, MONTEREY, CALIF.

INPUT.
N = DESIRED FOURIER COEFFICIENT.
K = HIGHEST ORDER EXPONENT ON X FOR WHICH AN AND BN ARE REQUIRED.
X1 = 1/2 INTERVAL OVER WHICH X**K IS EXPANDED.
X2 = 1/2 SPACING BETWEEN EXPANSIONS.

OUTPUT.
AN = VECTOR OF A-COEFFICIENTS FOR FOURIER EXPANSION. THE
COEFFICIENT FOR X**I IS STORED IN THE I+1 LOCATION OF AN,
FOR I=0, 1, 2, . . . K.
BN = VECTOR OF B-COEFFICIENTS FOR FOURIER EXPANSION. THE
COEFFICIENT FOR X**I IS STORED IN THE I+1 LOCATION OF BN,
FOR I=0, 1, 2, . . . K.
NOTE - ALTHOUGH ONLY THE COEFFICIENTS FOR X**K MAY BE REQUIRED, THE
COEFFICIENTS FOR EXPANSION ON 1, X, X**2, . . . X**(K-1) ARE
ALSO PROVIDED SINCE THESE ARE OBTAINED AS A CONSEQUENCE OF
CALCULATING THE REQUIRED INFORMATION.

CONSTANTS:
PI=3.1415927
KMP1=K+1
IF (N.GT.0) GO TO 20
C SPECIAL CASE, N = 0.
A(N,K) AND B(N,K) ARE THE FOURIER COEFFICIENTS A-SUB-N AND B-SUB-N
RESPECTIVELY FOR THE EXPANSION X**K, K = 0, 1, . . .
A(0,K) = .5*(X1**(K+1))*(1+(-1)**K)/(X2*(K+1))
B(0,K) = 0
SIGN=-1.
C1=.5/X2
DO 10 KP1=1,KMP1
C1=C1*X1
AN(KP1)=C1*(1.-SIGN)/FLOAT(KP1)
SIGN=-SIGN
10 BN(KP1)=0.
RETURN
C GENERAL CASE, N.GT.0.
A(N,K) = {X1**K}*{(1+(-1)**K)*SIN(N*PI*X1/X2)/(N*PI) -
(K*X2/(N*PI))*B(N,K-1)}
B(N,K) = {X1**K}*{(-1+(-1)**K)*COS(N*PI*X1/X2)/(N*PI) +
(K*X2/(N*PI))*A(N,K-1)}
WHERE A(N,-1) = B(N,-1) = 0
PI = 3.1415927

```

SUBROUTINE FXTOK

SEPT. 77

C	SOLUTION BEGINS WITH K = 0 AND USES THE ABOVE RECURSION FORMULAS	510
C	TO CALCULATE A(N,K) AND B(N,K).	520
C		530
C	CONSTANTS:	540
20	ANPI=FLOAT(N)*PI	550
	ANPIX=ANPI*X1/X2	560
	SN1=STN(ANPIX)/ANPI	570
	CS1=COS(ANPIX)/ANPI	580
C	K = 0.	590
	AN(1)=2.*SN1	600
	BN(1)=0.	610
	IF (K.EQ.N) RETURN	620
C	K = 1, 2, . . . K	630
	SIGN=-1.	640
	CC=X2/ANPI	650
	C1=1.	660
	DO 30 KN=2,KMP1	670
	K=KN-1	680
	C1=C1*X1	690
	C2=FLOAT(K)*CC	700
	AN(KN)=C1*(1.+SIGN)*SN1-C2*BN(K)	710
	BN(KN)=C1*(SIGN-1.)*CS1+C2*AN(K)	720
30	SIGN=-SIGN	730
	RETURN	740
	END	750



# SUBROUTINE FXY34

SEPT. 77

SUBROUTINE FXY34(N,X,Y,Z,NER)

DIMENSION X(1),Y(1),Z(1),AA(4,4)

ROUTINE TO CALCULATE THE COEFFICIENTS OF A POLYNOMIAL

FUNCTION OF Z IN X AND Y.

BY G. N. VANDERPLAATS

MAY, 1977.

NAVAL POSTGRADUATE SCHOOL, MONTEREY, CALIF.

--INPUT.

N = NUMBER OF INTERPOLATION POINTS (N = 3 OR 4).

X,Y = X AND Y COORDINATES, I=1,N.

Z = Z = F(X,Y) = FUNCTION VALUES.

Z IS DESTROYED.

--OUTPUT.

Z = POLYNOMIAL COEFFICIENTS.

IF N = 3,  $Y = Z(1) + Z(2)*X + Z(3)*Y$ .

IF N = 4,  $Y = Z(1) + Z(2)*X + Z(3)*Y + Z(4)*X*Y$ .

NER = ERROR INDICATOR, 0 = NO ERROR, NER.GT.0 = ERROR DUE TO

TWO X,Y POINTS ARE THE SAME OR THREE X,Y POINTS ARE

COLLINEAR.

DIMENSION OF AA MATRIX AND NUMBER OF RHS VECTORS FOR EQUATIONS.

NDIM=4

NRHS=1

INSURE N = 3 OR 4.

IF(N.LT.3) N=3

IF(N.GT.4) N=4

SET UP COEFFICIENT MATRIX FOR SIMULTANEOUS EQUATION SOLUTION.

DO 10 I=1,N

AA(I,1)=1

AA(I,2)=X(I)

AA(I,3)=Y(I)

AA(I,4)=X(I)\*Y(I)

SOLVE EQUATIONS.

CALL GELIM2(AA,Z,N,NRHS,NDIM,NDIM,NDIM,NRHS,NER)

IF(N.EQ.3) Z(4)=0.

RETURN

END

## SUBROUTINE GELIM2

SEPT. 77

	SUBROUTINE GELIM2 (A,B,N,NLC,M1,M2,M3,M4,NER)	10
	DIMENSION A(M1,M2),B(M3,M4),K(10)	20
C	SOLUTION OF SIMULTANEOUS EQUATIONS WITH MULTIPLE CONSTANT VECTORS	30
C	BY GAUSS ELIMINATION, USING PIVOT SEARCH.	40
C	BY G. N. VANDERPLAATS, 9-25-70	50
C	A=COEF. MATRIX B=MATRIX CONTAINING NLC CONSTANT VECTORS	60
C	N=NO. OF EQUATIONS M1 AND M2 ARE DIMENSIONS AS GIVEN ABOVE	70
C	IF NER=1 ON RETURN, A IS SINGULAR.	80
	NER=1	90
	EPS=1.0E-20	100
C	INITIALIZE K TO ZERO	110
	DO 10 I=1,N	120
10	K(I)=0	130
C	BEGIN ELIMINATION	140
	DO 90 J=1,N	150
C	FIND BEST PIVOT ROW	160
	AA=0.	170
	II=0	180
	DO 20 I=1,N	190
	IF (K(I).NE.0) GO TO 20	200
	BB=ABS(A(I,J))	210
	IF (BB.LE.AA) GO TO 20	220
	AA=BB	230
	II=I	240
20	CONTINUE	250
	IF (II.EQ.0.OR.AA.LE.EPS) RETURN	260
	K(II)=J	270
C	PIVOT ON POSITION A(II,J)	280
C	REDUCE A(II,J) TO IDENTITY	290
	AA=1./A(II,J)	300
	DO 30 L=J,N	310
30	A(II,L)=A(II,L)*AA	320
	DO 40 L=1,NLC	330
40	B(II,L)=B(II,L)*AA	340
C	ELIM. COEF. OF JTH COL. FOR I.NE.II	350
	L1=J+1	360
	DO 80 I=1,N	370
	IF (I.EQ.II) GO TO 80	380
	BB=A(I,J)	390
	IF (ABS(BB).LE.EPS) GO TO 80	400
	IF (L1.GT.N) GO TO 60	410
	DO 50 L=L1,N	420
50	A(I,L)=A(I,L)-A(II,L)*BB	430
60	CONTINUE	440
	DO 70 L=1,NLC	450
70	B(I,L)=B(I,L)-B(II,L)*BB	460
80	CONTINUE	470
90	CONTINUE	480
C	RE-ORDER VARIABLES TO ORIGINAL POSITION	490
C	TEMPORARILY STORE SOLN. MATRIX IN A	500



SUBROUTINE GELIM2

SEPT. 77

```

DO 100 I=1,N
DO 100 J=1,NLC
100 A(I,J)=R(I,J)
C STORE VALUES BACK IN R IN PROPER ORDER
DO 110 I=1,N
L=K(I)
DO 110 J=1,NLC
110 B(L,J)=A(I,J)
NER=0
RETURN
END

```

510  
520  
530  
540  
550  
560  
570  
580  
590  
600  
610

## SUBROUTINE IZERN

SEPT. 77

	SUBROUTINE IZERN(IRB,RRI,IETA,ETAI,NETA,R,PD,A)	10
	DIMENSION RRI(1),ETAI(1),PD(1),A(1),RI(4),TI(4),POI(4)	20
	ROUTINE TO CALCULATE ZERNICKE FUNCTIONS OF SECTION OF BEAM WITH	30
C	FIRST NODE IRB, IETA.	40
C	BY G. N. VANDERPLAATS	50
C	NAVAL POSTGRADUATE SCHOOL, MONTEREY, CALIF.	60
C	IF IRB = 1 AND IETA = 1, THIS IS THE FIRST CALL TO IZERN.	70
C	THEREFORE ZERO OUT VECTOR A.	80
	IF (IRB.GT.1. OR IETA.GT.1) GO TO 10	90
	DO 20 I=1,10	100
20	A(I)=0.	110
10	CONTINUE	120
C	RADIAL COORDINATES.	130
	RI(1)=0.	140
	IRB1=IRB-1	150
	IF (IRB.GT.1) RI(1)=RBI(IRB1)	160
	RI(4)=RI(1)	170
	RI(2)=RBI(IRB)	180
	RI(3)=RI(2)	190
C	ETA COORDINATES.	200
	TI(1)=ETAI(IETA)	210
	TI(2)=TI(1)	220
	IETA1=IETA+1	230
	TI(3)=ETAI(1)+6.2831854	240
	IF (IETA.LT.NETA) TI(3)=ETAI(IETA1)	250
	TI(4)=TI(3)	260
C	PHASE DISTORTION.	270
	N1=(IRB-2)*NETA+IETA	280
	N2=N1+NETA	290
	N3=N2+1	300
	IF (IETA.EQ.NETA) N3=N3-NETA	310
	N4=N1+1	320
	IF (IETA.EQ.NETA) N4=N4-NETA	330
	POI(1)=0.	340
	IF (N1.GT.0) POI(1)=PD(N1)	350
	POI(2)=PD(N2)	360
	POI(3)=PD(N3)	370
	POI(4)=0.	380
	IF (N1.GT.0) POI(4)=PD(N4)	390
C	CALCULATE INTERPOLATION COEFFICIENTS.	400
	N=4	410
	IF (IRB.EQ.1) N=3	420
	CALL FXYS4(N,RI,TI,POI,NER)	430
C	INTEGRATION.	440
	R1=RI(1)	450
	T2=TI(2)	460
	N2=RI(2)	470
	T3=TI(3)	480
	AZ=POI(1)	490
	A1=POI(2)	500



SUBROUTINE ZERN

SEPT. 77

A2=PD1(3)  
A3=PD1(4)  
CALL ZERN(R,R1,R2,T2,T3,AZ,A1,A2,A3,A)  
RETURN  
END

510  
520  
530  
540  
550

## SUBROUTINE PHDIST

SEPT. 77

```

SUBROUTINE PHDIST (X,R,THETA,EPSM,XM,PHI,GAMMA,RHO,Y,Z,PHIPP,U,V,C 10
1P,ABAR,BBAR,AL,ACL,THMAX,EPS,RINDEX,RB,ETA,AN,BN,MAXK,MAXP,NMAX,MM 20
2AX,KTRAP,A,R,T,DELPL,IBEHM,REFDPL,NAVEL,RFUS,ETA1,RB1,GAMMA1,PHI1 30
3,NETAI,NRRT,TDENRT,PDISTI,DENGAM,AMACH,DENRTU,AKPRIM,IPRINT,IPLT) 40
DIMENSION ARAR(1),BRAR(1),AN(1),BN(1),T(1),ETA1(1),RB1(1),GAMMA1(1 50
1),PHI1(1),PDISTI(1) 60
DIMENSION AT(32),XP(100),YP(100),ZP(100) 70
ROUTINE TO CALCULATE PHASE DISTORTION FOR THE IBEAM TURRET 80
ORIENTATION. 90
BY G. N. VANDERPLAATS NOV., 1976 100
NAVAL POST GRADUATE SCHOOL, MONTEREY, CALIF. 110
REFDPL = REFERENCE DELTA PATH LENGTH ALONG CENTER OF BEAM. 120
NEXTA=3 130
C BEAM ORIENTATION. 140
PHI=PHI1(1) 150
GAMMA=GAMMA1(1) 160
A1=57.29578*PHI 170
A2=57.29578*GAMMA 180
IF(IPRINT.GT.0) WRITE(6,90) IBEAM,A1,A2,AMACH 190
C CALCULATE REFERENCE PHASE DISTORTION. 200
RB=0. 210
ETA=0. 220
C TURRET SURFACE INTERCEPT. 230
CALL SRFINI (XM,EPSM,PHI,GAMMA,A,RB,ETA,X,R,THETA,ABAR,BBAR,EPS,MA 240
1XK,MAXP,A1,THMAX) 250
A1REF=A 260
C REFERENCE CHANGE IN PATH LENGTH DUE TO DISTORTION. 270
CALL DUPL (X,R,THETA,EPSM,XM,PHI,GAMMA,RHO,Y,Z,PHIPP,U,V,CP,ABAR,B 280
1BAR,AL,ACL,THMAX,EPS,RINDEX,RB,ETA,AN,BN,MAXK,MAXP,NMAX,MMAX,KTRAP 290
2,A,B,T,DELPL,TDENRT,DENGAM,AMACH,DENRTU,AKPRIM,DELPLA) 300
REFDPL=DELPL*RFUS/NAVEL 310
A1=57.29578*ETA 320
A2=0. 330
XP(1)=0. 340
YP(1)=0. 350
ZP(1)=0. 360
IF(IPRINT.GT.0) WRITE(6,100)RB,A1,A2,A2,A,A2 370
C CHANGE IN PATH LENGTH DUE TO DISTORTION FOR SPECIFIED VALUES OF 380
RB AND ETA. 390
C INCREMENT RB. 400
NN=0 410
MM=1 420
DO 60 IRB=1,NRBI 430
RB=RB1(IRB) 440
C INCREMENT ETA. 450
DO 50 IETA=1,NETAI 460
ETA=ETA1(IETA) 470
C SURFACE INTERCEPT. 480
CALL SRFINI (XM,EPSM,PHI,GAMMA,A,RB,ETA,X,R,THETA,ABAR,BBAR,EPS,MA 490
1XK,MAXP,A1,THMAX) 500

```



## SUBROUTINE PHDIST

SEPT. 77

```

C      CHANGE IN PATH LENGTH DUE TO DISTORTION.
      CALL DOPL (X,R,THETA,EPSh,XM,PHI,GAMMA,RHO,Y,Z,PHIPP,U,V,CP,ABAH,B
      IBAR,AL,ACI,THMAX,EPs,RINDEX,HR,ETA,AN,BN,MAXK,MAXP,NMAX,MMAX,KTRAP
      Z,A,B,T,DEI,DPL,TDENRT,DENGAM,AMACH,DENRTO,AKPRIM,DELPLA)
      DPL=DELOP1*RFUS/NAVEL
      NN=NN+1
      MM=MM+1
      AI(NN)=A
      PDISTI(NN)=DPL-REFDPL
      A1=57.29578*ETA
      XX=RB*SIN(ETA)
      YY=RB*COS(ETA)
      XP(MM)=XX
      YP(MM)=YY
      ZP(MM)=PDISTI(NN)
      IF (IPRINT.GT.0) WRITE(6,100)RB,A1,XX,YY,A,PDISTI(NN)
      IF (IRB.LT.NRAI) GO TO 40
      IF (IETA.GT.1) GO TO 10
      X11=XP(MM)
      Y11=YP(MM)
      DP11=PDISTI(NN)
      ETA1=ETA+.6.2831854
      GO TO 40
C      INTERPOLATE FOR MORE BOUNDARY POINTS.
10  NCOUNT=0
      MM1=MM+NEXTRA
      XP(MM1)=XP(MM)
      YP(MM1)=YP(MM)
      ZP(MM1)=ZP(MM)
      DETA=(ETA-ETA1)/(FLOAT(NEXTRA)+1.)
      DPD=PDISTI(NN)-PDIST1
      DX=XP(MM)-X11
      DY=YP(MM)-Y11
20  CONTINUE
      IF (ABS(DX).LT.1.0E-10) DX=1.0E-10
      IF (ABS(DY).LT.1.0E-10) DY=1.0E-10
      DO 30 INT=1,NEXTRA
      ETA1=ETA1+DETA
      XX=RB*SIN(ETA1)
      YY=RB*COS(ETA1)
      XP(MM)=XX
      YP(MM)=YY
      ZP(MM)=PDIST1+DPD*(YY-Y11)/DY
30  MM=MM+1
      NCOUNT=NCOUNT+1
      IF (IETA.LT.ETA1) GO TO 40
      IF (NCOUNT.GT.1) GO TO 40
      DETA=(ETA1-ETA)/(FLOAT(NEXTRA)+1.)
      PDIST1=PDISTI(NN)
      DPD=DP11-PDIST1

```

1000

## SUBROUTINE PHDIST

SEPT. 77

	ETA1=ETA	1010
	XIM1=XP(MM)	1020
	YIM1=YP(MM)	1030
	DX=X11-XIM1	1040
	DY=Y11-YIM1	1050
	MM=MM+1	1060
	GO TO 20	1070
40	CONTINUE	1080
	ETA1=ETA	1090
	PDIST1=PDIST1(NN)	1100
	XIM1=XP(MM)	1110
	YIM1=YP(MM)	1120
50	CONTINUE	1130
60	CONTINUE	1140
	MM=MM+1	1150
	PHI=57.29578*PHI1(IREAM)	1160
	GAMMA=57.29578*GAMMA1(IREAM)	1170
	IF (IPLOT.GT.0) CALL MAPS (MM, PHI, GAMMA, NETAI, NRBI, XP, YP, ZP)	1180
	IF (IPRINT.EQ.0) RETURN	1190
C	CALCULATE ZERNICKE COEFFICIENTS.	1200
C	VECTOR ZP IS USED TO STORE ZERNICKE COEFFICIENTS, A.	1210
	RBMAX=RB1(NRBI)	1220
	DO 62 IRH=1, NRBI	1230
	DO 62 IETA=1, NETAI	1240
62	CALL IZERN (IRH, RB1, IETA, ETA1, NETAI, RBMAX, PDIST1, ZP)	1250
	WRITE (6, 63) (ZP(I), I=1, 10)	1260
63	FORMAT (//, 5X, 22#ZERNICKE COEFFICIENTS//, 5X, 9#AVERAGE =, E13.5/5X,	1270
	* 9#TILT, X =, E13.5, 10X, 3#Y =, E13.5/5X, 9#FOCUS =, F13.5/5X,	1280
	* 9#ASTIG =, 2E13.5/5X, 9#COMA =, 4E13.5)	1290
	RETURN	1300
C		1310
90	FORMAT (//, 5X, 29#PHASE DISTORTION CALCULATIONS//, 5X, 25#BEAM ORIENTA	1320
	TION NUMBER =, I5/5X, 25#AZMUTH ANGLE =, F10.2, 8# DEGREES/	1330
	* 5X, 25#ELEVATION ANGLE =, F10.2, 8# DEGREES/5X,	1340
	* 11#MACH NUMBER, 15X, 1# =, F10.2/10X, 1#R, 9X, 3#ETA, 8X, 1#X, 11X, 1#Y, 11X,	1350
	* 1#A, 11X, 1#N)	1360
100	FORMAT (5Y, E10.4, 2X, F7.2, 6E12.4)	1370
	END	1380



## SUBROUTINE PHIUV

SEPT. 77

	SUBROUTINE PHIUV (X,THETA,R,AMACH,AL,ACL,THMAX,MAXK,MAXP,NMAX,MMAX	10
	,ABAR,BBAR,EPS,AN,BN,PHI,U,V)	20
	DIMENSION ABAR(1),BBAR(1),AN(1),BN(1)	30
C	ROUTINE TO CALCULATE POTENTIAL FUNCTION, PHI, AND PERTURBATION	40
C	VELOCITIES U AND V.	50
C	BY G. N. VANDERPLAATS	60
C	NAVAL POSTGRADUATE SCHOOL, MONTEREY, CALIF.	70
C		80
C	CONSTANTS	90
	DEL1=1.0E-4	100
	DEL2=1.0E-4	110
	BETA=1.-AMACH**2	120
	BETA=ABS(BETA)	130
	BETA=SQRT(BETA)	140
	PI=3.1415927	150
	BPI=PI/ACL	160
	BPIRL=BPI*AL	170
	NMAX1=NMAX+1	180
C	INITIALIZE PHI, U AND V.	190
	PHI=0.	200
	U=0.	210
	V=0.	220
C	CALCULATE POTENTIAL AND VELOCITIES.	230
C	M = LOOP.	240
	DO 40 M=1,MMAX	250
	AM=FLOAT(M)	260
	AMPIRL=AM*BPIRL	270
	AMPIRL=AM*BPIRL	280
	IF (AMACH.GT.1.) GO TO 10	290
C	SUBSONIC.	300
C	K-BESSEL FUNCTIONS FOR N=-1 AND N=0.	310
	N=1	320
	CALL BESK (AMPIRL,N,BKRN,IER)	330
	CALL BESK (AMPIRL,N,BKN,IER)	340
	N=0	350
	CALL BESK (AMPIRL,N,BKRNPI,IER)	360
	CALL BESK (AMPIRL,N,BKNPI,IER)	370
	GO TO 20	380
10	CONTINUE	390
C	SUPERSONIC.	400
C	J-BESSEL FUNCTIONS FOR N=-1 AND N=0.	410
	PRECIS=.0001	420
	N=1	430
	CALL BESJ (AMPIRL,N,BJRN,PRECIS,IER)	440
	CALL BESJ (AMPIRL,N,BJN,PRECIS,IER)	450
	BJRN=BJRN	460
	BJN=BJN	470
	N=0	480
	CALL BESJ (AMPIRL,N,BJRNPI,PRECIS,IER)	490
	CALL BESJ (AMPIRL,N,BJNPI,PRECIS,IER)	500

## SUBROUTINE PHIUV

SEPT. 77

C	Y-BESSEL FUNCTIONS FOR N=1 AND N=0,	510
	N=1	520
	CALL BESY(AMPIRL,N,BYRN,IER)	530
	CALL BESY(AMPIL,N,BYN,IER)	540
	BYRN=BYRN	550
	BYN=BYN	560
	N=0	570
	CALL BESY(AMPIRL,N,BYRNP1,IER)	580
	CALL BESY(AMPIL,N,BYNP1,IER)	590
20	CONTINUE	600
C	N = LOOP.	610
	DO 30 NP1=1,NMAX1	620
	NP1=1	630
	IF(AMACH.GT.1.) GO TO 25	640
C	SUBSONIC.	650
	BKNM1=BKN	660
	BKRNM1=BKRNM	670
	BKN=BKNP1	680
	BKRNM=BKRNP1	690
C	N+1 BESSEL FUNCTIONS BY RECURSION.	700
	BKNP1=2.*FLOAT(N)*BKN/AMPIL+BKNM1	710
	BKRNP1=2.*FLOAT(N)*BKRNM/AMPIRL+BKRNM1	720
	GO TO 27	730
25	CONTINUE	740
C	SUPERSONIC.	750
	BYNM1=BYN	760
	BYRNM1=BYRNM	770
	BYN=BYNP1	780
	BYRNM=BYRNP1	790
	BJNM1=BJN	800
	BJRNM1=BJRNM	810
	BJN=BJNP1	820
	BJRNM=BJRNP1	830
C	N+1 BESSEL FUNCTIONS BY RECURSION.	840
	BYNP1=2.*FLOAT(N)*BYN/AMPIL-BYNM1	850
	BYRNP1=2.*FLOAT(N)*BYRNM/AMPIRL-BYRNM1	860
	BJNP1=2.*FLOAT(N)*BJN/AMPIL-BJNM1	870
	BJRNP1=2.*FLOAT(N)*BJRNM/AMPIRL-BJRNM1	880
27	CONTINUE	890
C	N,M COMPONENT OF PHI, U AND V.	900
	CALL PHUVNM(N,M,X,THETA,AMACH,AL,ACL,THMAX,BKNM1,BKNP1,BKRNM1,BKRNP1,MAXK,MAXP,ABAR,BRAR,EPS,N,BN,PHINM,UNM,VNM,	910
	* BJNM1,BJN,BJNP1,BJRNM1,BJRN,BJRNP1,BYNM1,BYN,BYNP1,BYRNM1,	920
	* BYRN,BYRNP1)	930
C	UPDATE PHI, U AND V.	940
	PHI=PHI+PHINM	950
	U=U+UNM	960
	V=V+VNM	970
C	CHECK CONVERGENCE.	980
	IF(N.EQ.0) GO TO 30.	990
		1000



SUBROUTINE PHIUV

SEPT. 77

```

      IF (ABS(PHINM),LT,DEL1,AND,(ABS(UNM),LT,DEL1,AND,ABS(VNM),LT,DEL1))
      * GO TO 35
30  CONTINUE
35  CONTINUE
      IF (M.EQ.1) GO TO 36
      DPHI=ABS(PHI-PHIA)
      DU=ABS(U-UIA)
      DV=ABS(V-VA)
      IF (DPHI,LT,DEL2,AND,(DU,LT,DEL2,AND,DV,LT,DEL2)) GO TO 45
36  PHIA=PHI
      UA=U
      VA=V
40  CONTINUE
45  CONTINUE
      RETURN
      END

```

## SUBROUTINE PHUVNM

SEPT. 77

```

SUBROUTINE PHUVNM (N,M,X,THETA,AMACH,AL,ACL,THMAX,BKNM1,BKNP1,BKRN
1M1,BKRN,BKRNPI,MAXK,MAXP,ABAR,BBAR,EPS,AN,BN,PHINM,UNM,VNM,
* BJNM1,BJN,BJNP1,BJRNMI,BJRN,HJRNPI,BYNM1,BYN,BYNP1,BYRNMI,
* BYRN,BYRNPI)
COMMON /CMLC2/AMX(10,15),BMX(10,15),ANT(10,15)
DIMENSION ABAR(1),BBAR(1),AN(1),BN(1)
ROUTINE TO CALCULATE N,M COMPONENTS OF POTENTIAL, PHINM, AND
PERTURBATION VELOCITIES UNM AND VNM FOR A TURRET DEFINED BY A
DOUBLE POLYNOMIAL.
BY G. N. VANDERPLAATS                                OCT., 1976
NAVAL POSTGRADUATE SCHOOL, MONTEREY, CALIF.
INPUT
N,M          - SUBSCRIPTS ON PHI, U AND V.
X            - LONGITUDINAL COORDINATE ALONG TURRET.
THETA       - CIRCUMFERENTIAL COORDINATE AROUND TURRET.
BETA        - ARS(1.-AMACH**2)
AL,ACL      - 1/2 LENGTH OF TURRET AND 1/2 PERIOD BETWEEN TURRETS.
THMAX       - 1/2 CIRCUMFERENCE OF FUSELAGE OCCUPIED BY TURRET.
BKNM1, BKNP1 - K BESSEL FUNCTIONS AT N-1 AND N+1.
BKRNMI, BKRN, BKRNPI - K BESSEL FUNCTIONS OF R AT N-1, N AND N+1.
MAXK, MAXP   - MAX EXPONENT OF X AND THETA POLYNOMIALS.
ABAR, BBAR   - X AND THETA POLYNOMIAL COEFFICIENTS.
AN, BN       - DUMMY STORAGE DIMENSIONED MAX(MAXK+1,MAXP+1)
OUTPUT
PHINM        - PERTURBATION POTENTIAL.
UNM          - U PERTURBATION VELOCITY.
VNM          - V PERTURBATION VELOCITY.
CONSTANTS:
PI=3.1415927
AMPL=FLOAT(M)*PI/ACL
BETA=ARS(1.-AMACH**2)
BETA=SQRT(BETA)
BMPL=BETA*AMPL
SM=AMPL*X
CM=COS(SM)
SM=SIN(SM)
SN=FLOAT(N)*THETA
CN=COS(SN)
MAXKPI=MAXK+1
MAXPPI=MAXP+1
10  CALCULATE A-BAR TIMES A-SUB=M AND A-BAR TIMES B-SUB=M.
    AAM=0.
    ARM=0.
    DO 10 I=1,MAXKPI
    AAM=AAM+ARAR(I)*AMX(I,M)
    ARM=ARM+ARAR(I)*BMX(I,M)
    CALCULATE B-BAR TIMES A-SUB=N.
    BAN=0.
    BBN=0.

```



## SUBROUTINE PHUVNM

SEPT. 77

	NP1=N+1	510
	DO 20 I=1,MAXPP1	520
20	BAN=BAN+BRAR(I)*ANT(I,NP1)	530
C	CALCULATE F-SUB-N OF THETA.	540
	FN=BAN*CN	550
	IF(AMACH.GT.1.) GO TO 30	560
C	SUBSONIC.	570
C	CALCULATE PHINM.	580
	C1=AAM*SM-ARM*CM	590
	C2=BETA*(BKNP1+BKNM1)	600
	C3=2.*EPS*FN+BKRN	610
	PHINM=C3*C1/C2	620
C	CALCULATE UINM.	630
	UINM=C3*AMPL*(AAM*CM+ARM*SM)/C2	640
C	CALCULATE VNM.	650
	VNM=-AMPL*EPS*FN*(BKRN*1+BKNM1)+C1/(BKNP1+BKNM1)	660
	RETURN	670
30	CONTINUE	680
C	SUPERSONIC.	690
	ANM=BYNP1-BYNM1+BJNP1-BJNM1	700
	RNM=BYNP1-BYNM1-BJNP1+BJNM1	710
	APB=ANM+RNM	720
	AMB=ANM-RNM	730
	AB2=ANM**2+RNM**2	740
	A1=APB*SM-AMB*CM	750
	A2=AMB*SM+APB*CM	760
	A3=AAM*BYRN+ARM*BJRN	770
	A4=AAM*BJRN-ARM*BYRN	780
	A5=2.*EPS*FN/(AB2*BETA)	790
C	PHINM.	800
	PHINM=A5*(A1+A3+A2+A4)	810
C	UINM.	820
	UINM=A5*AMPL*(A2+A3-A1+A4)	830
C	VNM	840
	VNM=-EPS*FN*AMPL*((A1+AAM-A2*ARM)*(BYRN*1-BYRN*1)+	850
	\$(A1*ARM+A2*AAM)*(BJRN*1-BJRN*1))/AB2	860
	RETURN	870
	END	880

SUBROUTINE REFIND

SEPT. 77

	SUBROUTINE REFIND (X,R,THETA,EPsm,XM,PHI,GAMMA,RHO,Y,Z,PHIPP,U,V,C	10
	IP,ABAR,BUAR,AL,ACL,THMAX,EPs,RINDEX,RB,ETA,AN,BN,MAXK,MAXP,NMAX,MM	20
	ZAX,DENGAM,AMACH,DENRTO,AKPRIM)	30
	DIMENSION ABAR(1),BBAR(1),AN(1),BN(1)	40
C	ROUTINE TO CALCULATE INDEX OF REFRACTION -1 FOR A SPECIFIED POINT	50
C	ON A BEAM	60
C	BY G. N. VANDERPLAATS	70
C	NOV., 1976.	80
C	NAVAL POSTGRADUATE SCHOOL, MONTEREY, CALIF.	90
C	GIVEN AZMUTH, ELEVATION AND DISTANCE ALONG BEAM, CALCULATE	100
C	X, THETA AND R-COORDINATES.	110
C	CALL XRTPOH (XM,EPsm,PHI,GAMMA,RHO,RB,ETA,X,R,THETA,Y,Z)	120
C	CALCULATE POTENTIAL AND PERTURBATION VELOCITIES.	130
C	CALL PHIUV (X,THETA,R,AMACH,AL,ACL,THMAX,MAXK,MAXP,NMAX,MMAX,ABAR,	140
C	BBAR,EPs,AN,BN,PHIPP,U,V)	150
C	INDEX OF REFRACTION.	160
	CP=2.*U-V*V	170
	C1=1.+5*DENGAM*AMACH*AMACH*CP	180
	RINDEX=AKPRIM*DENRTO/(C1*DENGAM)	190
	RETURN	200
	END	



## SUBROUTINE RMBINT

SEPT. 77

```

SUBROUTINE RMBINT (T,K,K1)
DIMENSION T(1)
ROUTINE TO PERFORM ROMBERG INTEGRATION.
BY G. N. VANDERPLAATS
NAVAL POSTGRADUATE SCHOOL, MONTEREY, CALIF.
C
C INPUT
C T = VECTOR CONTAINING RESULTS OF TRAPEZOIDAL RULE INTEGRATION.
C IF T(1) CONTAINS TRAP. RULE RESULTS FOR N INTERVALS, T(2)
C CONTAINS RESULTS FOR 2N INTERVALS, T(3) CONTAINS RESULTS FOR
C 4N INTERVALS AND T(I) CONTAINS RESULTS FOR (2**(I-1))N
C INTERVALS.
C K = NUMBER OF TRAPEZOIDAL RULE RESULTS CONTAINED IN T.
C K1 = K ON LAST CALL TO RMBINT. FIRST TIME RMBINT IS CALLED K1=1.
C OUTPUT.
C T = VECTOR CONTAINING LAST ROW OF ROMBERG TABLE IN REVERSE ORDER.
C THE HIGHEST ORDER APPROXIMATION TO THE INTEGRAL IS IN T(1).
C T(2) GIVES THE 2ND HIGHEST ORDER APPROXIMATION SO THE
C DIFFERENCE BETWEEN T(1) AND T(2) IS AN ACCURACY ESTIMATION.
C T(K) IS THE HIGHEST ORDER TRAP. RULE APPROXIMATION AND IS NOT
C DESTROYED.
C NOTES
C 1) IF ACCURACY IS NOT SATISFACTORY, THE NUMBER OF TRAP RULE
C STATIONS CAN BE DOUBLED AND A NEW SOLUTION STORED IN K+1 OF T.
C THEN SET K1=K AND K=K+1 AND CALL RMBINT AGAIN FOR NEW SOLUTION.
C 2) ALL INITIAL ENTRIES OF T UP TO K-1 ARE DESTROYED.
C REFERENCE, CONTE, ELEMENTARY NUMERICAL ANALYSIS, MCGRAW-HILL,
C 1965, PP 126-133.
C IF (K.LE.1) RETURN
C K1=K+1
C BUILD ROW_KK OF ROMBERG TABLE.
C DO 10 KK=K1,K
C KM1=KK-1
C A=1.
C I=KK
C PUT ROW_KK IN T(I), I=1, KK IN REVERSE ORDER. T(KK) DOES NOT CHANGE.
C DO 10 II=1, KM1
C I=I-1
C A=4.*A
C 10 T(I)=(A*T(I+1)-T(I))/(A-1.)
C RETURN
C END

```

## SUBROUTINE RSURF

SEPT. 77

	SUBROUTINE RSURF (ABAR,BBAR,EPS,MAXK,MAXP,X,THETA,AL,THMAX,R)	10
	DIMENSION ABAR(1),BBAR(1)	20
C	ROUTINE TO CALCULATE THE NON-DIMENSIONAL TURRET RADIUS AT	30
C	X AND THETA.	40
C	BY G. N. VANDERPLAATS	50
C	NOV., 1976.	60
C	NAVAL POSTGRADUATE SCHOOL, MONTEREY, CALIF.	70
C	SPECIAL CASE - THETA OR X NOT ON TURRET, POINT IS ON CYLINDRICAL	80
C	FUSELAGE.	90
	R=1.	100
	IF (ABS(THETA).GE.THMAX.OR,ABS(X).GE.AL) RETURN	110
C	CONSTANTS	120
	MAXKPI=MAXK+1	130
	MAXPPI=MAXP+1	140
C	POINT ON TURRET.	150
C	EVALUATE F(X)	160
	FX=ABAR(1)	170
	IF (MAXK.FQ.0) GO TO 20	180
	XI=1.	190
	DO 10 IX=2,MAXKPI	200
	XI=XI*X	210
	IF (ABS(XI).LT.1.0E-20) GO TO 20	220
10	FX=FX+ABAR(IX)*XI	230
20	CONTINUE	240
C	EVALUATE F(THETA)	250
	FTH=BBAR(1)	260
	IF (MAXP.FQ.0) GO TO 40	270
	THI=1.	280
	DO 30 ITH=2,MAXPPI	290
	THI=THI*THETA	300
	IF (ABS(THI).LT.1.0E-20) GO TO 40	310
30	FTH=FTH+BBAR(ITH)*THI	320
40	CONTINUE	330
C	R=1.0 + F(X)*F(THETA)*EPS	340
	R=1.+FX*FTH*EPS	350
	RETURN	360
	END	



## SUBROUTINE SLOPE

SEPT. 77

	SUBROUTINE SLOPE (MAXK,ABAR,AL,SLOPEX,NVAL,AMULTS)	10
	DIMENSION ABAR(1),SLOPEX(1)	20
C	ROUTINE TO CALCULATE SLOPE OF A POLYNOMIAL AT NVAL POINTS	30
C	BETWEEN X = -AL AND X = AL.	40
	IF (NVAL.(T.2) RETURN	50
	DX=2.*AL/(FLOAT(NVAL)-1.)	60
	X=-AL-DX	70
	MAXK1=MAXK+1	80
	DO 30 I=1,NVAL	90
	X=X+DX	100
	SLOPEX(I)=0.	110
	IF (MAXK.(T.1) GO TO 30	120
	SLOPEX(I)=ABAR(2)	130
	IF (MAXK.FQ.1) GO TO 20	140
	AMULT=1.	150
	XI=1.	160
	DO 10 J=3,MAXK1	170
	XI=XI*X	180
	AMULT=AMULT*XI	190
10	SLOPEX(I)=SLOPEX(I)+AMULT*ABAR(J)*XI	200
20	SLOPEX(I)=AMULTS*SLOPEX(I)	210
30	CONTINUE	220
	RETURN	230
	END	240

## SUBROUTINE SRFINT

SEPT. 77

```

SUBROUTINE SRFINT (XM, EPSM, PHI, GAMMA, A, RB, ETA, X, R, THETA, ABAR, BBAR,
1EPS, MAXK, MAXP, AL, THMAX)
DIMENSION ABAR(1), BBAR(1)
ROUTINE TO CALCULATE DISTANCE ALONG BEAM FROM MIRROR TO TURRET
SURFACE.
BY G. N. VANDERPLAATS
NAVAL POST GRADUATE SCHOOL, MONTEREY, CALIF.
NOV., 1976
OUTPUT.
A = DISTANCE FROM MIRROR TO TURRET SURFACE.
IF A = -1.0E-6 ON RETURN, MIRROR SURFACE IS OUTSIDE TURRET
SURFACE.
IF A = 1.0E-6 ON RETURN, NO INTERCEPT COULD BE FOUND AT A LESS
THAN 10. THIS PROBABLY RESULTS FROM UNREALISTIC TURRET SHAPE.
METHOD.
FOR VARIOUS VALUES OF RHO, CALCULATE X, RR AND THETA FOR A POINT
ON THE BEAM. FOR EACH X AND THETA, CALCULATE RS FOR RADIUS TO
THE SURFACE. INTERPOLATE TO GET RR=RS. THE CORRESPONDING VALUE
OF RHO IS A.
DRHO=.2
RADIUS OF BEAM RAY AT POINT ON MIRROR SURFACE.
RHO=0.
A1=0.
RR1=EPSM
X=XM
THETA=0.
IF (RR.GT.1.0E-4) CALL XRTPOB (XM, EPSM, PHI, GAMMA, RHO, RB, ETA, X, RR1,
1THETA, Y, Z)
SURFACE RADIUS OF POINT AT X AND THETA, FOR RHO=0.
CALL RSURF (ARAR, BBAR, EPS, MAXK, MAXP, X, THETA, AL, THMAX, RS1)
DR1=RS1-RR1
A=-1.0E-6
IF DR1.LT.0, BASE OF MIRROR IS OUTSIDE TURRET.
IF (DR1.LT.0.) RETURN
PICK ARBITRARY NEW RHO AND INTERPOLATE.
10 RHO=RHO+DRHO
RADIUS OF POINT ON BEAM.
CALL XRTPOB (XM, EPSM, PHI, GAMMA, RHO, RB, ETA, X, RR2, THETA, Y, Z)
TURRET SURFACE.
CALL RSURF (ABAR, BBAR, EPS, MAXK, MAXP, X, THETA, AL, THMAX, RS2)
DR2=RS2-RR2
DDR=DR2-DR1
IF (ABS(DDR).LT.1.0E-10) DDR=1.0E-10
A=A1-DRHO*DR1/DDR
IF (A.LE.0.OR.RHO.GT.10.) GO TO 20
A IS EXTRAPOLATED POINT, UPDATE AND INTERPOLATE AGAIN.
RR1=RR2
RS1=RS2
DR1=DR2
A1=RHO
GO TO 10

```



SUBROUTINE SRFINI

SEPT. 77

20 CONTINUE  
IF (A.LT.0.) A=1.0E-6  
RETURN  
END

510  
520  
530  
540

## SUBROUTINE SURPRT

SEPT, 77

```

SUBROUTINE SURPRT (ABAR,BBAR,MAXK,MAXP,EPS,AL,THMAX)
DIMENSION ABAR(1),BBAR(1)
ROUTINE TO PRINT SURFACE FUNCTION ORDINATES FOR POLYNOMIAL TURRET,
BY G. N. VANDERPLAATS
NAVAL POSTGRADUATE SCHOOL, MONTEREY, CALIF.
INPUT.
C ABAR = VECTOR OF POLYNOMIAL COEFFICIENTS IN X-DIRECTION. ABAR
C MUST BE DIMENSIONED AT LEAST MAXK+1 IN CALLING ROUTINE.
C BBAR = VECTOR OF POLYNOMIAL COEFFICIENTS IN THETA-DIRECTION.
C BBAR MUST BE DIMENSIONED AT LEAST MAXP+1 IN CALLING
C ROUTINE.
C MAXK = ORDER OF X-POLYNOMIAL.
C MAXP = ORDER OF THETA-POLYNOMIAL.
C EPS = SCALAR SURFACE MULTIPLIER. SURFACE = EPS*F(X)*F(THETA).
C AL = 1/2 TURRET LENGTH.
C THMAX = 1/2 TURRET ANGLE.
C OUTPUT.
C POLYNOMIAL FUNCTION COORDINATES IN TERMS OF X AT THETA = 0 AND
C THETA AT X = 0.
MAXK1=MAXK+1
MAXP1=MAXP+1
WRITE (6,70) EPS
WRITE (6,80) (ABAR(I),I=1,MAXK1)
WRITE (6,90)
WRITE (6,90) (BBAR(I),I=1,MAXP1)
C X-DIRECTION.
WRITE (6,100)
DX=.1*AL
X=-1.2*AL
DO 30 I=1,23
X=X+DX
Z=ABAR(1)
AMULT=0.
ZPRIM=0.
IF (MAXK.FD.0) GO TO 20
XI=1.
DO 10 J=2,MAXK1
AMULT=AMULT+1.
ZPRIM=ZPRIM+AMULT*ABAR(J)*XI
XI=XI*X
10 Z=Z+ABAR(1)*XI
20 CONTINUE
Z=EPS*BBAR(1)*Z
ZPRIM=EPS*BBAR(1)*ZPRIM
IF (1.EQ.1.OR.1.EQ.23) Z=0.
IF (1.EQ.1.OR.1.EQ.23) ZPRIM=0.
WRITE (6,110) X,Z,ZPRIM
30 CONTINUE
C THETA-DIRECTION.
WRITE (6,120)

```



## SUBROUTINE SURPRT

SEPT. 77

```

01      DTH=1*THMAX
02      TH=-1.2*THMAX
03      DO 60 I=1,23
04          TH=TH+DTH
05          Z=BAR(1)
06          IF (MAXP.F0.0) GO TO 50
07          TH=1.
08          AMULT=0.
09          ZPRIM=0.
10          DO 40 J=2,MAXP1
11              AMULT=AMULT+1.
12              ZPRIM=ZPRIM+AMULT*BAR(J)*THI
13              THI=THI+TH
14          40 Z=Z+BAR(1)*THI
15          50 CONTINUE
16              Z=EPS+ABAR(1)*Z
17              ZPRIM=EPS+ABAR(1)*ZPRIM
18              IF (I.EQ.1.OR.I.EQ.23) Z=0.
19              IF (I.EQ.1.OR.I.EQ.23) ZPRIM=0.
20              THR=TH*57.29578
21              WRITE (6,110) TH,THR,Z,ZPRIM
22          60 CONTINUE
23          RETURN

```

C

```

70      FORMAT (///5X,18H SURFACE DEFINITION,5X,6H(EPS =,F7.3,1H)/5X,54HPOL
71      YNOMIAL COEFFICIENTS (A(I), I=0,MAXK) IN X-DIRECTION)
72      80 FORMAT (5X,5E12.5)
73      90 FORMAT (/5X,54HPOLYNOMIAL COEFFICIENTS (B(I), I=0,MAXP) IN THETA-DI
74      RECTION)
75      100 FORMAT (/5X,11HCOORDINATES/RX,1HX,11X,1HZ,9X,7HZ-PRIME)
76      110 FORMAT (5X,F7.3,5X,F8.4,5X,F8.4,5X,F8.4)
77      120 FORMAT (/12X,5H THETA/5X,7H RADIANS,6X,7H DEGREES,8X,1HZ,9X,7HZ-PRIME
78      1)
79      END

```

510  
520  
530  
540  
550  
560  
570  
580  
590  
600  
610  
620  
630  
640  
650  
660  
670  
680  
690  
700  
710  
720  
730  
740  
750  
760  
770  
780  
790  
800  
810  
820  
830  
840

## SUBROUTINE TINPUT

SEPT. 77

C	SUBROUTINE TINPUT	10
C	INPUT CARD FORMAT	20
C		30
C	* TITLE(I), I=1,20	40
C	ANYTHING MAY BE TYPED IN COL. 2-80	50
C		60
C	AERODYNAMICS - OPTICS	70
C	* AMACH, DENRT0, TDENRT, DENGAM, AKPRIM, WAVEL	80
C	AMACH = FREESTREAM MACH NUMBER	90
C	DENRT0 = FLIGHT DENSITY/SEA LEVEL DENSITY	100
C	TDENRT = DENSITY INSIDE TURRET/SEA LEVEL DENSITY	110
C	DENGAM = EXPONENT ON PRESSURE-DENSITY RELATIONSHIP	120
C	AKPRIM = INDEX OF REFRACTION CONSTANT	130
C	WAVEL = BEAM WAVELENGTH	140
C		150
C	GEOMETRY	160
C	TURRET	170
C	* RFUS, AL, THMAX, ACL, EPS	180
C	RFUS = FUSELAGE RADIUS	190
C	AL = TURRET NON-DIMENSIONAL HALF LENGTH	200
C	THMAX = TURRET HALF ANGLE (RAD)	210
C	ACL = HALF TURRET SPACING	220
C	EPS = TURRET HEIGHT MULTIPLIER	230
C	* MAXK, MAXP, NSBC, NTHBC	240
C	MAXK = ORDER OF X-POLYNOMIAL SHAPE FUNCTION	250
C	MAXP = ORDER OF THETA-POLYNOMIAL	260
C	NSBC = NUMBER OF SETS OF Y AND Y-PRIME BOUNDARY	270
C	CONDITIONS IN X-DIRECTION, EXTERNALLY IMPOSED.	280
C	NTHBC = NUMBER OF SETS OF Y AND Y-PRIME BOUNDARY	290
C	CONDITIONS IN THETA-DIRECTION, EXTERNALLY IMPOSED.	300
C	NOTE. AT X=THETA=0, Y=EPS IS AUTOMATICALLY IMPOSED.	310
C		320
C	* ABAR(I), I=1, MAXK+1	330
C	ABAR(I) = I-1 COEFFICIENT OF X-POLYNOMIAL	340
C	* YYPXBC(I,J), J=1,3	350
C	YYPXBC(I,J) = X, Y AND Y-PRIME BOUNDARY CONDITIONS IN THE	360
C	X-DIRECTION.	370
C	* BBAR(I), I=1, MAXP+1	380
C	BBAR(I) = I-1 COEFFICIENT OF THETA-POLYNOMIAL	390
C	* YYPTBC(I,J), J=1,3	400
C	YYPTBC(I,J) = X, Y AND Y-PRIME BOUNDARY CONDITIONS IN THE	410
C	THETA-DIRECTION.	420
C		430
C	MIRROR CENTER	440
C	* EPSM, XM	450
C	EPSM = Z-LOCATION OF CENTER OF MIRROR	460
C	XM = X-LOCATION OF CENTER OF MIRROR	470
C		480
C	PHASE DISTORTION CALCULATION POINTS	490
C	* NETAI, NRBI	500



## SUBROUTINE TINPUT

SEPT. 77

```

C      NETAI = NUMBER OF ETA ANGLES                      510
C      NRBI = NUMBER OF RADIUS POINTS                    520
C * ETAI(I), I=1, NETAI                                FORMAT(8F10) 530
C      ETAI(I) = ANGLE (DEGREES)                          540
C * RBI(I), I=1, NRBI                                    FORMAT(8F10) 550
C      RBI(I) = RADIUS                                      560
C                                                        570
C      BEAM ORIENTATION                                    580
C * NBEAM                                                FORMAT(8I10) 590
C      NBEAM = NUMBER OF DIFFERENT BEAM ORIENTATIONS ANALYZED 600
C * PHII(I), GAMMA(I), AMACHI(I), WGTI(I) NBEAM CARDS  FORMAT(8I10) 610
C      PHII(I) = AZMUTH ANGLE (DEGREES)                    620
C      GAMMA(I) = ELEVATION ANGLE (DEGREES)                 630
C      AMACHI(I) = MACH NUMBER. DEFAULT = AMACH.           640
C      WGTI(I) = WEIGHTING COEFFICIENT. DEFAULT = 1.       650
C      SUBROUTINE TINPUT
C      COMMON /G1ORCH/ ABAR(20), ACL, AKPRIM, AL, AMACHI(30), UBAR(20), DENRTO,
C      * DENGAM, EPS, EPSM, GAMMA(30), PHII(30), RFUS, SLUPEX(30), SUMPD2,
C      * TDENRT, THMAX, WVEL, WGTI(30), XM
C      COMMON /CMLOC/ ETAI(16), MAXK, MAXP, NBEAM, NETAI, NRBI, NTHMC, NXRC,
C      * RBI(10), TITLE(20), YYPXRC(10,3), YYPTBC(10,3)
C      ROUTINE TO READ INPUT FOR LASER TURRET PHASE DISTORTION ANALYSIS. 720
C      BY G. N. VANUERPLAATS NOV., 1976
C      NAVAL POSTGRADUATE SCHOOL, MONTEREY, CALIF.
C      TITLE.
C      READ (5,70) (TITLE(I), I=1,20)
C      IPNPUT=0
C      IF (IPNPUT.EQ.0) WRITE (6,140) (TITLE(I), I=1,20)
C      AERO-OPTICS.
C      READ (5,80) AMACH, DENRTO, TDENRT, DENGAM, AKPRIM, WVEL
C      IF (IPNPUT.EQ.0) WRITE (6,150) AMACH, DENRTO, TDENRT, DENGAM, AKPRIM, W
C      1AVEL
C      GEOMETRY.
C      TURRET.
C      READ (5,80) RFUS, AL, THMAX, ACL, EPS
C      IF (IPNPUT.EQ.0) WRITE (6,160) RFUS, AL, THMAX, EPS, ACL
C      THMAX=THMAX/57.29578
C      READ (5,90) MAXK, MAXP, NXRC, NTHMC
C      NXRC=NXRC+1
C      NTHMC=NTHMC+1
C      MAXK=MAXK+1
C      MAXP=MAXP+1
C      IF (IPNPUT.EQ.0) WRITE (6,170) MAXK
C      READ (5,80) (ARAR(I), I=1, MAXK)
C      ARAR(I)=1
C      IF (IPNPUT.EQ.0) WRITE (6,180) (ARAR(I), I=1, MAXK)
C      YYPXRC(1,1)=0.
C      YYPXRC(1,2)=EPS
C      YYPXRC(1,3)=200.

```

## SUBROUTINE TINPUT

SEPT. 77

```

IF (NXBC.EQ.1) GO TO 20
IF (IPNPUT.EQ.0) WRITE (6,100)
DO 10 I=2,NXBC
  READ (5,80) (YYPXBC(I,J),J=1,3)
10 CONTINUE
20 CONTINUE
IF (IPNPUT.EQ.0) WRITE (6,110) ((YYPXBC(I,J),J=1,3),I=1,NXBC)
IF (IPNPUT.EQ.0) WRITE (6,190) MAXP
READ (5,80) (BBAR(I),I=1,MAXP1)
C IMPOSE HOJNDARY CONDITION R(0)=1.
  BBAR(1)=1.
  IF (IPNPUT.EQ.0) WRITE (6,180) (BBAR(I),I=1,MAXP1)
  YYPTBC(1,1)=0.
  YYPTBC(1,2)=EPS
  YYPTBC(1,3)=200.
  IF (NTHBC.EQ.1) GO TO 40
  IF (IPNPUT.EQ.0) WRITE (6,120)
  DO 30 I=2,NTHBC
    READ (5,80) (YYPTBC(I,J),J=1,3)
30 CONTINUE
40 CONTINUE
  IF (IPNPUT.EQ.0) WRITE (6,130) ((YYPTBC(I,J),J=1,3),I=1,NTHBC)
C MIRROR CENTER.
  READ (5,80) EPSM,XM
  IF (IPNPUT.EQ.0) WRITE (6,200) XM,EPSM
C PHASE DISTORTION CALCULATION POINTS.
  READ (5,90) NETAI,NRBI
  READ (5,80) (ETAI(I),I=1,NETAI)
  IF (IPNPUT.EQ.0) WRITE (6,230)
  IF (IPNPUT.EQ.0) WRITE (6,240) (ETAI(I),I=1,NETAI)
  DO 50 I=1,NETAI
50 ETAI(I)=ETAI(I)/57.29578
  READ (5,80) (RRI(I),I=1,NRBI)
  IF (IPNPUT.EQ.0) WRITE (6,250)
  IF (IPNPUT.EQ.0) WRITE (6,240) (RRI(I),I=1,NRBI)
C BEAM ORIENTATIONS.
  READ (5,90) NREAM
  IF (IPNPUT.EQ.0) WRITE (6,210)
  DO 60 I=1,NREAM
  READ(5,80) PHII(I),GAMMAI(I),AMACHI(I),WGHTI(I)
  IF(AMACHI(I).LT.0.001) AMACHI(I)=AMACH
  IF(ARS(WGHTI(I)).LT.0.001) WGHTI(I)=1.
  IF(IPNPUT.EQ.0) WRITE(6,220)I,PHII(I),GAMMAI(I),AMACHI(I),WGHTI(I)
  PHII(I)=PHII(I)/57.29578
  GAMMAI(I)=GAMMAI(I)/57.29578
60 CONTINUE
  RETURN
C
70 FORMAT (2A4)
80 FORMAT (8F10.2)

```

1010  
1020  
1030  
1040  
1050  
1060  
1070  
1080  
1090  
1100  
1110  
1120  
1130  
1140  
1150  
1160  
1170  
1180  
1190  
1200  
1210  
1220  
1230  
1240  
1250  
1260  
1270  
1280  
1290  
1300  
1310  
1320  
1330  
1340  
1350  
1360  
1370  
1380  
1390  
1400  
1410  
1420  
1430  
1440  
1450  
1460  
1470  
1480  
1490  
1500



## SUBROUTINE TINPUT

SEPT. 77

```

90 FORMAT(8I10) 1510
100 FORMAT (/5X,19HBOUNDARY CONDITIONS/5X,3HX/L,6X,1HY,4X,7HY-PRIME) 1520
110 FORMAT (3F9.3) 1530
120 FORMAT (/5X,19HBOUNDARY CONDITIONS/5X,11HTheta/THMAX,4X,1HY,4X,7HY 1540
1-PRIME) 1550
130 FORMAT (5X,3F9.3) 1560
140 FORMAT (1H1,4X,21HTURRET ANALYSIS INPUT//5X,5HTITLE/5X,20A4) 1570
150 FORMAT (/5X,11HAERO-OPTICS/5X,36HMACH NUMBER, AMACH 1580
1 =,F6.3/5X,36HEXTERNAL DENSITY RATION, DENRTO =,F6.3/5X,36HIN 1590
2TERNAL DENSITY RATIO, TDENRT =,F6.3/5X,36HPRESSURE-DENSITY EXP 1600
3ONENT, DENGAM =,F6.3/5X,36MPHASE DISTORTION CONSTANT, AKPRIM =,E 1610
411.4/5X,36HWAVELENGTH, WAVEL =,E11.4) 1620
160 FORMAT (/5X,17HGEOMETRY/5X,27HFUSELAGE RADIUS, RFUS =,F7.3/5X, 1630
127HTURRET HALF-LENGTH, =,F7.3/5X,27HTURRET HALF-ANGLE, THMAX 1640
2 =,F7.3,9H DEGREES/5X,27HTURRET HEIGHT FACTOR, EPS =,F7.3/5X,27H 1650
3TURRET HALF-SPACING, ACL =,F7.3) 1660
170 FORMAT (/5X,35HTURRET POLYNOMIAL SHAPE COEFFICIENTS/5X,24HX-DIRECT 1670
1ION, ORDER =,I5/5X,11HCOEFFICIENTS) 1680
180 FORMAT (4X,5E13.5) 1690
190 FORMAT (/5X,24HTheta-DIRECTION, ORDER =,I5/5X,11HCOEFFICIENTS) 1700
200 FORMAT (/5X,28HLOCATION OF CENTER OF MIRROR/5X,6HXM =,F7.3,5X,6 1710
1HEPSM =,F7.3) 1720
210 FORMAT (/5X,17HBEAM ORIENTATIONS/5X,18HBEAM PHI GAMMA,4X, 1730
12HMACH WEIGHT) 1740
220 FORMAT(18,2F8.2,2F8.3) 1750
230 FORMAT (/5X,35HPHASE DISTORTION CALCULATION POINTS/5X,6HANGLES) 1760
240 FORMAT (5X,5F10.3) 1770
250 FORMAT (/5X,5HRAU11) 1780
END 1790

```

## SUBROUTINE TRAP2N

SEPT. 77

```

SUBROUTINE TRAP2N (IGOTO,A,B,N2,X,FX)
ROUTINE TO PERFORM TRAPEZOIDAL RULE INTEGRATION FOR F(X)2N,
BEGINING WITH F(X)N.
BY G. N. VANDERPLAATS
NAVAL POSTGRADUATE SCHOOL, MONTEREY, CALIF.
NOV., 1976
C INPUT
C IGOTO = CALCULATION PARAMETER. INITIALLY CALL TRAP2N WITH
C IGOTO = 0.
C A = LOWER BOUND ON INTEGRATION.
C B = UPPER BOUND ON INTEGRATION.
C N2 = NUMBER OF INTERVALS USED IN THIS SOLUTION. N2 = 1
C IF INTEGRATION IS JUST BEGINING. OTHERWISE N2 = 2*N
C OF PREVIOUS SOLUTION.
C FX = F(X)N ON FORST CALL (IGOTO=0) AND F(X) ON SUBSEQUENT CALLS
C (IGOTO=1).
C OUTPUT
C IGOTO = CALCULATION CONTROL. IF IGOTO.NE.0, CALCULATE F(X) AND
C CALL AGAIN. IF IGOTO=0 ON RETURN, INTEGRATION IS COMPLETE
C X = X-VALUE FOR NEW FUNCTION EVALUATION (IF IGOTO.NE.0)
C FX = F(X)2N IF IGOTO=0. THIS IS FINAL SOLUTION.
C USAGE K IS TOTAL NUMBER OF TRAPEZOIDAL SOLUTIONS DESIRED.
C DO 20 I = 1,K
C N2=2*(I-1)
C IGOTO = 0
C 10 CALL TRAP2N((IGOTO,A,B,N2,X,FX)
C IF(IGOTO.FO.0) GO TO 20
C FX = F(X)
C GO TO 10
C 20 CONTINUE
C SOLUTION IS COMPLETE.
C IF (IGOTO=1) 10,20,40
C CONSTANT.
C 10 H=(B-A)/FLOAT(N2)
C FN=0.
C A1=1.
C A2=1.
C IF (N2.GT.1) GO TO 20
C SPECIAL CASE, 1 INTERVAL.
C A1=H
C A2=.5
C X=A
C IGOTO=1
C RETURN
C GENERAL CASE, N2.GE.1
C 20 FN1=.5*FX+A1
C I=1
C 30 I=I+2
C IF (I.GT.N2) GO TO 50
C X=A+FLOAT(I)*H

```



11.1982

SUBROUTINE TRAP2N

SEPT. 77

SUBROUTINE TRAP2N

IGOTO=2

RETURN

40 FN=FN+FX

GO TO 30

50 FN=A2\*FN

FX=FN1+FN,H

IGOTO=0

RETURN

END

510  
520  
530  
540  
550  
560  
570  
580  
590

## SUBROUTINE YRTPOB

SEPT. 77

```

C SUBROUTINE YRTPOB (XM, EPSM, PHI, GAMMA, RHO, RB, ETA, X, R, THETA, Y, Z) 10
C ROUTINE TO CALCULATE COORDINATES, X, R, THETA OF A POINT ON A BEAM. 20
C BY G. N. VANDERPLAATS NOV., 1976 30
C NAVAL POSTGRADUATE SCHOOL, MONTEREY, CALIF. 40
C INPUT. 50
C XM = X-LOCATION OF CENTER OF MIRROR. 60
C EPSM = Y-LOCATION OF CENTER OF MIRROR. 70
C PHI = AZMUTH ANGLE MEASURED FROM POSITIVE X-AXIS. 80
C GAMMA = ELEVATION ANGLE MEASURED FROM X-Y PLANE. 90
C RHO = DISTANCE ALONG BEAM. 100
C RB = RADIAL DISTANCE FROM CENTER OF BEAM. 110
C ETA = ANGULAR LOCATION MEASURED FROM LINE IN THE X-Y PLANE. 120
C OUTPUT. 130
C X = X-CYLINDRICAL AND CARTISIAN COORDINATE. 140
C Y = Y-CARTISIAN COORDINATE. 150
C Z = Z-CARTISIAN COORDINATE. 160
C R = RADIAL LOCATION TO POINT FROM X-AXIS. 170
C THETA = CIRCUMFERENTIAL LOCATION OF POINT FROM Z-AXIS. 180
C NOTE - ALL ANGLES ARE IN RADIAN. 190
C 200
C 210
C 220
C 230
C 240
C 250
C 260
C 270
C 280
C 290
C 300
C 310
C 320
C 330
C 340
C 350
C 360
C 370
C 380
C 390
C 400
C 410
C 420
C 430
C 440
C 450
C 460
C 470
C 480
C 490
C 500

```

CONSTANTS:  
 SNP=SIN(PHI)  
 CNP=COS(PHI)  
 SNG=SIN(GAMMA)  
 CNG=COS(GAMMA)  
 SNE=SIN(ETA)  
 CNE=COS(ETA)

CARTISIAN COORDINATES.  
 $X = XM - RHO * COS(GAMMA) * COS(PHI) - RB * SIN(ETA) * SIN(PHI) +$   
 $RB * COS(ETA) * SIN(GAMMA) * COS(PHI)$   
 $Y = XM - RHO * CNG * CNP - RB * (SNE * SNP - CNE * SNG * CNP)$   
 $Y = RHO * COS(GAMMA) * SIN(PHI) - RB * SIN(ETA) * COS(PHI) -$   
 $RB * COS(ETA) * SIN(GAMMA) * SIN(PHI)$   
 $Z = EPSM + RHO * SNG * CNP + RB * (SNE * CNP + CNE * SNG * SNP)$   
 $Z = EPSM + RHO * SNG * RB * CNE * CNG$

POLAR COORDINATES.  
 $R = \sqrt{Y^2 + Z^2}$   
 $THETA = ARCTAN(-Y/Z)$   
 GUARD AGAINST ZERO DIVIDE.  
 IF (ABS(Z).LT.1.0E-6) Z=1.0E-6  
 YZ=ABS(Y/Z)  
 THETA=ATAN(YZ)  
 ANGLE GREATER THAN PI/2.  
 IF (Z.LT.0.) THETA=3.1415927-THETA  
 NEGATIVE ANGLE.  
 IF (Y.GT.0.) THETA=-THETA



10			
20			
30			
40			
50			
60			
70			
80			
90			
100			
110			
120			
130			
140			
150			
160			
170			
180			
190			
200			
210			
220			
230			
240			
250			
260			
270			
280			
290			
300			
310			
320			
330			
340			
350			
360			
370			
380			
390			
400			
410			
420			
430			
440			
450			
460			
470			
480			
490			
500			
510			
520			
530			
540			
550			
560			
570			
580			
590			
600			
610			
620			
630			
640			
650			
660			
670			
680			
690			
700			
710			
720			
730			
740			
750			
760			
770			
780			
790			
800			
810			
820			
830			
840			
850			
860			
870			
880			
890			
900			
910			
920			
930			
940			
950			
960			
970			
980			
990			
1000			

SUBROUTINE YRTPOB  
 RETURN  
 END

SEPT. 77

510  
 520

## SUBROUTINE ZERN

SEPT. 77

	SUBROUTINE ZERN(R,R1,R2,T1,T2,AZ,A1,A2,A3,A)	10
	DIMENSION A(10),Z(10)	20
C	ROUTINE TO CALCULATE OPTICAL PROPERTIES OF PHASE DISTORTION IN	30
C	TERMS OF ZERNICKE POLYNOMIALS.	40
C	BY G. N. VANDERPLAATS	50
C	NAVAL POSTGRADUATE SCHOOL, MONTEREY, CALIF.	60
C	PHASE DISTORTION IS ASSUMED OF THE FORM $AZ + A1 \cdot R + A2 \cdot T + A3 \cdot R \cdot T$	70
C	WHERE R = RADIUS AND T = THETA IN RADIANS.	80
C	--- INPUT.	90
C	R = BEAM RADIUS.	100
C	R1, T1 = LOWER LIMITS OF INTEGRATION.	110
C	R2, T2 = UPPER LIMITS OF INTEGRATION.	120
C	AZ, A1, A2, A3 = POLYNOMIAL COEFFICIENTS.	130
C	A = VECTOR OF ZERNICKE COEFFICIENTS. ON FIRST CALL TO ZERN A MUST	140
C	BE ZERO.	150
C	---- OUTPUT.	160
C	A = UPDATED VECTOR OF ZERNICKE COEFFICIENTS.	170
C		180
	DO 20 I=1,4	190
	GO TO (21,22,23,24),I	200
21	CALL ZINT(R,R1,T1,AZ,A1,A2,A3,Z)	210
	SIGN=1.	220
	GO TO 25	230
22	CALL ZINT(R,R1,T2,AZ,A1,A2,A3,Z)	240
	SIGN=-1.	250
	GO TO 25	260
23	CALL ZINT(R,R2,T1,AZ,A1,A2,A3,Z)	270
	SIGN=-1.	280
	GO TO 25	290
24	CALL ZINT(R,R2,T2,AZ,A1,A2,A3,Z)	300
	SIGN=1.	310
25	CONTINUE	320
	DO 30 J=1,10	330
30	A(J)=A(J)+SIGN*Z(J)	340
20	CONTINUE	350
	RETURN	360
	END	370



SEPT. 77

230

## DATA FORMS



**COPEs DATA**

**DATA BLOCK A**

TITLE	FORMAT
	20A4

**DATA BLOCK B**

+	\$							COMMENT
	NCALC	NDV	NSV	N2VAR	IPNPUT	IPSENS	IP2VAR	FORMAT
*								8I10

DATA BLOCK C - OMIT IF NDV = 0

[illegible]

**DATA BLOCK D - OMIT IF NDV = 0**

[illegible]

DATA BLOCK E - OMIT IF NDV = 0

+	\$			COMMENT
	NDVTOT	IOBJ	SGNOBJ	FORMAT
*				2I10,F10

## DATA BLOCK F - OMIT IF NDV = 0

[illegible]

DATA BLOCK A

DATA BLOCK A

[illegible]

DATA BLOCK H - OMIT IF NDV = 0

+	\$		COMMENT
	NCONS		FORMAT
*			I10

**DATA BLOCK I - OMIT IF NDV = 0 OR NCONS = 0**

[illegible]



**COPES DATA CONT.**

**DATA BLOCK I - CONT.**

[illegible]

## DATA BLOCK J - OMIT IF NSV = 0

[illegible]

**DATA BLOCK K - OMIT IF NSV = 0**

[illegible][illegible]

# COPES DATA - CONT.

## DATA BLOCK K - CONT.

+	\$							COMMENT
		ISENS	NSENS					FORMAT
*								2I10
+	\$							COMMENT
		SNS1	SNS2	SNS3	SNS4	SNS5	SNS6	FORMAT
*								8F10

+	\$							COMMENT
		ISENS	NSENS					FORMAT
*								2I10
+	\$							COMMENT
		SNS1	SNS2	SNS3	SNS4	SNS5	SNS6	FORMAT
*								8F10

## DATA BLOCK L - OMIT IF N2VAR = 0

+	\$					COMMENT
		N2VX	M2VX	N2VY	M2VY	FORMAT
*						4I10

## DATA BLOCK M - OMIT IF N2VAR = 0

+	\$							COMMENT
		NZ1	NZ2	NZ3	NZ4	NZ5	NZ6	FORMAT
*								8I10

## DATA BLOCK N - OMIT IF N2VAR = 0

+	\$							COMMENT
		X1	X2	X3	X4	X5	X6	FORMAT
*								8F10

## DATA BLOCK O - OMIT IF N2VAR = 0

+	\$							COMMENT
		Y1	Y2	Y3	Y4	Y5	Y6	FORMAT
*								8F10

## DATA BLOCK P

	END							FORMAT
*	END							3A1



# LASER TURRET ANALYSIS DATA

## DATA BLOCK A

TITLE	FORMAT
	20A4

## DATA BLOCK B

AMACH	DENRTO	TDENRT	DENGAM	AKPRIM	WAVEL	FORMAT
						6F10

## DATA BLOCK C

RFUS	AL	THMAX	ACL	EPS	FORMAT
					5F10

## DATA BLOCK D

MAXK	MAXP	NXBC	NTHBC	FORMAT
				4I10

## DATA BLOCK E

ABAR0	ABAR1	ABAR2	ABAR3	ABAR4	ABAR5	ABAR6	ABAR7	FORMAT
								8F10

## DATA BLOCK F

X	YBC	YPBC	FORMAT
			3F10

## DATA BLOCK G

BBAR0	BBAR 1	BBAR2	BBAR3	BBAR4	BBAR5	BBAR6	BBAR7	FORMAT
								8F10

## DATA BLOCK H

THETA	YBC	YPBC	FORMAT
			3F10

**LASER TURRET ANALYSIS DATA - CONT.**

**DATA BLOCK I**

	EPSP	XM		FORMAT
*				2F10

**DATA BLOCK J**

	NETAI	NRBI		FORMAT
*				2I10

**DATA BLOCK K**

[illegible]

**DATA BLOCK L**

[illegible]**DATA BLOCK M**

	NBEAM	FORMAT
*		I10

**DATA BLOCK N**

[illegible]



# DISTRIBUTION LIST

- |    |  |   |
|----|--|---|
| 1. | Lt. Col. K. Gilbert<br>AFWL/LRO<br>Kirtland AFB, NM 87117  | 5 |
| 2. | Library<br>Naval Postgraduate School<br>Monterey, CA 93940   | 1 |
| 3. | Research Administration<br>Naval Postgraduate School<br>Monterey, CA 93940   | 1 |
| 4. | Defense Documentation Center<br>Cameron Station<br>Alexandria, VA 22314  | 2 |
| 5. | Dr. Garret Vanderplaats<br>Code 1613<br>David Taylor Naval Ship Research<br>and Development Center<br>Bethesda, MD 20084     | 4 |
| 6. | Professor Allen E. Fuhs, Chairman<br>Department of Mechanical Engineering<br>Naval Postgraduate School<br>Monterey, CA 93940 | 4 |
| 7. | Dr. B. Hogge<br>AFWL<br>Kirtland AFB, NM 87117   | 1 |